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Title: Crack characterisation using invariable feature extraction in stainless steel specimen used for absorber tubes of CSP applications via EMAT

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Abstract: Absorber tubes are one of the most critical components of parabolic trough Concentrated Solar Plants. Due to the high temperatures where these tubes perform at with concentration of sunlight, it is very likely to get damaged such as crack etc and lose functionality. Therefore, the monitoring via NDT techniques is essential for preventing them from being significantly defective and reducing maintenance cost. Non-contact method is one of the best inspection candidates, which is more reliable to the tubes at high temperature through a review and the access to the absorber tubes is limited. In this paper, the crack detection and quantification for stainless steel specimen used for absorber tube using Electromagnetic Acoustic Transducers is presented. Through numerical and experimental studies, features are extracted to quantify the crack. Among these features, the ratio between the first edge echo and the second crack echo ( $A_{c2}/A_{e1}$ ) is investigated as invariable feature to factors such as electromagnetic coupling, lift-off distance between EMATs and the specimen etc. In addition, the feature  $A_{c2}/A_{e1}$  has linear relationship with the depth of the crack when the depth is more than 0.75mm, which proves the feature  $A_{c2}/A_{e1}$  is invariable for crack quantification via both numerical modelling and experimental studies.

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03/02/2016

Dear whom it may concerns

I would like to submit the attached manuscript, "Crack characterisation using invariable feature extraction in stainless steel specimen used for absorber tubes of CSP applications via EMAT" for consideration for possible publication in the Renewable Energy. The follows are the abstract of this manuscript. Thanks a lot.

### Crack characterisation using invariable feature extraction in stainless steel specimen used for absorber tubes of CSP applications via EMAT

Absorber tubes are one of the most critical components of parabolic trough Concentrated Solar Plants (CSPs). Due to the high temperatures where these tubes perform at with concentration of sunlight, it is very likely for them to get damaged such as crack and mass loss *etc.*, and lose functionality of power generation. Therefore, the monitoring of their structural health via Non-Destructive Testing (NDT) techniques is regarded as essential for preventing them from being significantly defective and thereby reducing maintenance cost. Non-contact method is one of the best inspection candidates, which is more reliable to the tubes at high temperature through a review and the access to the absorber tubes is limited. In this paper, the crack detection and quantification for stainless steel specimen used for absorber tube using Electromagnetic Acoustic Transducers (EMATs) is presented. Through numerical and experimental studies, features are extracted to quantify the crack. Among these features, the ratio between the first edge echo and the second crack echo ( $A_{c2}/A_{e1}$ ) is investigated as invariable feature to factors such as electromagnetic coupling, lift-off distance between EMATs and the specimen *etc.* In addition, the feature  $A_{c2}/A_{e1}$  has linear relationship with the depth of the crack when the depth is more than 0.75mm, which proves the feature  $A_{c2}/A_{e1}$  is invariable for crack quantification via both numerical modelling and experimental studies.

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**Best Regards**

Liang Cheng

### Highlights:

- Crack detection and quantification for stainless steel specimen used for absorber tubes of CSP plants using Electromagnetic Acoustic Transducers (EMATs) is developed
- An feature invariable to factors such as electromagnetic coupling, lift-off distance between EMATs and the specimen is found
- The feature  $A_{c2}/A_{e1}$  has linear relationship with the depth of the crack

1 Crack characterisation using invariable feature extraction in stainless steel  
2 specimen used for absorber tubes of CSP applications via EMAT

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11 very likely for them to get damaged such as crack and mass loss *etc.*, and lose functionality of power  
12 generation. Therefore, the monitoring of their structural health via Non-Destructive Testing (NDT)  
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14 reducing maintenance cost. Non-contact method is one of the best inspection candidates, which is  
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24 **Key words:** Electromagnetic acoustic transducer, stainless steel, crack, non-  
25 destructive testing and evaluation, invariable feature

26 **1. Introduction**

27 As the greenhouse effect on the climate across the world grows, there is an increasing  
28 need for more stable renewable sources of energy as alternative means for environmentally  
29 friendly power generation. Concentrated Solar Power (CSP) plant is a promising technology  
30 for renewable energy production. By the end of 2014 there were thirty-five CSP plants  
31 producing more than 2.5 GW of power in Europe [1]. This represented more than 55% of the  
32 total global CSP capacity amounted to a total CSP production capacity of 4.4 GW. Outside  
33 Europe there were eleven CSP plants in the US with four of the biggest ones having been  
34 completed in 2014, three in China and twelve in the rest of the world. As of early 2015, there  
35 were twenty-two CSP plants under construction around the world which will add another 2.5  
36 GW of capacity by 2015 (265 MW installed in Europe) [2]. Several more CSP projects have  
37 been announced around the world. If all of them materialise they will add another 9 GW of  
38 CSP capacity by 2025. At the moment, Spain is the European and world leader in the  
39 exploitation of CSP technology with the U.S. and China following. In the U.S. the total  
40 installed CSP capacity saw a significant increase in 2014 with more than 1 GW connected to

41 the grid. By 2020 it is anticipated that the U.S. and China will have closed the gap with  
 42 Europe considerably. Nonetheless, it is expected that at least in the medium term Spain will  
 43 retain its global leadership in total installed CSP capacity.

44 CSP plants consist of several kilometres of solar absorber tubes and insulated pipes  
 45 working at a high temperature up to 550°C. The inspection of CSP tubing and piping,  
 46 absorber tubes in particular, is currently very challenging due to their complex design which  
 47 offers poor access to the surfaces requiring inspection. Mahoney from Sandia National  
 48 Laboratories reported a failure rate of 30-40% in solar absorbers at the Solar Energy  
 49 Generating Systems within a decade of operation [3]. The price of replacement was  
 50 expensive resulting in a significant extra maintenance cost on an annual basis [3]. Failures  
 51 can also result in significant leaks and fires due to combustion of the oil commonly used as  
 52 working fluid in the majority of CSP plants resulting in further significant infrastructure  
 53 damage [4]. Pitting and general corrosion of the solar receiver and insulated pipes is one of  
 54 the most common structural degradation mechanisms. Operation at a wide temperature range  
 55 involving repeated heating and cooling cycles may result in more aggressive forms of  
 56 corrosion [5]. Corrosion and Stress Corrosion Cracking (SCC) can lead to sudden and  
 57 catastrophic failure, especially in plants operating at high pressure [6]-[7]. Local pitting  
 58 corrosion can cause initiation of stress corrosion cracking or result in small-scale leaks. Most  
 59 stainless steel pipes are vulnerable to pitting corrosion and stress corrosion cracking [8].

60 At the moment there is no reliable methodology for the inspection of in-service solar  
 61 receivers and insulated pipes and therefore CSP plant maintenance procedures are corrective  
 62 rather than preventive. Therefore there is an urgent need to increase the reliability of CSP  
 63 infrastructure and optimise maintenance procedures by using efficient and cost-effective  
 64 inspection methods. In ref [9], a comparison of the advantages and disadvantages of the non-  
 65 destructive testing and evaluation (NDT&E) methods available to CSP plant operators is  
 66 shown in Table 1. The compared NDT&E methods include magnetic flux leakage (MFL)  
 67 [10]-[12], eddy current testing (ECT) [13]-[16], alternating current field measurement  
 68 (ACFM) [17]-[18], radiographic inspection (RI) [19]-[21], ultrasonic testing (UT) [22]-[23],  
 69 long range ultrasonic testing (LRUT) [24]-[26], electromagnetic acoustic transducer (EMAT)  
 70 [27]-[28], infrared thermography (IR) [29]-[32], acoustic emission (AE) [33]-[34], *etc.*

71 **Table 1** Comparison of advantages and disadvantages of NDT&E methods for CSP plant inspection

Technique	Advantages	Disadvantages	Detection capability
MFL	Fast, sensitive to transverse cracks and corrosion, applicable for surface and hidden defects, applicable on some ferrous pipes and storage tanks walls and floor, can be automated, low lift-off sensitivity, pigging compatible	Only ferrous pipes and storage tanks, defect geometry influences quantification, parallel cracks can be missed, if wall thickness loss is gradual can go undetected, local inspection, requires good magnetisation to avoid underestimation or missed defects, bulky equipment	Surface and hidden corrosion and fatigue cracks, inclusions

ECT	Inexpensive, sensitive to microstructural, electric and magnetic properties, sensitive to small defects, applicable to any conductive material, pigging compatible, can be automated, can operate at significant lift-offs	Very lift-off sensitive, inspection penetration depth and resolution dependent on frequency, local inspection, more efficient for surface and near-surface inspection, low resolution in high lift-offs	Surface and near-surface defects (cracks and pitting corrosion), general corrosion, microstructural changes
ACFM	Inexpensive, sensitive to small defects, capable of quantifying depth and length of surface-breaking defects, pigging compatible, can be automated, can operate at significant lift-offs	Only surface-breaking defects, local inspection, quantification only possible for fatigue cracks	Surface-breaking defects including pitting corrosion and fatigue cracks
RI	Accurate, does not require removal of the insulation of glass envelope, provides permanent record, can be digitised, can quantify wall loss in insulated pipes, can inspect weld quality, applicable to all components	Health and safety issues, time consuming, local inspection, requires access from both sides, bulky and expensive equipment if digital detectors and portable X-ray sources are used, very difficult to detect cracks	Internal and surface defects associated with corrosion and weld inclusions
UT	Relatively inexpensive unless phased arrays are used, capable of detecting hidden defects and quantifying both hidden and surface-breaking defects, can be applied to any type of material	Not applicable to solar absorber tubes, requires removal of insulation, local inspection	Internal and surface defects including fatigue cracks and corrosion
LRUT	Relatively fast, capable of detecting large hidden and surface breaking defects, can be applied to any type of material, can inspect long sections up to several tens of metres in one go, requires removal of insulation only in the area of installation	Only simple geometries can be inspected (i.e. pipes), considerable dead zone, defects need to be relatively large to be detectable, signal to noise ratio can be affected by the inspection conditions (e.g. presence of tight insulation, working fluid, etc.)	Relatively severe corrosion and transverse cracks
EMAT	Inexpensive, non-contact, no material limitation as long as it is conductive, can detect both hidden and surface-breaking defect, can be local or long range, can be applied at high temperature, easy to produce specific waves and modes	Low signal to noise ratio, sensor requires cooling at high temperatures, bulky sensors, lift-off cannot exceed 2 mm	Surface and hidden defects including corrosion and fatigue cracks
IR	Fast and global, excellent for the detection of heat losses, can detect leaks	Difficult to detect structural defects, can be affected by surroundings, expensive equipment	Heat losses and leak detection
AE	Continuous monitoring, can be applied for detection of crack initiation and propagation,	No quantitative information of damage, influenced adversely by noise sources, can be expensive,	Corrosion, cracking, leaks

	detection of corrosion debris, long term monitoring, can be used at high temperature	complicated data management	
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72

73 Among the NDT techniques mentioned above, non-contact UGW can be seen as a good  
74 candidate for this application, because it is capable of non-contact ultrasound generation and  
75 reception with a large coverage of inspection range. EMAT transducers are highly  
76 appropriate for the generation of UGW without contact, because the ultrasound is directly  
77 generated within the material. Due to the couplant-free feature, EMATs are particularly  
78 useful for automated inspection, and hot, cold, clean, or dry environments. EMAT are ideal  
79 transducers for generating Shear Horizontal (SH) bulk wave mode [27], Surface Wave, Lamb  
80 waves [28] and all sorts of other guided-wave modes in metallic and/or ferromagnetic  
81 materials.

82 Several attempts to measure the size of cracks using EMAT have been carried out [35]-  
83 [39]. Wilcox *et al* [35] extracted voltages in frequency domain of A0 and S0 wave mode to  
84 quantify the depth of the crack in aluminium plate. Dixon *et al* [36] used the amplitude of A1,  
85 S1 and A2 extracted from frequency response to describe the crack in different thickness of  
86 the plate. Bemstein *et al* also used the magnitude in frequency domain for crack  
87 characterisation. Her *et al* [38] found the crack leads to the time delay between the two pulses  
88 deduced from the frequency spectrum. Edwards *et al* [39] measured the depth of cracks using  
89 the amplitude of the first received echo in time-domain, showing the amplitude of the first  
90 echo decreases when the depth of the crack increases. All above approaches using amplitudes  
91 as a feature are valid with an assumption of the lift-off or coupling between EMAT and  
92 specimen are constant. Once the lift-off or coupling condition varied during the experiment  
93 due to surface condition or local property variation of the specimen, the amplitude will be  
94 changed significantly. In that case, the feature is no longer valid for accurate approximation  
95 of the depth of the crack. In this paper, an invariable feature to the variation of lift-off or  
96 coupling condition will be investigated and validated via both simulation and experiment.

97 The rest of the paper is organised as follows: Section 2 introduces the theoretical  
98 background of EMAT; Sections 3 and 4 focus on the simulation and experimental studies on  
99 the sizing of the depth of crack via selected features, respectively. A conclusion of the  
100 findings from the studies is drawn in Section 5.

## 101 **2. Theoretical background of EMAT**

102 There are two mechanisms to generate waves through magnetic field interaction. One is  
103 Lorentz force, when the material is conductive, and the other is magnetostriction when the  
104 material is ferromagnetic. In this application, the materials used for absorber tubes are  
105 austenitic stainless steel grades which are paramagnetic. Therefore, Lorentz force is the  
106 dominant factor for the generation of ultrasound.

107 Lorentz force is generated by the alternating current (AC) in the electric coil inducing  
108 eddy currents on the thin skin of the material. Due to skin effect, the distribution of eddy  
109 current is only at a thin layer of the material. Eddy current in the magnetic field experiences

110 Lorentz force. The Lorentz force is applied on the surface region of the material, governed by  
111 following equation:

$$112 \quad \mathbf{f} = q\mathbf{E} + \mathbf{J} \times \mathbf{B} \quad (1)$$

113 where  $\mathbf{f}$  is Lorentz force density (force per unit volume),  $q$  is the charge density (charge  
114 per unit volume),  $\mathbf{E}$  is the electric field,  $\mathbf{J}$  is the current density and  $\mathbf{B}$  is the magnetic flux  
115 density. When there is no charge, the equation can be rewritten as:

$$116 \quad \mathbf{f} = \mathbf{J} \times \mathbf{B} \quad (2)$$

117 Then the tensor of  $\mathbf{f}$  can be written in x, y, z coordination as:

$$\begin{aligned} 118 \quad f_x &= J_y B_z - J_z B_y \\ 119 \quad f_y &= J_z B_x - J_x B_z \\ 120 \quad f_z &= J_x B_y - J_y B_x \end{aligned} \quad (3)$$

121 A ferromagnetic material undergoing a dimensional change due to an external magnetic  
122 field being applied is referred to as magnetostrictive. The phenomenon is called  
123 magnetostriction, and the change is affected by the magnitude and direction of the field [40].  
124 The AC in the electric coil induces an AC magnetic field and thus produces magnetostriction  
125 at ultrasonic frequency in the material. The disturbances caused by magnetostriction then  
126 propagate in the material as an ultrasound wave.

127 In this application, the major material for absorber tube is paramagnetic austenitic  
128 stainless steel as mentioned earlier with a relative magnetic permeability  $\mu_r=1.008$ . Therefore,  
129 Lorentz force is the main consideration for ultrasound generation.

### 130 **3. Numerical modelling for crack detection and quantification**

#### 131 **a. Finite element method using COMSOL Multi-physics**

132 Finite Element Method (FEM) techniques, based on numerical solutions of Partial  
133 Differential Equations (PDEs), offer a method for finding approximate numerical solutions of  
134 the coupling of electromagnetics and ultrasonics for EMAT. The solution approach involves  
135 either eliminating the differential equations completely or rendering the PDEs into an  
136 approximating system of ordinary differential equations, which are then solved numerically  
137 by integration using standard techniques such as Euler's method [41]. By using COMSOL,  
138 models of the Lorentz force generated by coupled electromagnetic field excited by magnets  
139 and coils and the propagation of ultrasound is possible to be determined with a reasonable  
140 accuracy. This can be performed using time dependant analysis in the magnetic field,  
141 magnetic field without currents and structural mechanics modules.

- 142 • Magnetic field module is used to solve the induced currents  $\mathbf{J}_e$  (or called eddy currents) in  
143 the test sample excited by coils.
- 144 • Magnetic field without currents module is implemented to calculate the static magnetic  
145 field  $\mathbf{B}_s$  generated by magnets.

146 • Lorentz force can be calculated according to  $\mathbf{J}_e$  and  $\mathbf{B}_s$  governing equation  $\mathbf{f} = \mathbf{J}_e \times \mathbf{B}_s$ .  
 147 Then, solid mechanics module is utilised to model the ultrasound generated in the test  
 148 sample with respect to Lorentz force.

149 According to the convergence study completed [42], the minimum number of elements  
 150 required for a wavelength to obtain the best approximation of results is 10. Therefore the  
 151 maximum element size to be used to define the mesh can be calculated as follows:

$$152 \quad \text{Max element size} = \lambda_{min}/10 \quad (4)$$

153 where  $\lambda_{min}$  is the minimum separation between mesh elements.

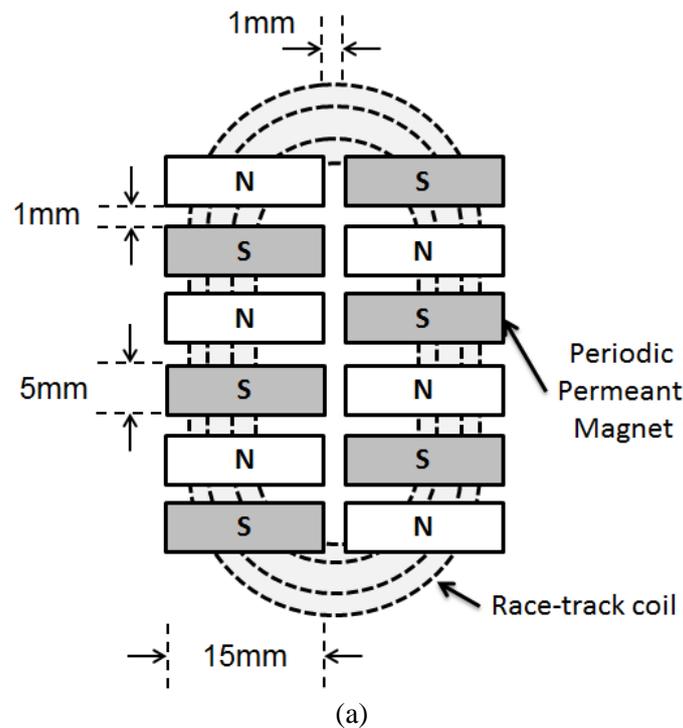
154 For the maximum time step, it is governed following equation:

$$155 \quad \text{Max time step} = \lambda_{min} / (10v) \quad (5)$$

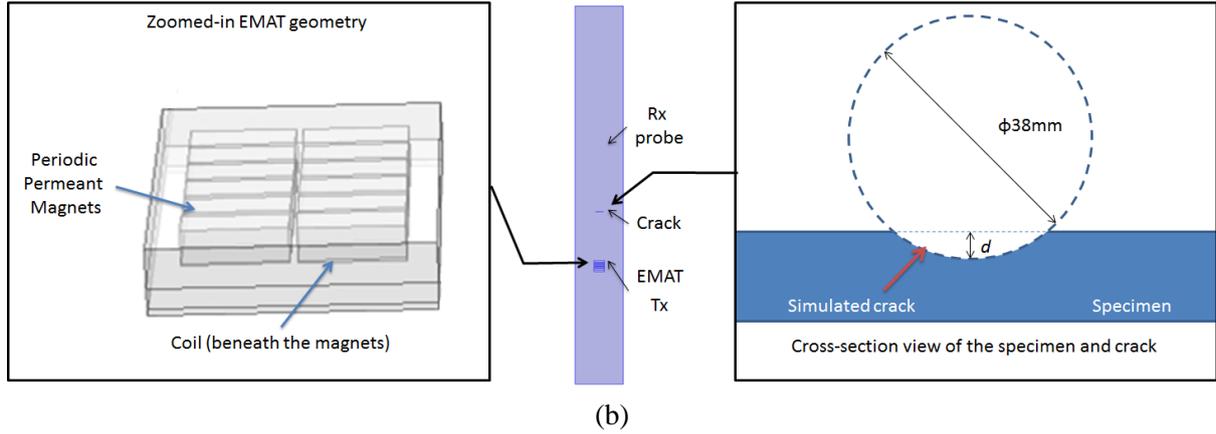
156 where  $v$  is speed of wave.

### 157 b. Numerical model

158 In order to focus the energy to cover the inspection of the whole sample up to several metres,  
 159 Shear Horizontal (SH) mode SH0 is determined as the best option for this application. As a trade-off  
 160 of propagation range and sensitivity to the defect, a wavelength of 12mm is fixed via Periodic  
 161 Permeant Magnet (PPM) design. SH EMAT working at 256kHz for the wavelength of 12 mm is  
 162 determined for 3mm-thick plate made of stainless steel 316L. Then an array of PPM and a race-track  
 163 coil are selected for the EMAT transducers. The design of the EMAT is shown in Figure 1a and this  
 164 configuration of EMAT developed in COMSOL as shown in Figure 1b, where a Hanning window  
 165 centred at 256 kHz with five cycles is used for excitation of the coil.  
 166



167  
 168



**Figure 1.** (a) Design of the EMAT transducer in top view; (b) EMAT model developed in COMSOL for the investigation of cracks.

As illustrated in Figure 1a, the dimension of each magnet is 15mm x 5mm x 5mm. The distance between magnets is 1mm. In addition, the magnetic strength of each magnet is 0.3T. The diameter of coil is 0.4 mm and the width and length are 15 mm and 35 mm respectively with a lift-off distance 0.1mm to the sample. The excitation current density  $J(t)$  is defined as follow:

$$J(t) = \begin{cases} J_0 \sin(2\pi ft) [1 - \cos(2\pi ft/N)], & \text{for } t \leq N/f \\ 0, & \text{for } t > N/f \end{cases} \quad (6)$$

where  $J_0 = 1\text{A}/(0.4 \text{ mm})^2$  (diameter of coil is 0.4 mm),  $f = 256 \text{ kHz}$ ,  $N = 5$  cycles. The material properties is summarised in Table 2.

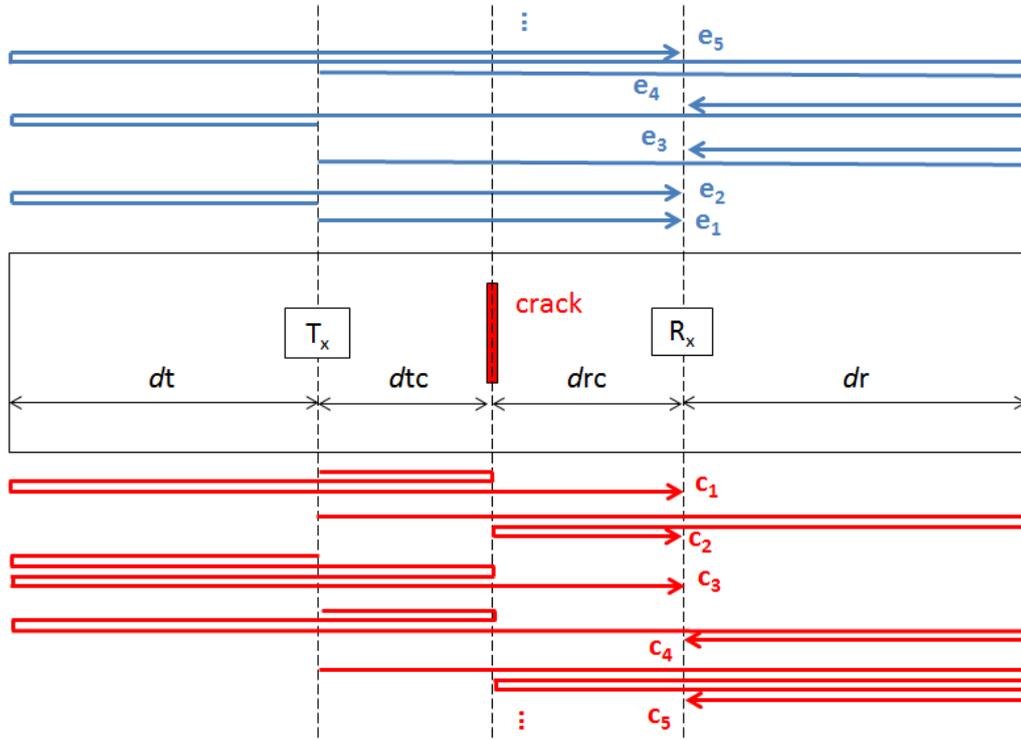
**Table 2** Material properties of stainless steel 316Ti at room temperature

Young's Modulus (GPa)	195
Poisson Ratio	0.285
Density ( $\text{kg/m}^3$ )	8000
Electrical conductivity (S/m)	1.45e6
Permeability	1.008
Permisivity	1

As shown in Figure 1b, the simulated crack is created by making the difference between the specimen and a circle disk with the diameter of 38mm. The circle disk is to simulate the slitting disk of a grinder used in experiment to cut the crack with a certain depth. The depth of crack  $d$  can be modified by moving the circle disk upwards or downwards in the numerical model. The geometric shape in the cross-sectional view is shown in Figure 1b. The width of the crack is 1mm and the tip-to-tip length of the crack can be calculated via the geometric configuration.

In the numerical model, the length and thickness of the stainless steel plate are 1.25m and 3mm, respectively, which matches the dimension of the specimen used in experimental studies discussed in Section 4. To reduce the calculation cost, the width is set as 50cm while the two boundaries on the

195 width sides are set as “continuity” to avoid any reflections from these boundaries. The positions of  
 196 EMAT transmitter, crack and reception probe are defined in Figure 2. Under this pitch-catch  
 197 configuration, the expected echoes reflected from only edges of the defect-free sample (denoting as  $e_1$ ,  
 198  $e_2$ , *etc.*) and additional ones from the crack of defective sample (denoting as  $c_1$ ,  $c_2$ , *etc.*) are shown in  
 199 the Figure 2.



200

201 **Figure 2** Illustration of the configurations of EMATs and crack with potential echoes to be received  
 202 for the case  $dt < dr$  and  $dtc = drc$

203 From above figure, the distances of flight for the echoes  $e_1$  to  $e_5$  can be summarised as  
 204 follows:

205 
$$D_{e1} = dtc + drc$$

206 
$$D_{e2} = 2 * dt + dtc + drc$$

207 
$$D_{e3} = 2 * dr + dtc + drc$$

208 
$$D_{e4} = 2 * dt + dtc + drc + 2 * dr$$

209 
$$D_{e5} = 2 * dt + 3 * dtc + 3 * drc + 2 * dr \quad (7)$$

210 And the distances of flight for the echoes  $c_1$  to  $c_5$  are:

211 
$$D_{c1} = 2 * dt + 3 * dtc + drc$$

212 
$$D_{c2} = dtc + 3 * drc + 2 * dr$$

213 
$$D_{c3} = 4 * dt + 3 * dtc + drc$$

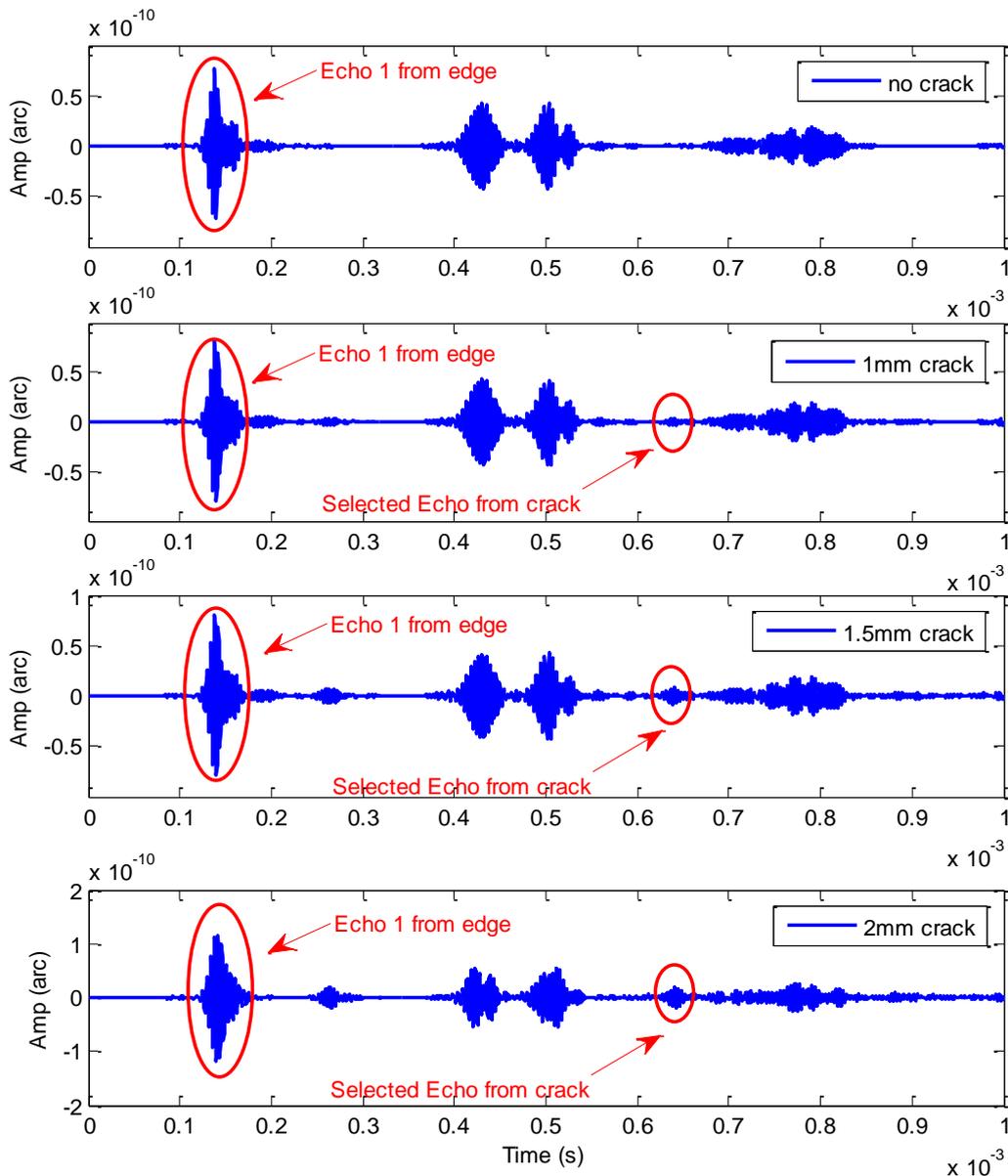
214 
$$D_{c4} = 2 * dt + 3 * dtc + drc + 2 * dr$$

215 
$$D_{c5} = dtc + 3 * drc + 4 * dr \quad (8)$$

216

217 **c. Simulation results for cracks**

218 In the simulation, the values of  $dt$ ,  $dte$ ,  $drc$  and  $dr$  are set as 39cm, 18cm, 18cm and 50cm,  
 219 respectively to match the configuration in experimental studies. In total, 18 simulations are  
 220 carried out, including 17 trials with various depths of the crack from 0.25mm to 2.9mm and  
 221 one reference for crack-free case. The simulated received signal for crack-free, 1mm, 1.5mm  
 222 and 2mm deep crack are shown in Figure 3. A time span to obtain the echoes of  $e_1$  to  $e_4$  and  
 223  $c_1$  to  $c_5$  is selected as 1ms.



224  
 225 **Figure 3** Simulation results of received signals for crack-free, 1mm, 1.5mm and 2mm  
 226 deep cracks

227 Taking  $c_2$  marked as ‘selected echo from crack’ in Figure 3 as an example, it can be  
 228 noticed that the amplitude of  $c_2$  ( $A_{c2}$ ) increases and the amplitude of  $e_1$  ( $A_{e1}$ ) decreases when  
 229 the depths of the crack increases for the four cases shown in Figure 3. The reduction on  $A_{e1}$

230 means that a part of ultrasound energy is reflected backwards and dissipated at the crack  
 231 when the wave propagates through the crack from the EMAT transmitter to the receiver. As  
 232 the increase on the crack depth, the energy loss due to the reflection and dissipation at the  
 233 crack becomes larger, resulting in a reduction on  $A_{e1}$ . For  $c_2$ , the ultrasound propagates  
 234 through the crack firstly, reflects on the right edge of the sample, propagates towards the  
 235 crack and then reflects back at the crack and finally received at the receiver, which can be  
 236 seen in Figure 2. For various depths of the crack, the difference is the propagation through and  
 237 the reflection at the crack for echo  $c_2$ . The overall production of these is an enhancement on  
 238 the  $A_{c2}$  from the simulation results. It means the ratio of the reflection at the crack is larger  
 239 than the loss when the wave propagates through the crack. Therefore,  $A_{e1}$  and  $A_{c2}$  can be used  
 240 as features to quantify the crack.

241 However, there is not the only factor of the crack contributing to the amplitudes of the  
 242 echoes such as  $A_{e1}$  and  $A_{c2}$ . In reality, the coupling between EMAT transmitter or receiver  
 243 and the specimen for different measurement is not always the same. The variation of the  
 244 coupling conditions will strongly affect the amplitudes of the received echoes, in term of  
 245 voltages in experimental studies. In addition, the amplitude of echoes is very sensitive to the  
 246 lift-off distance between the EMAT and the specimen. The amplitude will reduce  
 247 exponentially when the lift-off increases. Moreover, the temperature also has an influence on  
 248 the amplitudes. All above factors in the experiment lead to non-reliable  $A_{e1}$  and  $A_{c2}$  as  
 249 features to quantify the crack, especially the curved specimen introduces more uncertainties  
 250 of coupling and lift-off when EMATs are placed on a pipe. In the next sub-section, invariable  
 251 features will be investigated for the crack quantification regardless with those external  
 252 factors.

#### 253 **d. Feature extraction and crack quantification**

254 As discussed in Section 3c, features such as  $A_{e1}$  and  $A_{c2}$  can only work at ideal situation.  
 255 The most common and inevitable influence in experimental studies is the lift-off variation. In  
 256 order to eliminate the influence from external factors, such as lift-off, on the stability of the  
 257 feature, the ratio between  $A_{c2}$  and  $A_{e1}$  in dB ( $20 \cdot \log(A_{c2}/A_{e1})$ ) is selected. It can assume that  
 258 the loss of propagation through and dissipation at the crack when the ultrasound pass through  
 259 the crack is a function of the depths of defect, denoting as  $P(d)$  in dB; the attenuation of the  
 260 reflection at the crack is also a function of the depths of defect, denoting as  $R(d)$  in dB. The  
 261 amplitudes of  $e_1$  and  $c_2$  can be expressed as follows according the propagation paths shown in  
 262 Figure 2:

$$263 \quad A_{e1} \text{ (dB)} = A_0 \text{ (dB)} + \sigma * d_{tc} + P(d) + \sigma * d_{rc} \quad (9)$$

$$264 \quad A_{c2} \text{ (dB)} = A_0 \text{ (dB)} + \sigma * d_{tc} + P(d) + \sigma * d_{rc} + \sigma * d_r + \sigma_{\_edge} + \sigma * (d_r + d_{rc}) + R(d) + \sigma * d_{rc} \quad (10)$$

266 where  $A_0$  denotes the amplitude of the ultrasound generated at the EMAT transmitter and  
 267 propagating towards the right edge of the specimen in Figure 2;  $\sigma$  is a linear function denoting  
 268 the attenuation ratio against distance (unit dB/m);  $\sigma_{\_edge}$  is a constant, denoting the  
 269 attenuation of the reflection at the edge of the specimen. The value of  $A_0$  takes into account  
 270 of lift-off variation, coupling *etc.* Because  $A_{e1}$  and  $A_{c2}$  are obtained in the same measurement,

271 the values of  $A_0$  in equation (9) and (10) are the same.  $A_{c1}$  is not selected because the wave  
 272 generated from the transmitter propagates towards the left edge of the specimen rather the  
 273 right edge. In that case, the values of  $A_0$  for  $A_{e1}$  and  $A_{c1}$  are not necessarily the same even for  
 274 the same measurement. Subtracting (10) by (9), it comes:

$$275 \quad A_{c2}/A_{e1} \text{ (dB)} = R(d) + \sigma_{\text{edge}} + \sigma^*(2*dr+2* drc) \quad (11)$$

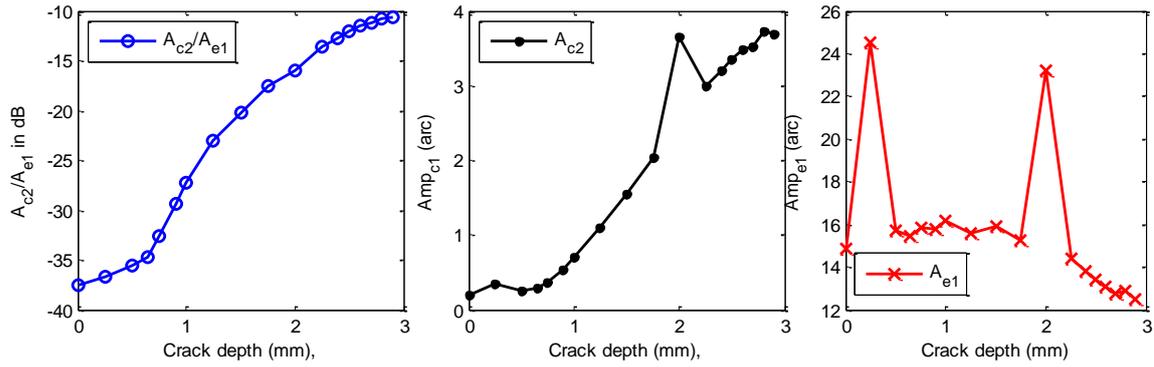
276 For the different depths of the crack, the components  $\sigma_{\text{edge}}$  and  $\sigma^*(2*dr+2* drc)$  stay  
 277 unchanged in the equation (11). Therefore, the feature  $A_{c2}/A_{e1}$  reflects the function  $R(d)$  for  
 278 the quantification of the crack without influence of other factors discussed above. Table 3 and  
 279 Figure 4 show the comparison of three features  $A_{c2}/A_{e1}$ ,  $A_{c2}$  and  $A_{e1}$  against the depth of the  
 280 crack  $d$  via numerical modelling.

281 It can be noticed that  $A_{c2}/A_{e1}$  monotonically increases as the  $d$  increases without any  
 282 influence on the variation of lift-offs, where the features  $A_{c2}$  and  $A_{e1}$  suffer from strong  
 283 influence on the variation of lift-offs for  $d=0.25\text{mm}$  and  $2\text{mm}$  cases. In particular, the trace of  
 284  $A_{c2}/A_{e1}$  against  $d$  is likely linear when  $d \geq 0.75\text{mm}$ , which shows the feature  $A_{c2}/A_{e1}$  is an  
 285 invariable and efficient feature for crack quantification. The feature  $A_{c2}/A_{e1}$  will be also used  
 286 in experimental studies for the validation in the Section 4.

287 **Table 3** Features of the amplitude of 1<sup>st</sup> edge echo ( $A_{e1}$ ), 2<sup>nd</sup> crack echo ( $A_{c2}$ ) and the ratio between  
 288  $A_{c2}$  and  $A_{e1}$  ( $A_{c2}/A_{e1}$ ) against varied crack depths

$d$ (mm)	0	0.25 *	0.5	0.65	0.75	0.9	1.0	1.25	1.5
$A_{c2}/$ $A_{e1}$ (dB)	-	-	-	-	-	-	-	-	-
$A_{c2}$ (arc)	0.19 622	0.35 61	0.26 09	0.28 51	0.37 44	0.53 44	0.70 73	1.09 67	1.55 67
$A_{e1}$ (arc)	14.8 53	24.5 2	15.7 03	15.4 78	15.8 77	15.7 91	16.1 8	15.6 04	15.8 85
$d$ (mm)	1.75	2.0*	2.25	2.4	2.5	2.6	2.7	2.8	2.9
$A_{c2}/$ $A_{e1}$ (dB)	-	-	-	-	-	-	-	-	-
$A_{c2}$ (arc)	2.03 8	3.65 9	3.00 2	3.20 2	3.34 6	3.47 9	3.51 9	3.72 6	3.68 1
$A_{e1}$ (arc)	15.2 49	23.2	14.4 05	13.8 06	13.4 07	13.1 07	12.7 72	12.8 97	12.5 02

289 \* The lift-off distances for two cases for  $d=0.25\text{mm}$  and  $d=2\text{mm}$  are intentionally reduced  
 290 compared with other cases.



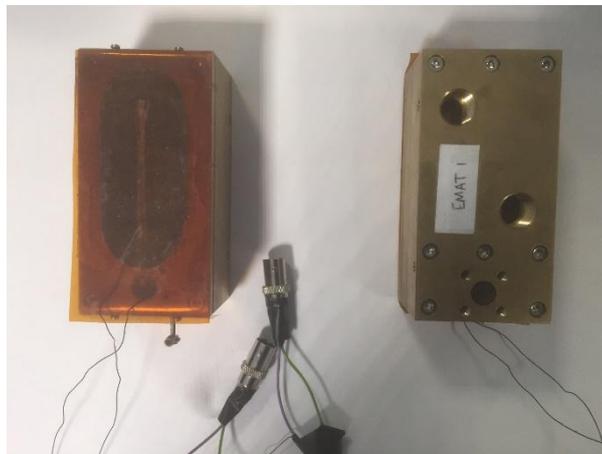
291

292 **Figure 4** Simulation features of the amplitude of 1<sup>st</sup> edge echo ( $A_{e1}$ ), 2<sup>nd</sup> crack echo ( $A_{c2}$ ) and the ratio  
 293 between  $A_{c2}$  and  $A_{e1}$  ( $A_{c2}/A_{e1}$ ) against varied crack depths

294 **4. Experimental validation**

295 a. EMAT transducers

296 A pair of EMATs specifically designed for the inspection of the absorber tubes needs to  
 297 withstand high temperatures and generate/receive waves of in plane displacement  
 298 propagating axially over long distances. In this case, SH0 (or T(0,1) in pipes) wave mode can  
 299 be used due to its non-dispersive nature and, therefore, a PPM racetrack coil EMAT can be  
 300 used for excitation/reception, with the same magnet and coils structure as the design in  
 301 numerical modelling. The integrated EMAT transducers are shown in Figure 5.



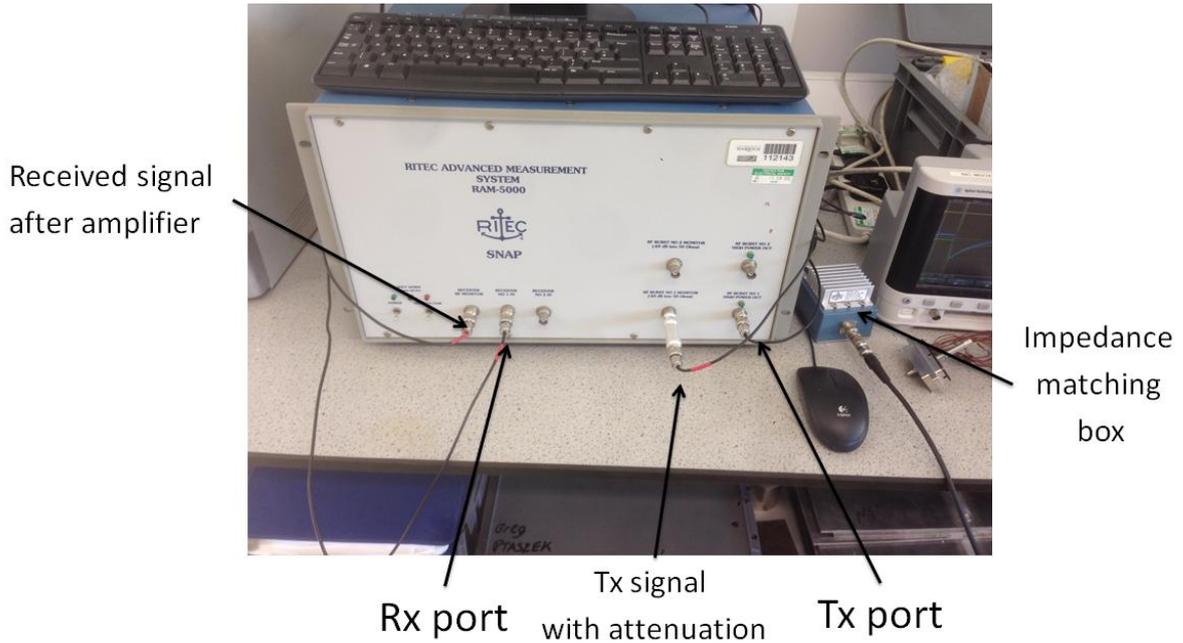
**Figure 5** Developed EMAT transducers

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304 b. Experiment setups

305 EMAT transducers work in pitch-catch mode. EMAT Tx was connected to the  
 306 transmission port of Ritec via BNC cable and EMAT Rx was linked with the reception port  
 307 of Ritec. The ports of Ritec along with impedance matching box are shown in Figure 6. A  
 308 316Ti stainless steel plate with the dimension of 1.25m × 1.25m × 3mm was used to  
 309 investigate the cracks within stainless steel specimen. The Ritec pulser-receiver was  
 310 controlled by a computer with Ritec software to set-up the excitation waveform using  
 311 Hanning with 6 cycles at 256 kHz with maximum power output around 1000V peak-to-peak  
 312 voltage. At reception side, an 80dB gain amplifier is used.



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**Figure 6** Picture of Ritec pulser-receiver

315 There are three cracks were manufactured using a grinder with a slitting disk with the  
 316 diameter of 38mm. Because the cracks were manufactured at different locations, the  $dt$  and  $dr$   
 317 values are slightly different. The depths of each crack and its experimental configuration are  
 318 listed in Table 4. The differences of the values of  $dt$  and  $dr$  are relatively small, therefore, the  
 319 differences in  $\sigma^*(2*dr+2* drc)$  can be ignored.

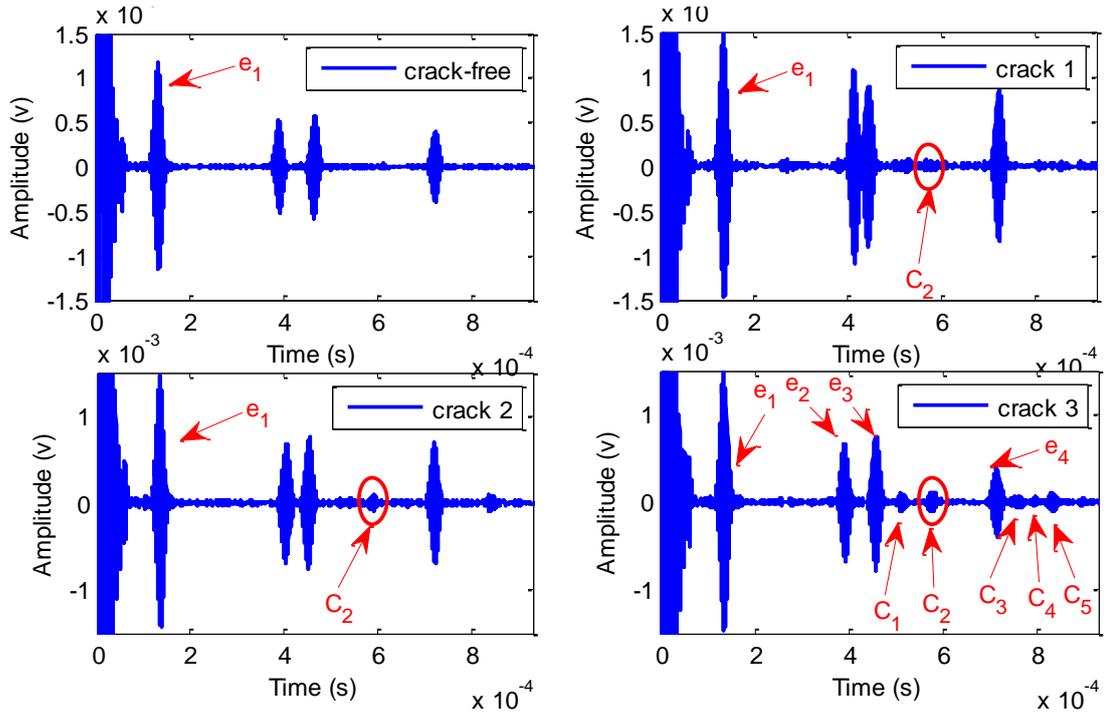
320 **Table 4** Depths of crack and experiment configuration.

Crack #	$d$ (mm)	$dt$ (cm)	$dr$ (cm)	$drc$ (cm)	$drc$ (cm)
1	1.130	42	46.5	18	18
2	1.442	41.5	46.5	18	18
3	1.727	39	48.5	18	18

321

322 c. Results and validation

323 The experiments in pitch-catch mode for crack-free and cracks 1-3 were carried out under  
 324 the configuration shown in Table 4. The received EMAT signals for these 4 cases are shown  
 325 in Figure 7. Echoes  $e_1$ - $e_4$  and  $c_1$ - $c_5$  can be seen within the time slot shown in Figure 7. From  
 326 the results, it can also be noticed that all three cracks can be detected via cracked echo  $c_2$ . In  
 327 addition, the amplitude of  $c_2$  for the crack 3 is the largest due to the largest depth of the crack.  
 328 As the values of  $dt$  and  $dr$  are slightly different for the cracks, the time of arrivals of  $e_1$  to  $e_4$   
 329 are slightly different. The difference between  $dt$  and  $dr$  is the smallest for crack 1,  $e_2$  and  $e_3$   
 330 are more close to each other for the signal of crack 1 according to equation (7).



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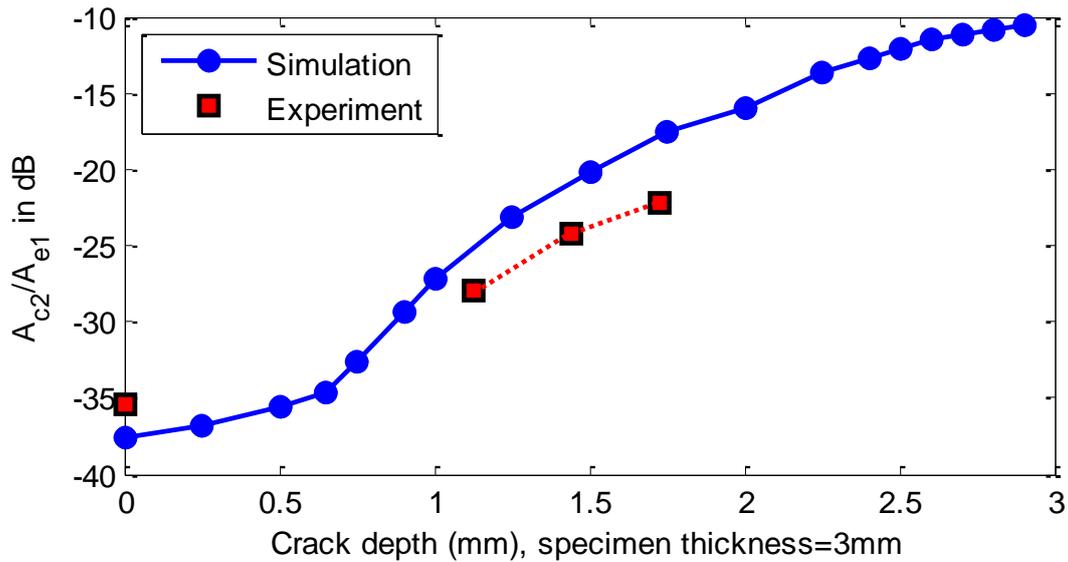
336

**Figure 7** Experiment results of received EMAT signals for crack-free, cracks 1, 2 and 3  
 After identifying the echoes, the feature  $A_{c2}/A_{e1}$  for above four experimental case studies is calculated in Table 5. A comparison of  $A_{c2}/A_{e1}$  derived from numerical and experiment against the depths of the crack is drawn in Figure 8.

**Table 5** Experimental feature of the ratio between  $A_{c2}$  and  $A_{e1}$  ( $A_{c2}/A_{e1}$ ) against varied crack depths

$d$ (mm)	0	1.130	1.442	1.727
$A_{c2}/A_{e1}$ (dB)	-35.3431	-28.0003	-24.2059	-22.1565

337



338

339

340

**Figure 8** Experimental and numerical results of the feature  $A_{c2}/A_{e1}$  in dB against varied depths of the crack

341 From Figure 8, it can be noticed that  $A_{c2}/A_{e1}$  monotonically increases as the  $d$  increases for  
342 both simulation and experiment. In particular, the trace of  $A_{c2}/A_{e1}$  against  $d$  is likely linear  
343 when  $d \geq 0.75\text{mm}$  in simulation and for all three experiment data points. This comparison  
344 shows an agreement between simulation and experiment results that the feature  $A_{c2}/A_{e1}$  is an  
345 invariable and efficient feature for crack quantification.

346 It worth to notice that the experimental result for  $d = 0$  mm (crack-free) is larger than that  
347 of simulation result. When  $d = 0$  mm, the C2 does not exists because there is no reflection  
348 from the crack. The feature  $A_{c2}/A_{e1}$  is actually the ration between amplitude of noise and  $A_{e1}$ ,  
349 in another word,  $A_{c2}/A_{e1}$  in dB becomes  $-\text{SNR}_{e1}$  (the negative value of the signal-to-noise  
350 ratio for echo 1). The SNR for e1 in simulation is always larger than that of experimental  
351 value; therefore the absolute value of  $A_{c2}/A_{e1}$  of simulation is larger. Moreover, simulation  
352 does not take into account or much less than reality of the attenuation of the specimen,  
353 dissipation at crack, energy loss at the edge of specimen *etc.* That leads to the features  $A_{c2}$   
354  $/A_{e1}$  for cracks in experiment is smaller than the simulation results.

## 355 5. Conclusion

356 In this paper, EMAT has been developed for the monitoring of the stainless steel absorber tubes  
357 used in CSPs. The periodic permanent magnet (PPM) and race track coil are designed to generate  
358 shear horizontal SH0 mode for plate and torsional mode T(0,1) for pipe.

359 Through the numerical modelling and experiment, cracks with varied depths in a 3mm thick  
360 stainless steel specimen can be detected via observation additional echoes ( $c_1$ - $c_5$ ) compared the signal  
361 from crack-free specimen.

362 In order to quantify the cracks, three features: the amplitude of the first edge echo ( $A_{e1}$ ), the  
363 second crack echo ( $A_{c2}$ ) and the ratio between  $A_{c2}$  and  $A_{e1}$  ( $A_{c2}/A_{e1}$ ) have been investigated. Results  
364 showed that  $A_{c2}/A_{e1}$  monotonically increases as the  $d$  increases without any influence on the variation  
365 of lift-offs, where the features  $A_{c2}$  and  $A_{e1}$  suffer from strong influence on the variation of lift-offs.  
366 The results showed the feature  $A_{c2}/A_{e1}$  is an invariable and efficient feature for crack quantification.  
367 A validation has been conducted by experiments with three different depths of crack. The comparison  
368 of the feature  $A_{c2}/A_{e1}$  between numerical modelling and experiment showed an agreement on the  
369 monotonically increasing relationship of the feature and depth of the crack, in particular, the trace of  
370  $A_{c2}/A_{e1}$  against  $d$  is likely linear when  $d \geq 0.75\text{mm}$ .

371 In the future work, more depths of the crack will be manufactured and tested experimentally in  
372 order to make the look-up table for the crack quantification. In addition, studies of cracks in the  
373 absorber tubes under operation conditions will be carried out to compare the difference between plate  
374 and pipe specimens. During INTERSOLAR project, the EMATs were demonstrated in a test rig built  
375 in Spain. In the future, an array of EMATs will be validated on the absorber tubes in CSP plants in  
376 order to achieve full coverage of the tubes for defect detection, localisation and monitoring.

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