

## Loss of Control of Light Aircraft: A cost effective approach to Flight Test

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ith the

To the 41st SFTE International Symposium





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## What's the question?

- GASCo 28 year review of GA fatal accidents
  - 35-50% stall/spin
  - Significant type variations
- Why type variations
  - C150 rate >> C152 rate (~17:1)
  - PA28 "Hershey Bar" wing ~population mean, tapered wing <u>no</u> fatalities
- Understand the reasons
  - For flying training
  - For design



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## Spot the difference...?



## 10 events: C150/C152 stall/spin fatalities





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# **Methods & Equipment**





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# Flight Test Programme Build-up

|                              |      |             | Pha            | se 1           |         |            | Phase 2 |                 |                  |                | Phase 2 |                  |                 |  | Phase 3 |            |                 |
|------------------------------|------|-------------|----------------|----------------|---------|------------|---------|-----------------|------------------|----------------|---------|------------------|-----------------|--|---------|------------|-----------------|
|                              |      |             |                |                |         |            |         |                 |                  |                |         |                  | ļ               | Vic 2  |         |            | A/c 2           |
| Baseline (Aircraft 1)        |      | CG1<br>Mid  |                | CG2<br>Mid-Aft |         | CG3<br>Aft |         | Aircraft 2      | CG2<br>Mid       |                |         | Aircraft 3       | CG2<br>Mid      |  |         | Aircraft 4 | CG2             |
|                              |      |             |                |                |         |            |         |                 |                  |                |         |                  |                 |  |         |            | Mid             |
| C152                         |      | G-BOFL      | G-FOFL         |                | 4       |            |         | C152            | G-BPEO           |                |         | F152             | G-BFLU          |  |         |            |                 |
| TOW@% MAC:                   | 1637 | bs@23.81%   | <b>* 1</b> lb: | @25.289        |         |            |         | TOW 2% MAC      | 1670 lbs         | @23.39%        |         | TOW@% AC:        | 1655 lb         | @23.78%  |         |            |                 |
| Flt Test/Sortie:             | BTP  | 2008-06-04  | 8TP-20         | 08-06-05       | B7-2    | 08-06-06   |         | Flt Test/Sortie | BTP-20           | 8-06-09        |         | Flt Test/Sortie: | BTP-2           | 08-06-10   |         |            |                 |
| 54541                        | (BTF | 2008-06-01  |                | _              |         |            |         | <u> </u>        |                  |                |         |                  |                 | _  |         |            |                 |
| F10UL TOWOW MAC              | 4500 | -GBLR       |                |                |         |            | ┝       |                 |                  |                |         |                  |                 |  |         |            |                 |
| TUW@% MAC:                   | 1099 | DS@20_20%   |                | 10             |         | la         | ┝       | /               |                  |                |         |                  |                 |  |         |            |                 |
| Fit resu sortie:             | DIF  | 2000-70-02  | '              | a              |         | la         | Η       |                 |                  |                |         |                  |                 |  |         |            |                 |
| F150M                        |      | -PCRT       | G.I            | CP             | G.      | CRT        |         | C150M           | G.N              | VFA            |         | F150M            | G.              | СИН  |         |            |                 |
| TOW@% MAC:                   | 1600 | Us@25.68%   | 1425 lb        | 7,27.229       | 1598 lb | @27.90%    | -       | TOW@% MAC       | 1580 lbs         | D27.00%        |         | TOW@% MAC:       | 1599 LE         | a25.87%  |         |            |                 |
| Flt Test/Sortie:             | BT   | 2008-06-03  | BTP-2          | 3-06-08        | BTP-    | 78-06-07   |         | Flt Test/Sortie | BTP-20           | 8-06-11        |         | Flt Test/Sortie: | BTP-2           | 08-06-12   |         |            |                 |
|                              |      | •           |                |                |         |            |         |                 |                  |                |         |                  |                 |  |         |            |                 |
| F150G                        |      |             |                |                |         |            |         |                 |                  |                |         |                  |                 |  |         | F150G      | G-AVGU          |
| TOW@% MAC:                   |      | •8 aircraft |                |                |         |            |         |                 |                  |                |         |                  |                 |  |         | TOW@% MAC. | 1591 LBS@26.57% |
| Flt Test/Sortie:             |      | 1           | 71             |                |         | $\frown$   |         |                 | Flt Test/Sortie: | BTP-2008-06-14 |         |                  |                 |  |         |            |                 |
|                              |      | _ • /       | / [            | esi            | SO      | rtie       | S       | · + 3 C         | neo              | СКО            | U       | TS               |                 |  | Ч       |            |                 |
| Crew:                        |      | + 1 decline |                |                |         |            |         |                 |                  |                |         |                  |                 |  | 2       |            |                 |
|                              |      | _ • 2       | 5h             | rc             | 25n     | ning       | 2       | flight          | t∆c              | <i>t</i>       |         |                  | STATE OF STREET | Concession of the local division of the loca |         |            |                 |
| RISCO 2011/3 JOINING IL LESI |      |             |                |                |         |            |         |                 |                  |                |         | 51               | Br              | une  |         |            |                 |
| GENERAL AVIATION SAFETY COUN |      |             |                |                | ww      | w.brune    | el.a    | ac.uk/about/    | acad/s           | sed/sec        | dre     | es/cem/bfsl      |                 |  |         | LABC       | RATORY          |

## Flight Test using Rental Aircraft

- Aircraft Variables (ageing?)
  - maintenance, performance, W&CG
- Organisation
  - Different priorities
    - Flight test .v. Flight School
      - Aft CG stalling at 3000ft anybody?
- Local environment
  - Area & procedures, controlled airspace, ATC, Wx





### Stick Force to Change Airspeed Cessna 150L,M & 152 with Flaps a) UP b) DOWN (L30)



## **Climbing Flight: Cessna 150M**

BTP/Sortie: 2008-06-08 A/C= Cessna F150M G-BCRT Date: 12/06/09 Gross Wt.= 1425lbs CG = 36.2" AoD (72% Mid/Aft-CG/27% MAC) Vtrim = 75 MIAS (65 KCAS)

### Climb & Point Track - Full Test



### Time (hh:mm:ss) >>>



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# **Flight Test Results**

- Variability between aircraft, C150 consistently lower pitch forces
  - Mean gradient factors ~2-3
  - C150 sometimes neutral (e.g. L40)
- Stick force dependent on:-
  - (1) Flaps, (2) Power, (3) Trim, (4) CG
- Low stick force >>high mental w'load >>poor airspeed mgt
- Stall warning
  - C150 non-compliant with (current) part 23 in certain configs.
  - Simultaneous stall warning + aerodynamic stall





# Simulation Experiments

- Scenario-based Testing

## **Simulator**

- PC7 Fixed-base research simulator
  - Controllable force feedback
  - 150 x 40 deg. Visuals

## **Scenarios**



- 20 pilots x 5 scenarios x 3 stick force gradients
  - Circuits, EFATO, Climb-out, Go-around, Base to Finals Turn

## Data /Analysis

- Workload (Heart Rate, NASA TLX)
- Flight dynamics + RT/intercom



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## Typical simulator test results: margin of safety & effect of stick forces



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## Example: Pilot 24 Go-around, Full Pwr / Full Flap



## **Conclusions – Causes of LoC**

- Stall Boundary crossing (not point tracking)
  - Handling characteristics & workload
- Type training
  - A C150 <u>is not</u> a C152
- Design factors
  - Certification standards are too subjective
  - Consider Human Factors tools
- Further work needed
  - Historical certification standards for stalling
  - Apparent LSS results and models (FCMC in longstab calcs)





# **Lessons Learned**

- Independence
  - Manage FTI, check maintenance
  - Always check W&CG reports
- Workup
  - Essential but spread across sites and aircraft
  - Use learning curve
- Efficiency
  - Critical test points
  - Concentrate flying periods
    - But always review data between sorties
- Best practice
  - Qualified pilot opinion
  - Redundancy in data collection



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## **Any Questions?**

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