

1 **Experimental Study of Organic Rankine Cycle System and Expander**

2 **Performance for Heavy-Duty Diesel Engine**

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7 **Abstract**

8 A small scale organic Rankine cycle system capable of generating electric power
9 using exhaust gas of a 7.25 ℓ heavy duty diesel engine was built and tested. A
10 custom-designed radial inflow turbine was used as an expansion machine, and
11 NOVEC649 was used as the working fluid. In order to maintain steady state
12 operation, a thermal oil loop was installed in the system as an intermediate circuit
13 between the exhaust gas and organic Rankine cycle loop. Compared to the previous
14 study by the authors, the operating conditions were further extended. In addition,
15 the effects of cooling water temperature and working fluid superheating
16 temperature on turbine performance were explored in the current study. The
17 coupled engine-organic Rankine cycle system presented an electrical power,
18 turbine efficiency and thermal efficiency of 9 kW, 35% and 4%, respectively. The
19 results showed that both cooling water temperature and working fluid
20 superheating temperature had a negative impact on the radial turbine
21 performance (generated power and efficiency). The average decrement of the
22 generated power and turbine efficiency were 2.4% and 1.7%, respectively, when
23 increasing the cooling water temperature by 2°C, and 2.5% and 7.3% when
24 increasing the working fluid superheating temperature by 5°C. Moreover, the

25 extended tests were beneficiary for validating the proposed performance prediction
 26 meanline model developed by the authors in a previous study. The maximum
 27 deviation between the measured and predicted turbine efficiency was 3.5%.

28 Keywords: Waste heat recovery; heavy-duty diesel engine; organic Rankine cycle; experimental
 29 testing; cooling water temperature; superheating temperature

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Nomenclature			
Variables		abbreviation	
h	Enthalpy [kJ/kg]	BK	Blockage factor
m	Mass flow rate [kg/s]	BSFC	Break specific fuel consumption
N	Rotational speed [RPM]	CO ₂	Carbon dioxide
P	Pressure [kPa]	CTRC	CO ₂ -based transcritical Rankine cycle
T	Temperature [K]	EU	European Union
		EoS	Equation of state
	Subscript	FGT	Fixed geometry turbine
0	Stagnation property	GWP	Global warming potential
cr	critical	HDD	Heavy duty diesel engine
is	isentropic	HT-RC	high-temperature loop Rankine cycle
		ICE	Internal combustion engine
	Greek Symbols	LT-RC	Low-temperature loop Rankine cycle
η	Efficiency [-]	MFP	Mass flow parameter
		ODP	Ozone depletion potential
		ORC	Organic Rankine cycle

PDE	Positive displacement expander
PR	Pressure ratio
RC	Rankine cycle
WHR	Waste heat recovery

34

35 1. Introduction

36 Carbon dioxide (CO₂) emissions of the transportation sector have a
37 significant global impact on air quality and the environment. Therefore, EU
38 legislation sets mandatory emission reduction targets, as the fleet average to be
39 achieved by all heavy duty diesel (HDD) engines was 9% in 2017 compared to 2010
40 [1]. Among fuel-based applications, transportation burns most of the world's fuel,
41 accounting for more than 67% of the total fuel consumption in the United States
42 [2] and 73% in the United Kingdom in 2013 [3]. Therefore, manufacturers are
43 required to produce more efficient combustions engines. In this regard, waste heat
44 recovery (WHR) technology is one of the promising technologies in recovering the
45 wasted fuel energy

46 Recently, organic Rankine cycle (ORC) systems as waste heat recovery
47 (WHR) systems in internal combustion engines (ICEs) have received increasing
48 interest. This technology is most widely used in low- to medium-temperature heat
49 sources, typically between 80°C and 350°C, due to the low boiling point of organic
50 fluids compared to steam, which is widely used in large-scale applications.
51 However, further development is still required to implement such technology in
52 modern passenger cars because of the need for compact integration and
53 controllability in the engine [4]. Engine waste heat can be transferred directly
54 through the evaporator to the ORC loop, but in some studies, an intermediate

55 thermal oil loop between the exhaust gases and the ORC is used [5]. Direct heat
56 transfer from the exhaust gases to the organic fluid is often preferred in transport
57 applications as it increases the heat transfer efficiency and reduces the weight of
58 the WHR system, while the thermal oil loop requires an extra heat exchanger and
59 pump. However, cycles with an intermediate oil loop guarantees steady-state
60 conditions for the ORC operation, while any potential decomposition of the working
61 fluid at high exhaust enthalpy conditions can be avoided [5].

62 Many theoretical studies regarding integrated ORC systems in vehicle
63 powertrain present thermal efficiencies between 6% and 20%. This variation in
64 ORC thermal efficiency mainly depends on the heat sources used in engine
65 operating conditions. In 2012, Katsanos et al. [6] performed a thermodynamic
66 analysis of an ORC system applied on a six-cylinder heavy-duty two stage
67 turbocharged truck diesel engine. The results presented a 20% thermal efficiency
68 of the cycle. The following year, Shu et al. [7] compared three regenerative dual-
69 loop organic Rankine cycle systems are proposed to compare with the simple dual
70 loop organic Rankine cycle, using the wasted heat of the exhaust and engine
71 coolant of a diesel engine. A maximum net output power of 39.67 kW was obtained
72 with the simple dual loop organic Rankine cycle while the other loops presented
73 slightly lower performance due to higher system irreversibilities. Two years later
74 Song and Gu [8] obtained a thermal efficiency equals to 11.8% when recovering the
75 wasted heat in the exhaust gas of an inline six-cylinder turbocharged engine. The
76 following year, the authors [9] investigated the effects of variable geometry turbine
77 performance on the performance on organic Rankine cycle-internal combustion
78 engine (ORC-ICE) system. At the same engine operating point, the cycle with
79 variable geometry turbine presented a 13.9% thermal efficiency compared to 10.5%

80 when using a fixed geometry turbine. In 2018, several studies were published in
81 the open literature. Yang et al. [10] developed an ORC model to harvest the wasted
82 heat in the exhaust gas of a heavy-duty diesel engine, and presented a thermal
83 efficiency of 6.6%. Rashwan et al. [11] presented a thermodynamic analysis of an
84 organic Rankine cycle integrated with a cascaded closed loop cycle. According to
85 the authors, the cascaded closed loop cycle is considered one of the advanced heat
86 recovery technologies that enhances thermal efficiency significantly. The results
87 showed that thermal efficiency of the cascaded closed loop cycle was 21%, while it
88 was 11% with simple ORC system. Mashadi et al. [12] investigated the feasibility
89 of ORC systems to recover the wasted heat in a 4 cylinder gasoline engine coolant.
90 A thermal efficiency of 18.45% was obtained. More recently, Li et al. [13]
91 investigated the effects of turbine efficiency and working fluid type on the
92 performance of the system. The results showed a maximum thermal efficiency of
93 12.5% with R236ea as the working fluid. Therefore, as mentioned earlier, thermal
94 efficiencies of ORC systems could not exceed 20% in ICEs. The following paragraph
95 presents a summary of the experimental work of ORC-ICE systems.

96 Several experimental studies investigating the feasibility of ORC systems
97 as WHR systems in ICEs have been published. In 2007, Honda [14] installed an
98 ORC system on a hybrid vehicle with the vehicle running at constant speed. The
99 results presented a 13.2% improvement in the thermal efficiency compared to the base
100 vehicle. Five years later, Zhang et al. [15] installed a Rankine cycle system on a marine
101 2-stroke diesel engine and claimed that a 10% efficiency improvement was achieved.
102 In the same year, Hossain and Bari [16] conducted an experiment to measure the
103 available exhaust heat from a 40 kW diesel generator using Rankine cycle. At 40%
104 part load, the additional power developed was 3.4% which resulted in 3.3%

105 reduction in BSFC. In 2014, Zhang et al. [17] built an experimental system to
106 recover wasted heat in the exhaust gas of a 336 horsepower diesel engine. A single-
107 screw expander and R123 were selected as the expansion machine and working
108 fluid respectively. The results indicated that the maximum power output, ORC
109 efficiency and overall system efficiency were respectively 10.38kW, 6.48% and
110 43.8%. In the same year, Furukawa et al. [18] conducted an experimental test on the
111 ORC on order to recover the wasted heat in the engine coolant. The fuel consumption
112 decreased by 7.5%. A year later, Galindo et al. [19] tested an ORC system integrated
113 in a 2 liter turbocharged gasoline engine using ethanol as working fluid and swash-
114 plate expander as the expansion machine. A maximum real Rankine efficiency
115 value of 6% was obtained. In 2016, three experimental studies were published. Yu
116 et al. [20] constructed a cascaded system that comprises a steam Rankine cycle
117 (RC) as the high-temperature loop (HT-RC) and an organic Rankine cycle as the
118 low-temperature loop (LT-ORC) for waste heat recovery from an in-line, six
119 cylinders diesel engine. Comparing to the basic diesel engine, the power increment
120 reaches up to 5.6% by equipping the cascaded system. Guillaume et al. [21] used
121 exhaust gases of a truck diesel engine as the heat source for their ORC system.
122 They used a radial inflow turbine as the expansion machine and two working
123 fluids: R245fa and R1233zd. However, the employed turbine was developed mainly
124 using components modified from truck turbocharger designs. Also, the heat wasted
125 by the truck through the exhaust gases is simulated using an electric oil boiler
126 coupled to the ORC loop. The maximum electric power and turbine efficiency were
127 2.8 kW (using R245fa) and 32% (using R1233zd), respectively. AVL, FPT and Iveco
128 [22] built an ORC system to harvest the wasted heat in the exhaust gas of a 4-stroke
129 diesel engine. The tests were run on public roads and the results showed that the fuel

130 consumption could be reduced by 2.5–3.4%. The following year, two experimental
131 studies on ORC-ICE systems were published. Sellers [23] evaluated the benefits of
132 ORC systems in recovering the wasted heat in the jacket water of a 12 cylinder ship
133 engine. The results showed that the largest kilowatt hour value of 78,001 was
134 produced during the first voyage from Asia to the USA east coast. Shi et al. [24]
135 constructed four CO₂-based transcritical Rankine cycle (CTRC) systems (basic ORC,
136 ORC with regenerator, ORC with coolant preheater and ORC with both the preheater
137 and the regenerator) with kW-scale power output to recover waste heat from both
138 exhaust gas and coolant water of an in-line, six cylinders diesel engine. However, the
139 authors applied expansion valve instead of turbine. Among the four configurations,
140 the ORC with both the preheater and the regenerator showed the highest net power
141 output and thermal efficiency, whose estimation reach up to 3.47 kW and 7.8%,
142 respectively. In 2018, the authors [25] have built an ORC system to recover the wasted
143 heat in a a 7.25ℓ Yuchai engine at highly off-design conditions. An intermediate
144 thermal oil loop has been placed between the heavy duty diesel engine the ORC system
145 in order to ensure steady operation while keeping the fluid temperature below the
146 decomposition one. The maximum obtained thermal efficiency has been 4%, and an
147 electrical power of 6 kW has been generated. More recently, Linnemann et al. [26]
148 tested an ORC system, toluene as a working fluid, driven by biogas waste heat with
149 focusing on the design and testing of multi-coil helical evaporator performance.
150 According to the authors, the turbine was not operational thus, the working fluid was
151 carried through a bypass and expanded with an orifice plate, before entering the
152 recuperator. that the predicted values of the overall heat transfer coefficient and the
153 shell side Nusselt number are in good agreement with experimental data, showing a
154 maximum deviation of 5.5%. The brief literature survey indicates that ORC system is

155 a promising WHR technology An extensive review of automotive ORC systems can
156 be found in a previous work by the authors [27].

157 Radial turbine is a component within a larger system (i.e., ORC). Therefore,
158 cycle analysis should be considered during the turbine design stage. Moreover, in
159 heat sources such as ICEs, the thermodynamic parameters of the exhaust gas, such
160 as mass flow rate and temperature, can vary widely with time. This variance
161 causes heat sources to become unstable and uncontrollable. Therefore, the
162 performance behaviour of a turbine when the machine runs at off-design rotational
163 speeds, mass flow rates, and pressure ratios should be accurately predicted.
164 However, there is a scarcity of information in open literature regarding the mean-
165 line modelling to obtain the off-design performance of ORC turbines, a point
166 confirmed in White [28] and Wong [29]. Several air turbine models were developed
167 in literature such as [30-35]. However, no consideration of real fluid properties
168 were taken into account in the aforementioned studies, rather, ideal gas
169 correlations were applied. The thermodynamic properties of high-density fluids,
170 such as organic fluids, are different from those of ideal gases. For instance, organic
171 fluids have high molecular weight, low boiling points and low speed of sound. In
172 addition, ORC radial turbines present high expansion ratio and Mach number at
173 the stator exit due to the frequently changing specific volume, which results in
174 supersonic flows. This outcome necessitates the use of a real EoS rather than the
175 traditionally adopted Mach relationships for ideal gas. The thermodynamic
176 properties at each station must likewise be checked simultaneously through the
177 turbine stage. For this investigation, a novel performance prediction method for
178 ORC radial-inflow turbine was developed by the authors in a previous work [36].

179 The brief literature review in the 3rd and 4th paragraphs, and the literature
180 review study by the authors [27] show that only Guillaume et al. [21] and the
181 previous study of the authors [25] examined the feasibility of ORC systems in
182 automotive applications with radial turbines as expansion machines. Even though,
183 the turbine in Guillaume et al. [21] was developed using components of truck
184 turbochargers. Also, they applied an electric oil boiler as the heat source rather
185 than real engine exhaust gas. The coupling of ORC systems with real heavy duty
186 diesel engine an area in which little available literature exists. Moreover,
187 according to the literature review study [27] and the brief literature review in the
188 3rd and 4th paragraphs, effects of cooling water temperature and working fluid
189 superheating temperature on turbine performance have not yet been discussed,
190 neither theoretically nor experimentally. Therefore, further testing is performed
191 in the current study in order to study the effects of cooling water temperature on
192 the radial turbine performance. In addition, one of the objectives of the current
193 study is to accurately validate the proposed meanline model in Alshammari et
194 al.[36]. The latter is very essential since ORC radial turbines experience choking
195 conditions due to high pressure ratio and low speed of sound of organic fluids.
196 Besides validating the meanline model, the present study highlights the effects of
197 input parameters on the turbine off-design performance and mass flow rate.

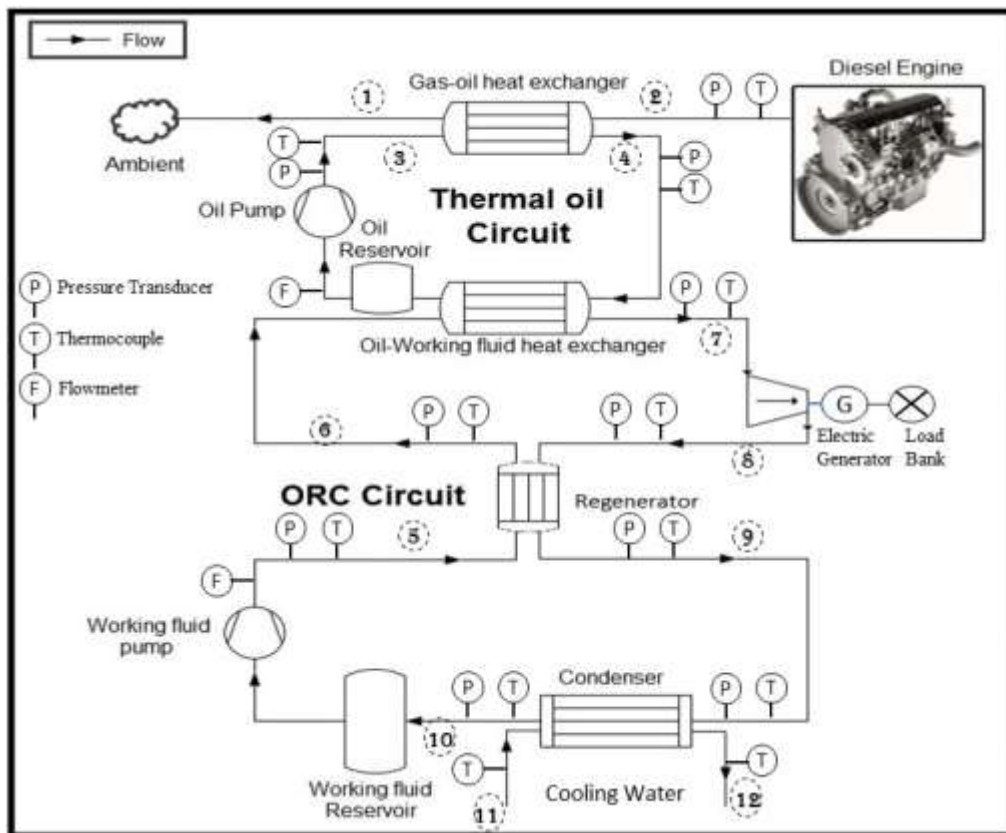
198 **2. Test Rig**

199 An experimental facility of an Organic Rankine Cycle coupled to an Internal
200 Combustion Engine is tested in order to investigate the feasibility of ORC as WHR
201 system in ICEs. The results of tests are also used to validate the novel meanline
202 model presented by the authors in [36]. The WHR system is presented in Fig. 1. A

203 photograph of the ORC skid with main components identified is presented in Fig.
204 2.

205 The test rig consists of the heavy duty diesel engine, thermal oil loop, ORC
206 loop and cooling loop. The diesel engine and ORC loop are briefly presented in this
207 study, as they are detailed in Alshammari et al. [25].

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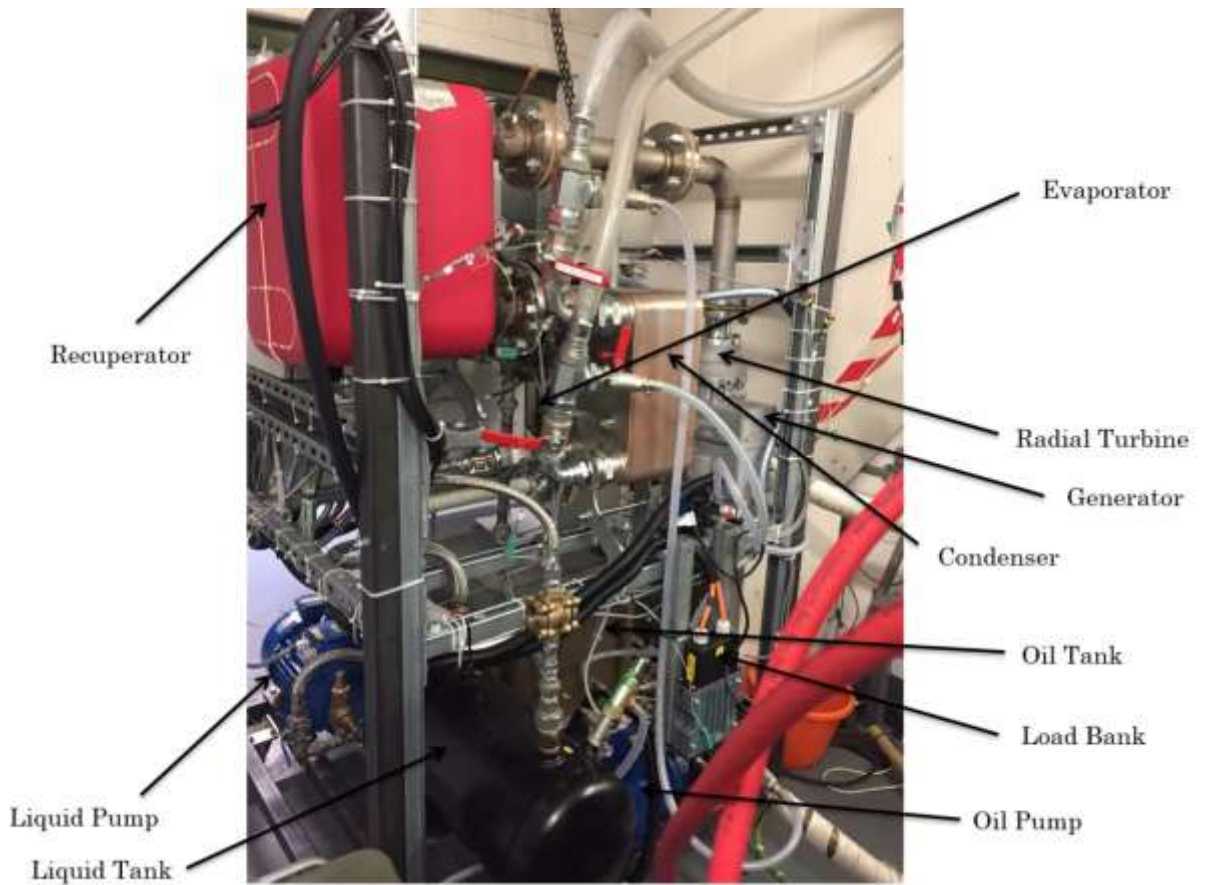


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Fig. 1: Experimental ORC installation diagram

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Fig. 2: Photograph of experimental ORC installation

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215 2.1 Fluid Selection

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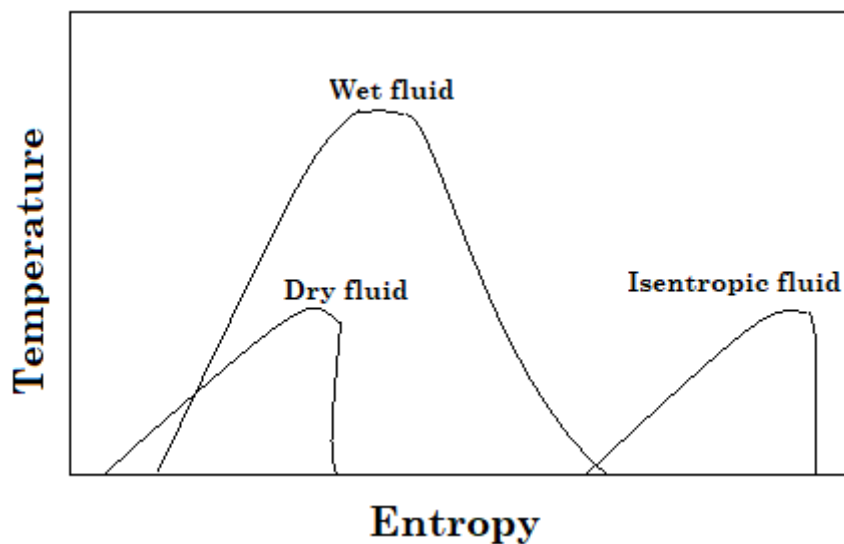
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The choice of working fluid for an ORC system is of great importance for the cycle efficiency and net work. ORC systems should only utilise working fluids with low global warming potential (GWP) and ozone depletion potential (ODP) [39]. Compared to steam (in conventional Rankine cycle), organic fluids exhibit unique advantages because they are better adapted to low heat source temperatures, enabling ORC systems to efficiently produce shaft work from low to medium temperature heat sources of up to 370 °C [40]. Compared to conventional Rankine cycles, a smaller plant size will be produced when organic fluids are used because of their high density. The higher the density is, the lower the volumetric flow rate

225 is and, subsequently, the smaller the component size becomes. The selection of
226 working fluid is determined by the application and the waste heat level [37]. Based
227 on the slope of the saturation vapour line, as shown in Fig. 3, working fluids can
228 be classified into three groups: wet, dry and isentropic. Dry and isentropic fluids
229 have enormous advantages for turbo-machinery expanders because they leave the
230 expander as superheated vapour and eliminate the corrosion that results from
231 liquid droplets that impinge on the turbine blades during the expansion [38].
232 Another advantage is that overheating the vapour before entering the expander is
233 not required, which means a small and cheap heat exchanger can be used.
234 Moreover, a superheated apparatus is not required when using dry and isentropic
235 fluids [39]. However, if the fluid is too dry, the expanded vapour will leave the
236 turbine with substantial superheat, which is a waste and adds to the cooling load
237 in the condenser.



238

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Fig. 3: Types of working fluids

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241 In order to come up with the optimum fluid for the current application,
 242 the authors [40] proposed novel method for selecting the proper working fluid
 243 for the current application considering thermodynamic properties, radial
 244 turbine speed and evaporator heat transfer surface. Using the aforementioned
 245 method, the authors further investigated the potential of NOVEC 649 as a
 246 working fluid for the current application [41]. The results showed that
 247 NOVEC649 produced lower back pressure at the evaporator exit and lower
 248 turbine rotational speed, which positively affects the electric generator cost. In
 249 addition, NOVEC649 is an effective heat transfer fluid that can be utilized in
 250 applications such as ORC where non-flammability or environmental factors are
 251 a consideration [42]. The thermos-physical properties of NOVEC649 are
 252 presented in Table 1.

253 Table 1: Properties of NOVEC649

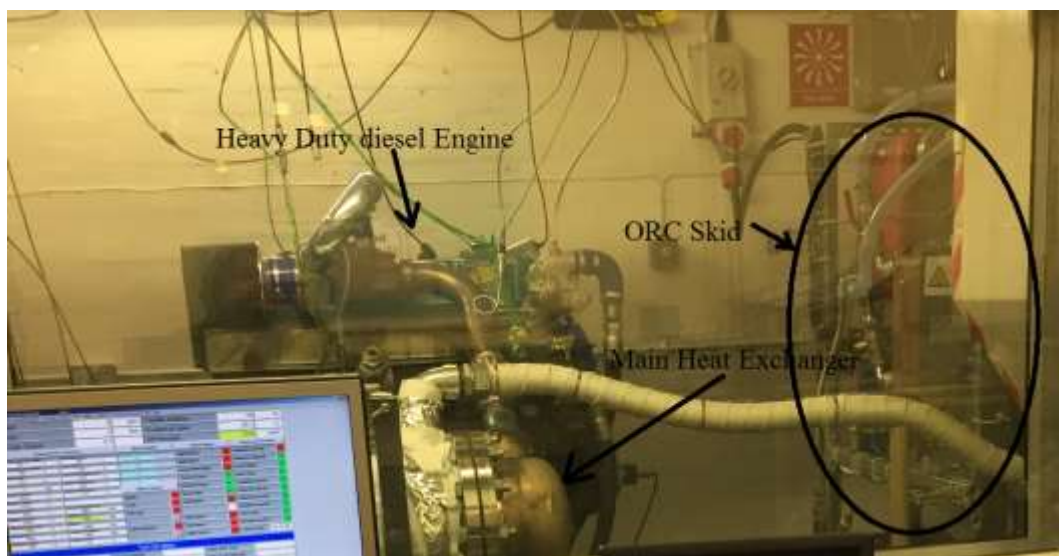
Fluid	Chemical Formula	Molecular Weight (g/mol)	T_{cr} (K)	P_{cr} (bar)	Boiling Point (K)	Molar Mass (g/kmol)	GWP	ODP
Novec649	<chem>CF3CF2C(O)CF(CF3)2</chem>	316	441.81	18.69	322.2	316.04	1	0

254

255 2.2 The heavy duty engine (Heat source)

256 A photograph of the heavy duty diesel engine applied in the current study
 257 is shown in Fig. 4. The diesel engine used in tests is a 7.25ℓ turbocharged, direct
 258 injection Yuchai engine with a 17.5:1 compression ratio. The maximum engine
 259 torque is 1100 Nm at 1400-1600 rpm and 100% load. The maximum engine power
 260 is 206 kW at 2300 rpm and 100% load. However, the engine was not able to run at

261 full load because of the technical issues of the dynamometer. The maximum
262 obtained power output during the tests was nearly 40% of the maximum engine
263 power. This technical issue can be considered as a positive point and more practical
264 since the ORC system rarely operates at the design conditions in automotive
265 applications. In this study, the engine exhaust gas is used as the heat source since
266 it contains the largest portion of wasted heat, which is approximately 20%–42% of
267 the total wasted heat [43], and high exergetic content [44].



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Fig. 4: Photograph of the 7.2 l heavy duty diesel engine (heat source)

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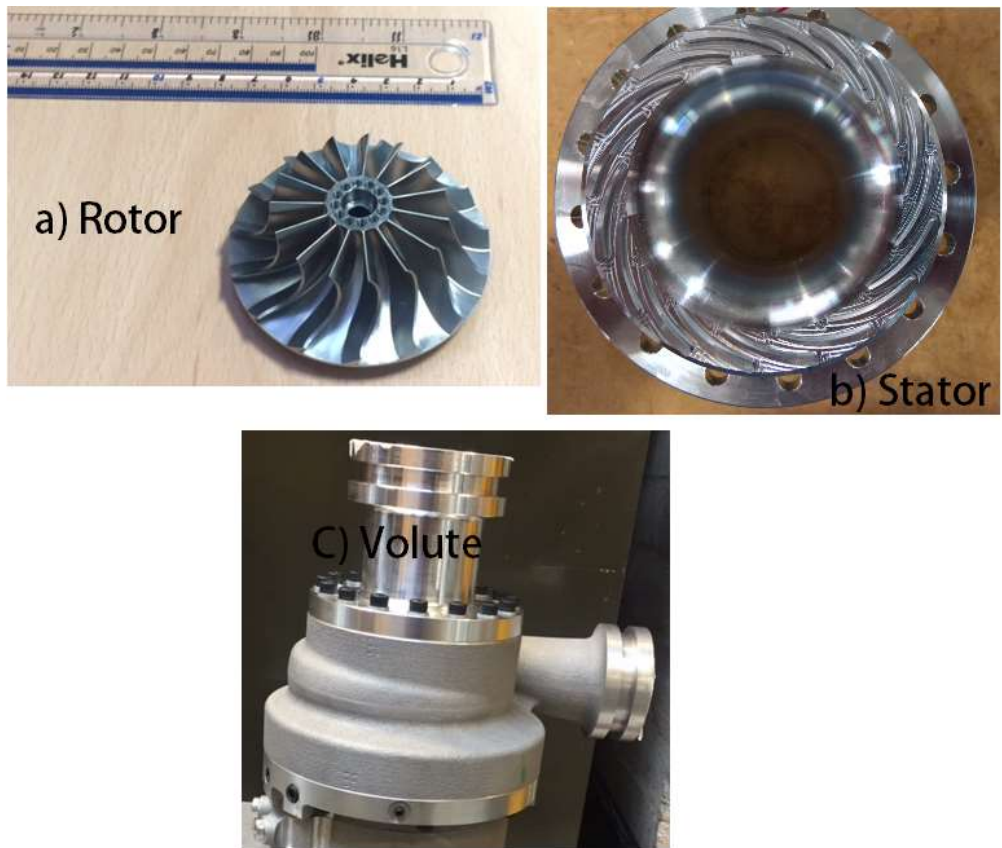
271 2.3 Thermal oil loop

272 The intermediate thermal oil loop is placed between the exhaust gas of the
273 engine and the ORC system via the main heat exchanger as shown in Fig. 4. The
274 thermal oil loop requires two more components i.e. heat exchanger and pump
275 which results in heavier system. However, the thermal oil loop assures steady-state
276 conditions for the ORC operation, and is beneficial in order to avoid any potential
277 decomposition of the working fluid at high exhaust enthalpy operations. In addition,

278 the combination of ORC-thermal oil assures stabilizing the thermal oil temperature
279 in the evaporator. The thermal oil is a synthetic organic heat transfer fluid contains
280 a mixture of diphenylethane and alkylated aromatics. It exhibits better thermal
281 stability, particularly at the upper end of hot oil's use range, and significantly
282 better low-temperature pumpability. It's critical temperature and pressure are
283 $489^{\circ}C$ and $24\ bar$, respectively.

284 **2.4 Organic Rankine cycle loop**

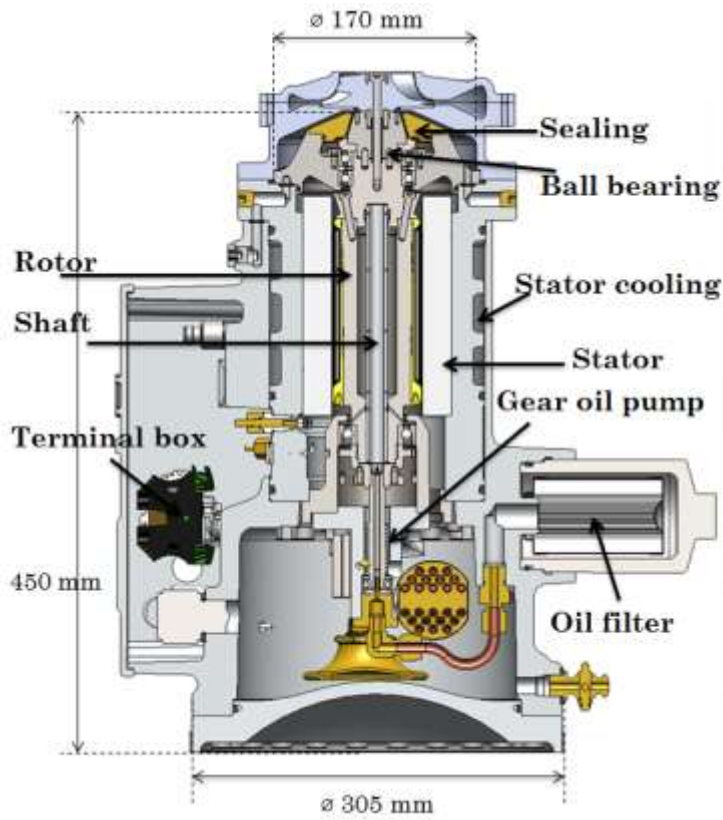
285 Fig. 2 presents a photograph for the experimental set up of the ORC system
286 used in this study. In the main heat exchanger (Fig. 4), heat transfers from the
287 exhaust gas of the diesel engine to thermal oil. Then, the hot thermal oil passes
288 through the evaporator to exchange heat with working fluid. After that, the vapour
289 working fluid flows into the radial turbine where the enthalpy is converted to
290 effective work. Then, the fluid enters the recuperator to use the still heat in the
291 working fluid. In order for the fluid to be transformed back to the liquid phase, it
292 flows into the condenser where the cooling circuit starts. The liquid NOVEC649 is
293 then pumped to the evaporator and the cycle starts again. Fig. 5 shows the radial
294 turbine parts i.e. volute, stator and rotor. Fig. 6 presents the coupled turbo-
295 generator unit. More information about the specifications of the ORC components
296 and the instrumentations can be found in Alshammari et al. [25].



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298

Fig. 5: Manufactured radial inflow turbine



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300

Fig. 6: Electrical generator (coupled to turbine)

301 **2.5 Cooling loop**

302 The cooling loop is a water circuit linked to the condenser in order to remove
303 the heat from condenser to environment, where the state of fluid changes from
304 vapour to liquid. The cooling water inlet temperature changes based on the outside
305 temperature. The tests were run from mid of June to end of July at Brunel
306 University London. The average ambient temperature ranges from $18^{\circ}C$ to $26^{\circ}C$.
307 The condenser unit is a counter current flow, brazed plate heat exchanger. The
308 counter current configuration in the condenser is beneficial to ensure that
309 saturated liquid leaves the condenser, thereby, allowing the working fluid pump
310 to operate more efficiently.

311

312 **3. Mean-line Modelling**

313 Although the ORC system is a promising WHR technology, its cycle
314 efficiency is low due to the low working temperatures. Therefore, designing an
315 efficient turbine and predicting its performance are of great importance to avoid
316 further efficiency reductions. In addition, the thermodynamic properties of heat
317 sources are variable, thus making the prediction of turbine performance at
318 different operating conditions even more important at the design phase.

319 Supersonic flow is likely to take place in ORC turbines operating due to the
320 high operating pressure ratio and low speed of sound of the organic fluid.
321 Therefore, the corrected mass flow parameter (MFP) is applied instead of the real
322 mass flow rate. In this case, the mass flow rate relies only on the corrected mass

323 flow rate and the operating conditions regardless of the turbine speed and enthalpy
324 drop, as shown in equation (1). This is because, for Mach numbers greater than
325 unity, the mass flow rate remains constant for any pressure ratio equal or greater
326 than the choked value. So, when choking takes place at the blade row, the choked
327 mass flow is kept fixed. Then, the velocity triangle is solved using the choking mass
328 flowrate value without the described iteration process presented in Fig. 7.

$$MFP = \frac{MFR\sqrt{T_{01}}}{P_{01}} \quad (1)$$

329 The performance of the turbine is measured using bulk properties for the
330 total-to-total isentropic efficiency definition as shown in equation (2), where “in”
331 and “ex” indicates turbine inlet and exit, respectively. The detailed mathematical
332 model can be found in Alshammari et al. [36].

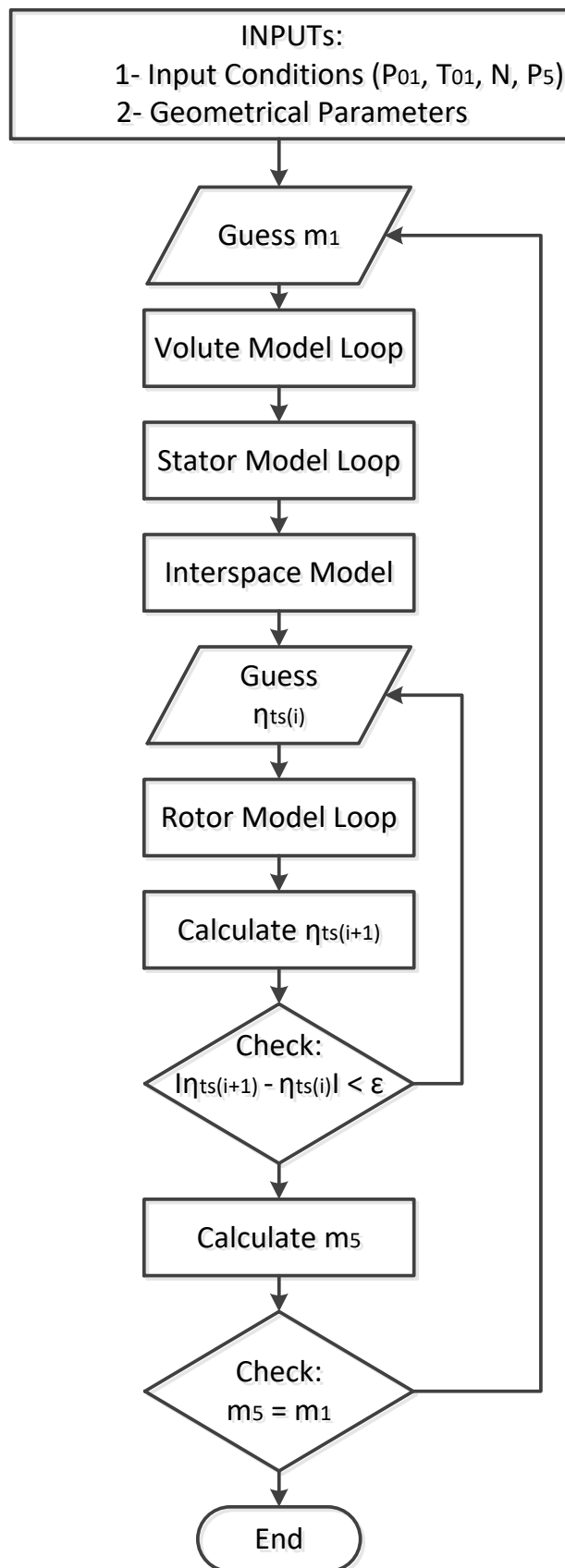
$$Turbine\ Efficiency = \frac{h_{01} - h_{05}}{h_{01} - h_{05,is}} \quad (2)$$

333

334 Table 2 presents input parameters and performance of the custom-designed radial
335 inflow turbine at the design point.

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337



338

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Fig. 7: Flowchart of the performance prediction meanline model [36]

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Table 2: Turbine parameters at design point

parameter	Value	Unit
Turbine inlet total pressure	9	bar
Turbine inlet total temperature	471.5	K
Turbine exit static pressure	1.30	bar
Turbine speed	40,000	rpm
Working fluid mass flow rate	0.8	Kg/s
Turbine efficiency	74.4	%
Turbine power output	13.6	kW

342

343 4. Results and Discussion

344 The first part of this section covers the results of the ORC testing, and the second
345 part covers the validation of the performance prediction meanline model.

346 4.1 Experimental results of the organic Rankine cycle system

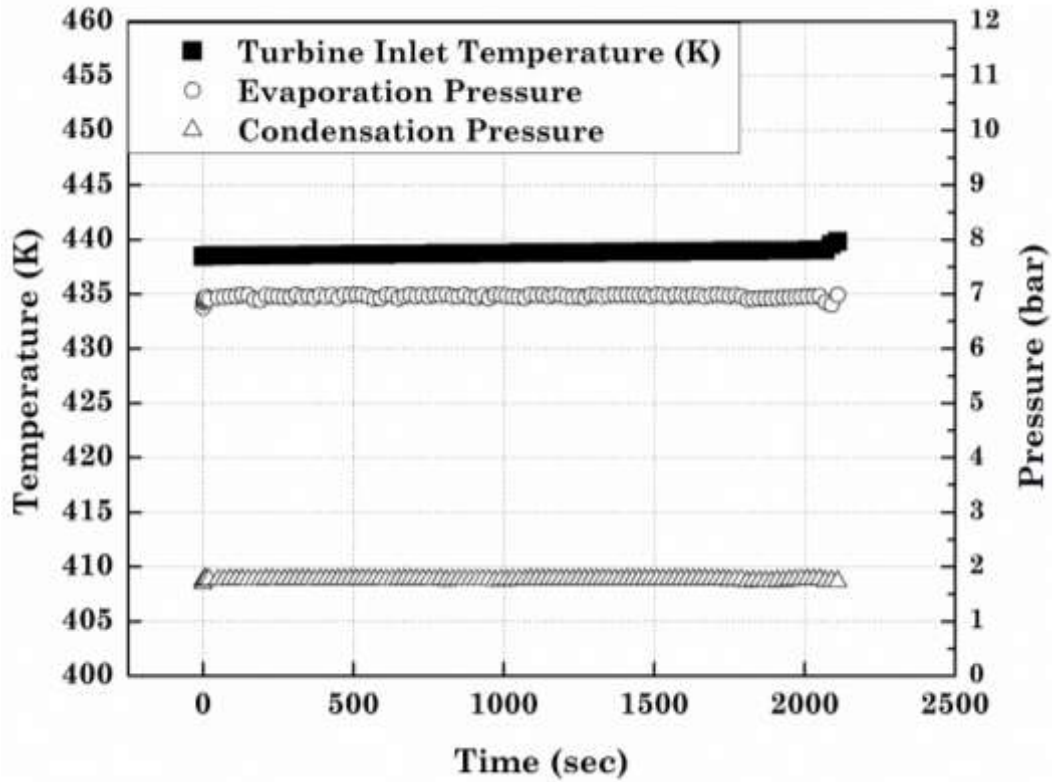
347 It is worth mentioning that the recording of the test data was initiated once
348 thermal equilibrium was achieved. Therefore, the time (x-axis) shown in the
349 figures in this section is the time after recording and not the time from the start of
350 the test.

351 4.1.1 Steady state operation

352 Fig. 8 assures the steady-state operation of the system through entire period
353 of testing. The figure presents the temperature at the turbine inlet, evaporation
354 pressure, and condensation pressure. The maximum variation of the temperature
355 is 0.65% for about 37 minutes of testing which is negligible. The pressure, on the

356 other hand, is oscillating but with very small variation. The maximum variations
357 of evaporation and condensation pressures are 2.5% and 1.6%, respectively.

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Fig. 8: Steady state conditions during testing

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4.1.2 Generated power and pump power

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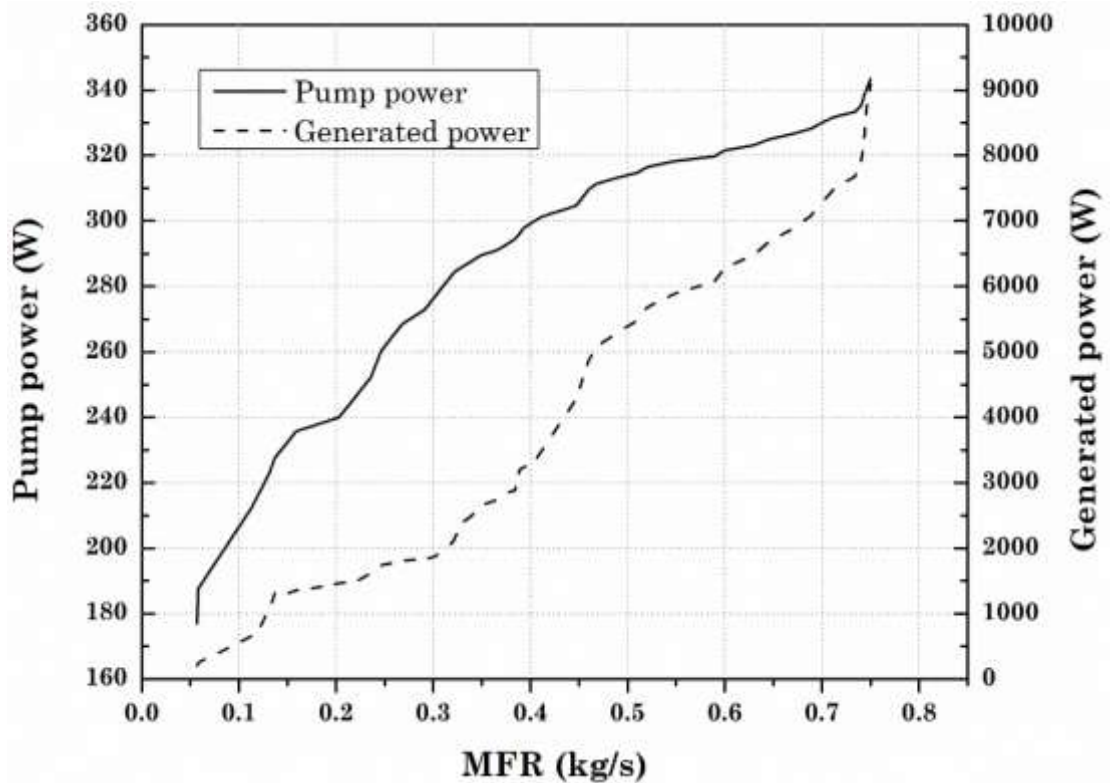
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Fig. 9 shows the variations of pump power and generated power by the generator with mass flow rates. The working fluid pump is connected with an inverter to adjust the pump rotational speed by converting the frequency, which controls the flow rate of the working fluid through the cycle. Obviously, the figure shows an increasing trend for both pump power consumption and generated power with increasing mass flow rate of the NOVEC649. The pump power increases from 177 W to 343.5 W as the mass flow rate of the working fluid increases from 0.057 kg/s to 0.75 kg/s. The generated power increased from 200 W to 9100 kW. Further

370 increase in mass flow rate (beyond 0.75 kg/s), results in a gradual decrease in the
371 generated power due to the increased pump power while the generated power
372 remains constant.

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375 Fig. 9: Generator power and pump power with mass flow rate

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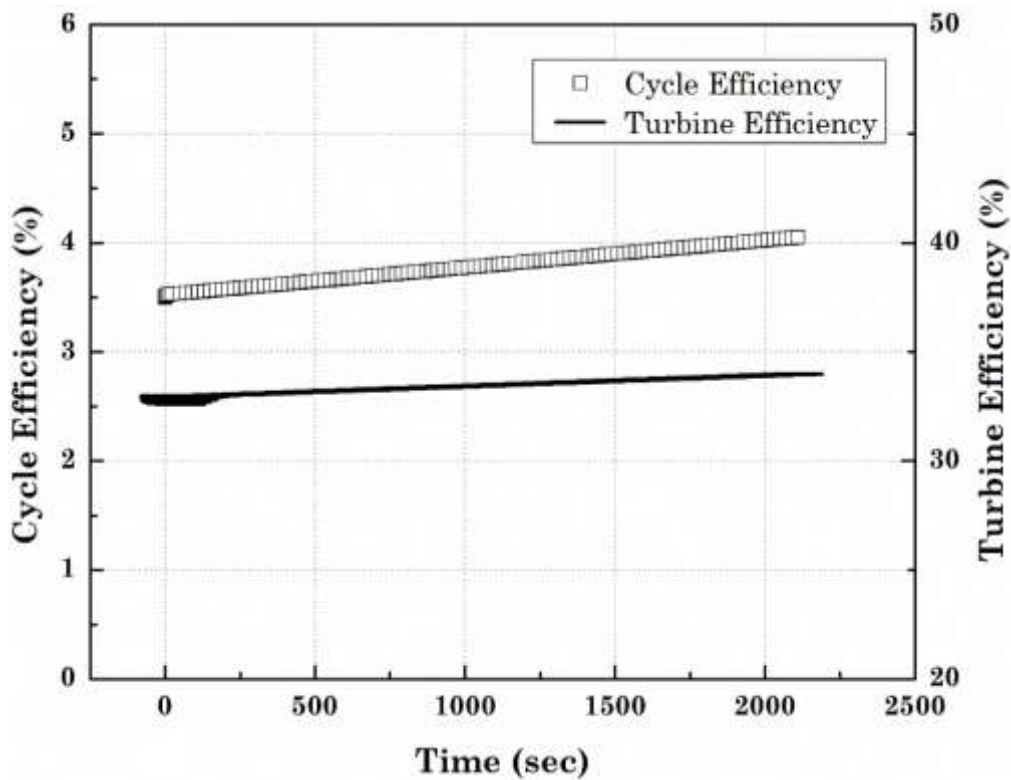
377 4.1.3 Cycle and turbine efficiencies

378 The cycle and turbine efficiencies with time are depicted in Fig. 10. The
379 average cycle thermal efficiency is 4% ($\pm 0.5\%$), while the turbine efficiency
380 presents larger variations with an average value of 34% ($\pm 1.38\%$). The cycle and
381 turbine efficiencies are recorded with pressure ratio equals to 4.85. The recorded
382 efficiencies are way below the designed values (Table 2). However, this is expected
383 since the system runs at substantially off-design conditions (due to the technical

384 issue of the engine dynamometer as mentioned in section 2.2), at which the peak
385 efficiency of the radial turbine was 34% at 20,000 rpm (instead of 74.4% at 40,000
386 rpm at the design point). Therefore, the maximum thermal efficiency of the cycle
387 was 4.3% instead of 9.3% at the design point.

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391 Fig. 10: cycle and turbine efficiencies during entire period of testing

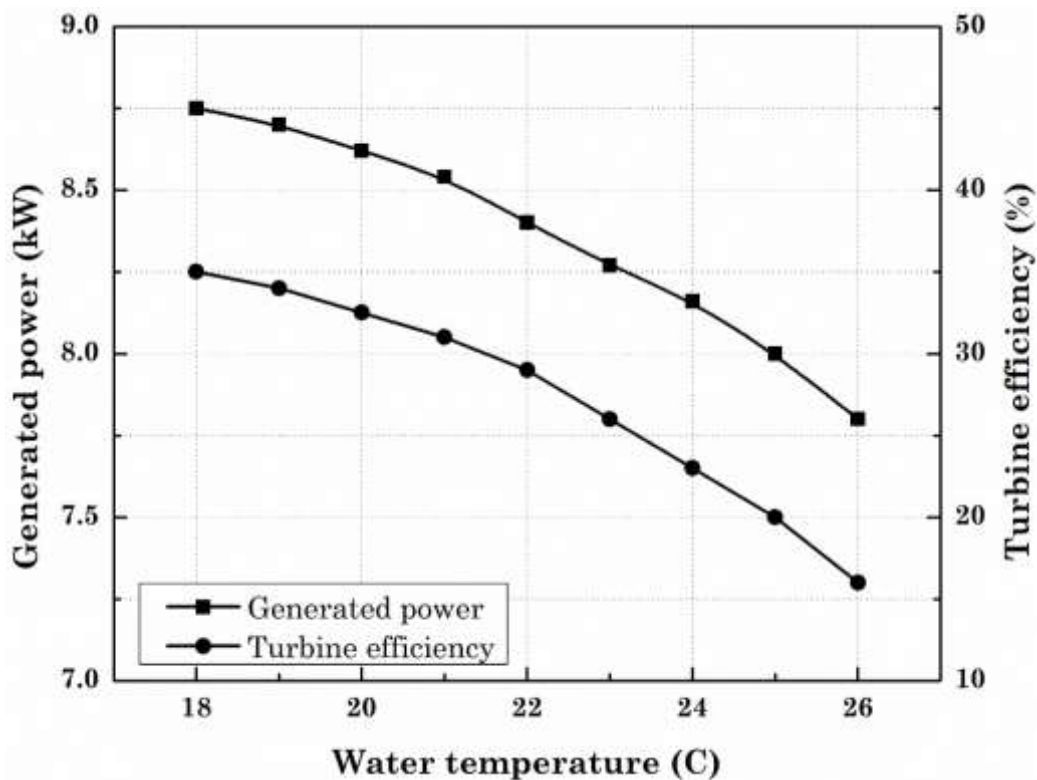
392

393 4.1.4 Effects of cooling water temperature

394 The cooling water temperature depends on the outside temperature since
395 the cooling tower is roof-mounted. Therefore, the tests were run at various times
396 of the day from mid of June to end of July, where the temperature ranged from

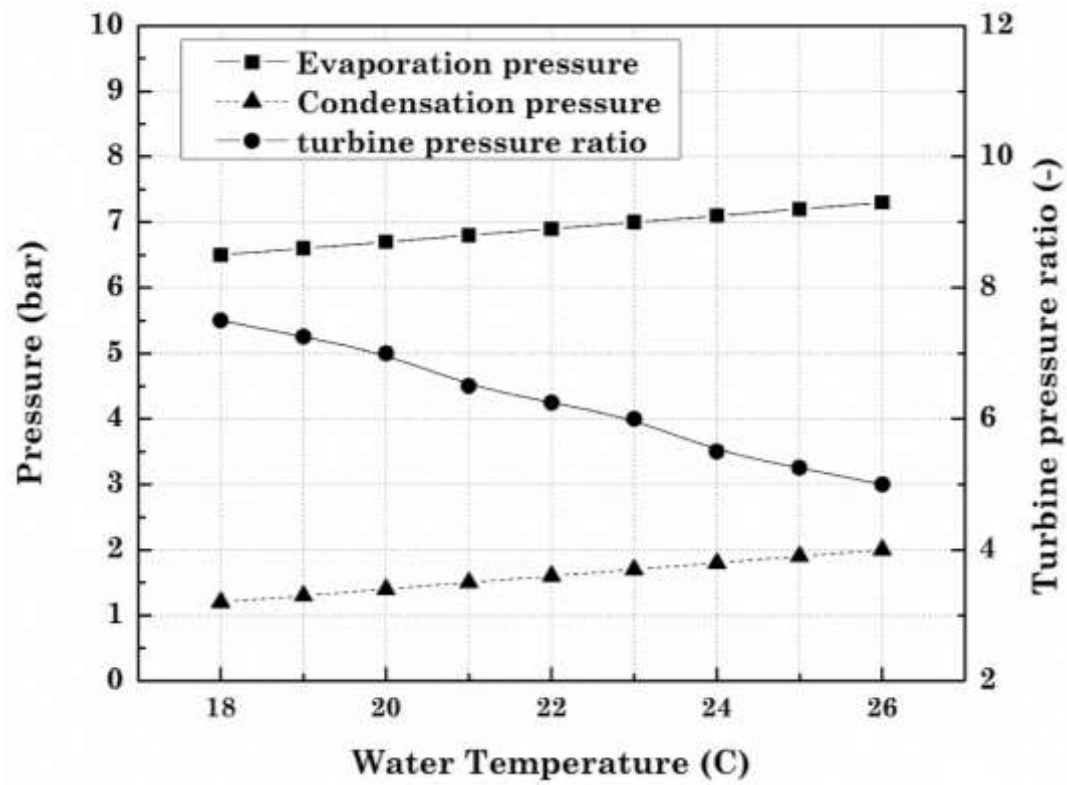
397 18°C to 26°. To ensure accurate results, the heavy-duty diesel engine, and hence
398 the thermal oil, was run at constant conditions i.e. constant temperature and mass
399 flow rate for all tests. In addition, the cooling water flow rate was kept constant.
400 The results are presented in Fig. 11 and Fig. 12.

401 Fig. 11 obviously shows that both generated power and turbine efficiency
402 decreases with increasing the cooling water temperature. The generated power
403 decreases due to the decreasing pressure ratio as shown in Fig. 12. The turbine
404 pressure ratio decreases since the condensing pressure increases (as the cooling
405 water temperature increases) which increases the turbine exit pressure. Although
406 the evaporation pressure also increases, the increase rate is lower than the
407 condensation pressure. The turbine efficiency decreases due to the increased
408 pressure at the turbine exit. Fig. 11 shows that increasing cooling water
409 temperature by 2° C resulted in average decrement of 2.4% in the generated
410 electrical power and 1.7% in the turbine efficiency.



412 Fig. 11: Generated power by generator and turbine efficiency with increasing cooling water
413 temperature (heat sink).

414



415

416 Fig. 12: Evaporation pressure, condensation pressure and turbine pressure ratio with
417 increasing cooling water temperature (heat sink).

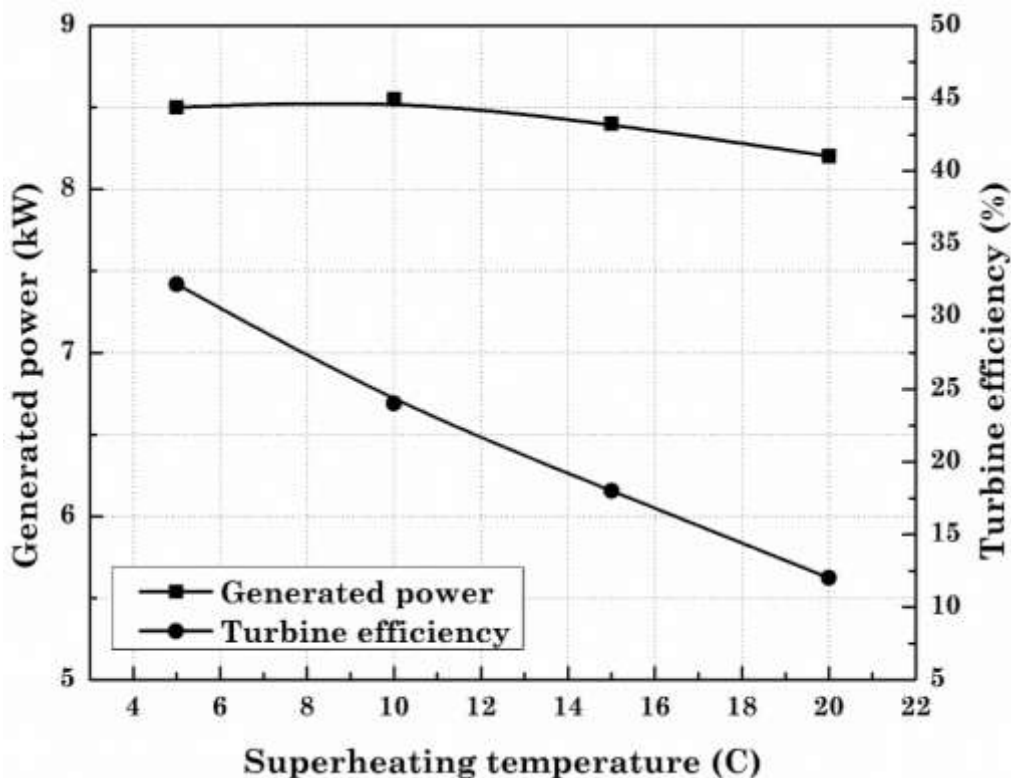
418 4.1.5 Effects of superheating temperature

419 The working fluid used in the testing (NOVEC649) is a dry fluid. Therefore,
420 more attention should be paid to the superheating temperature. In case of
421 excessive superheating, the expanded vapour would leave the turbine with a
422 substantial superheat, which is a waste and adds to the cooling load in the
423 condenser. The effects of the superheating temperature on the system is shown in
424 Fig. 13 and Fig. 14. To ensure accurate results, the working fluid mass flow rate,

425 and cooling water flow capacity and temperature were kept constant during all
426 tests.

427 In order to increase the superheating temperature, the evaporation pressure
428 should be increased. This can be obtained by decreasing the speed of working fluid
429 pump. Decreasing the evaporation pressure results in decreasing turbine pressure
430 ratio since superheating temperature increases as shown in Fig. 14. As a result of
431 the decreasing pressure ratio, the generated power and turbine efficiency also
432 decrease as shown in Fig. 13. It is worth mentioning that the condensation
433 pressure is nearly constant due to the constant temperature and flow capacity of
434 cooling water. Fig. 13 shows that increasing NOVEC 649 superheating
435 temperature by 5°C resulted in an average decrementt of 2.5% in the generated
436 electrical power and 7.3% in the turbine efficiency.

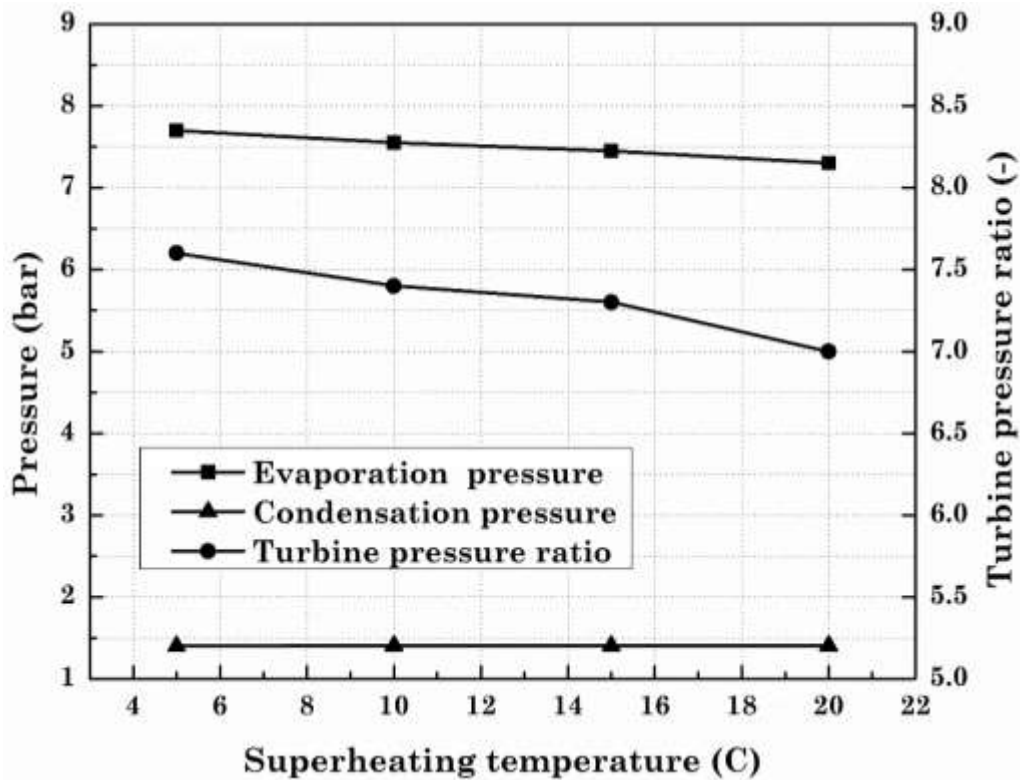
437



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439 Fig. 13: Generated power by generator and turbine efficiency with increasing superheating
440 temperature

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443 Fig. 14: Evaporation pressure, condensation pressure and turbine pressure ratio with
444 increasing superheating temperature

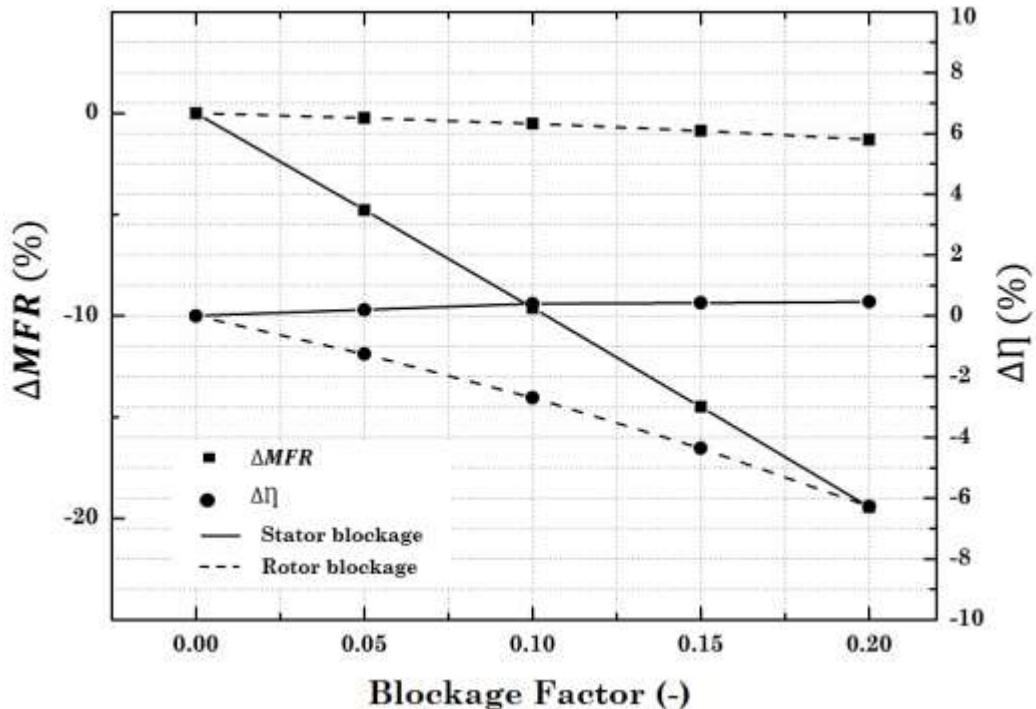
445 4.2 Parametric study of the meanline model

446 It is of great importance to carry out a parametric study of the meanline
447 model prior to validation in order to evaluate the effects of the empirical input
448 parameters that are controlled by the user. These input parameters are stator and
449 rotor blockage factors, stator and rotor deviation angles, and rotor incidence angle.

450 Fig. 15 presents the effect of the blockage factors of the stator and rotor on
451 the turbine mass flow rate and efficiency. Clearly, increasing the stator blockage
452 has critical impact on the MFR since the flow area decreases by increasing the

453 stator blockage factor. Increasing the blockage factor from 0 to 0.2 results in 20%
 454 reduction in the mass flow rate (MFR). On the other hand, turbine efficiency is
 455 insignificantly affected by increasing the stator blockage factor with a maximum
 456 reduction of 0.46% when the blockage factor increases from 0 to 0.2. This slight
 457 effect is related to the change in the incidence angle which results in higher
 458 absolute velocity of the flow at the stator exit.

459 Rotor blockage factor has much less but non-negligible effect on the mass
 460 flow rate. Larger blockage factors results in narrower flow path, and hence, lower
 461 flow capacities. The efficiency, on the other hand, is relatively sensitive to the rotor
 462 blockage due to, beside the reduction in MFR, the increased kinetic energy loss at
 463 the rotor exit. As the blockage increases from 0 to 0.2, the MFR and efficiency
 464 decrease by 1.3% and 6.27%, respectively.



465

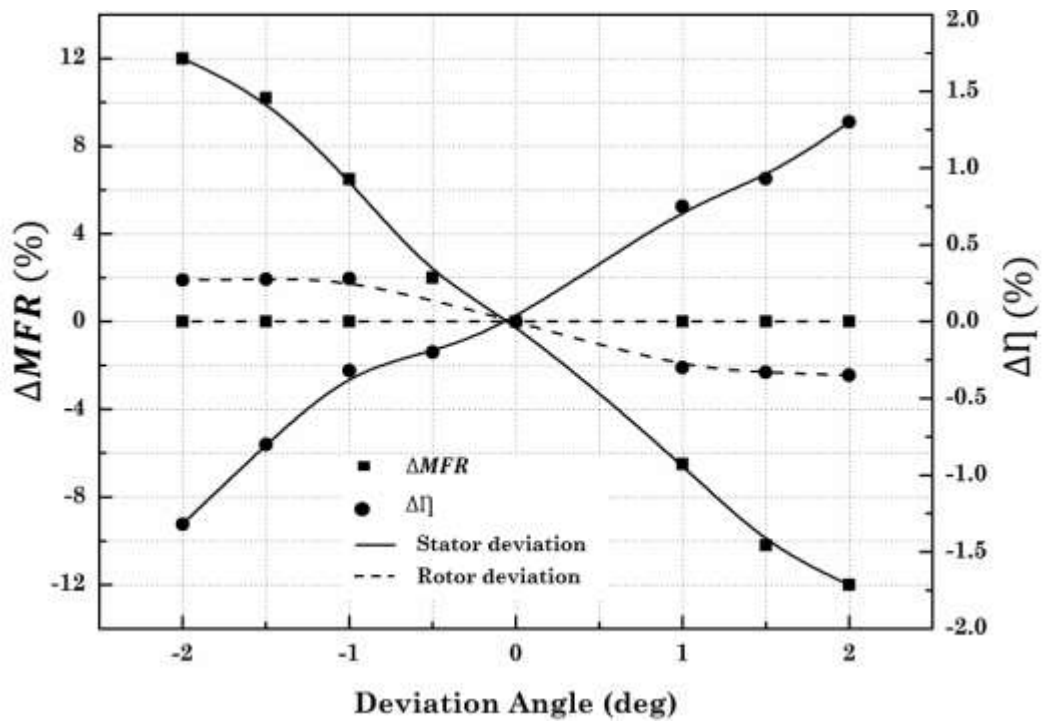
466

Fig. 15: Deviation of mass flow rate and turbine efficiency versus blockage factors

467 Fig. 16 presents the effect of the deviation angles on the turbine flow
468 capacity and efficiency. The figure obviously shows that the turbine mass flow rate
469 is very sensitive to the stator deviation angle. As the angle increases from 0 to 2,
470 the mass flow rate decreases by 12%. Changing the deviation angle in the negative
471 direction, from 0 to -2, the mass flow rate increases by 12%. The relationship
472 between the mass flow rate and the stator deviation angle is related to the
473 definition of the mass flow rate at the stator exit. The deviation angle is defined as
474 the difference between the vane angle (setting angle) and the flow angle. As the
475 flow angle moves towards the negative direction (the deviation moves towards the
476 positive direction), the mass flow rate significantly decreases, and vice versa. Fig.
477 16 also indicates that the stator deviation angle has moderate impact on the
478 turbine efficiency. As the deviation angles moves to the negative direction, the
479 Mach number increases at stator exit, as stated by Benson [45] and Pullen et
480 al.[46], which results in lower turbine efficiency.

481 The turbine flow rate and efficiency are also investigated by changing the rotor
482 deviation angle by $\pm 2^\circ$. It is obvious that the rotor deviation angle has no impact
483 on the mass flow rate while its impact on the turbine efficiency is small but not
484 negligible. The efficiency decreases with positive rotor deviation angles and
485 increases with positive rotor deviation angles. This is explained by the increase
486 and decrease of the swirl at the rotor exit. As the deviation angle moves to the
487 positive direction, the swirl increases which results in slightly lower efficiencies.

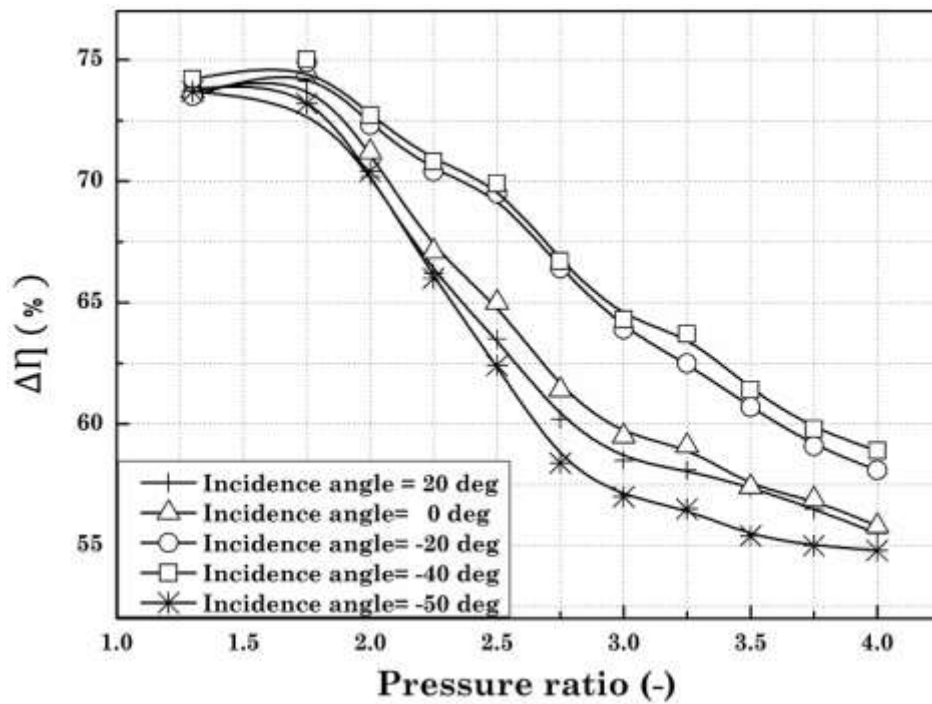
488



489

490 Fig. 16: Deviation of mass flow rate and turbine efficiency versus deviation angles

491 Last but not least, the rotor incidence angle is explored and the results
 492 are plotted in Fig. 17. According to the measurement of Woolley and Hatton [47],
 493 the flow becomes more uniform as the incidence angle moves from 0° to negative
 494 values down to -40° . Beyond -40° , the flow separation appears again on the
 495 pressure side. Baines [48] concurred with Wooley and Hatton on the optimum
 496 incidence angle. He also stated that zero or positive incidence angle has the effect
 497 of reducing the cross-passage pressure gradient which results in flow separation
 498 on the suction surface. Moreover, Kline et al. [49] stated that positive incidence
 499 results in higher exit energy loss, leading to lower turbine efficiency. The results
 500 shown in Fig. 17 confirm the findings mentioned above. The best performance is
 501 achieved with the incidence angle in the range of -20° to -40° . At -50° , the
 502 efficiency drops dramatically. At low pressure ratios (up to 1.75), the performance
 503 of the turbine is similar for different incidence angles. At higher values ($PR > 1.75$),
 504 the incidence angles -40° , -20° , and 0° , respectively, show best performance.



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Fig. 17: Effects of incidence angle on turbine efficiency

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Based on the results of the parametric study, optimum turbine performance is obtained with zero blockage factor. However, blocking is mandatory due to the blade thickness. Therefore, a value of $BK = 0.1$ is assigned for both stator and rotor as recommended by Moustapha et al. [56]. This value is considered realistic, although it affects the turbine flow capacity and efficiency, since it accounts for geometric blockage and boundary layers. Based on Fig. 16, the deviation angles for both stator and rotor are kept 0° in order to maintain the same flow capacity through the turbine stage. An incidence angle of -40° is chosen in the meanline model since it presents the optimum turbine performance as shown in Fig. 17.

516

4.3 Validation of the Method

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Although the meanline model has been validated in Alshammari et al. [36], further validation is necessary due to some significant missing parameters in the testing case by Shao et al. [50]. One of the missing parameters is the rotor blade

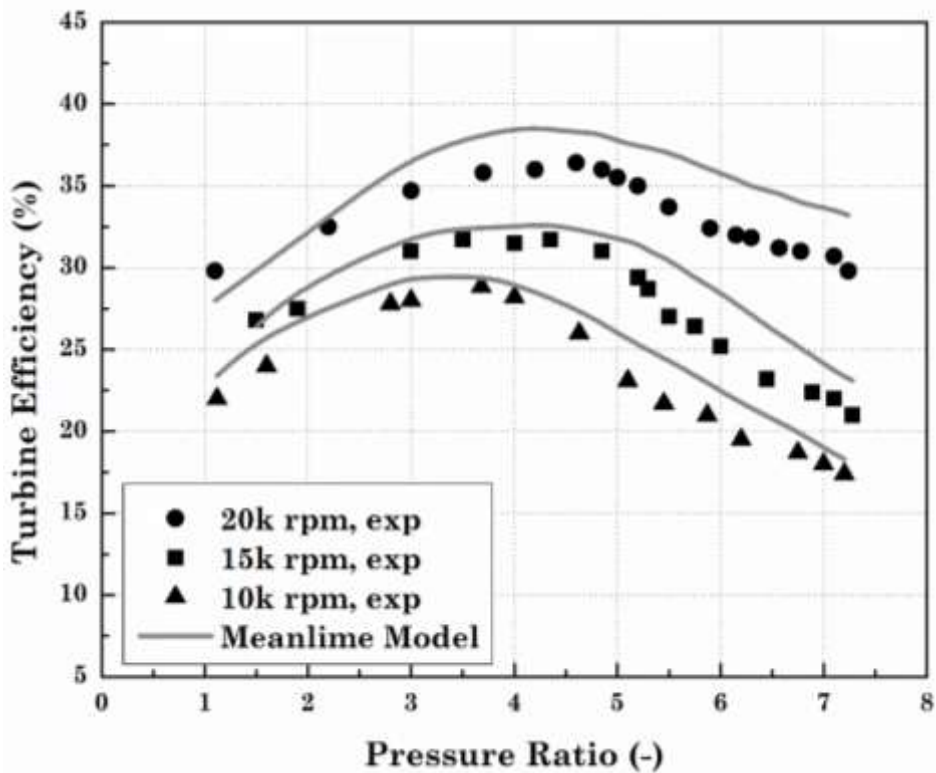
520 height which plays a vital role in the prediction of the flow capacity. The flow
521 capacity or mass flow rate is the control parameter in the meanline model since it
522 is firstly assumed at the beginning of the process and then validated at each
523 station. Therefore, the mass flow rate is also a critical parameter at the prediction
524 of the turbine efficiency. Another missing parameter is the stator opening (stator
525 throat) which is very essential in estimating the flow velocity at the stator outlet,
526 and the nature of the flow whether subsonic or supersonic. Unlike air turbines,
527 radial turbines usually choke in ORC systems. In the previous testing case (Shao
528 et al. [50]), mass flow rates were not plotted against pressure ratio. Therefore,
529 there was no chance to assess the developed mass model. Although these data were
530 missing, the maximum deviation between the predicted efficiency and measured
531 efficiency using Shao et al. [50] was 7%.

532 The results of the meanline model are compared against the test
533 results of the current study for three various speed lines i.e. 10,000 rpm, 15,000
534 rpm and 20,000 rpm. It is worth mentioning that the radial inflow turbine was
535 designed mainly for this application considering the exhaust gas temperature of
536 the engine at full load as the heat source for the thermal oil loop. In such
537 consideration, the rotational speed of the design point is 40,000 rpm (Table 2).
538 However, due to the limitation of the engine dynamometer, the engine is operating
539 at partial load (81 kW instead of 206 kW) with a torque of 450 N.m and maximum
540 speed tested (1700 rpm). Therefore, the turbine is tested at highly off-design
541 conditions.

542 In Fig. 18, the model is validated against the experimental data for the three
543 speed lines. Although efficiencies are mostly overestimated by the proposed
544 meanline model, the general trend of the turbine efficiency is correctly reproduced.

545 The maximum deviation between the predicted and measured data is 3.50 % in the
546 15000 rpm speed line as depicted in Fig. 19. The turbine has been designed to
547 operate at high pressure ratio (PR = 6). Fig. 18 shows that the meanline model is
548 capable of estimating the turbine performance at pressure ratios near the design
549 point value with relative error less 2.8%. At higher pressure ratios, the deviation
550 is relatively high which suggests that the tested turbine may suffer some unusual
551 effects at high pressure ratios.

552

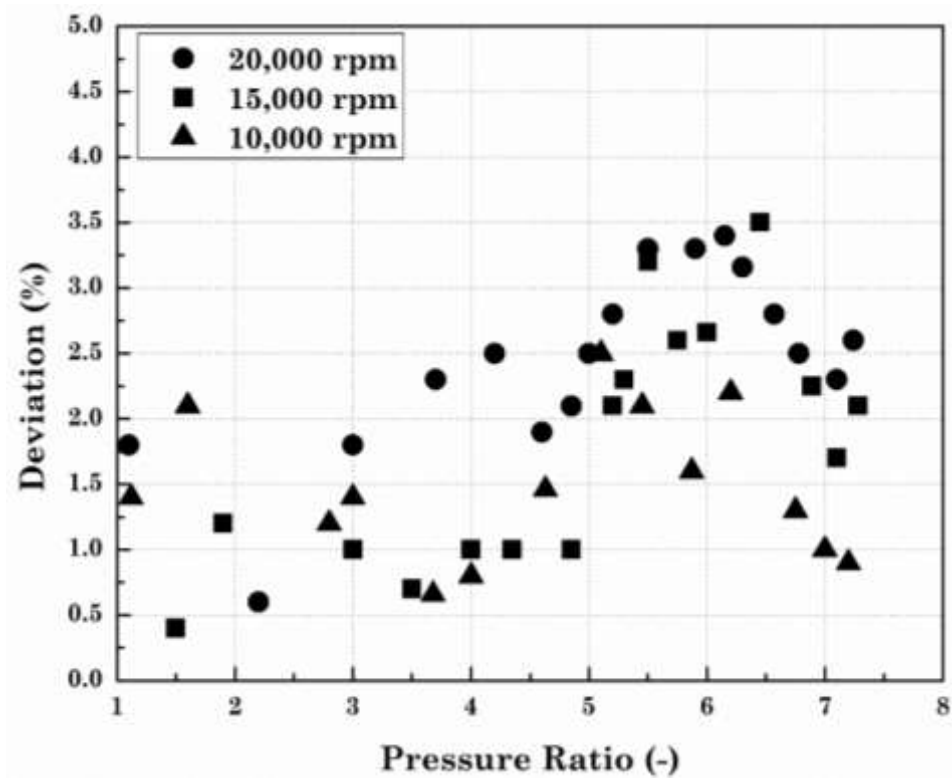


553

554 Fig. 18: Validation of the proposed meanline model

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Fig. 19: Deviations between the tested and predicted results

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In order to evaluate the proposed mass model, the comparison between

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tested and predicted mass flow parameter equation (1) versus pressure ratio is

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depicted in Fig. 20. The corrected mass flow rate is also known as mass flow

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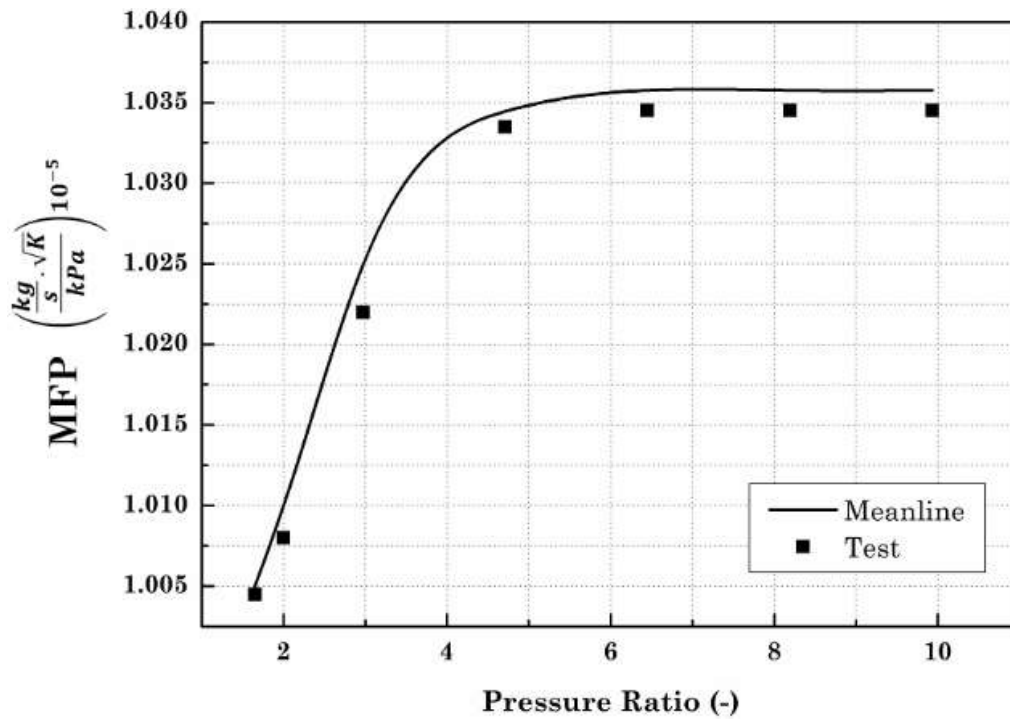
parameter (MFP). As expected, the turbine chokes at $PR \geq 4.8$. The figure also

563

assures the capability of the proposed mass model in predicting the occurrence of

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flow choking.



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Fig. 20: Tested and predicted mass flow parameter versus pressure ratio

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Since the experimental results confirmed the ability of the meanline model

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to produce turbine maps, the performance of the current turbine is estimated at

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extended range of operating points. Using the design point data available in Table

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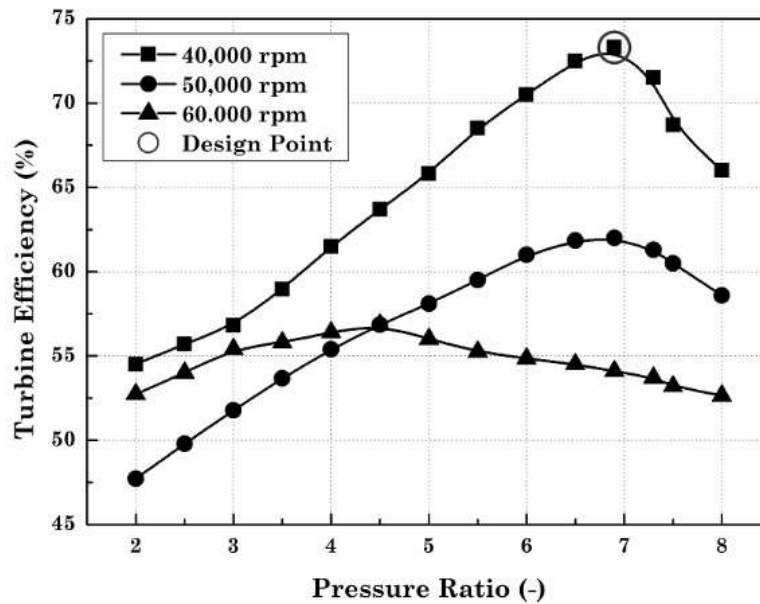
2, the turbine map is built as shown in Fig. 21. The figure clearly shows that the

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40,000 rpm speed line (design point speed) presents more efficient turbine

572

performance for the full range of operating pressure ratios.



573

574 Fig. 21: Performance map of the custom-designed radial inflow turbine using the meanline
 575 model

576

577 **5. Conclusion**

578 In this paper, a compact ORC system, with a custom-designed radial turbine
 579 (and generator), coupled to heavy duty diesel engine was tested. The tests were
 580 run at engine partial load conditions considering the fact that off-design point is
 581 the frequent engine operating point. The ORC and the custom-designed radial
 582 inflow turbine presented a nearly constant efficiencies of 4% and 35%, respectively.

583 One of the main objectives of the paper was to explore the effects of cooling
 584 water temperature and fluid superheating temperature on the cycle performance.
 585 The results showed that increasing the cooling water temperature had a negative
 586 impact on the turbine performance. This was due to the decreased pressure ratio
 587 which affected both generated power and turbine efficiency. Similarly, increasing
 588 the superheating temperature, while fixing water flow capacity and temperature,

589 deteriorated the turbine efficiency and generated power since the turbine pressure
590 ratio presented a decreasing trend.

591 The results of the tests were applied in order to validate the previously
592 developed performance prediction meanline model by authors [38]. The results of
593 model were in good agreement with experimental results with a maximum
594 deviation of 3.5% at 15000 rpm.

595

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599 Limited in the preparation of the ORC test rig and subsequent experimental
600 testing.

601

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