

1 Modelling of Electrically-Assisted Turbocharger 2 Compressor Performance

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8 **Abstract:** For the purposes of design of a turbocharger centrifugal compressor, a one-dimensional
9 modelling method has been developed and applied specifically to electrically-assisted
10 turbochargers (EAT). For this purpose, a mix of authoritative loss models was applied to determine
11 the compressor losses. Furthermore, an engine equipped with an electrically-assisted turbocharger
12 was modelled using commercial engine simulation software (GT-Power) to assess the performance
13 of the engine equipped with the designed compressor. A commercial 1.5L gasoline, in-line, 3-
14 cylinder engine was selected to be modelled. In addition, the simulations have been performed for
15 an engine speed range of between 1000 and 5000 rpm. The design target was an electric turbocharger
16 compressor that could meet the boosting requirements of the engine with noticeable improvement
17 in transient response. The results from the simulations indicated that the EAT improved the overall
18 performance of the engine compared to the equivalent conventional turbocharged engine model.
19 Moreover, the electrically-assisted turbochargers (EAT) equipped engine with power outputs of
20 1kW and 5kW EAT was increased by an average of 5.96% and 15.4%, respectively, ranging from
21 1000 rpm to 3000 rpm engine speed. For the EAT model of 1kW and 5kW, the overall net reduction
22 of the BSFC was 0.53% and 1.45%, respectively from the initial baseline engine model.

23 **Keywords:** Centrifugal compressor; electrically-assisted turbocharger; 1D modelling; compressor
24 design; transient response.

25

26 1. Introduction

27 1.1 General Turbocharging Advantages and Limitations

28 The impact of global warming has forced regulatory authorities around the world to establish
29 strict regulations on CO₂ emissions and other greenhouse gas emissions. Although the regulations
30 created several technical challenges for automotive industry, this led to technology development as
31 there was a demand of decreasing the fuel consumption of engines, increasing their performance as
32 result more power and more sustainable products had to be designed to reduce their environmental
33 impact [1-3]. Among various technologies that have been developed, the most effective way to
34 increase the fuel economy and decrease the CO₂ emissions of passenger vehicles is engine
35 downsizing[4][5]. The purpose of engine downsizing is the reduction of the throttling and friction
36 losses associated with a larger engine by using a smaller, high power-density engine. Thus, the engine
37 efficiency is improved by operating in the more efficient regions, with the boosting device
38 (turbocharger or supercharger) being the crucial component that allows this significantly increased
39 power density to occur.

40 However, these conventional boosting devices have some limitations. Traditionally, the main
41 drawback of the turbocharger is the delay in generating boost pressure which results in insufficient

42 torque at low engine speeds, known as turbo lag [1][6]. Conversely, the mechanical supercharger has
43 a fast response at low engine speed, but it creates parasitic losses to the engine as the required power
44 for its operation is taken from the engine. One feasible solution for avoiding the drawbacks of the
45 conventional boosting technologies is the electrification of the boosting system [7] [8]. Not only do
46 electrification systems reduce the transient response (turbo lag) by 50-400% with respect to time-to-
47 torque depending on the operating point of the turbocharger and engine, but they can also increase
48 the overall engine efficiency for a substantial part of the turbocharged-engine operating range when
49 compared to equivalent mechanical turbocharger systems [9][10]. Therefore, the aim of the paper is
50 to model an advanced electric turbocharger compressor to improve the transient response of the
51 turbocharger at low engine speed and in turn to enhance the performance, efficiency and fuel
52 economy of the engine.

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54 *1.2 Electrically Assisted Turbocharger (EAT)*

55 Katrasnik et al. [11] evaluated the electrical assisted turbocharger in which the torque is applied
56 to the turbocharger using a high speed electric motor. The motor can be integrated into the
57 turbocharger bearing housing, or as an extension to the turbocharger shaft on the compressor side.
58 On the other hand, Baines [12] claimed this type of arrangement would be a less severe working
59 environment for the motor than that of electrically driven compressor in a combined charging system.
60 However, the unit size will be increased and the effects on the shaft dynamics could be problematic.
61 Generally, more weight will be added by the motor, thus, it will affect bearing loads, and also will
62 increase the base inertia of the turbocharger. Nonetheless, the motor can be operated as generator to
63 recover the exhaust energy as electric power which is an additional feature. Katrasnik et al. [11]
64 conducted an experiment in which the transient response and load acceptance of a 6.9L 6 cylinder
65 commercial diesel engine were simulated with the original fixed geometry turbocharger installed and
66 also with two electrically assisted variants. After collaborating and validating the engine model with
67 the experiment data, the transient response of the engine at a fixed speed tip-in was decreased by up
68 to 55% with an electrically assisted turbocharger, the acceptance load was considerably developed.

69

70 *1.3 Preliminary Design of Centrifugal Compressors*

71 Yang et al [7] developed a radial compressor for an electric supercharger in a 2.0L petrol engine
72 by using a mean-line design process. In the process, the operating conditions of the compressor were
73 provided and a set of geometrical parameters were selected to start the design. Then, computational
74 fluid dynamics (CFD) simulation was applied in a detailed 3D design to analyse the compressor's
75 performance. The results of computational fluid dynamics (CFD) simulation method showed a
76 moderate reduction in total pressure ratio and total impeller efficiency compared to the predicted
77 corresponding values in the meanline method. However, it is worth noting that the scroll was
78 excluded in the computational fluid dynamics (CFD) simulation, hence in this method, the predicted
79 values of total pressure ratio and total efficiency were slightly increased as the scroll loss was not
80 included. Therefore, there is a significant difference in the results obtained from 1D modelling
81 method and the computational fluid dynamics (CFD) simulation.

82

83 On the other hand, Harley et al [13] developed a novel meanline modelling method for radial
compressors that is applied in automotive turbochargers. In this modelling approach, the pressure

84 ratio and the recirculation flow of the compressor are linked. Furthermore, Harley et al [13] stated
85 that the results obtained from the existing meanline methods would not be accurate in comparison
86 with test data as the pressure ratio is predicted and thus it could rise and lead to surge. However, the
87 new method deals with the issue by modelling the recirculation effects. Three different diameter
88 centrifugal compressors with vaneless diffusers and scroll collectors were used for the process [13].
89 The procedure starts with the analysis of the inlet blockage creation by recirculation flow and how
90 this affects the induced flow into the impeller. Galvas [14] loss model collection was selected for the
91 analysis as according to Harley et al [13][15][16] not only is a robust model, but it also operates best
92 across a variety of automotive turbocharger radial designs. Then, computational fluid dynamics
93 (CFD) modelling was applied to verify all the results. The results indicate that with the proposed
94 meanline modelling process, the prediction methods are improved as the flow conditions are
95 represented more accurately in the impeller. A major positive outcome of this study is that the novel
96 method could be used for any centrifugal compressor to improve the blade angle selection when
97 recirculation occurs.

98 For further validation and exposing the weakness of the novel meanline modelling method,
99 another experiment was performed by Harley et al[7]. Although the modelling process was for the
100 same variety of turbocharger, automotive, the investigation was done on two new centrifugal
101 compressors. For the purpose of this study, the meanline modelling was used.

102 On the one hand, the results obtained from the single-zone modelling process were predicted
103 accurately for the larger compressor at low to medium tip speeds. However, there was a slightly
104 difference in the prediction at the highest tip speeds. Moreover, the results obtained from the smaller
105 compressor were not predicted accurately as the characteristic of the pressure ratio was insufficient
106 towards surge. As far as the single passage computational fluid dynamics (CFD) method is
107 concerned, the results have shown that the inlet circulation can be predicted with high accuracy;
108 however, the method is incapable to predict the performance in the impeller trailing edge [7].

109 In a recent paper by Samkit et al[17], the preliminary design of a centrifugal compressor was
110 investigated using non-dimensional method. The parameters' dimensions were calculated by various
111 equations and correlations such as Weisner, Lame Ovals equations and Rodger and Sapiro[9], and
112 Johnston and Dean [18] correlations. Moreover, a 3D centrifugal compressor was designed to validate
113 the results of the theoretical analysis. The analysis and simulations indicated that there was a slight
114 difference, about 9% in the results between CFD and the theoretical method. Although the obtained
115 results were in a good agreement, the design losses were neglected from the theoretical calculations.
116 Thus, the preliminary design procedure of the radial compressor is not accurate enough as the
117 variation in the results would increase if the design losses were included.

118 Kerres et al [19] compared the performance of two types of 1D prediction models with numerical
119 data for an automotive turbocharger centrifugal compressor with a vaneless diffuser and a scroll
120 (Figure 1). The prediction models used for the comparison were suggested by Aungier [20] and Oh
121 et al [21]. Moreover, the numerical results obtained by a 3D Reynolds-averaged Navier–Stokes
122 (RANS) simulation for a small turbocharger compressor. The results of the comparison indicated that
123 the 1D model were less accurate at low impeller speeds and choke than the computational fluid
124 dynamics (CFD) 3D method; however, at high impeller speeds, they could predict precisely with a
125 slight variation from the predictions of CFD calculations, see Figure 1.

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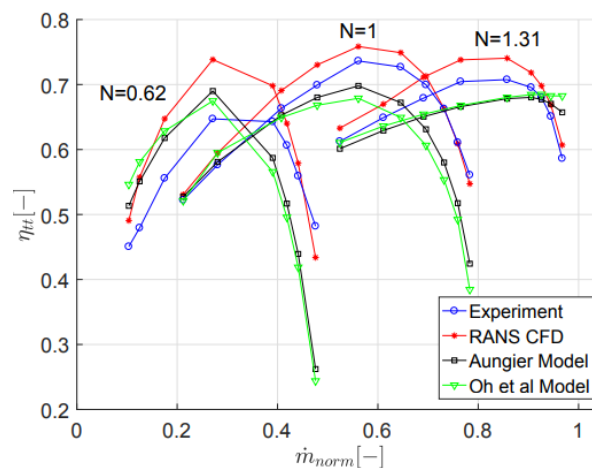


Figure 1: Isentropic Efficiency of Experiment compared with Reynolds-averaged Navier–Stokes (RANS) computational fluid dynamics (CFD), and the two prediction models.

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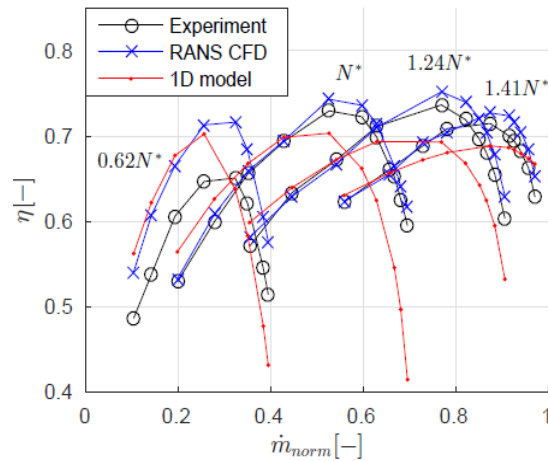
131 Furthermore, Kerres et al [19] found that the loss model by Aungier could predict with higher
132 accuracy than Oh et al [21] model. In addition, the results showed that the loss model collections could
133 be improved, especially for jet-wake mixing and skin friction at high mass flows.

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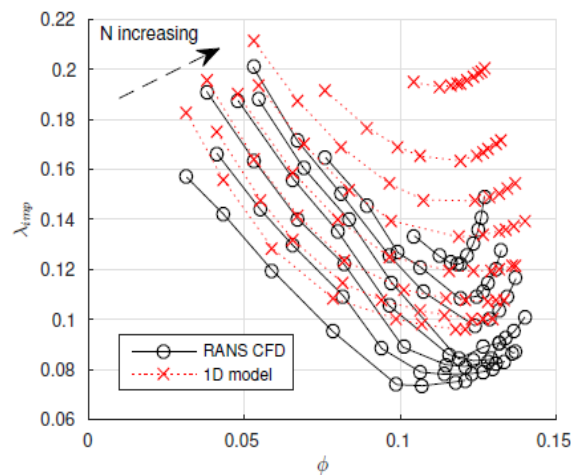
Moreover, an assessment of high-performance radial compressor parameters by 1D modelling
and Reynolds-averaged Navier–Stokes (RANS) 3D computational fluid dynamics (CFD) simulation
was done by Sundström et al [22]. A loss model and a steady state RANS model by Aungier [23] were
chosen for the 1D modelling and computational fluid dynamics (CFD) calculations, respectively.
Also, the centrifugal compressor examined was for automotive turbochargers, especially lightweight
vehicles. The aim of the study was not only to quantify the variation between the predicted and the
experimental data, but also to check the validation of the methodologies that were used.

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Sundström et al [22] in Figure 2 found that the 1D method generated accurate results at design
conditions; however, at higher speedlines towards surge and choke, there was a significant variation
in the results. Figure 2 presents these variations at higher speedlines; N indicates the design speed of
the compressor. Furthermore, the results for the validation of the used methodologies indicated that
the impeller external loss was considerably higher in the computational fluid dynamics (CFD)
calculations towards surge, but this loss could not be captured by 1D model, see Figure 3. Also,
another significant difference was that the scroll loss was predicted larger in the 1D method. Finally,
the authors suggested that by improving these major disparities in 1D modelling method, it would
be beneficial as the accuracy of the prediction method could be improved.



151
152 **Figure 2.** Comparison on the isentropic efficiency among experiment, Reynolds-averaged Navier–
153 Stokes (RANS) computational fluid dynamics (CFD) and 1D model.
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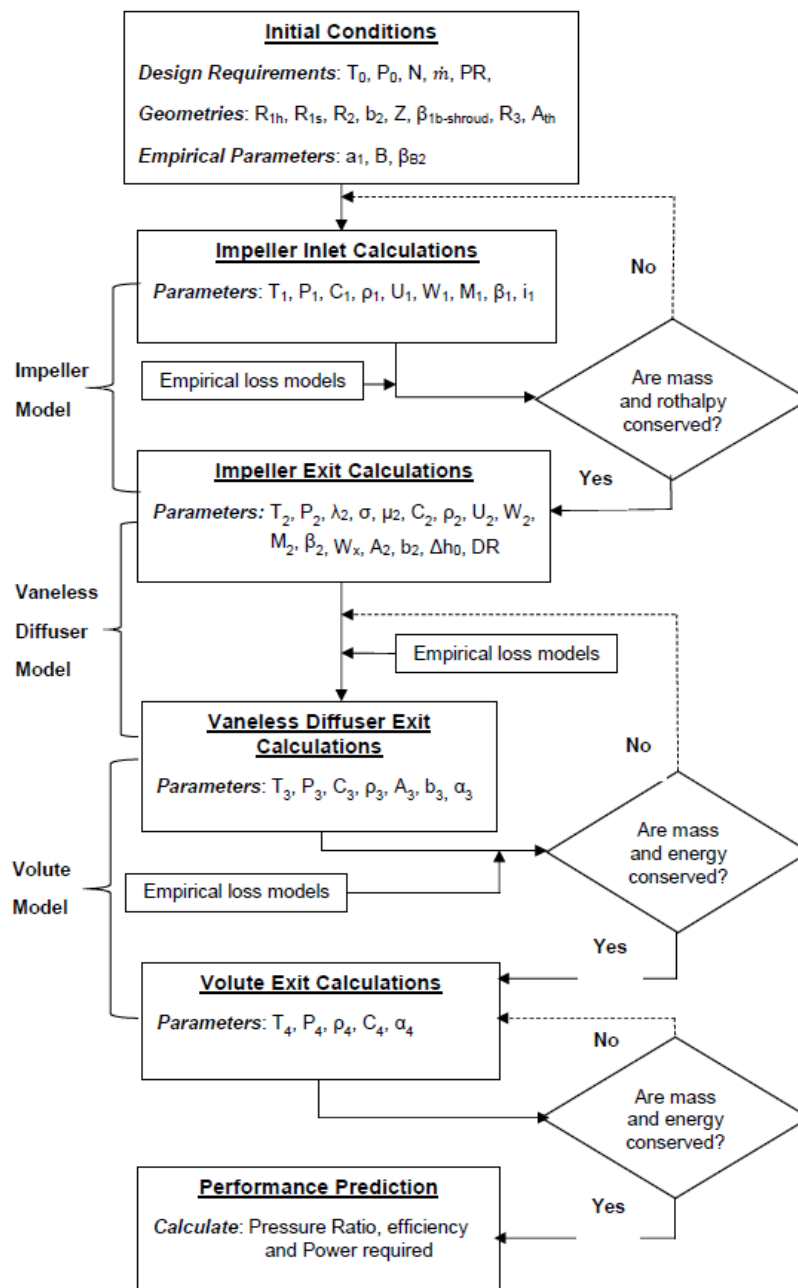
155
156 **Figure 3:** Impeller Losses of RANS CFD and 1D Model.
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158 The present paper, therefore, is attempting to implement leading compressor preliminary
159 performance zero-dimensional and pseudo one-dimensional techniques to the problem of
160 implementation of these for turbocharger compressors which are electrically assisted.

161 2. Centrifugal compressor design methodology

162 The design target was an electric turbocharger compressor that could meet the boosting
163 requirements of the engine at the specified engine speeds while offering a noticeable improvement
164 in transient response. For the analysis of the turbocharger compressor flows, a one-dimensional
165 approach was applied as it is not only a fast and reliable prediction model for centrifugal compressor,
166 but also requires a small number of input parameters [19]. The 1D modelling method is based on the
167 fundamental fluid flow equations, thermodynamic equations, and empirical correlations [23]. These
168 equations were obtained from the Euler's turbomachinery theory and corresponding velocity

169 diagrams, while the empirical relationships were obtained from experimental data. Also, in the 1D
 170 modelling approach, it was assumed that the flow conditions were uniform as the air would behave
 171 as an ideal gas [24].



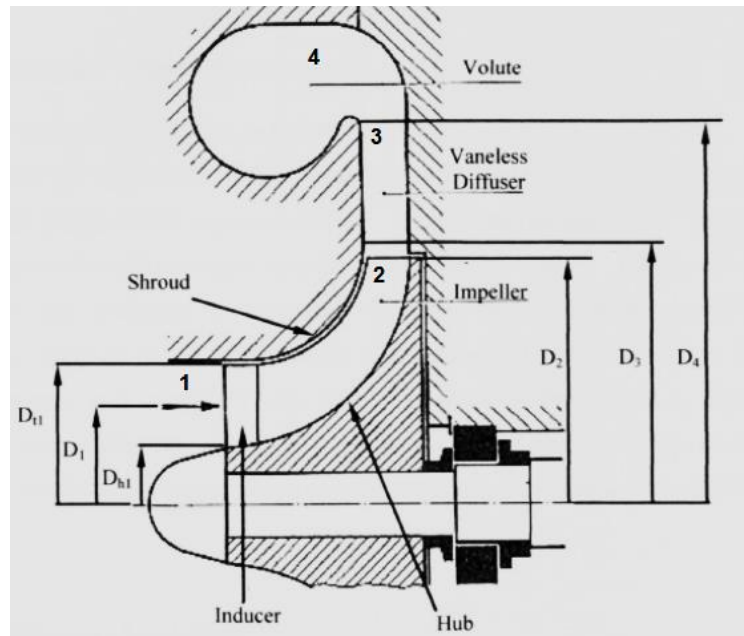
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Figure 4: Compressor Modelling Procedure Flow Diagram.

174 In Figure 4, the schematic flow diagram of the compressor modelling procedure that was used
 175 for this analysis is shown, implemented in the MATLAB programming language. In addition, a
 176 crucial step for the modelling and analysis of the compressor was separating its individual
 177 components into four stages. Each stage includes the inlet or exit characteristics of the component.
 178 The compressor consists of the impeller, vaneless diffuser and the scroll. Thus, stage one is the
 179 impeller inlet, while stage two is the impeller exit and the vaneless diffuser inlet. The impeller exit

180 and the vaneless diffuser operate under the same flow conditions. Moreover, stage three is the
 181 vaneless diffuser exit and the scroll inlet where the conditions are again the same. Finally, stage four
 182 is the scroll exit condition. In Figure 5, the cross-sectional schematic of the centrifugal compressor
 183 with the stage separation is shown.



184

185 **Figure 5.** Centrifugal Compressor cross-section schematic labelling the stages.

186 To start the design procedure of the centrifugal compressor, the input parameters and the
 187 compressor geometry are required. Thus, the parameters that were given as designed constraints
 188 would be the inlet conditions (ambient static pressure and temperature), maximum rotational speed,
 189 air mass flow rate, and pressure ratio. As far as the compressor geometries were concerned they were
 190 obtained from the centrifugal compressor that was designed by Yang et al[4]. Both input parameters
 191 and compressor geometries are shown below.

192

Table 1. Operating conditions for the centrifugal compressor

Input Parameters	Metric Units
Ambient Static Pressure	1.01325 bar
Ambient Static Temperature	298 K
Maximum Rotational Speed	120000 rpm
Air Mass Flow Rate	0.08 kg/s
Pressure Ratio	2

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Table 2. Main Compressor Geometry [4].

Geometrical Parameters	Values
Inlet hub radius R_{th}	6.2 mm
Inlet shroud radius R_{1s}	17.2 mm
Impeller exit radius R_2	29.1 mm
Exit blade height b_2	3.4 mm

Blade number Z	5+5
Blade angle at inlet tip $\beta_{1b-shroud}$	-62 deg
Diffuser exit radius R_3	40.75 mm
Scroll throat A_{th}	397^2

196

197 **3. Non-dimensional loss parameters**

198 In this section, the non-dimensional loss models, $\bar{\omega}$, of the compressor components that were
 199 used for the purposes of the study were presented. Specifically, the loss models for each component
 200 were individually described and determined.

201 *3.1 Impeller loss models*

202 According to Aungier[23], the impeller losses can be converted and expressed as a total relative
 203 pressure loss defined as:

$$\Delta P_{tr} = f_c (P_{tr1} - P_{s1}) \sum_i \bar{\omega}_i \quad (1)$$

204 Where f_c is a correction factor, which depends on the total relative density ρ_{tr}

205 The mathematical descriptions of these two functions according to Aungier[23] are described as:

$$f_c = \frac{\rho_{tr2} T_{tr2}}{\rho_{tr1} T_{tr1}} \quad (2)$$

$$\rho_{tr} = \frac{P_{tr}}{RT_{tr}} \quad (3)$$

206 As the impeller is the rotating component, it was essential to determine the total relative pressure
 207 and temperature, which are defined as follows [19]:

$$P_{tr} = P_s \left[1 + \left(\frac{\gamma - 1}{2} \right) M_r^2 \right]^{\frac{\gamma}{\gamma - 1}} \quad (4)$$

$$T_{tr} = T_s \left[1 + \left(\frac{\gamma - 1}{2} \right) M_r^2 \right] \quad (5)$$

208 Where M_r is the relative Mach number, defined as:

$$M_r = \frac{W}{\sqrt{\gamma R T_s}} \quad (6)$$

209 To calculate the actual conditions of the impeller, the non-dimensional losses need to be
 210 subtracted from the isentropic conditions. However, the total relative temperatures in isentropic and
 211 actual flow are the same because the rothalpy is conserved. Thus, as far as the total relative pressure
 212 in the actual flow at the second stage is concerned, it can be calculated by subtracting the total relative
 213 pressure losses from the total relative pressure in the isentropic process. The equation is shown
 214 below[23]:

$$P_{tr2} = P_{tr2,is} - \Delta P_{tr} \quad (7)$$

215 Furthermore, according to Bathie[25], the total pressure and temperature at the throat are
216 described as:

$$T_{th} = T_{t1} - \left[\frac{2N\pi(C_{u1}r_{m1} - C_{uth}r_{mth})}{60c_p} \right] \quad (8)$$

$$P_{th} = P_{t1} \left(\frac{T_{th}}{T_{t1}} \right)^{\frac{\gamma}{\gamma-1}} \quad (9)$$

217 Aungier[23] and Boyce [26] loss models were selected to determine the loss parameters of the
218 impeller. The chosen models that are analysed are presented in Table 1.

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221

Table 1. Impeller Loss Models used for the study[23][26].

Impeller Loss Models	
1. Shock	2. Incidence
3. Diffusion	4. Choke
5. Skin friction	6. Clearance gap
7. Blade loading	8. Hub-shroud loading
9. Wake mixing	10. Expansion
11. Supercritical Mach number	

222

223 Moreover, each loss parameter was described and analysed individually. Also, the loss model
224 was described as a function of inlet, throat, isentropic exit and boundary conditions such as the area,
225 the radius of the impeller, the rotational speed and the mass flow rate.

226

227 3.2 Vaneless diffuser loss models

228 According to Aungier[23], the non-dimensional vaneless diffuser losses can be converted and
229 expressed as a total pressure loss defined as:

$$\Delta P_{t3} = (P_{t2} - P_{s2}) \sum_i \bar{\omega}_i \quad (10)$$

230 The vaneless diffuser loss model process is similar to that in the impeller. However, the vaneless
231 diffuser is a stationary component and the rothalpy is decreased to a constant enthalpy, hence the
232 total temperature in the vaneless diffuser and the scroll is constant.

233 The actual conditions of the vaneless diffuser were calculated by subtracting the total pressure
234 loss from the isentropic total pressure. The equation is shown below[23]:

$$P_{t3} = P_{t3,is} - \Delta P_{t3} \quad (11)$$

235 Furthermore, two loss parameters were selected to model the losses for the vaneless diffuser, the
236 skin friction and diffusion[23].

237

Table 2. Vaneless diffuser loss models

Vaneless Diffuser Loss Models	
1. Skin friction	2. Diffusion

238

239 3.3 Scroll loss models

240 According to Aungier[23], the non-dimensional scroll losses can be converted and expressed as
241 a total pressure loss defined as:

$$\Delta P_{t4} = (P_{t3} - P_{s3}) \sum_i \bar{\omega}_i \quad (12)$$

242 As stated in the previous vaneless diffuser section, the scroll is in stationary form, hence the total
243 temperature in the scroll is constant. Furthermore, to calculate the actual pressure conditions of the
244 scroll, the non-dimensional pressure losses need to be subtracted from the isentropic conditions. The
245 equation is shown below[23]:

$$P_{t4} = P_{t4,is} - \Delta P_{t4} \quad (13)$$

246 Moreover, it has been stated that there are four non-dimensional scroll losses[23].

247

Table 3. Scroll loss models

Scroll Loss Models	
1. Meridional velocity loss	2. Tangential velocity loss
3. Skin friction	4. Exit cone loss

248

249 4. Modeling the engine

250 For the aim of this study, two models of an engine were designed in GT-Power. The first model
251 was sourced from an available model of the Ford Ecoboost 1.5L, an inline 3-cylinder engine which
252 served as the baseline model with the correct turbocharger map implemented in the model. The
253 second model had the same engine but with the EAT implemented. The second model is shown in
254 Figure .

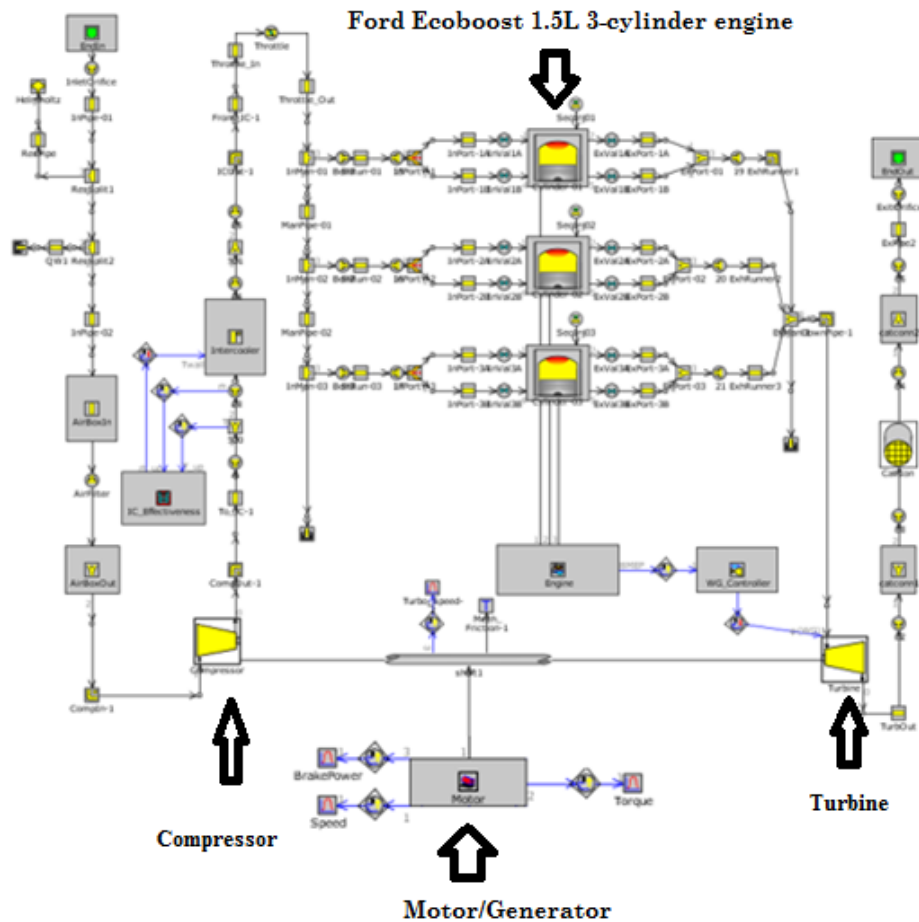


Figure 6: Model of the Ford EcoBoost 1.5L electrically assisted turbocharged engine.

In the above model the motor/generator object, which represents the electrical machine of the electrically assisted turbocharger, is attached to the turboshaft.

4.1 Motor/Generator specifications

The motor/generator parameters used for the simulation were based on an electrical machine developed by Lee and Ehsani[27]. Table 4 below shows the Brushless DC motor/generator specifications while Figure 7 illustrates the peak power and torque curves of the engine modelled.

Table 4: Motor/Generator Specifications.

Attribute	Object value	Units
Torque Constant	0.0042	N.m/A
Equivalent Resistance	3	Ohm
Equivalent Inductance	1.22	mH
Friction Torque	0.662	N.m
Stall Current	200	A
Inertia	8.2614e – 5	kgm ²

267

268 Furthermore, the applied voltage of the battery that was used to the engine model was 48V. The
 269 development of electric turbocharger was initially based (and limited by) the 12V architecture of
 270 automobiles. However, some automobile makers have been considering the use of 48V architectures
 271 to enhance the goal of improving transient response for a considerable time in order to accommodate
 272 mild-hybridization requirements. According to a simulation that was conducted by Nishiwaki et al
 273 [28] for a 48V battery system and compared against an equivalent 12V architecture, have shown a
 274 significant improvement in the transient response at low engine speeds (0.4s to reach a target 1.65
 275 pressure ratio against 1.0s for the 12V system).

276 Finally, for the analysis presented here, it is worth noting that the thermal behavior of the
 277 motor/generator was ignored.

278

279 4.2 Engine Specifications

280 For the simulation of the modeled engine, it was required to import the Ford Ecoboost 1.5L
 281 specifications for modeling the engine objects. The specifications are described below.

282

283

Table 5. Ford Focus, Ecoboost 1.5L, 3-cylinder petrol engine specification [29]

Attribute	Object value	Unit
Bore	83	Mm
Stroke	92.4	Mm
Connecting Rod Length	137	Mm
Compression Ratio	10.4	----
TDC Clearance Height	1	mm
Max.Brake Power	110	kW
Max. Brake Torque	240	Nm
Min. Brake Specific Fuel Consumption (@3000rpm)	234	g/kWh
Valves	4	

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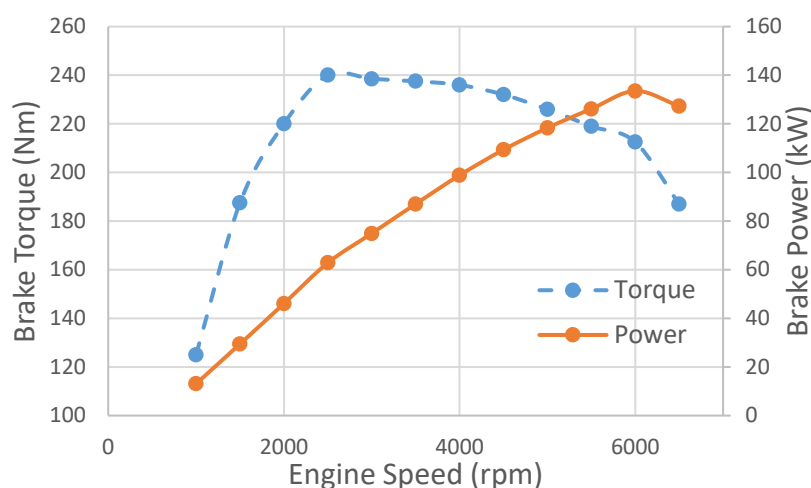


Figure 7: Performance of the based turbo-charged engine.

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288 4.3 Description of the simulation

289 Five cases were used for the simulation of the baseline model and electrically assisted
 290 turbocharged model. Each case had different engine speed which was varied from 1000 to 5000 rpm.
 291 Moreover, as initial conditions were given, the ambient pressure and temperature were 1.01325 bar
 292 and 298K, respectively.

293 It is worth to be mentioned that the model of the electrically assisted turbocharger was modified
 294 and analysed for five different power levels, varied from 1kW to 5kW. This is performed to study
 295 and examine how the power consumed or generated from or by the electrical machine affects the
 296 engine torque at low and high engine speed rates. The operating point used for the simulations are
 297 presented in Table 6.

298

299

Table 6. The operating points that were used for the simulation

Operating Point	Turbocharger	Maximum Power
Baseline	Conventional	-
Operating Point 1	EAT	1kW
Operating Point 2	EAT	2 kW
Operating Point 3	EAT	3 kW
Operating Point 4	EAT	4 kW
Operating Point 5	EAT	5 kW

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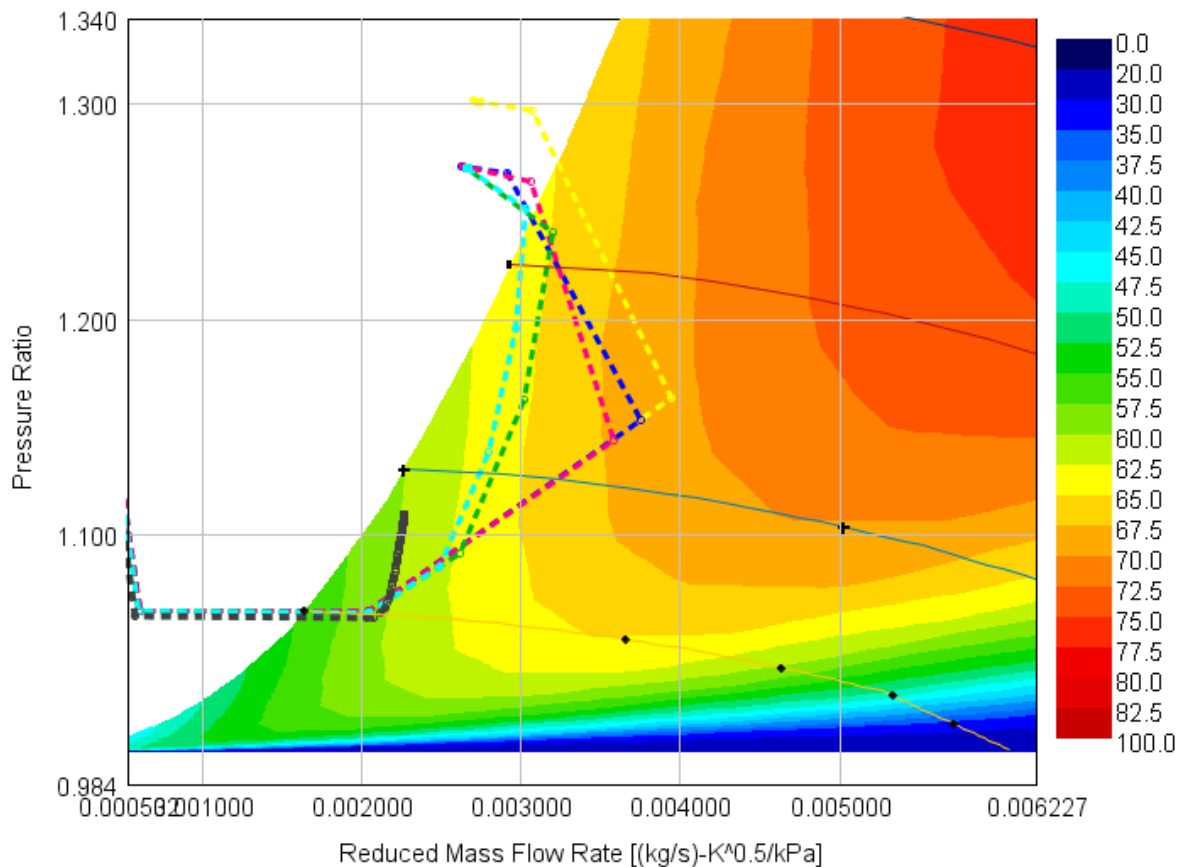
303 5. Results and Discussion

304 5.1 Compressor Efficiency

305 The operation of the centrifugal compressor for the baseline and EAT models at low engine
 306 speed rates, 1000 rpm and 2000 rpm, are presented in the following figures – the outcome of the
 307 engine simulation setup described in Section 5.

308 Figure shows that the operation of the compressor at 1000 rpm engine speed is moved
 309 significantly to the right side of the performance map with the assistance of the motor/generator.
 310 Thus, the compressor operation has moved to a more efficient region. Moderate improvement can be
 311 observed to the operation for the 1kW and 2kW EAT model compared to the baseline model. On the
 312 other hand, from 3kW to 5kW power models, the compressor operation has increased greatly
 313 compared to the baseline model.

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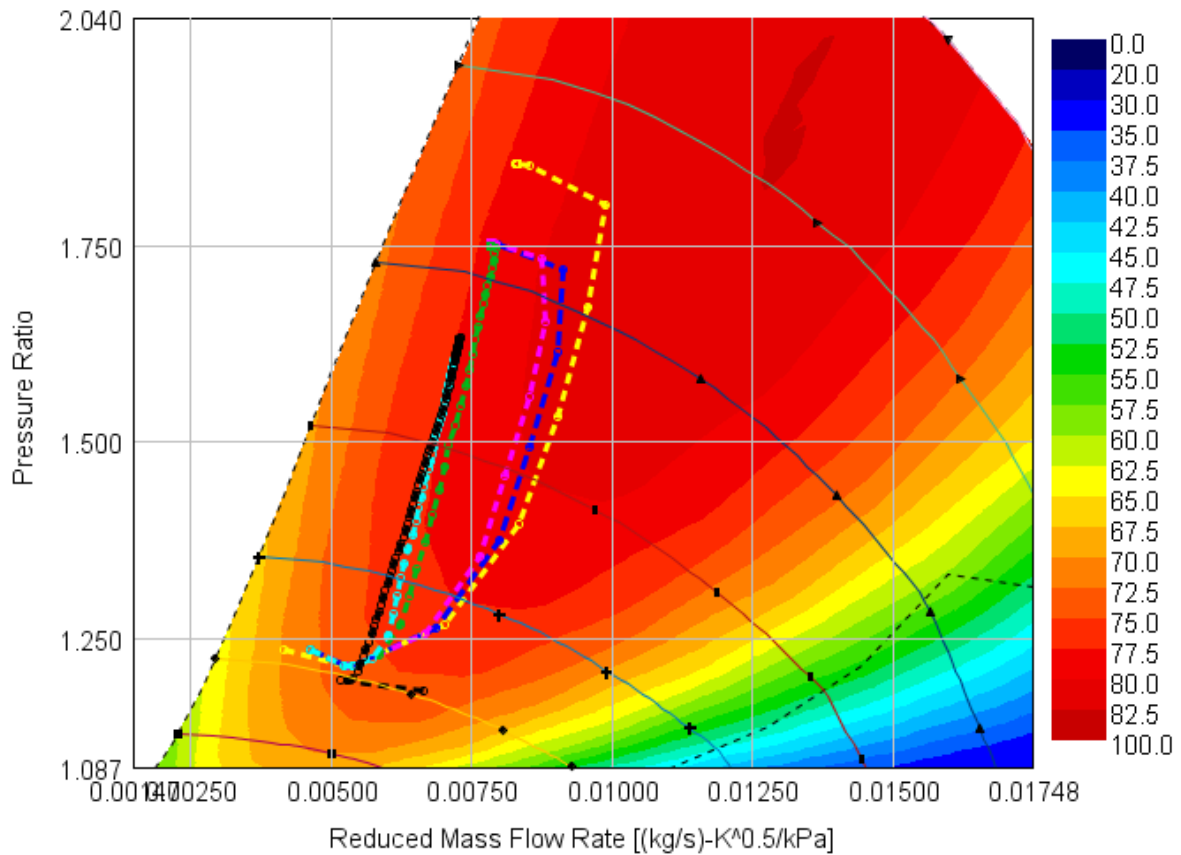
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316 **Figure 8:** Compressor Performance at 1000 rpm engine speed with efficiency contours.

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318 Furthermore, an important drawback is that the EAT forces the operation of the compressor to
 319 exceed the surge line of the efficiency map as is shown in Figure . Therefore, map enhancement

320 techniques should be applied to the compressor, such as casing treatment, to overcome this
 321 drawback.



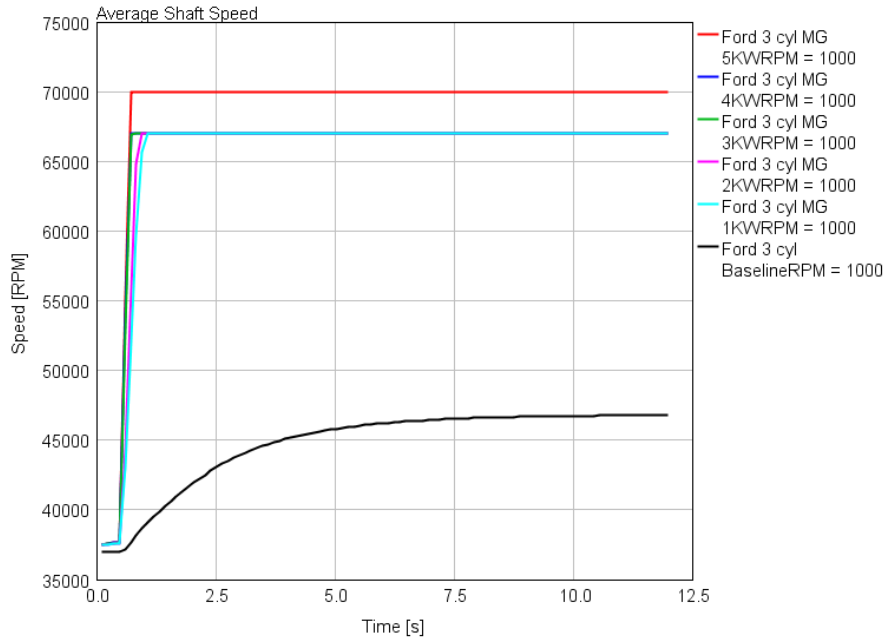
322 **Figure 9:** Compressor Performance at 2000 rpm engine speed.

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 324
 325 As far as the 2000 rpm engine speed is concerned, Figure illustrates that the application of
 326 the motor/generator to the turbocharger shaft pushes the operation of the centrifugal compressor
 327 towards the right side of the map. Thus, the compressor operates to a more efficient area. Also, Figure
 328 shows that the electrically-assisted turbocharger (EAT) models, except from the 1kW model, have
 329 significantly increased the pressure ratio of the compressor, resulting in further improvement of the
 330 engine performance.

331
 332 *5.2 Transient Response*

333 As can be seen from Figure 13, the transient response time of the baseline engine at 1000 rpm
 334 has been reduced by 85.6% with the electrical assistance of 5kW (for accelerations to 110,000 rpm out
 335 of a maximum turbocharger speed of 120,000 rpm). Moreover, the 3kW and the 4kW models have
 336 almost the same transient response time as the 5kW. On the other hand, the transient response time
 337 for 2kW configuration has been decreased by 64% compared to baseline model. It is worth to be

338 mentioned that the application of the electrical assisted turbocharger with 1kW reduced only 20% of
339 the response time for the engine compared to the conventional turbocharger. Therefore, it is essential
340 to install at least a 2kW EAT to improve the performance of the engine considerably.

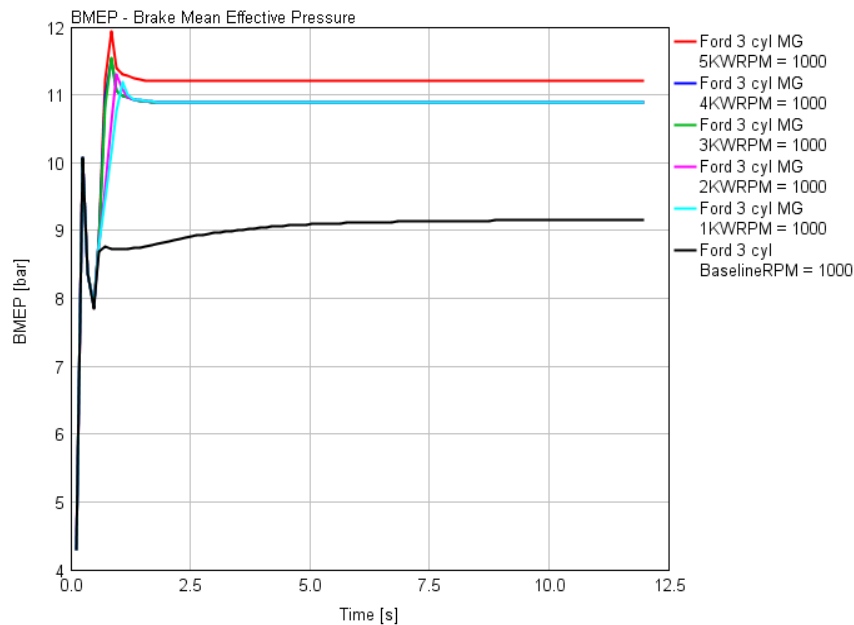


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Figure 10: Turbocharger shaft speed at 1000 rpm engine.

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Figure 11: Brake Mean Effective Pressure (BMEP) response time at 1000 rpm engine speed.

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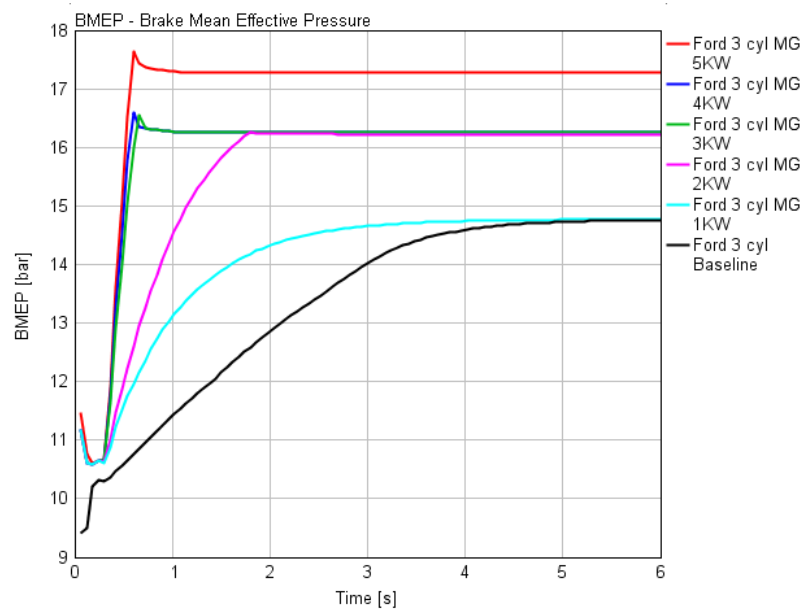
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The results in Figure 10 indicate that the electrical assisted turbocharger of 1kW at 1000 rpm of engine speed, can decrease the transient response time of the engine by 78.4% whereas with higher

350 power modifications the response time can be reduced up to 83.2%. Moreover, the fluctuations that
 351 are displayed on the graphs in Figure 11 were a result of the mass flow rate variation due to the
 352 throttle valve opening. In addition, as can be seen from Figure 11, there is an insignificant effect on
 353 the reduction of the response time among the 3kW and the higher power level models.

354 Furthermore, the results from Figure 12 indicate that as the power levels rise, not only is the
 355 response time of the engine decreased, but the BMEP is also increased significantly.

356



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Figure 12: Brake Mean Effective Pressure response time at 2000 rpm engine speed.

358

359 In Figure , the average turbocharger shaft speed over time is illustrated. Overall, the shaft speed
 360 is rapidly increased with the electrical assistance. Especially, the maximum shaft speed that was
 361 achieved by the 5kW model peaked at 110 krpm, while the maximum speed for the models from 2kW
 362 to 4kW was at 105 krpm. Although the shaft speed of 1kW configuration has been increased faster
 363 than the conventional turbocharger, the maximum speed that was achieved was lower than that of
 364 the baseline model. Thus, the 1kW model is not a preferable turbocharger for the engine.

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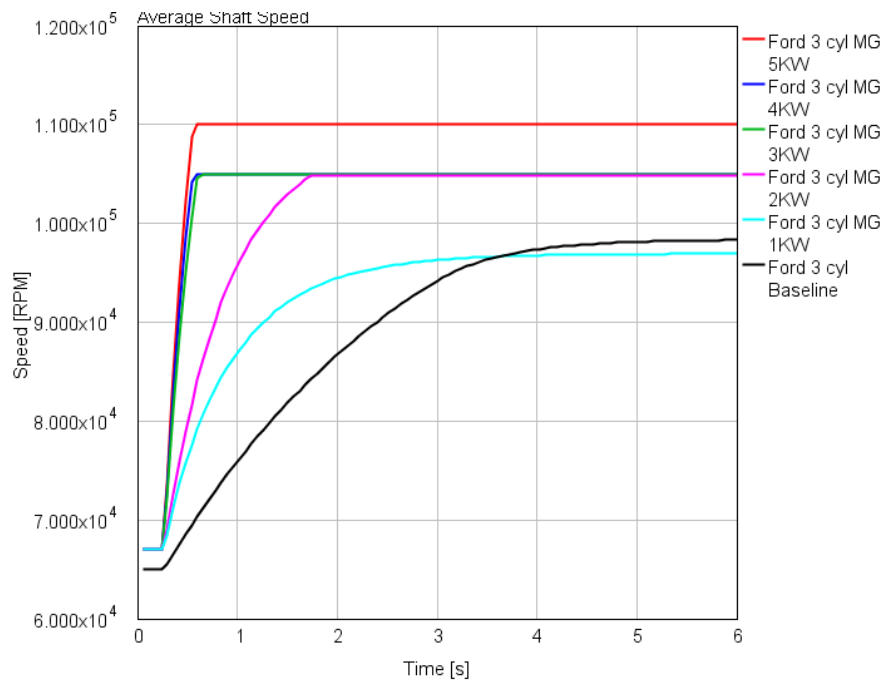


Figure 13: Turbocharger shaft speed at 2000 rpm engine speed.

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370 5.3 Brake Torque and BSFC graphs

371 In this section, the brake torque and the brake specific fuel consumption of the GT-Power models
372 over the engine speed graphs are presented and discussed. The results in Figure 14 indicate that
373 engine torque was significantly improved with the electrically assisted turbocharger. Especially, the
374 highest value of torque achieved by the 5kW model was at 240 Nm at 3000 rpm engine speed.
375 Moreover, the torque graphs of the 3kW and 4kW model were almost the same, but, nonetheless, the
376 difference can be spotted in the netBSFC map in Figure (after power required to run the motor was
377 subtracted). The BSFC of the 4kW power level is slightly reduced for the engine speed range from
378 3000 to 4000 rpm compared to the 3kW EAT model.

379 Furthermore, as far as the 1kW and 2kW cases are concerned, the engine torque has been
380 increased for both models, however, only for the engine range of 1000-2000 rpm and 1000-3000 rpm,
381 respectively. It is worth noting that at 4000 to 5000 rpm, the torque of the EAT models has been
382 decreased and have slight differences with the engine torque of the baseline model as can be seen
383 in Figure 14. This occurs because the electrical assisted turbocharger extracts torque from the shaft at
384 high engine speed to charge the battery of the electric machine.

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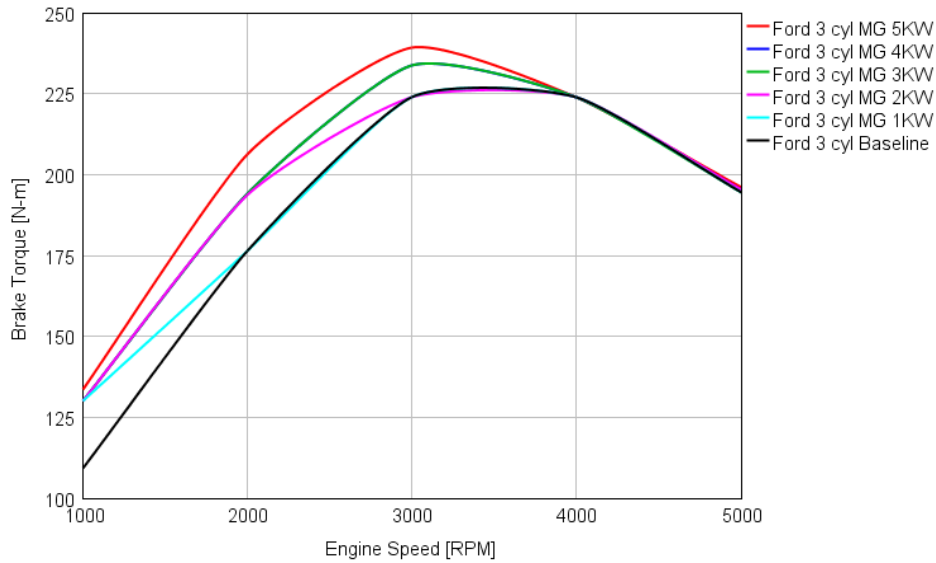


Figure 14. Engine Torque results from the engine model.

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The results in Figure depict the BSFC of GT-Power models. It is evident that when the EAT is applied, the BSFC of the engine is significantly improved. Specifically, for the 5kW model, the BSFC has been reduced by an average of 1.4% compared to the baseline model. In addition, the 1kW has considerably decreased the BSFC for 1000-4000 rpm speed range whereas the BSFC has been further reduced with the electrical assistance of 2kW. Finally, at high engine speed, there is a slight fall in the BSFC graphs of the electrically-assisted turbocharger (EAT) models and the baseline model despite the fact that additional torque is extracted from the engine.

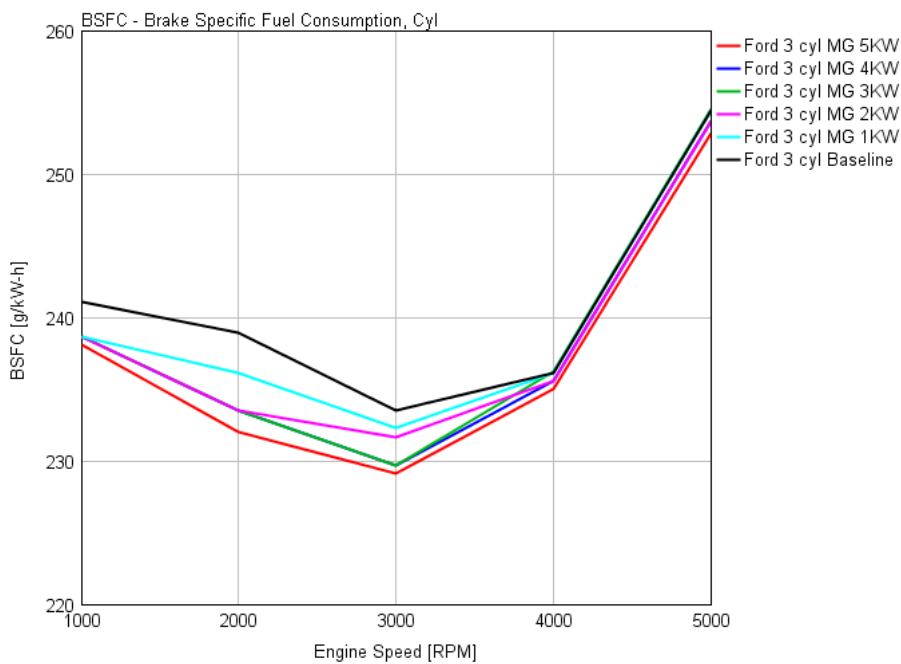


Figure 15: BSFC map results from the engine model.

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399 6. Results from the optimised compressor

400 In this section, the results obtained from GT-Power simulation for the optimised compressor
 401 which was found by using the Matlab code are presented and discussed. Table 8 includes the
 402 optimised geometries of the centrifugal compressor which were obtained from the Matlab code.

403

404

Table 8: Optimisation results for B=0.02 and N=120,000 rpm.

$r_{1h}[m]$	0.001	$Z_r[-]$	10
$r_{1t}[m]$	0.0113	$\beta_{B2}[^\circ]$	-45
$M_{1t}[-]$	0.7015	$DR[-]$	1.6682
$\beta_1[^\circ]$	15.1548	$n_{stage}[-]$	0.4361
$A_{f1}[m^2]$	3.9555e-4	$b_2[m]$	0.0047
$U_{1t}[m/s]$	141.5642	$r_2[m]$	0.0294
$W_{1t}[m/s]$	231.0347	$U_2[m/s]$	369.6808
$C_{m1}[m/s]$	223	$W_2[m/s]$	138.4913
$C_{u1}[m/s]$	81.1653	$M_2[-]$	0.6807
$C_1[m/s]$	237.3116		

405

406 These geometries were applied to the GT-Power models to evaluate the effect of the optimised
 407 compressor to the engine performance. In addition, to identify the differences between the initial and
 408 the optimised compressor a one by one comparison of the models was conducted.

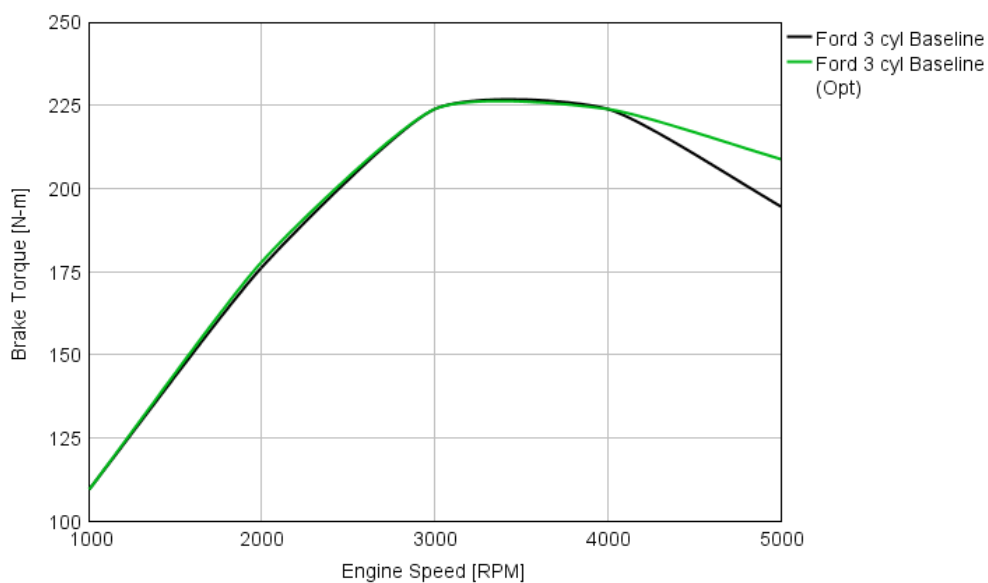
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410 6.1 Comparison of the baseline models

411 The results in Figure 16 shows that overall, there was a slight increase in the engine torque
 412 with the optimised compressor. However, at high engine speed and especially from 4000 to 5000
 413 rpm the engine torque has been moderately risen. Similarly, in Figure 17 the BSFC for the optimised
 414 compressor has been slightly improved compared with the BSFC of the initial compressor.

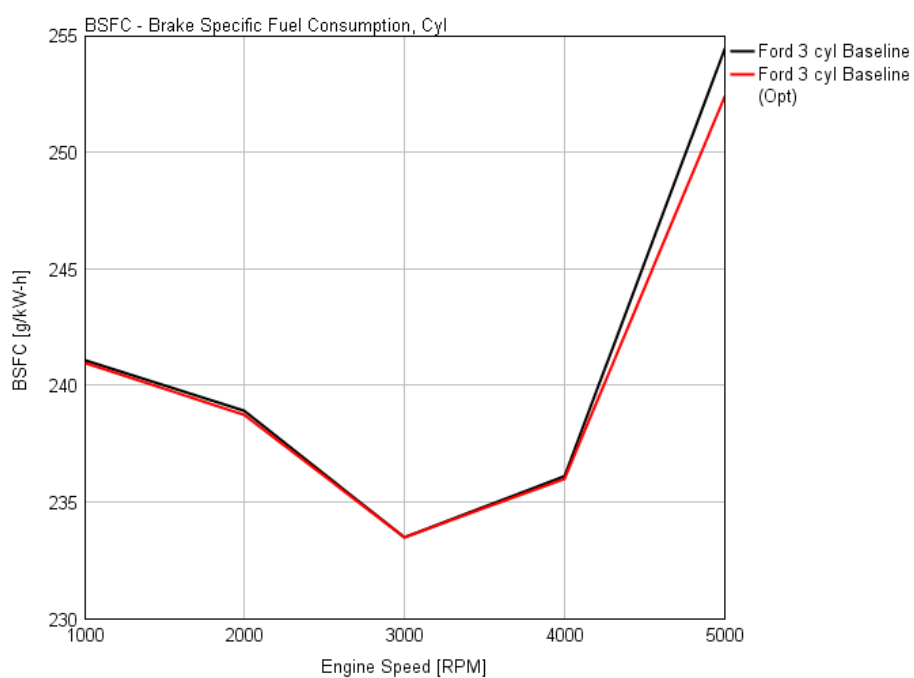
415 Moreover, there was not any reduction to the transient response time of the engine with the
 416 optimised compressor. However, the turbocharger shaft speed for 1000rpm engine speed has been
 417 peaked to a higher value than the initial compressor design as is shown in Figure 18.

418



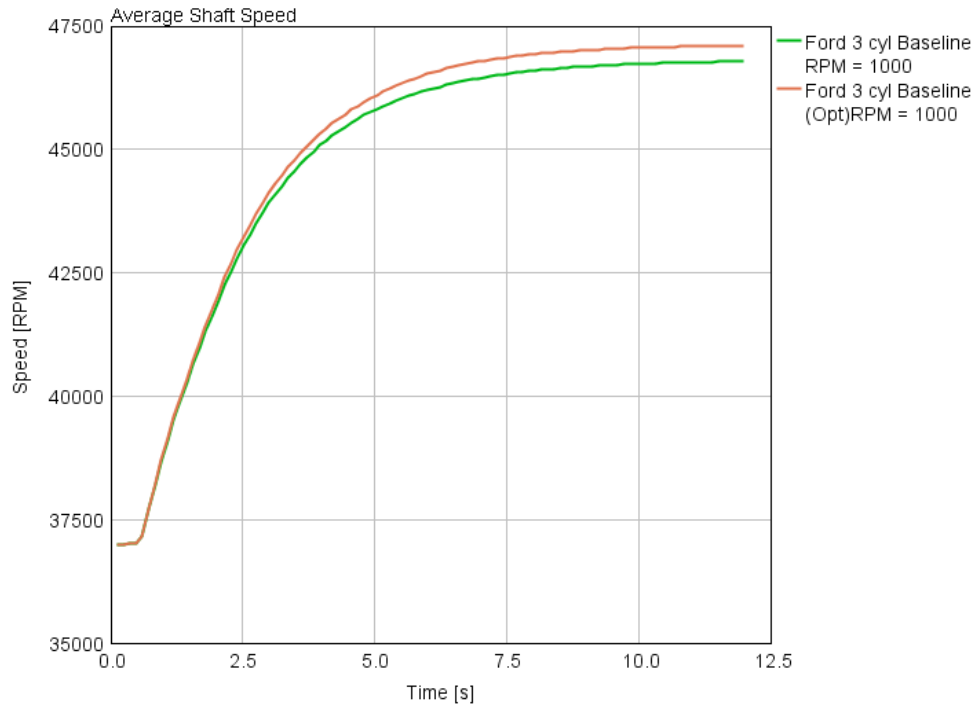
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Figure 16: Brake torque versus engine speed for the initial and the optimised baseline models.



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Figure 17: BSFC graphs for the initial and optimised baseline models.



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Figure 18: Average Shaft speed of the baseline models at 1000 rpm engine speed.

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427 *6.2 Comparison of the EAT with 1kW power level models*

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The results obtained from the simulations for the initial and optimised compressor of the

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electrical assisted turbocharger models with power level 1kW indicate that there is no significant

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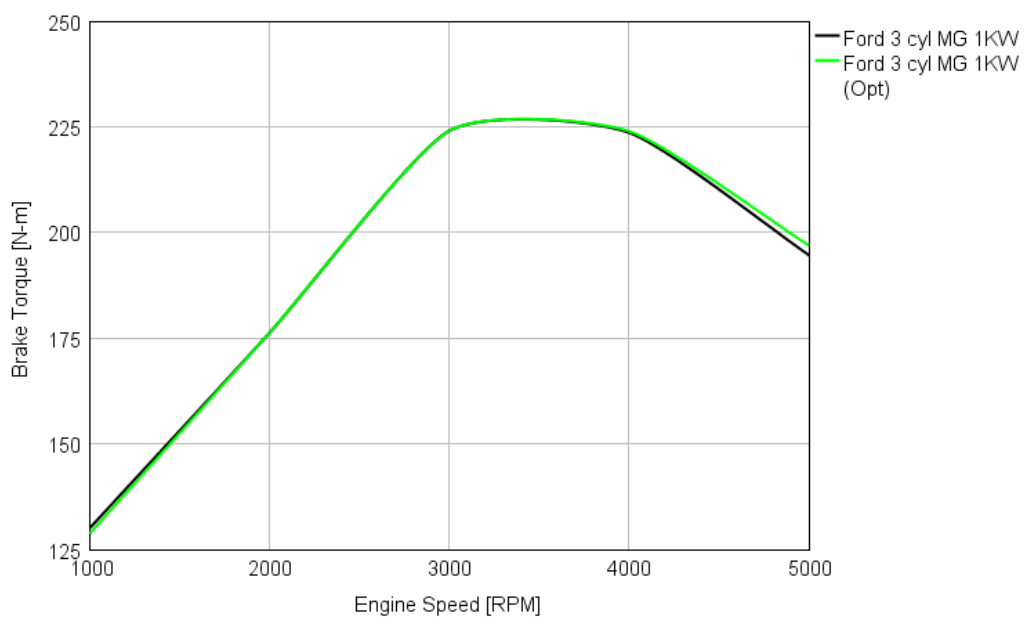
difference between the two compressors. Figure 19 shows the engine torque versus engine speed

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for both centrifugal compressors. Even though the two graphs are very similar, there is a slight rise

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of the torque for the optimised compressor at engine speed range of 4000-5000rpm.



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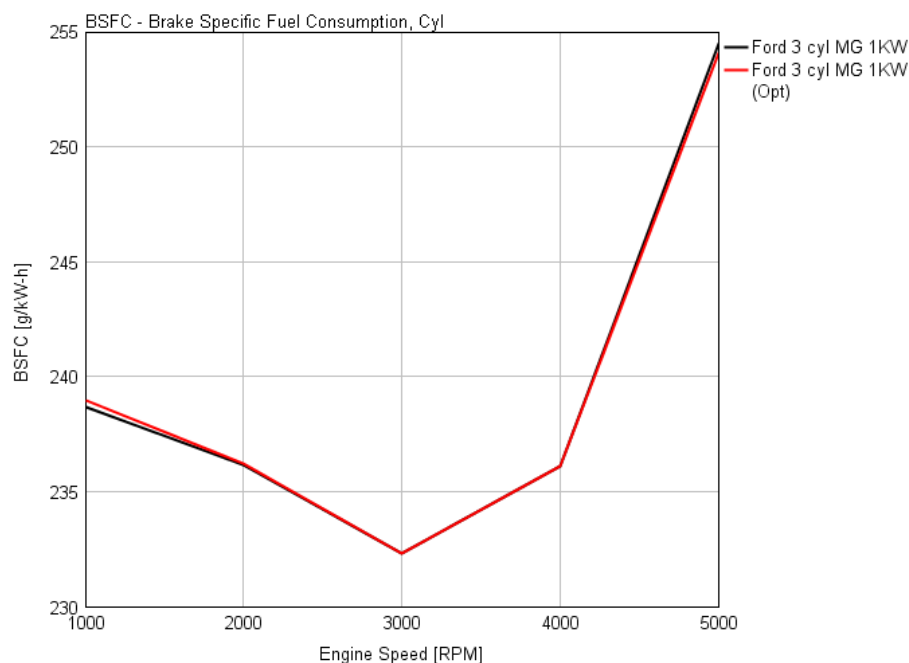
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Figure 19: Brake torque over engine speed for the initial and the optimised EAT 1kW models.

435

436 Similarly, as can be seen from Figure 20 the BSFC for both compressors are almost the same
437 except for the engine speed range of 1000-2000 rpm in which the BSFC of the optimised compressor
438 has been slightly increased. However, the results indicated that with the optimised compressor
439 there is an insignificant improvement in the engine performance.

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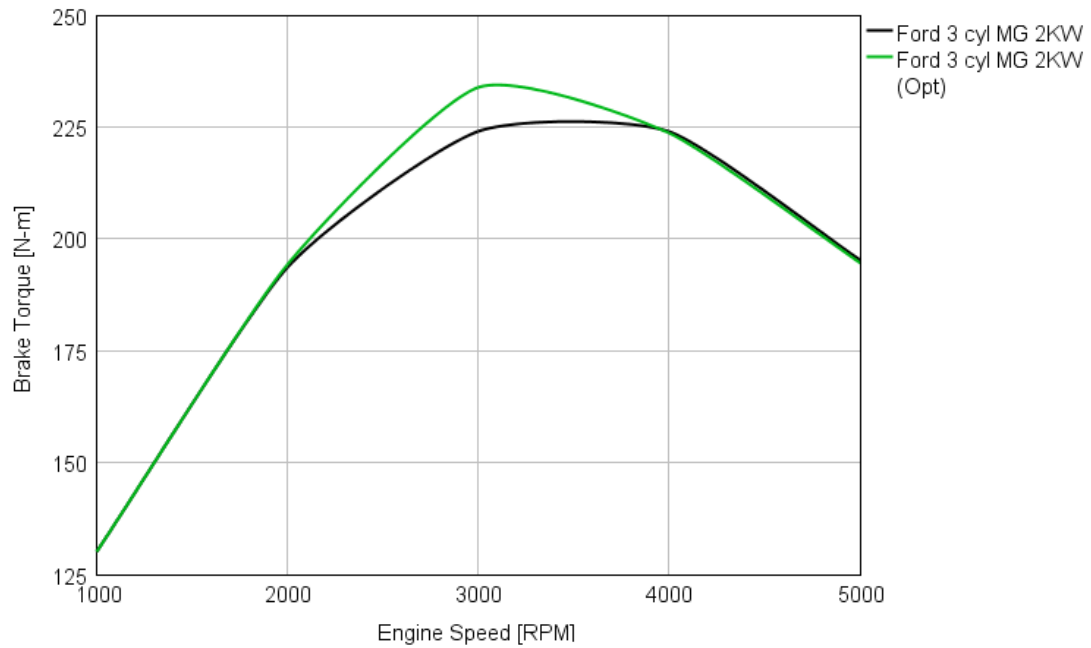
Figure 20: BSFC graphs for the initial and optimised EAT 1kW models.

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444 6.3 Comparison of the EAT with 2kW power level models

445 The optimised compressor in the EAT model with 2kW power level configuration has
446 significantly improved the overall performance of the engine. Especially, as shown in Figure 21,
447 from 2000 to 4000 rpm the engine torque of the optimised compressor has increased moderately
448 compared to the initial design of the compressor. On the other hand, at low engine speed the brake
449 torque for both compressors was the same while at high engine speed the initial compressor had
450 greater engine torque than the optimised compressor.

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Figure 21: Brake Torque graphs for the initial (black) and optimised (green) EAT 2kW models.

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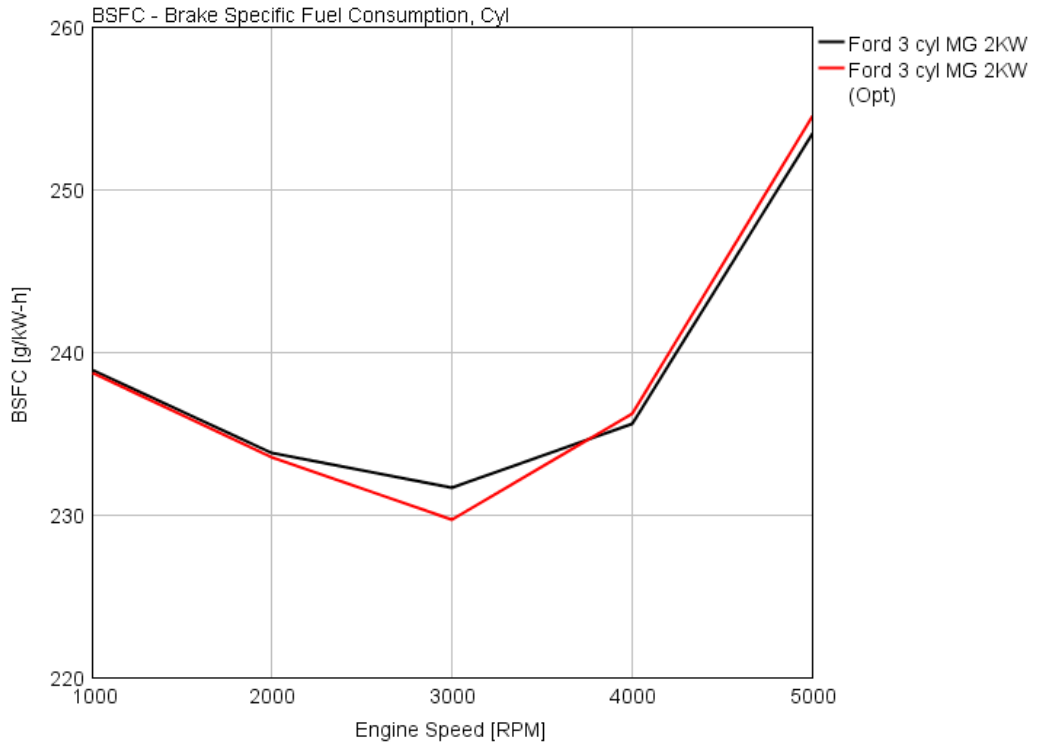
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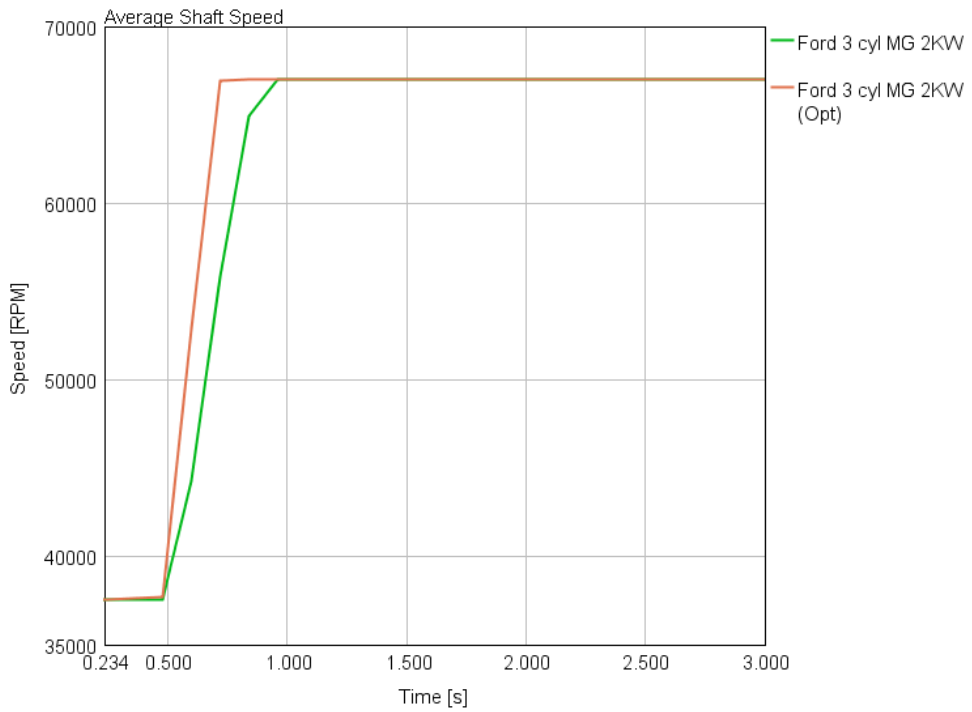
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The graphs in Figure 22 present the BSFC over the engine speed for the initial and the optimised compressor. It is evident that at the engine speed range of 2000 rpm-4000 rpm the brake specific fuel consumption (BSFC) has been considerably reduced with the optimised compressor while at high engine speed the brake specific fuel consumption (BSFC) is slightly increased compared to the initial compressor. Moreover, the results in Figure 23 indicate that the transient response time of the engine model with the optimised compressor has been reduced by 25%.



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Figure 22: BSFC graphs for the initial and optimised EAT 2kW models.

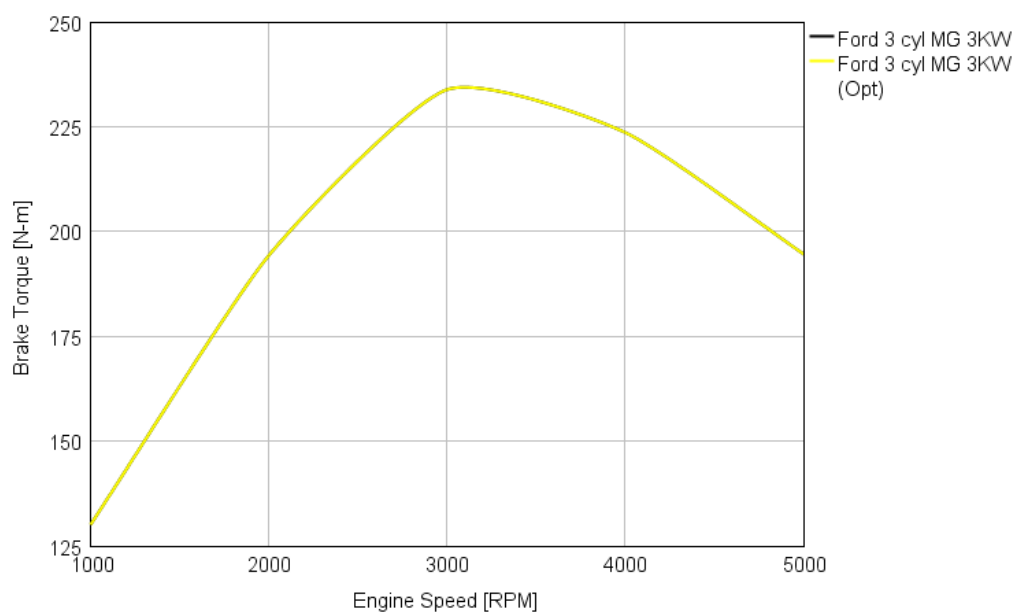


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Figure 23: Average Shaft speed of the electrically-assisted turbocharger (EAT) models with 2kW power level at 1000 rpm engine speed.

469 6.4 Comparison of the EAT with 3kW and 4kW power level models

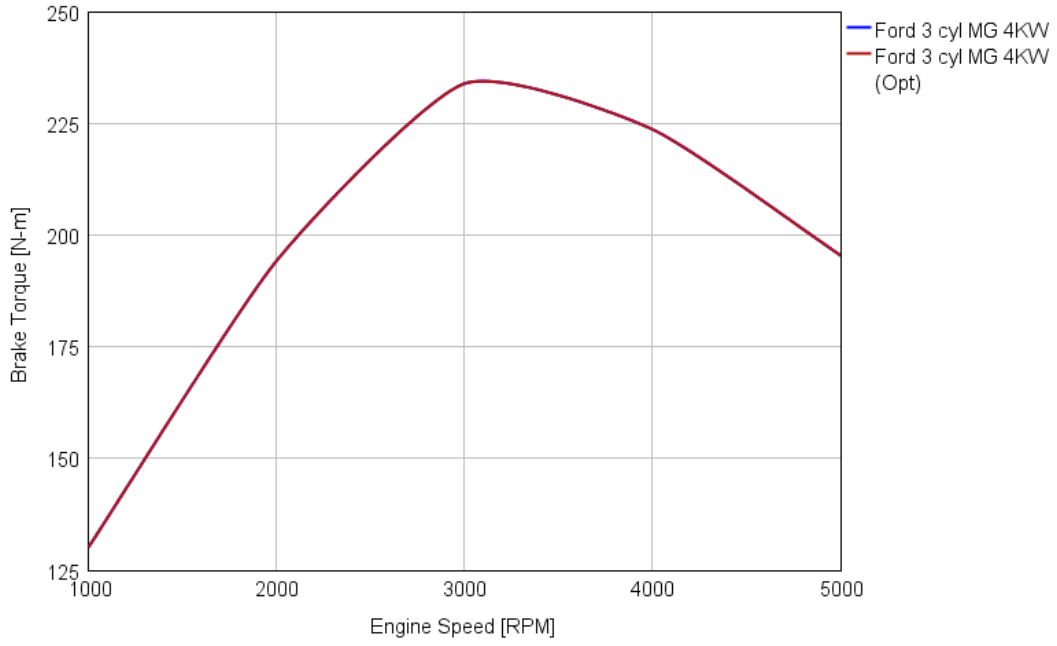
470 The results obtained from the GT-Power simulations show that in the electrically-assisted
471 turbocharger (EAT) models of 3kW despite the change on the compressor, the engine performance
472 remained steady. Similarly, it has been found that the electrically-assisted turbocharger (EAT)
473 model of 4kW with the optimised compressor has the same results with the initial compressor
474 model. In the following figures, some results of the 3kW as well the 4kW models are presented.
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477 **Figure 24:** Brake Torque of the electrically-assisted turbocharger (EAT) 3kW model for the initial and
478 optimised compressor.

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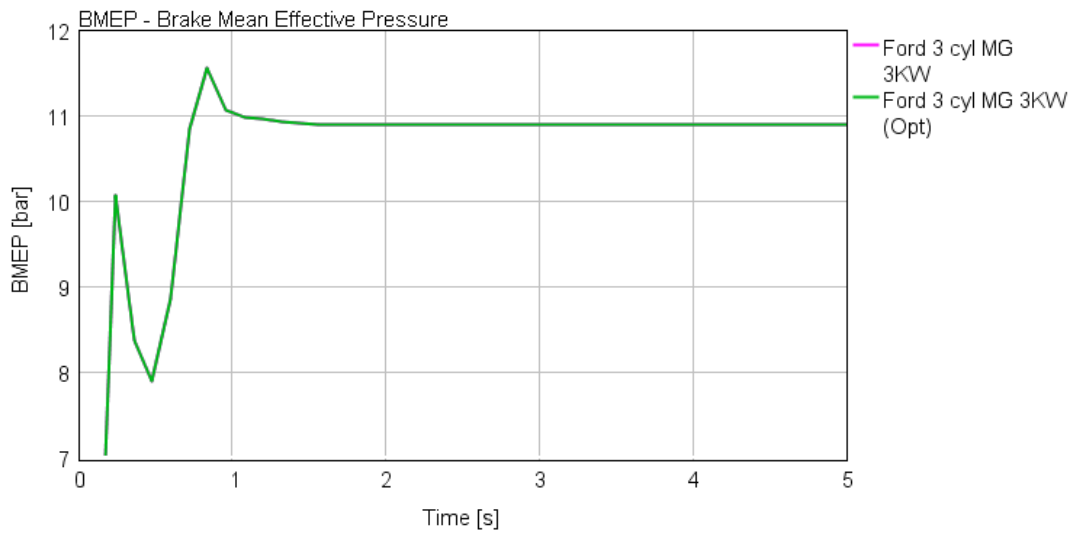
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Figure 25: Brake Torque of the electrically-assisted turbocharger (EAT) 4kW model for the initial and optimised compressor.

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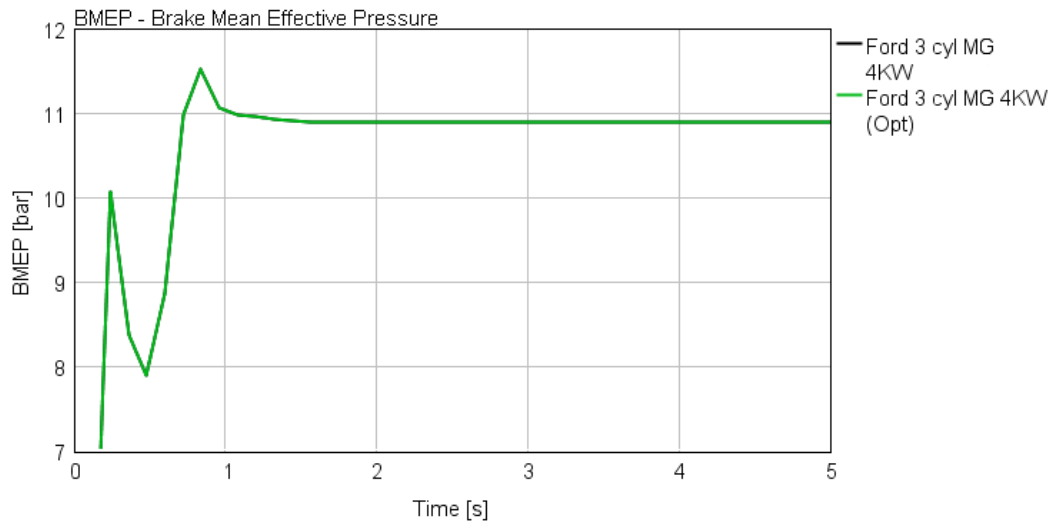
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Figure 26: Brake mean effective pump (BMEP) graphs of the EAT 3kW model for the initial and optimised compressor.

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Figure 27: Brake mean effective pressure (BMEP) graphs of the EAT 4kW model for the initial and optimised compressor.

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499 6.5 Comparison of the EAT with 5kW power level models

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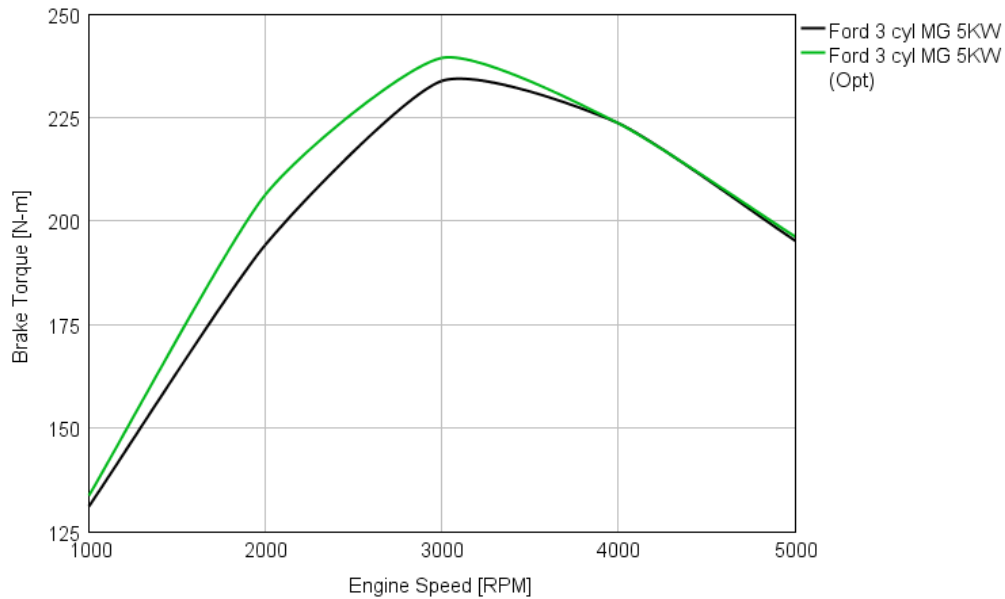
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As can be seen from the above figures, the results of the baseline and optimised compressor totally match for both modelling cases 3kW and 4kW. One plausible reason for the results that occurred could be an error in the simulations for these cases. Moreover, another explanation could be that in the optimised compressor the exit diameter of the impeller was slightly reduced, hence the change did not affect significantly the performance of the engine, as in the electrically-assisted turbocharger (EAT) 1kW case.

In this section, a comparison between the engine model with the optimised compressor and the engine model with the initial compressor was conducted. Figure 28, presents the brake torque of the two models versus engine speed. It is clear that the engine torque has considerably risen with optimised centrifugal compressor in place. Especially from 1000 rpm the engine torque is increased gradually until 3000 rpm where the engine torque is peaked at 293.2N. However, from 4000 to 5000 rpm the torque is decreased sharply and matches with the values of the torque of the engine model with initial compressor.



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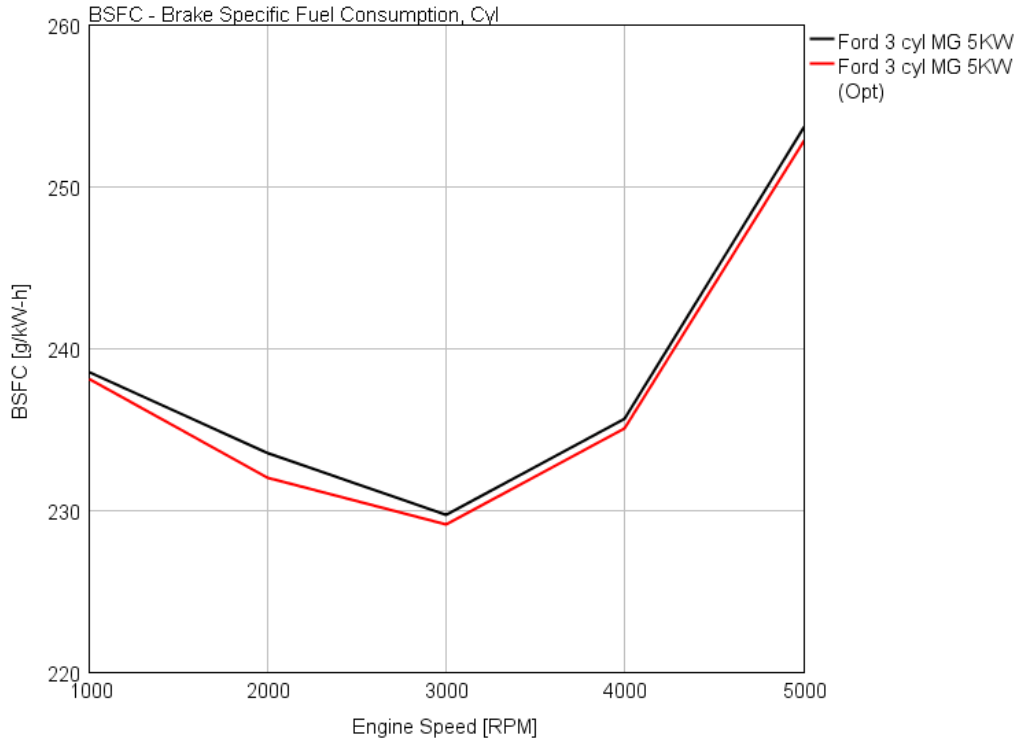
Figure 28: Brake Torque graphs for the initial (black) and optimised (green) EAT 5kW models.

510

511 In addition, the results in Figure 29 indicate that overall the brake specific fuel consumption
512 (BSFC) of the engine model with the optimised compressor has been considerably reduced
513 compared with the engine model with the initial compressor. Furthermore, the highest reduction in
514 the brake specific fuel consumption BSFC of the electrically-assisted turbocharger (EAT) model
515 occurred at low engine speed.

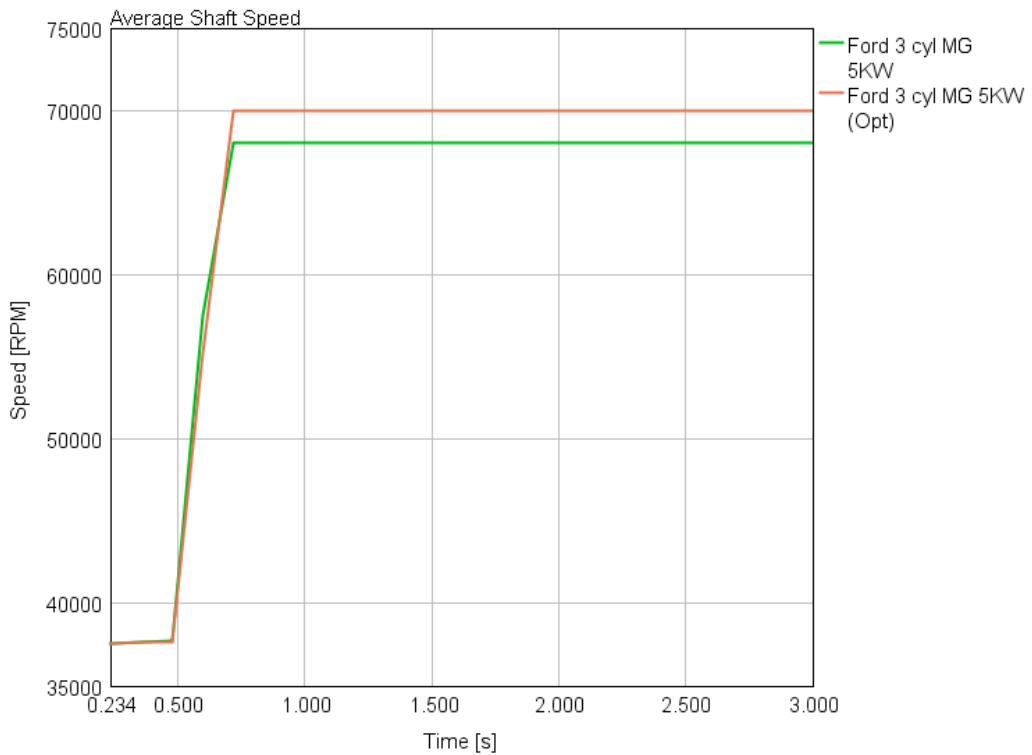
516 Moreover, Figure 30 outlines the average turbocharger shaft speed for the two engine models
517 against time. Although the transient response time remained steady with the optimised compressor
518 model, the maximum shaft speed has been increased by 3000 rpm as shown in Figure 28. Therefore,
519 the overall performance of the engine model with the optimised compressor has been improved.

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Figure 29: Brake specific fuel consumption (BSFC) graphs for the initial and optimized electrically-assisted turbocharger (EAT) 5kW models.



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Figure 30: Average Shaft speed of the electrically-assisted turbocharger (EAT) models with 5kW power level at 1000 rpm engine speed.

528 **7. Conclusion**

529 A design methodology is implemented based on authoritative loss models for the preliminary
 530 design of a centrifugal electrically-assisted turbocharger (EAT) compressor. The methodology
 531 extended to engine performance modeling for a conventionally and an electrically-assisted
 532 turbocharger (EAT)-equipped engine layout. The calculated compressor geometry arising from the
 533 proposed method was imported to the engine simulation software while the EAT-equipped engine
 534 was tested at five different power levels from 1kW to 5kW.

535 The results obtained from the engine simulations indicated that the electrically-assisted
 536 turbocharger (EAT) engine model improved the overall performance of the engine compared to the
 537 baseline engine. Moreover, the electrically-assisted turbocharger (EAT) equipped engine power
 538 output with 1kW and 5kW electrically-assisted turbocharger (EAT) power levels was increased by an
 539 average of 5.96% and 15.4%, respectively, from 1000 to 3000 rpm engine speed compared to the
 540 baseline model.

541 As far as the BSFC is concerned for the electrically-assisted turbocharger (EAT) model for 1kW
 542 and 5kW, there was an overall fuel consumption decrease of 0.53% and 1.45% depending on engine
 543 operating conditions, respectively, compared to the initial baseline engine model.

544 **Author Contributions:** Nikolaos Xypolitas and Mamdouh Alshammari were the research students that
 545 conducted the detailed study and wrote the first draft of this paper. Apostolos Pesyridis conceived of the project,
 546 created the layout of the investigations, and checked the computational outcome of the resultant modelling effort
 547 and subsequent discussion.

548 **Conflicts of Interest:** The authors declare no conflict of interest.

Nomenclature*Variables*

C_m	Meridional component of absolute velocity [m/s]	r	Radial velocity component
C_u	Tangential component of absolute velocity [m/s]	rms	Root mean square
C_p	Specific heat at constant pressure [J/(kg·K)]	Th	Throat parameter
N	Rotational speed of turbocharger [rpm]	Tr	Total relative

P Pressure [bar]

R Gas constant for air [J/(kg·K)]

r Radius [mm]

T Temperature [K]

Greek letters

γ Ratio of specific heats

Δ Difference

ρ Density [kg/m³]

$\bar{\omega}$ Non-dimensional pressure loss

Subscripts

0 Stagnation or total state; ambient condition

1-4 compressor stages

Acronyms

is, s isentropic
m Meridional velocity component

BMEP Brake mean effective pressure
BSFC Brake specific fuel consumption
EAT Electrically assisted turbocharger
MG Motor-Generator
RANS Reynolds Averaged Navier Stokes
rpm Revolution per minute

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Nomenclature

Variables

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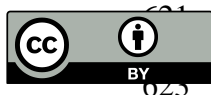
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