

Mechanical model based on a BVP for FRPs applied on flat and curved masonry pillars with anchor spikes

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Abstract

The use of fiber reinforced polymer (FRP) materials for strengthening interventions of existing constructions is a consolidated and widespread technique. In this context, although strengthening interventions generally involve curved masonry elements (arches, vaults, domes, etc.), only a few studies specifically concern the influence of the geometry curvature, or the effect of mechanical anchors (widely used in current practice for preventing premature failures), on the bond behavior of FRPs. The present paper proposes an interface exponential model for simulating the bond behavior of curved masonry pillars reinforced with FRP strips applied at the intrados or extrados by both epoxy adhesive and anchor spikes. The proposed model is based on a relatively simple boundary value problem (BVP) obtained by assuming for the spike a constitutive behavior under shear forces quantitatively deduced by post-processing the numerical data from a finite element micro-modeling approach previously proposed by the authors. The application of the proposed model to experimental cases carried out by the authors underline the stability of solution and the reliability of the proposed approach to account for the effect of both the curvature of the substrate and presence of the spike anchor on the bond behavior of FRPs.

Keywords: masonry; arches and vaults; CFRP reinforcement; anchor spike; non-linear Boundary Value Problem for ODEs; debonding.

1. Introduction

The preservation of the masonry-built heritage is nowadays considered a priority in highly civilized countries. Among the different typologies of masonry structural elements that the scientific community is trying to preserve against extreme events like earthquakes floods and storms, the most diffused is probably constituted by curved structures, such as arches, vaults, domes, etc..

41 Masonry behavior is well known for its main weakness in tension, exhibiting such kind of material
42 a very limited tensile strength with marked softening, an almost cohesiveness frictional behavior
43 and a fairly good resistance in compression, followed by a mild softening with good dissipation in
44 terms of inelastic energy for crushing.

45 As a natural consequence of such kind of behavior, for arches and curved structural elements in
46 general, it appears very suitable to insert in all those zones undergoing tensions, strengthening
47 elements capable to absorb tensile stresses that otherwise would result into the propagation of
48 flexural cracks, up to the formation of a failure mechanism which usually appears at early stages of
49 the application of the horizontal loads.

50 For this reason, gluing FRP strips on the surface of such kind of structures appears an extremely
51 interesting technology that deserves to be studied [1–6].

52 In general, at present, the literature in the field of FRP strengthening on masonry surfaces results
53 superabundant both from an experimental and numerical point of view [7,8,17–26,9,27–30,10–16],
54 having also at disposal a variety of technical recommendations especially useful for practitioners
55 involved in such kind of structural upgrading [31].

56 However, the studies on curved masonry substrates appear still relatively limited and
57 recommendations given by codes of practice are rather vague in this regard, leaving space for
58 further research in this field, both experimental and numerical [26-47].

59 In particular, it is still not very clear and quantitatively determined the role played by anchorage
60 devices applied near the free edges of a reinforcement.

61 The present paper moves its step from a previous combined experimental and numerical research
62 made on curved pillars reinforced with FRP and subjected to standard debonding tests, in presence
63 or absence of anchor spikes [43–46,48–53]. The main aim was to provide information on the
64 ultimate load carrying capacity and ductility of such kind of reinforcement when subjected to
65 standard debonding and to compare the global and local results maintaining the same geometries
66 and materials used in presence and absence of anchorage.

67 The paper here presented proposes an interface exponential model for masonry pillars reinforced
68 with FRP strips where the material properties of the interface are calibrated by best fitting of the
69 previously obtained experimental results. The presence of the anchor spike is modeled by assuming
70 for the spike a constitutive behavior under shear forces which is quantitatively deduced from post
71 processing of numerical data obtained by a previous research by the authors, where the debonding

72 process was modelled in ABAQUS FEM software assuming damage propagating in the bulk
73 substrate [44,52].

74 A relatively simple Boundary Value Problem (BVP) is obtained, where the presence of a
75 concentrated load in correspondence of the spike is dealt with a suitable and robust approach where
76 the concentrated load is substituted by a distributed tangential stress with equivalent properties.

77 A standard BVP solver based on finite difference is adopted, which showed excellent numerical
78 stability and efficiently in all the cases investigated.

79 Three different sets of mechanical properties are considered for the FRP/substrate interface and the
80 spike constitutive behavior. The first two models (called Set 1 and Set 3) assumes two interface
81 relationships formally equivalent to the data reported in Bertolesi and co-workers in [52] where
82 damage in the bulk was considered with different fracture energies in tension and compression for
83 bricks and mortar. The third model assumes for the interface Set 1 data and for the spike
84 constitutive behavior the trilinear relationship proposed by Grande et al. in [48], which turns out to
85 be independent from the curvature of the substrate.

86 In all cases, excellent agreement with experimental data and previously presented numerical models
87 is found [48,52], showing that the present simple approach can be used by any practitioner for a fast
88 and reliable prediction of the debonding behavior of FRP on masonry curved structures in presence
89 of anchor spikes. Finally, the role played by the spike is clearly reproduced by the procedure here
90 proposed, where it is underlined an evident activation at a relatively advanced state of deformation.

91 2. Brief overview of the experimentation carried out

92 A brief summary of the experimental study published in [46] is given below. The reader can refer to
93 the cited paper for further details. The experimental program involved twenty-five specimens,
94 subdivided into five series: CA-I-A, CA-E-A, CB-I-A, CB-E-A, C0-A; the first four series refer to
95 curved specimens while the last refers to flat specimens. Two different radii of curvature ($R = 1500$
96 mm, referred to as "CA" and $R = 3000$ mm, referred to as "CB") were considered in order to study
97 the behavior of the specimens in various geometric conditions.

98 All specimens were reinforced with a CFRP strip and equipped with a single anchor spike (referred
99 to as "-A" in the specimen's labelling). Some of the curved specimens were reinforced at the
100 intrados (labeled with "I") and the others at the extrados (labeled with "E"). All the specimens were
101 made of five bricks (dimensions $65 \times 120 \times 250$ mm³) by interposed four mortar joints of constant
102 thickness in flat specimens and variable thickness in curved specimens (minimum thickness 10
103 mm), as shown in Figure 1.

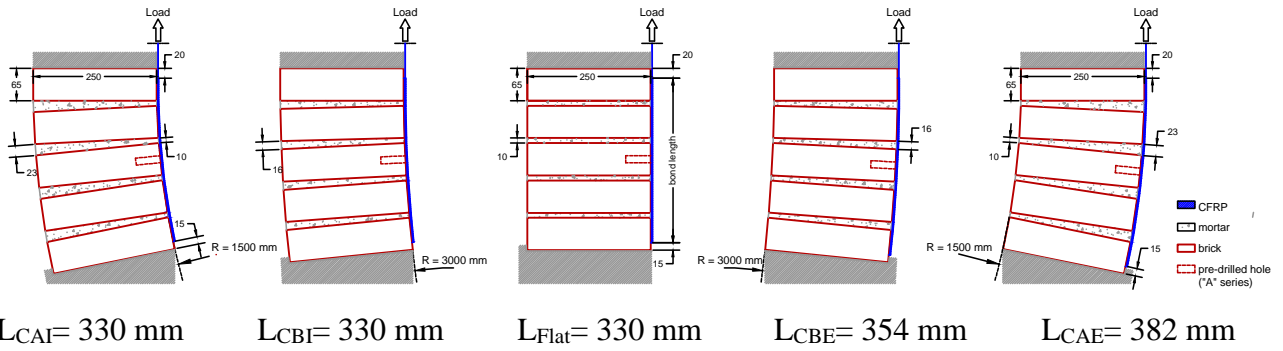


Figure 1: The five series of experimentally tested FRP reinforced curved and flat masonry pillars.

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106 The carbon fiber fabric sheet used to manufacture the specimens had a nominal thickness of 0.165
 107 mm (as declared by the supplier), a width of 100 mm and a length ranging from 330 mm to 382 mm
 108 according to the geometry of the surface to be bonded. Only a part of the sheet was bonded to the
 109 masonry, one part remained dry and the end part was glued to the steel loading device of the test
 110 machine. The length glued to the brick (L) is variable according to the geometry of the specimen
 111 (Figure 1) but is always greater than the effective bonding length ($L_{eff}=122\text{mm}$), calculated with the
 112 formula provided by CNR-DT200 [31] with reference to flat configurations.

113 The mechanical properties of the materials composing the reinforcement, declared by the
 114 manufacturer, are reported in Table 1 and those of the brick and mortar, obtained from an
 115 experimental investigation [46,51], are reported in Table 2.

116 Table 1: Mechanical properties of the reinforcing system components (declared by the
 117 manufacturer).

	Nominal thickness [mm]	Tensile elastic modulus [MPa]	Bending elastic modulus [MPa]	Ultimate tensile strain [%]	Characteristic tensile strength [MPa]	Shear strength [MPa]
Unidirectional carbon fiber fabric	0.165	240000	--	1.3	3200	--
Adhesive	--	--	2200	--	--	95

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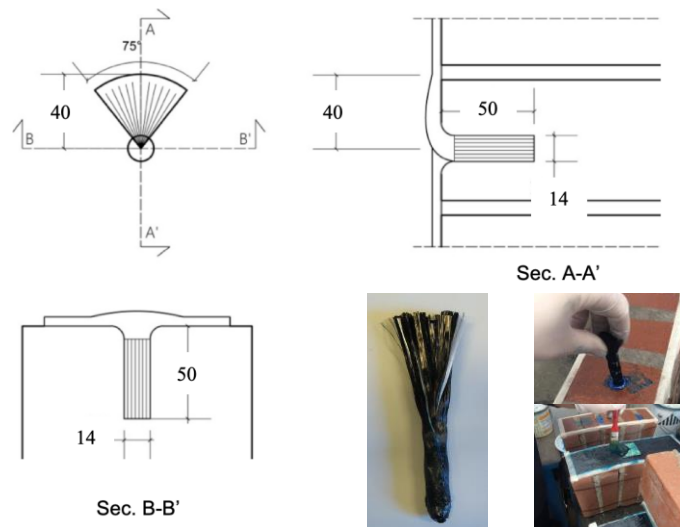
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120 Table 2: Mechanical properties of the bricks and mortar.

	n. specimens	Mean [MPa]	C.V. [%]
CLAY BRICK			
Compressive strength	6	19.90	5.11
Young modulus	6	8712	6.92
Direct tensile strength	6	2.49	16.90
Bending tensile strength	6	3.36	33.77
MORTAR			
Bending tensile strength	6	1.85	9.42
Compressive strength	12	5.18	8.212

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122 In all the specimens, the CFRP sheets were equipped with an anchor spike, also in CFRP, inserted
 123 at center of the central brick. The anchor was made by rolling up a rectangle (200x90 mm²) of
 124 carbon fiber fabric sheet. Before being rolled up, a part of the fabric measuring 35x200 mm² was
 125 pre-impregnated with resin. Thus, the cylinder obtained was inserted partly (impregnated portion)
 126 into a hole drilled in the center of the central brick and partly (dry portion) layered and fan-glued
 127 onto the carbon strip, as shown in Figure 2 [46].



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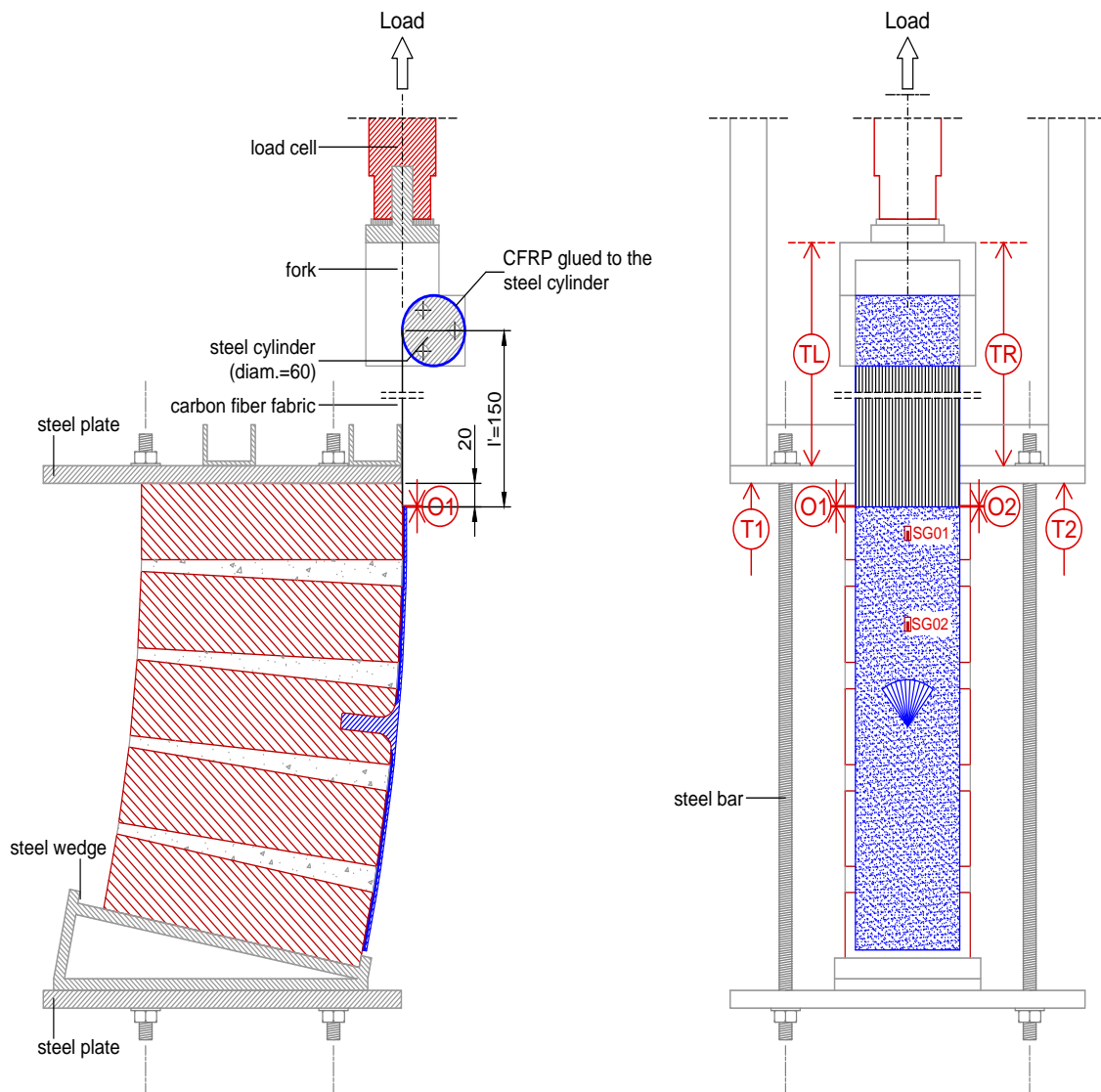
Figure 2: Spike anchors: geometric characteristics (measures in mm) [46].

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130 All the specimens were subjected to a single lap shear test: the masonry pillars were constrained by
 131 two steel plates and the force was applied to the end of the carbon sheet (see Figure 3).

132 The tests were carried out under displacement control; the displacement was increased
133 monotonically by means of an actuator equipped with a fork and a steel cylinder, to which the end
134 of the carbon fiber fabric sheet was glued.

135 The specimens were equipped with a load cell (50 kN), two “omega” transducers (“O1” and “O2”),
136 four displacement transducers (“TL”, “TR”, “T1” and “T2”) and two strain gauges (“SG01” and
137 “SG02”, applied to three specimens of each series) as showed in Figure 3.



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Figure 3: Test setup and instrumentation.

141 The overall behavior of the specimens can be described by analyzing the load-slip diagrams shown
142 in Figure 4, where \bar{s} in abscissa represents the sliding between the upper end (loaded side) of the
143 reinforcement and the masonry substrate. All the load-slip diagrams present a quasi-linear initial
144 branch, ending with the formation of the first cracks in the masonry substrate, near the loaded end

145 of the reinforcement (i.e. far from the anchor). After that, the diagrams show a load drop and a
146 subsequent ascending branch, much more scattered than the first one, until the maximum load is
147 reached. Then, the diagrams referring to flat specimens and to curved specimens reinforced at the
148 intrados show a sudden load decrease and a final post-peak branch corresponding to an average load
149 of about 30% to 55% (depending on the series) of the maximum load. The specimens of such series
150 mainly showed a cohesive failure mode, corresponding to the detachment of the CFRP sheet
151 because of fractures occurring in the substrate, a few millimeters below the composite
152 reinforcement, associated to the pull out of the anchor spike.

153 Differently, the diagrams referring to the curved specimens reinforced at the extrados show a
154 different behavior: these, in fact, exhibited a brittle failure mode occurring because of the tensile
155 failure of the dry carbon fiber fabric, outside the bonded zone. In this case, in fact, the anchor and
156 the stabilizing effect of the curvature prevented the detachment of the CFRP sheet reinforcement
157 during the test. It should however be noted that the average value of the maximum load of the
158 specimens reinforced at the extrados corresponds to about 50% of the nominal capacity of the dry
159 fabric declared by the manufacturer (see Table 3). This is due to a not perfectly uniform distribution
160 of the load in the dry fabric during the test.

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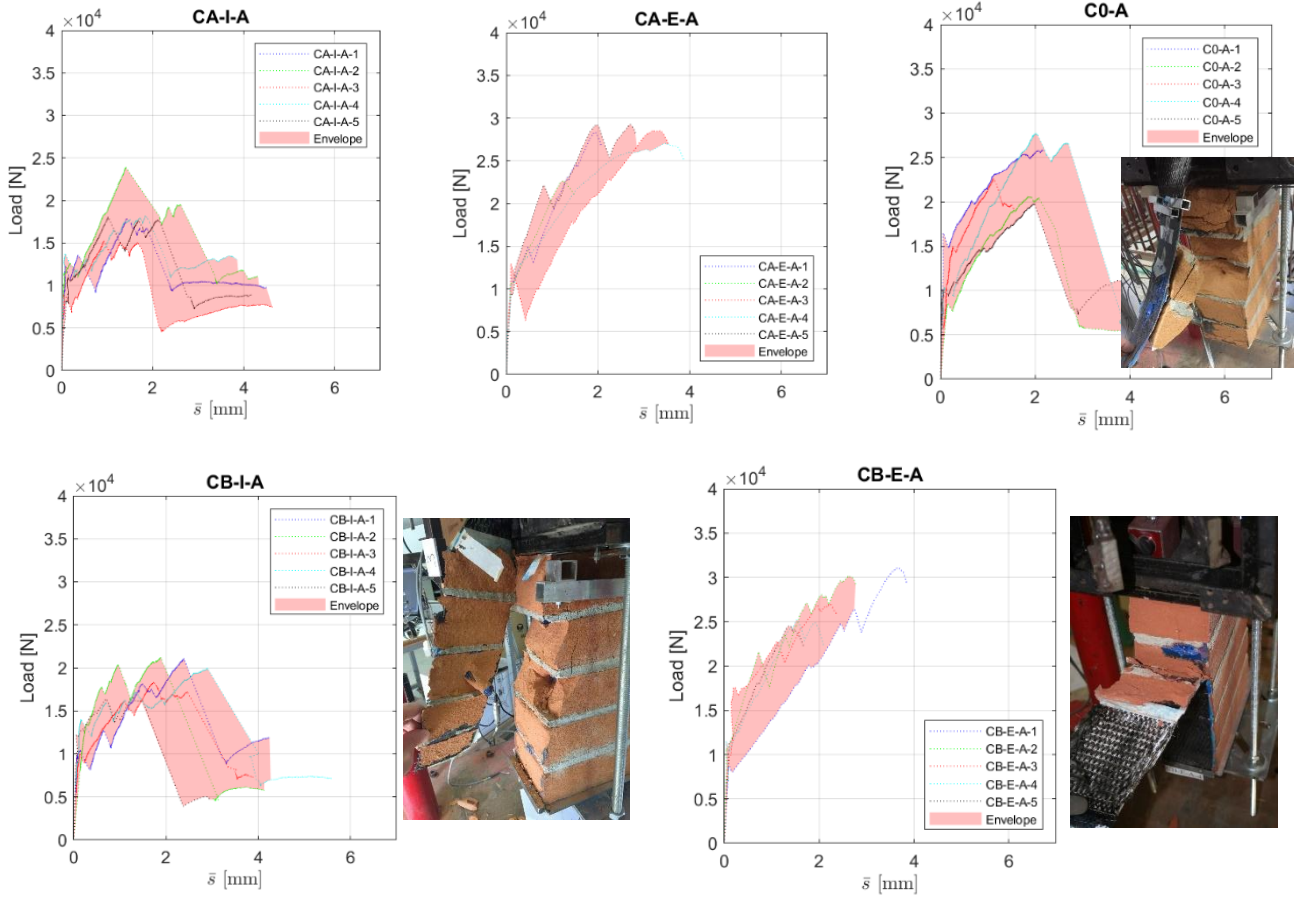


Figure 4: Experimental load-slip diagrams (slip \bar{s} refers to the loaded end of the reinforcement).

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2.1. Characterization of the CFRP-to-masonry interface

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The deformation values measured by the strain gauges externally glued to the reinforcement (SG01 and SG02 in Figure 3) were used to experimentally investigate the stress-slip behavior of the interface between the reinforcement and the substrate. Indeed, considering the schematization of the portion of reinforcement between SG01 and SG02 (see Figure 5), the average shear interface stress ($\bar{\tau}_{1-2}$) and normal reinforcement stress ($\bar{\sigma}_{1-2}$) components were evaluated enforcing simple equilibrium conditions through the following formulas:

$$\bar{\tau}_{1-2} = \frac{E_F(\varepsilon_1 - \varepsilon_2) t_F \cos \alpha}{\Delta x} \quad (1)$$

$$\bar{\sigma}_{1-2} = \frac{E_F(\varepsilon_1 + \varepsilon_2) t_F \sin \alpha}{\Delta x} \quad (2)$$

178

being:

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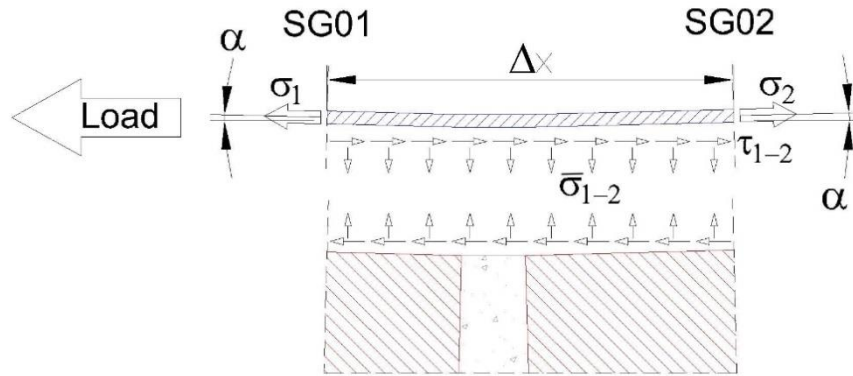
E_F = (homogenized) elastic modulus of the composite reinforcement in the fiber direction

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ε_1 e ε_2 = strain values measured with SG01 and SG02 respectively

181 t_F = thickness of the **strengthening material** (1 mm)

182 α and Δx = inclination of the reinforcement due to the curvature and distance between the
183 two strain gauges respectively



184

185 Figure 5: Scheme of the stress distribution at the reinforcement-masonry interface between SG01
186 and SG02.

187 Of course, normal stresses are negative (compressive) in the case of reinforcements bonded at the
188 extrados, positive (tensile) in the case of reinforcements bonded at the intrados and zero in the case
189 of flat specimens. The interface stress values referring to the portion of reinforcement between
190 SG02 and the anchor spike were evaluated using relations analogous to (1-2) assuming that the
191 deformation ε at the anchor was zero.

192 The slip values s_1 and s_2 corresponding respectively to the position of the strain gauges SG01 and
193 SG02 were evaluated assuming a linear deformation field between the two strain gauges and
194 between the anchor and SG02 and, in addition, that no slip occurred at the anchor. This last
195 hypothesis seems to be acceptable at least in the pre-peak phases of the tests, but it could lead to not
196 negligible errors in the final part, when the specimens are more damaged. For this reason, the data
197 corresponding to the last part of the tests were not considered for the evaluation of the slip and,
198 therefore, of the average interface stress components.

199 In agreement with the previously described hypotheses, it is easily calculated both the slips at the
200 strain gauges location:

$$s_2 = \frac{\varepsilon_2 \Delta x}{2} \quad (3)$$

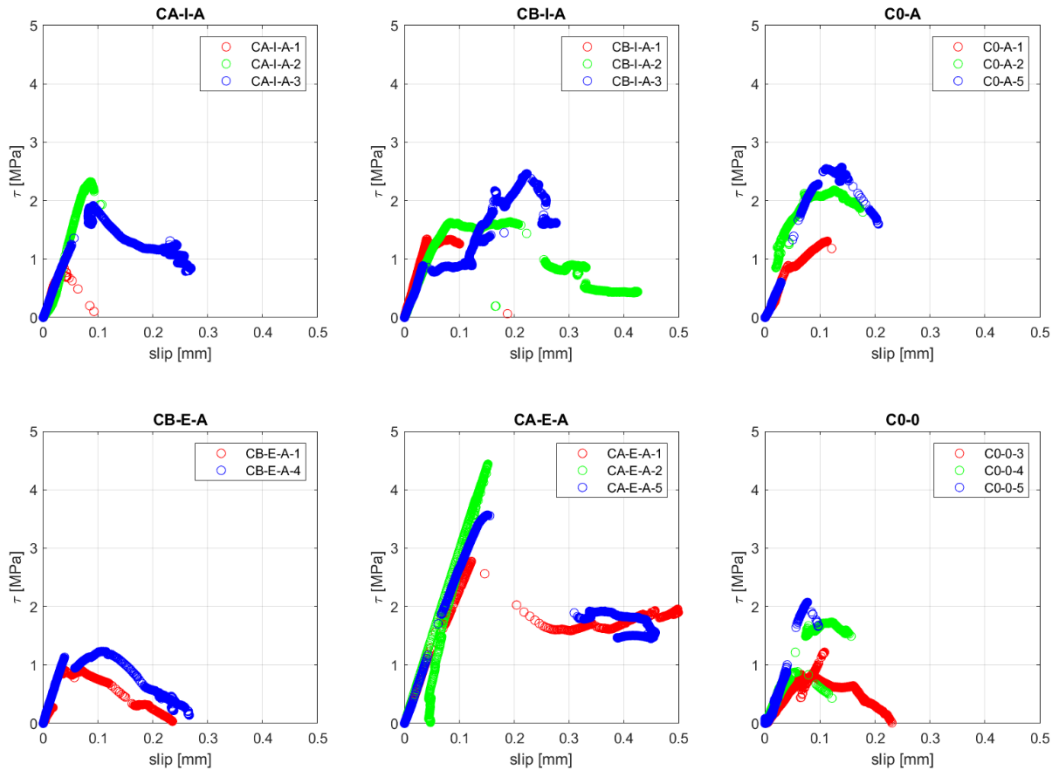
$$s_1 = s_2 + \frac{(\varepsilon_1 + \varepsilon_2)\Delta x}{2} \quad (4)$$

201 and so, the average slip values between SG01 and SG02 (\bar{s}_{1-2}) and between SG02 and the anchor
 202 (\bar{s}_{2-A}) have been evaluated as

$$\bar{s}_{1-2} = \frac{s_1 + s_2}{2} \quad (5)$$

$$\bar{s}_{2-A} = \frac{s_2}{2} \quad (6)$$

203 Shear stress-slip values obtained as described are represented in Figure 6, whilst in Figure 7 are
 204 reported the results in terms of normal stress-slip values. These results underline a good agreement
 205 with the analogous diagrams obtained by the Authors using the same procedure with reference to
 206 not anchored reinforcements [44,45,51]. Shear stress-slip diagrams of the non-anchored flat series
 207 (C0-0) have also been reported in Figure 6. Such diagrams have been obtained using five strain
 208 gauges, differently from C0-A series where only two strain gauges could be used. Therefore, since
 209 τ -slip experimental values corresponding to C0-0 series refer to a longer part of the reinforcement,
 210 these were considered suitable to calibrate the interface constitutive law as described below.



211
 212 Figure 6: Experimental τ -slip diagrams in presence of anchorage and without anchorage for only
 213 C0-0 series (flat case without anchorage).

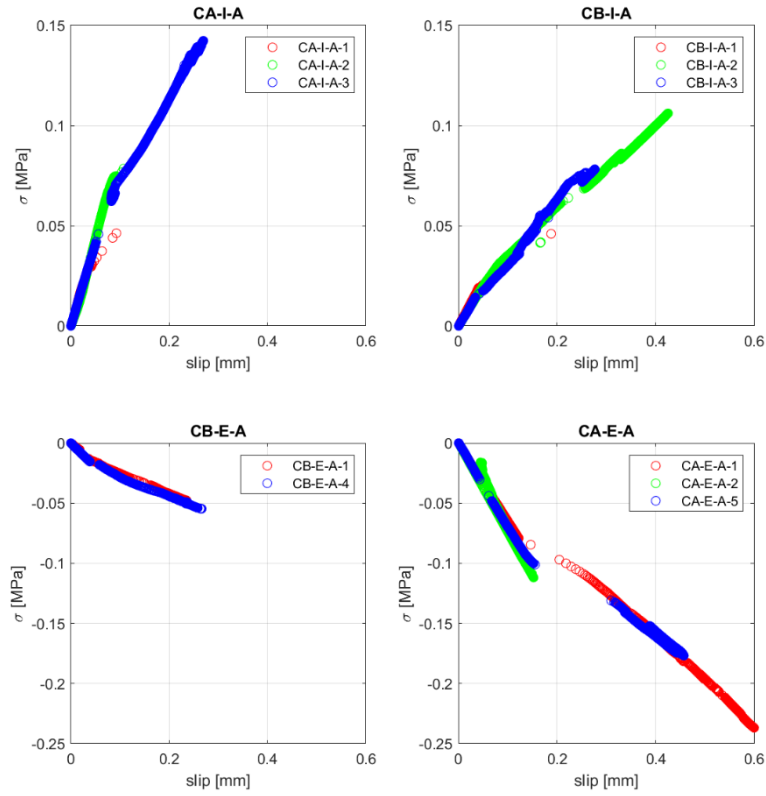


Figure 7: Experimental σ -slip diagrams.

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218 3. The mathematical model in brief

219 The mathematical model adopted in the computations presented in this paper is identical to that
220 proposed by the authors for the same specimens without anchor spikes, see [54]. The novelty here
221 relies on how the anchor spike is taken into account.

222 For the flat case without anchor spike, the Boundary Value Problem BVP which solves the
223 delamination problems is the following:

$$224 \quad t_F E_F \frac{d^2 s}{dx^2} - \tau(s, x) = 0 \quad \frac{ds}{dx} \Big|_{x=0} = 0 \quad s|_{x=L_F} = \bar{s}_0 \quad (7)$$

225 where t_F and E_F are FRP thickness and its Young Modulus, respectively, s is the slip between FRP
226 and substrate, x indicates the abscissa of a point of the FRP/substrate interface having defined the
227 free edge of the FRP strip as the origin of the frame of reference, $\tau(s, x)$ is the tangential stress field
228 acting at the FRP/substrate interface and L_F is the fiber bonded length; \bar{s}_0 is the slip at the loaded
end.

229 When an anchor spike of diameter d_s is connected to the support at a distance L_S from the x frame of
 230 reference origin, the same BVP of Eq. (7) must be solved, exception made that a fictitious
 231 tangential stress $\tau^*(s, x)$ is taken into account instead of $\tau(s, x)$ for all those points with abscissa
 232 x_p laying within the interval $L_S - \frac{d_s}{2} \leq x_p \leq L_S + \frac{d_s}{2}$.

233 $\tau^*(s, x)$ is evaluated knowing the force $F_s(s_s, L_s)$ acting on the spike in correspondence of a slip in
 234 the spike equal to s_s according to the following formula:

$$\tau^*(s, x) = \frac{F_s(s_s, L_s)}{B_F d_s} \quad (8)$$

235 where B_F it is FRP width.

236 It is worth mentioning that Eq. (8) is not easy to be solved numerically, because the computation
 237 of τ^* requires the knowledge of the slip in another position, albeit very near, i.e. that of the spike.
 238 However, assuming the dimension of the spike negligible when compared to the overall dimension
 239 of the glued length, it is possible to assume $s \equiv s_s$. Authors experienced that if the interval length
 240 where τ^* replaces τ is small enough (up to 2 times the diameter of the spike in the present
 241 simulations, i.e. up to roughly 1% of the length of the reinforcement), negligible differences in
 242 terms of global response are obtained among different choices of the length. On the other hand, very
 243 small values of the interval length lead to computational efforts of the BVP solver that grow
 244 exponentially. For this reason, a length equal to the diameter of the spike is adopted in the present
 245 simulations, thus obtaining a good balance between computational burden and reliability of the
 246 solver and, at the same time, maintaining geometrical consistency with the experimentation carried
 247 out.

248 As far as the curved cases without anchor spike are concerned, the Boundary Value Problem BVP
 249 which governs the delamination problem is the following:

$$t_F E_F \frac{d^2 s}{dx^2} - \tau(s, x, \sigma_n) = 0 \quad \left. \frac{ds}{dx} \right|_{x=0} = 0 \quad s|_{x=L_F} = \bar{s}_0 \quad (9)$$

$$\sigma_n = \frac{t_F E_F}{R} \frac{ds}{dx}$$

250 In Eq. (9), it is evident how the presence of a normal stress σ_n at the FRP/substrate modifies the
 251 BVP of the flat case only in the evaluation of tangential stresses τ , which typically follow a Mohr-
 252 Coulomb behavior, changing both peak strength and ultimate ductility depending if σ_n is positive or
 253 negative.

254 The presence of the anchor spike is managed exactly in the same way briefly discussed for the flat
 255 case. In absence of a comprehensive experimental characterization of the anchor spike behavior in
 256 presence of a tangential load (increase up to failure) and a constant normal stress to apply at the
 257 head of the spike, the constitutive behavior of the spike $F_s(s_s, L_s)$ is derived by post-processing the
 258 Abaqus simulations reported in [52], as it will be discussed later on.

259 The $\tau(s, x, \sigma_n)$ constitutive relationship assumed for the FRP/substrate interface is the following:

$$\tau(s) = (\tau_M - \tau^*) \frac{s}{s_0} e^{\frac{\rho}{2} \left[1 - \left(\frac{s}{s_0} \right)^2 \right]} + \tau_r \left[1 - e^{-\frac{\rho}{2} \left(\frac{s}{s_0} \right)^2} \right] \quad (10)$$

260 In Eq. (10), the symbols have the following meaning: τ_M and τ_r are the peak and residual stress, ρ
 261 is a non-dimensional parameter tuning softening (or alternatively fracture energy) and s_0 the slip at
 262 $\tau = (\tau_M - \tau^*)$. τ^* is a further stress constant value that tunes that the maximum stress τ_M occurs at
 263 a slip equal to s^* .

264 It can be easily shown that s^* is obtained as follows:

$$s^* = \frac{\tau_r \rho + \sqrt{(\tau_r \rho)^2 + 4\rho \left[(\tau_M - \tau^*) e^{\frac{\rho}{2}} \right]^2}}{2\rho(\tau_M - \tau^*) e^{\frac{\rho}{2}}} s_0 \quad (11)$$

265 According to [54], τ^* and s^* are obtained intersecting the following two functions:

$$f_1(s^*) \rightarrow \tau^* = \tau_M - \frac{\tau_M - \tau_r \left[1 - e^{-\frac{\rho}{2} \left(\frac{s^*}{s_0} \right)^2} \right]}{\frac{s^*}{s_0} e^{\frac{\rho}{2} \left[1 - \left(\frac{s^*}{s_0} \right)^2 \right]}} \quad (12)$$

$$f_2(s^*) \rightarrow \tau^* = \tau_M - \frac{\tau_r}{e^{\frac{\rho}{2} \left(\frac{s^*}{s_0} - \frac{1}{\rho \frac{s^*}{s_0}} \right)}}$$

266 A normal stress σ_n acting at the FRP/substrate interface modifies τ_M , s_0 and τ_r according to a
 267 Mohr-Coulomb relationship as follows:

$$\tau_M^* = \tau_M - \sigma_n \tan \Phi \quad s_0^* = \frac{\tau_M^*}{\tau_M} s_0 \quad \begin{cases} \tau_r^* = \tau_r - \sigma_n \tan \Phi & \sigma_n < 0 \\ \tau_r^* = 0 & \sigma_n \geq 0 \end{cases} \quad (13)$$

268 In practice, Eq. (13) replaces τ, s, τ_r with τ_r^*, s^*, τ_r^* in Eq. (10) when the normal stress at the
 269 interface is not zero.

270 It is interesting to point out how, from Eq. (13), slip s_0 varies linearly with τ_M^* , which is indeed a
 271 variation of the ductility which indirectly utilizes a Mohr-Coulomb failure criterion. As well known,
 272 indeed, displacements are independent from cohesive-frictional relationships, which refer
 273 exclusively to ultimate strengths. As a matter of fact, ruling slip parameter with the ratio $\frac{\tau_M^*}{\tau_M}$
 274 transfers on displacements the linear relationship adopted for stresses, which in this case is
 275 cohesive-frictional. Such approach was proved to be in agreement with the actual behavior of the
 276 FRP/substrate interface also as far as displacements are concerned, because ductility is increased
 277 linearly with σ_n according to a Mohr-Coulomb criterion.

278 $\tau(s)$ and $\tau^*(s)$ relationships can be suitably determined by means of numerical least-squares
 279 procedures where a best fitting of experimental data is performed, once the information provided by
 280 strain-gauges installed (if any) on the surface of the FRP strip is available.

281 As already discussed in the previous section, installing some strain gauges on the FRP surface
 282 allows to experimentally determine directly local strains of the fiber, say $\varepsilon_{F,SGi}$ on the i-th strain
 283 gauge SG , and hence indirectly the experimental $\tau(s) - s$ relationship to adopt.

284 The results of the non-linear best fitting have been already presented in [44,52], where the reader is
 285 referred. Here, we present only the obtained numerical parameters for the different coefficients τ_M ,
 286 τ_r, s_0, ρ and Φ (see Table 3) characterizing the $\tau(s)$ and its shape at $\sigma_n = 0; \pm 0.1; \pm 0.5 \text{ MPa}$,
 287 which correspond roughly to the flat case, CAI & CAE and CBI & CBE. The different curves are
 288 depicted in Figure 8 and they are compared with the bilinear relationships (which obey a Mohr-
 289 Coulomb failure criterion in the same way the present approach does) adopted by Grande and co-
 290 workers in [49].

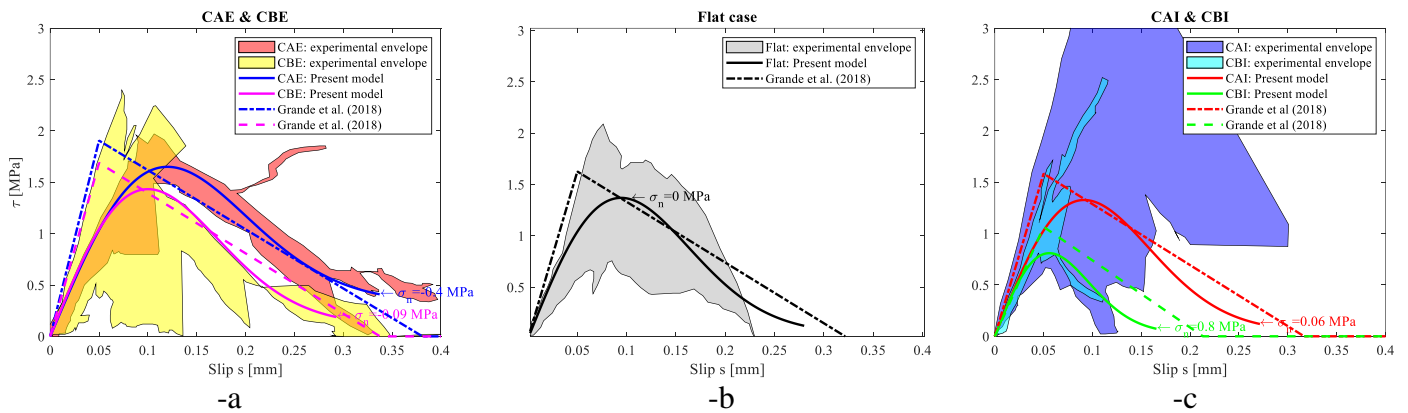


Figure 8: Interface τ -slip behavior assumed in the simulations and comparison with an existing bilinear relationship. -a: CAE &CBE. -b: Flat. -c: CAI & CBI

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Table 3: Parameters adopted to characterize the interface behavior by means of non-linear least squares optimization (corresponding to Set 1 mechanical properties adopted in [52]).

τ_M	τ_r	s_0	ρ	Φ
MPa	MPa	mm	-	°
1.42	0.02	0.031	0.088	35

294

295 It is also necessary to say a few words about the constitutive behavior to utilize for spike in the
 296 numerical simulations. First of all, it should be pointed out that only the shear behavior is required,
 297 according to Eqs. (7)(8). At present, it is still under study by the authors a comprehensive
 298 experimental campaign aimed at determining the constitute behavior of the spike, also in presence
 299 of different values of normal stress σ_n , in the range observed for CAI & CAE. However, in absence
 300 of such results, here the 3D Abaqus models used to analyze the pillars reinforced with FRP and
 301 anchor spike already presented in [52], are re-considered to find, through suitable post processing of
 302 the numerical results, the spike constitutive behavior provided by that models in terms of tangential
 303 force of the spike and slip of the point of application of the spike, i.e. $F_s - s_s$ relationship.

304 Indeed, the constitutive behavior of the spike anchor (i.e. $F_s - s_s$ relationship) can be extracted by
 305 considering that the external load applied to the FRP strip is transferred to the spike anchor inserted
 306 into the masonry pillar by a small area (i.e. red area depicted in Figure 9) connecting the external
 307 fan to the FRP anchor. Thus, $F_s - s_s$ relationship representing the spike behavior are obtained by
 308 trivial integration of the stresses on such section at different loading steps i:

$$F_s = \sum \sigma_i B_i t_i \quad (14)$$

309 Where σ_i is the vertical stress at node i acting on sections I, while $B_i * t_i$ is the influence area of
 310 node i located on the section (red area) depicted in Figure 9. Summation bounds and B_i depended
 311 on the FE mesh which in turn depends on the pillar geometry, whereas t_i was assumed equal to the
 312 FRP thickness. The slip s_s was obtained averaging the displacement monitored in each node of
 313 section I.

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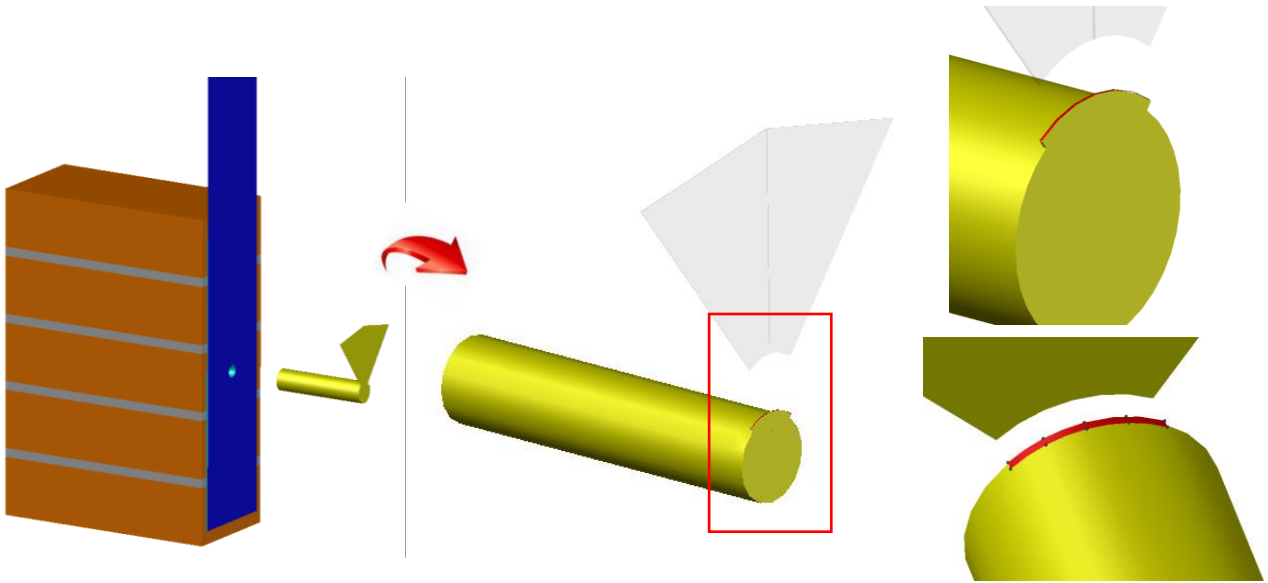


Figure 9: Post processing on Abaqus 3D FE models with spike to numerically deduce the spike constitutive behavior.

315

316 Applying the aforementioned procedure, $F_s - s_s$ relationships depicted in Figure 10 for the different
 317 cases are obtained. Only results for the Flat, CAE and CAI models are represented for the sake of
 318 conciseness. As it can be observed, in the Abaqus model proposed in [52], 3 different sets of
 319 mechanical properties are assumed for bricks and mortar, called Set 1, Set 2 and Set 3, see Figure
 320 11-a & -b. In the numerical applications here proposed, only Set 1 and Set 3 are taken into
 321 consideration, providing Set 2 intermediate results between the aforementioned ones. The three Sets
 322 of mechanical properties considered in [52] differ only for the fracture energies and the post-peak
 323 behavior adopted for the constituent materials. The remaining mechanical properties adopted
 324 correspond all to those experimentally evaluated in [46,51] (e.g. peak strength and elastic moduli).
 325 The behavior of the interface between FRP and substrate adopted here (see Table 3) is assumed in
 326 agreement to an experimental data fitting in absence of anchor spike, as discussed in detail in [44].
 327 Such approach does not have a direct link with the approach proposed in [44] for Set 1 and Set 3,
 328 but according to the results obtained, it may be considered reasonable to associate Set 1
 329 experimental data with the interface model of Table 3, since Set 1 data in [44] exhibit the best
 330 agreement with experimental global displacement curves. In order to find interface model
 331 parameters corresponding to Set 3, it may seem reasonable to proceed in terms of fracture energies.
 332 Consistently with such assumption, a new interface model is also used here which should
 333 correspond to Set 3 experimental data, assuming for the exponential model of Eq. (10) the needed
 334 parameters in terms of fracture energy. **Let us assume that the fracture energies in compression and**
 335 **tension for bricks in the model proposed in [44] are $\Gamma_{bc,3}$, $\Gamma_{bt,3}$ for Set 3 and $\Gamma_{bc,1}$, $\Gamma_{bt,1}$ for Set 1.**
 336 Their ratios ($\Gamma_{bc,3}/\Gamma_{bc,1}$ and $\Gamma_{bt,3}/\Gamma_{bt,1}$) turn out to be respectively equal to 3.02 and 0.88

337 respectively. It is assumed that for the interface model fracture energies corresponding to the Set 3
338 and Set 1 data are equal to $\Gamma_{bI,3}$ and $\Gamma_{bI,1}$. Their ratio $\Gamma_{bI,3}/\Gamma_{bI,1}$ is equated to
339 $\sqrt{\Gamma_{bc,3}\Gamma_{bt,3}/(\Gamma_{bc,1}\Gamma_{bt,1})}$, an assumption which allows to evaluate $\Gamma_{bI,3}$ and hence the corresponding
340 interface law, which should represent Set 3 data reported in [44]. The resulting parameters are
341 summarized in **Error! Reference source not found.** and the corresponding curve in absence of
342 normal stresses is shown in Figure 11-c, where also experimental data collected in absence of the
343 anchor spike are also represented with the trilinear law adopted in Grande et al. [49]. As it can be
344 noticed, the experimental data fitting is satisfactory also in this second case, meaning that all data-
345 sets assumed in [52] seem reasonable to describe the experimental behavior of the debonding
346 problem, assuming exclusively damage in the bulk of the substrate.

347

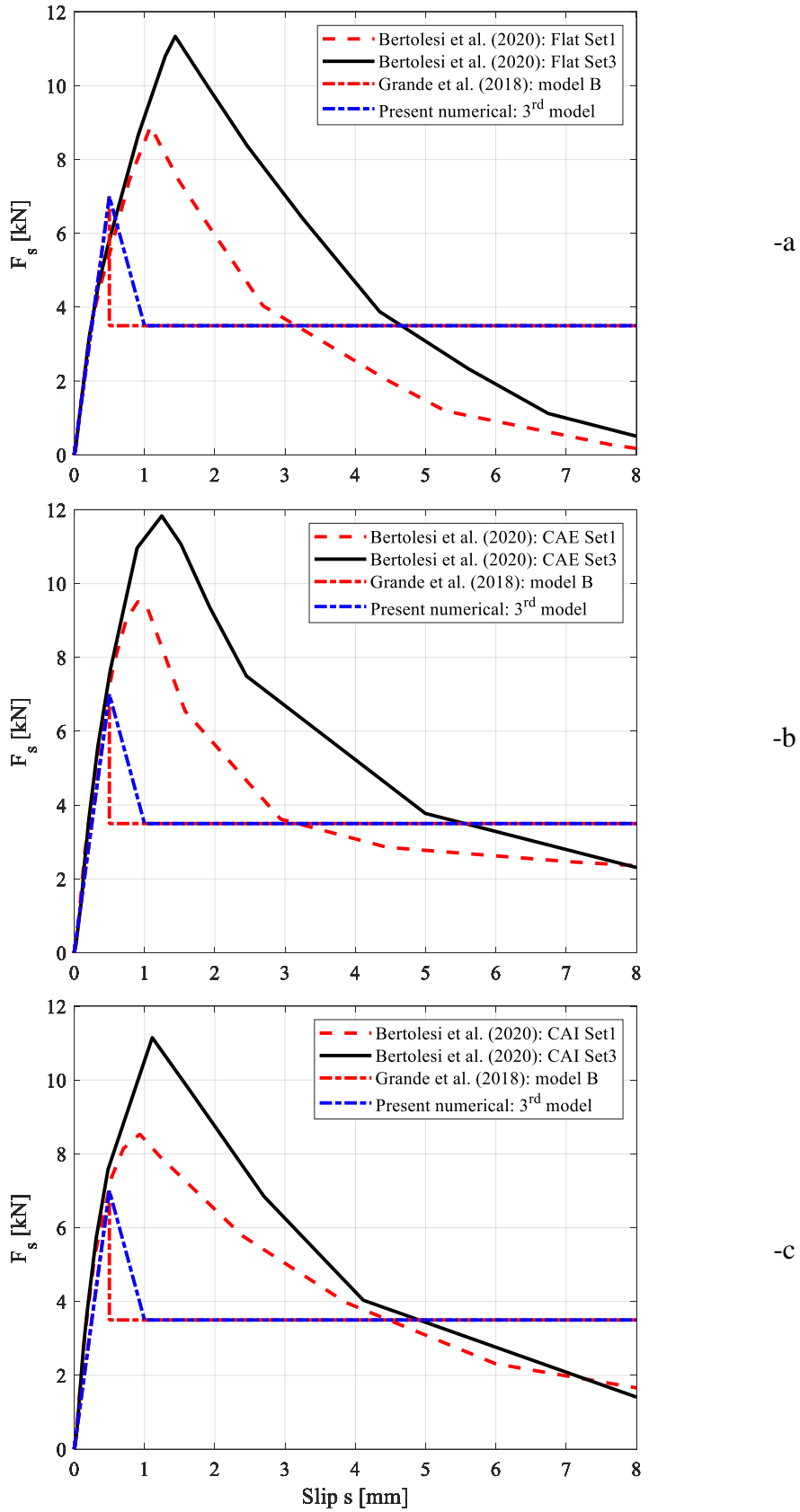


Figure 10: Spike $F_s - s_s$ shear behavior obtained from Abaqus FE post processed results (Set 1 and Set 3 mechanical properties) and hypotheses assumed here and by Grande et al. [48] -a: Flat. -b: CAE. -c: CAI.

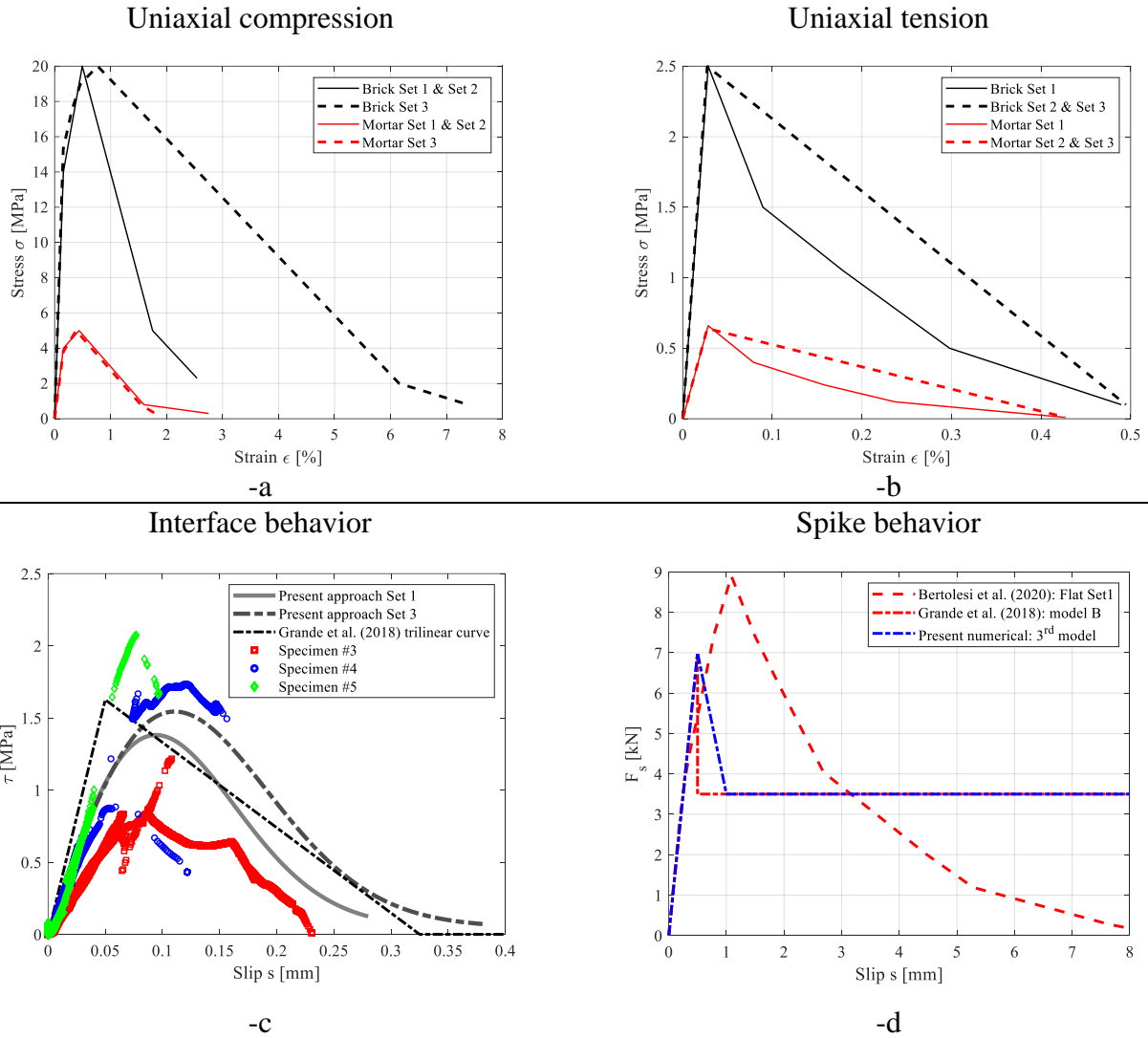


Figure 11:–a & b: assumptions made for bricks and mortar by Bertolesi et al. [52] in the heterogeneous FE approach (-a: compression; -b: tension). -c: interface behavior adopted in the present simulations and comparison with experimental data and previously presented trilinear approach. -d: Spike behavior adopted in the 3rd hypothesis in the present paper and comparison with Set 1 data by Bertolesi et al. [52] and numerical approach by Grande et al. [48].

349

350 4. Numerical results

351 In this Section, the results obtained with the numerical model previously discussed, in presence of
 352 anchor spike both for the flat and curved cases, are presented and critically compared with both
 353 experimental evidences and the results deduced from recent models developed by the authors to fit
 354 experimental data, relying on a FE discretization with springs [48]. In this latter approach, the
 355 presence of the spike in [48] is accounted for adding two uncoupled springs, one for the normal and
 356 one for shear action, respectively. Figure 10 depicts the spike shear force-slip ($F_s - s_s$) behavior
 357 obtained post processing Abaqus FE results from past investigation [52]. The interested reader is

358 referred to [52] for further detail of the model. Results are commented especially in light of their
359 dependence on the mechanical properties assumed for the anchor spike and the interface.

360 In Figure 12, the global force-displacement curve obtained with the proposed model for the flat case
361 (continuous black curve for Set 1 data and dashed black curve for Set 3) are compared with both
362 experimental data and two previously presented numerical approaches by the authors (one with
363 damage in the bulk [52], the other with non-linear interfaces between substrate and FRP based on a
364 FE discretization with springs [48]).

365 As it can be observed, the agreement with previously presented numerical approaches is very
366 satisfactory, meaning that both the FRP/substrate interface law and the spike anchor behavior are
367 model suitably grounding on convincing assumptions. It is particular evident the point of activation
368 of the spike, where a sudden change in the load carrying capacity and in its first derivative are
369 observed. Full delamination occurs obviously at different displacements because the interface laws
370 adopted differ for the fracture energies assumed, so that Set 3 is more ductile than Set 1. As
371 expected, the interface behavior is globally very similar to that obtained with damage in the bulk
372 (compare respectively the continuous black curve with the dashed red one and the dashed back
373 curve with the dashed green one, which correspond in pair to Set 1 and Set 3 mechanical properties
374 of [52]). In the Abaqus model proposed in [52], obviously there is not that limit in the ductility
375 which reflect in the global behavior with a sudden drop of the load carrying capacity, because of the
376 fact that damage occurs exclusive in the bulk of the substrate and softening spreads progressively
377 inside the volume, excluding the possibility of sudden drops of the global load carrying capacity. In
378 any case the global results appear in very good agreement with both Abaqus approach by Bertolesi
379 et al. [52] and the interface model by Grande et al. [48].

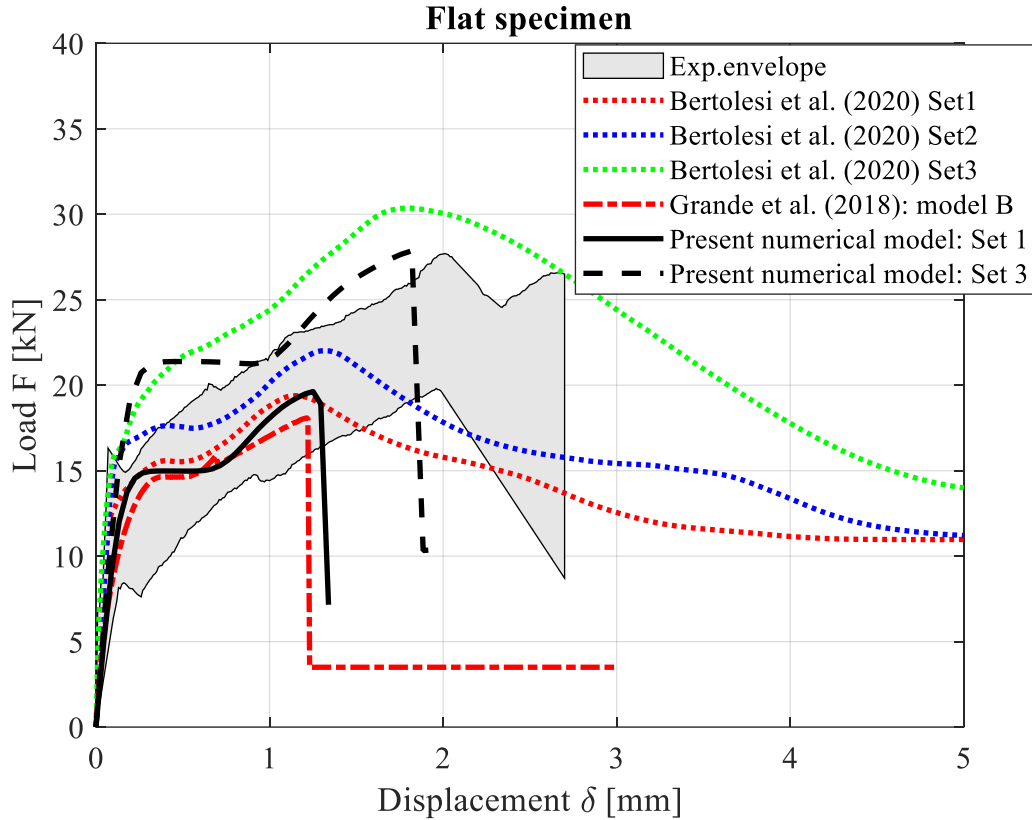


Figure 12: Flat specimen, global load-displacement curves. Comparison among present model, Grande et al. [48] numerical model, Bertolesi et al. [52] numerical model and experimental data.

380

381 As far as the evolution of interface slip and the stresses at the interface between FRP and substrate
 382 and on FRP are concerned, Figure 13 depicts such results at progressive values of displacement s_0
 383 applied at the loaded edge. In particular, subfigure -a depicts the interface slip function at increases
 384 s_0 values, subfigure -b tangential stress at the interface and subfigure -c fiber tensile stress. Left
 385 column refers to Set 1 mechanical properties, whereas right column to Set 3 mechanical properties.

386 First of all, the stability of the algorithm is worth noting, with an evident excellent robustness even
 387 for s_0 values corresponding to full debonding, i.e. in that range where global softening is
 388 particularly severe. This is certainly an intrinsic advantage of the numerical procedure adopted,
 389 especially when compared with a standard FE approach. Second, the role played by the anchor
 390 spike is particularly evident, especially as far as the normal stress plot in the FRP strip is concerned
 391 (subfigure -c). It is indeed quite noticeable the jump of the normal stresses in correspondence of the
 392 spike, which increases in a visible manner only at reasonably large values of the displacement
 393 applied at the loaded edge, i.e. near 0.7 and 1 mm for Set 1 and Set 3, as deducible from the global
 394 behavior reported in Figure 12.

395 Figure 14 shows for the flat specimen the numerically obtained force-slip curves on spike: Set 1
 396 results are represented by a continuous black curve, whereas Set 3 by a dashed line. As immediately

397 visible, the spike activates with its maximum load carrying capacity late during the deformation
398 process and failure occurs for the total debonding of the FRP from the substrate rather than for a
399 failure of the spike itself, which exhibits finally a drop in the load carrying capacity because the
400 local slip on the spike is not linearly dependent by the applied displacement s_0 at the loaded edge, a
401 feature which makes the model bypass the potential peak load carried by the spike.

402 Finally, in Figure 15 a comparison between present numerical predictions and experimental
403 evidence for strain gauges SG01 and SG02 are depicted, as usual representing Set 1 data on the left
404 and Set 3 on the right column. It is worth mentioning that experimental shear stresses on the interval
405 between spike and SG02, deduced from SG02 experimental data, are always linear because it was
406 made the simplistic hypothesis of assuming the spike subjected to a slip equal to zero. This
407 notwithstanding the experimentally deduced information is useful, because it provides a rough
408 estimation of the stiffness of the reinforcing system in correspondence of the interval between SG02
409 and the spike. Obviously, shear and normal stresses deduced from strain gauges information refer
410 for SG01 to the interval between SG01 and SG02, and for SG02 to the interval between SG02 and
411 the spike. Such procedure is obviously repeated identically for CAI, CAE, CBE and CBI specimens.
412 For the sake of brevity from now ongoing, the Authors will refer directly to strain gauges instead of
413 the aforementioned intervals.

414 After a careful analysis of the results obtained, it can be affirmed that Set 1 interface behavior
415 reproduces better the elastic phase, whereas Set 3 data seem much more in agreement with
416 experimental evidences in the non-linear range. According to authors opinion such result maybe
417 could be a consequence of the better quality of the specimens assembled in case of the anchor spike,
418 which exhibit also an experimental interface behavior better than that observed for the specimens
419 without anchor spike, compare for instance Figure 6 and Figure 8-b.

420

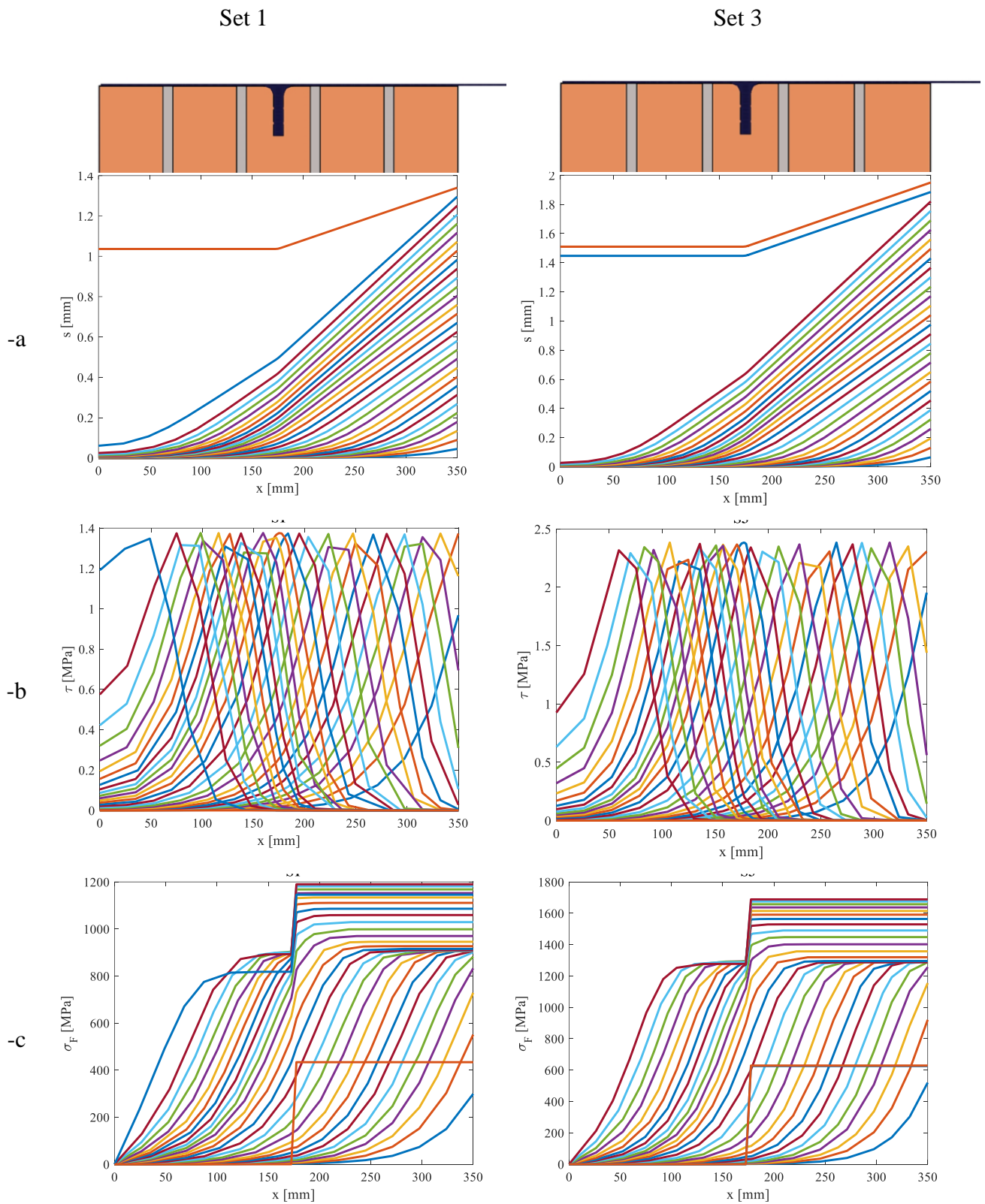


Figure 13: Flat specimen. –a: abscissa x-slip diagram –b: abscissa x- tangential stress at the FRP substrate interface. –c: abscissa x- normal stress in FRP reinforcement. Left column: Set 1 mechanical properties. Right column: Set 3 mechanical properties.

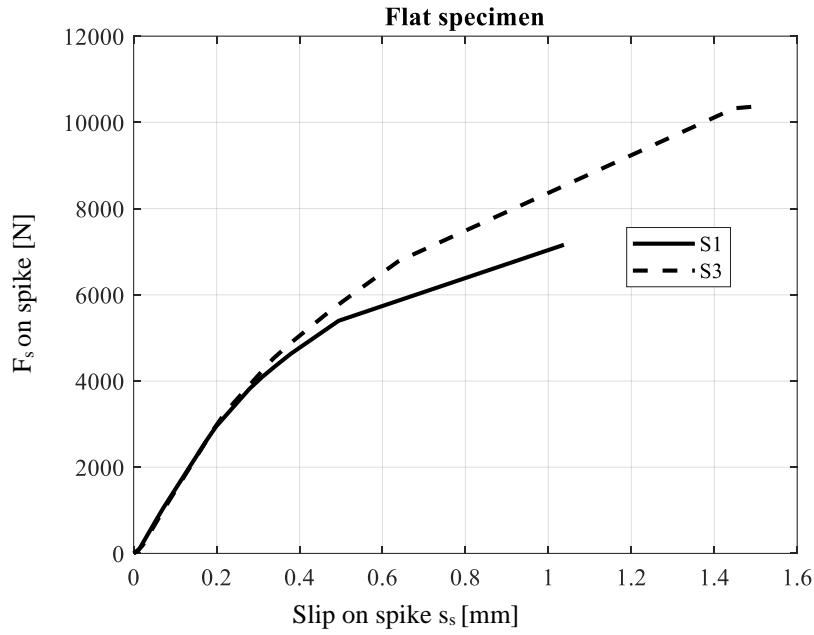


Figure 14: Flat specimen, force-slip curves on spike obtained numerically.

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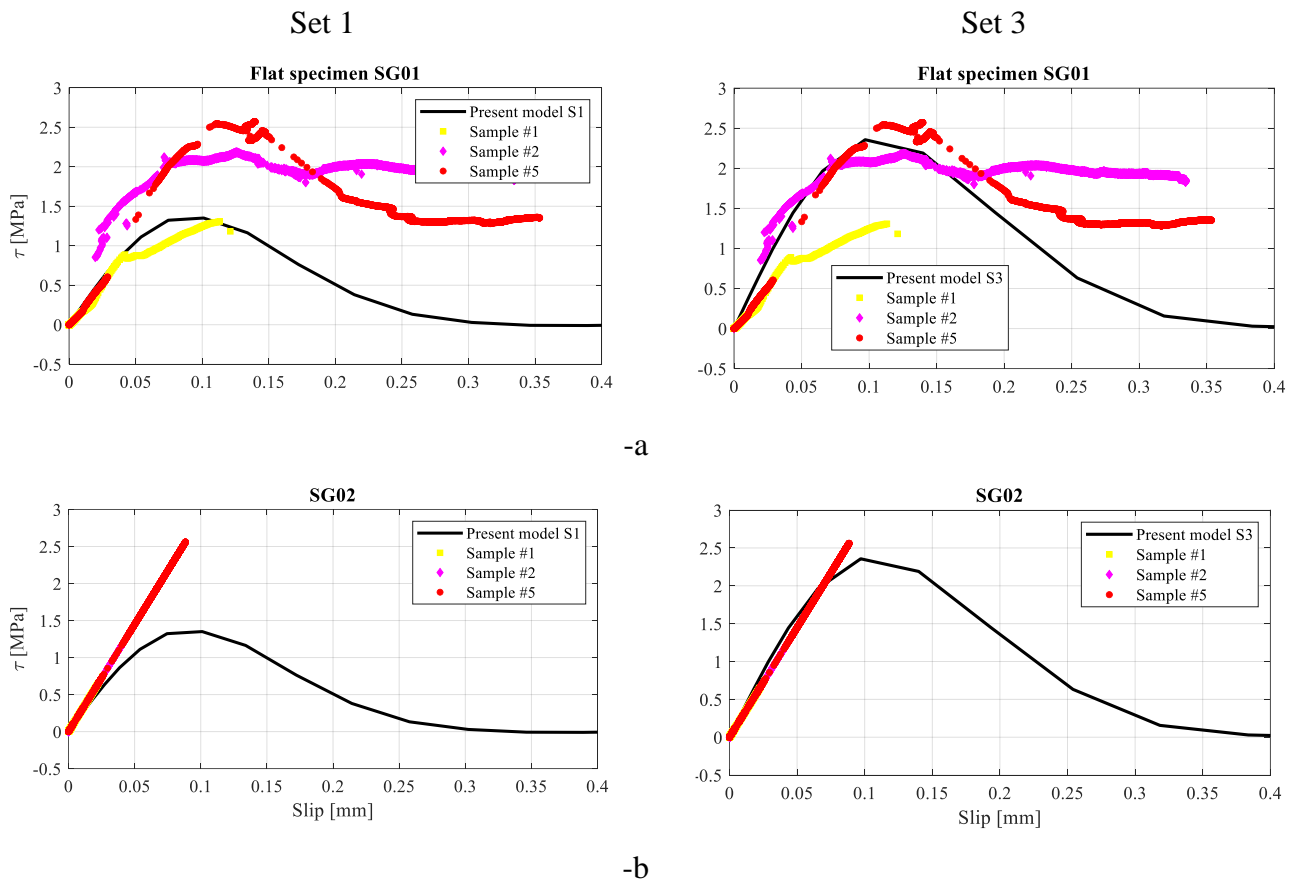


Figure 15: Flat specimen with anchor spike, tangential interface stresses. Comparison between numerical prediction (Set 1 left and Set 3 right) and experimental data provided by strain gauges SG01 (-a) & SG02 (-b).

423

424 As far as the specimens with flat configuration are concerned, global force-displacement curves
425 obtained for CAI, CAE, CBI and CBE are shown in Figure 16, again showing experimental data
426 envelopes and previously presented numerical models.

427 From an overall analysis of the global results obtained, the same considerations done for the flat
428 case can be repeated:

- 429 - The activation of the spike occurs late during the test, with a visible increase of the load
430 carrying capacity and in its first derivative.
- 431 - Set 1 and Set 3 data assumed in this paper for both the spike and the interface result in curve
432 appearing in very good agreement with those presented by Bertolesi and co-workers in [52].
433 This is not surprising, because the assumptions made in the present paper are in reasonable
434 agreement with those done in [52], where however a model with damage in the bulk of the
435 substrate was used, without any kind of non linearity at the FRP substrate interface.
- 436 - The sudden drop of the load carrying capacity visible in the present model is again a
437 consequence of the full debonding at the interface, which is obviously not visible in a model
438 where damage is diffused in the bulk.
- 439 - Set 1 results are also in excellent agreement with Grande et al. model [48], since the only
440 differences between the approaches are two: (i) a slight different behavior of the interface
441 (exponential here and trilinear in [48]) and (ii) a different constitutive model assumed for the
442 spike (here derived from Abaqus post processed results and in [48] assumed elasto-fragile).
- 443 - As far as the effect of the curvature is concerned, it may be pointed out how the intrados
444 reinforcement is less effective even in presence of anchor spike, with peak strength and
445 ductility sensibly lower than those found for the specimens reinforced at the extrados,
446 especially at high curvatures. The exploitation of the spike is indeed sensibly lower for CAI
447 if compared to that for CAE, see for instance Figure 17, where the numerically obtained
448 force-slip curves on spike are represented for both sets of material properties assumed in the
449 four different cases of curvatures analyzed. Such results are intuitively in agreement with a
450 Mohr-Coulomb behavior of the interface, which is subjected to tensile normal stresses for
451 CAI and CBI and to beneficial compression for CAE and CBE. Rather noticeable is finally
452 the agreement between present numerical predictions and experimental evidences reported
453 in [46], also considering the unavoidable scatter exhibited by experimental data in presence
454 of a limited number of replicates.

455 From Figure 18 to Figure 21, interface slip s (subfigures –a), tangential stress τ (subfigures –b),
456 FRP axial stress σ_F (subfigures –c) and interface normal stress σ_n (subfigures –d) at different

457 values of the applied displacement s_0 at the loaded edge are depicted. Figures refer in order
458 respectively to CAE (Figure 18), CBE (Figure 19), CAI (Figure 20) and CBI (Figure 21). The same
459 considerations done for the flat case can be repeated here, with an evident role played by the spike
460 in increasing locally the load carrying capacity at relatively large displacements applied at loaded
461 edge.

462 It is also interesting to notice how the curvature plays a beneficial role when the reinforcement is
463 applied at the extrados, because compression stresses arise at the FRP/support interface, especially
464 for CAE, with an observed peak normal stress exceeding 0.17 and 0.24 MPa at the loaded edge
465 immediately before delamination for Set 1 and Set 3 respectively, see Figure 18-d.

466 The opposite behavior is observed for the reinforcement glued at the intrados, especially for a large
467 curvature of the interface (CAI), where a positive normal stress of about 0.12 and 0.18 MPa is
468 present immediately before debonding for Set 1 and Set 3 respectively, see Figure 20-d. Since the
469 interface obeys as a Mohr-Coulomb failure criterion, both the load carrying capacity and the
470 ductility of the strengthened curved pillar become larger with reinforcement at the extrados, where
471 the interface compression normal stress plays a beneficial role.

472 The comparison between the numerical prediction deduced by accounting for Set 1 and Set 3
473 mechanical properties and experimental data provided by strain gauges is shown in Figure 22 to
474 Figure 25 for the specimens with anchor spike. The plots generally show, in the majority of cases, a
475 good agreement between numerical and experimental results, consistent with the outcomes of the
476 specimens without anchor spikes.

477 Finally in Figure 26, assuming for the spike the simplified tri-linear relationship shown in Figure
478 11-d and labeled as “3rd present numerical model”, the global load-displacement curves obtained for
479 the flat case, CAE, CAI, CBE and CBI are depicted and compared with both experimental envelopes
480 and numerical predictions by Grande et al. [48]. Not surprisingly, the results are almost
481 superimposable to those found in [48], because both the interface model and the spike shear force-
482 displacement relationship adopted in two approaches are very similar.

483 As can be observed in Figure 10, the constitutive model assumed in this latter case for the spike is
484 quite different from that deduced by Abaqus computations and there is no variability with the
485 curvature of the specimen, as it should be. This notwithstanding, the results appear fully in
486 agreement with the experimental data, simply because the activation of the strength of the spike
487 depends on the local slip of the anchorage in the different cases and the peak strength of the device
488 is never reached, as clearly visible by Figure 17 in the case of the four curved surfaces investigated.

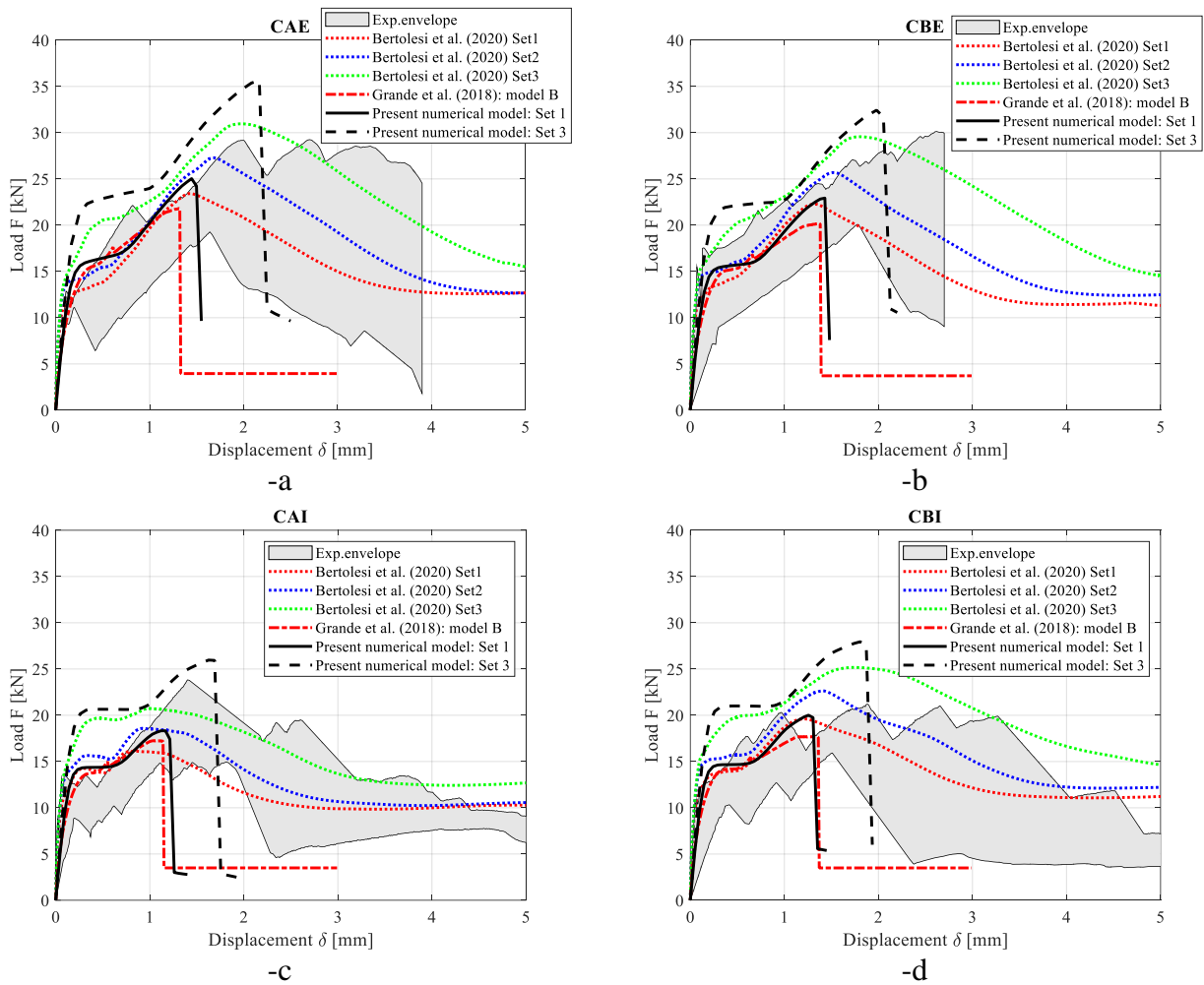


Figure 16: Global load-displacement curves obtained with the present approach (continuous black curve: Set 1; dashed black curve: Set 3). Comparison among present model results, experimental evidences and previously presented models. -a: CAE. -b: CBE. -c: CAI. -d: CBI.

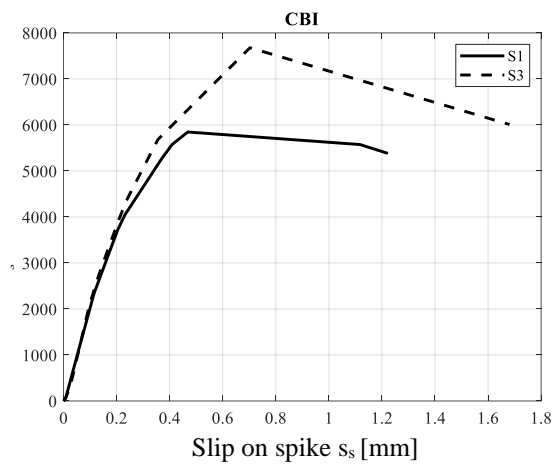
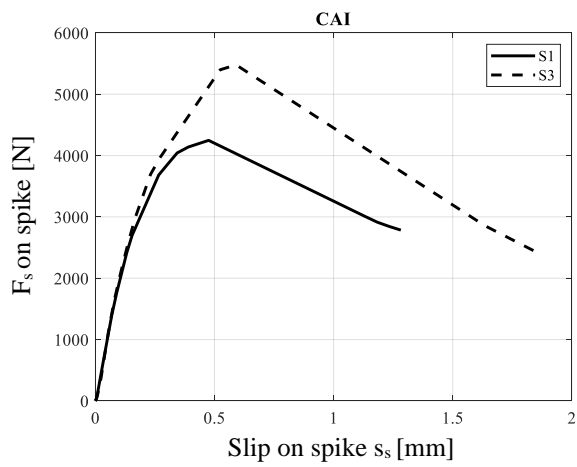
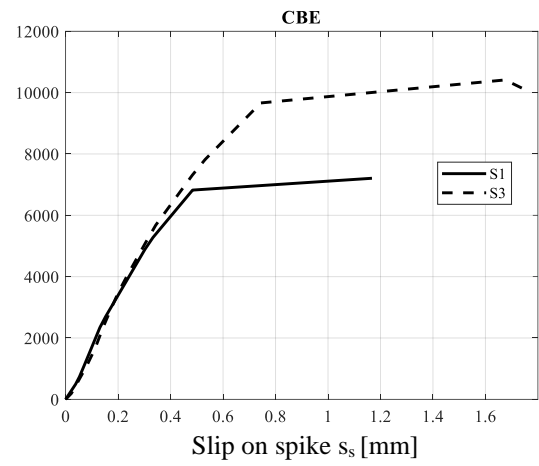
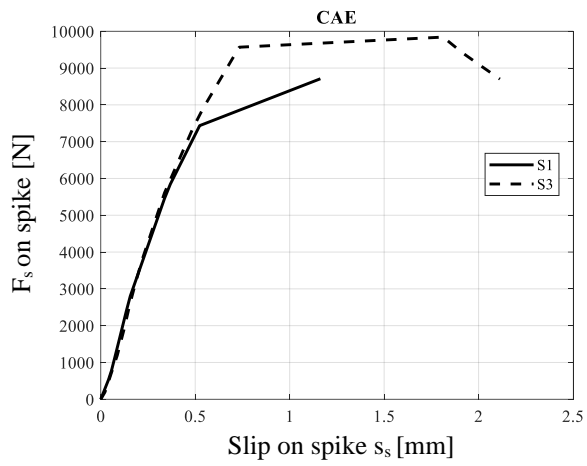


Figure 17: Spike slip-shear force behavior in the numerical model proposed (continuous black curve: Set 1; dashed black curve: Set 3).-a: CAE. -b: CBE. -c: CAI. -d: CBI.

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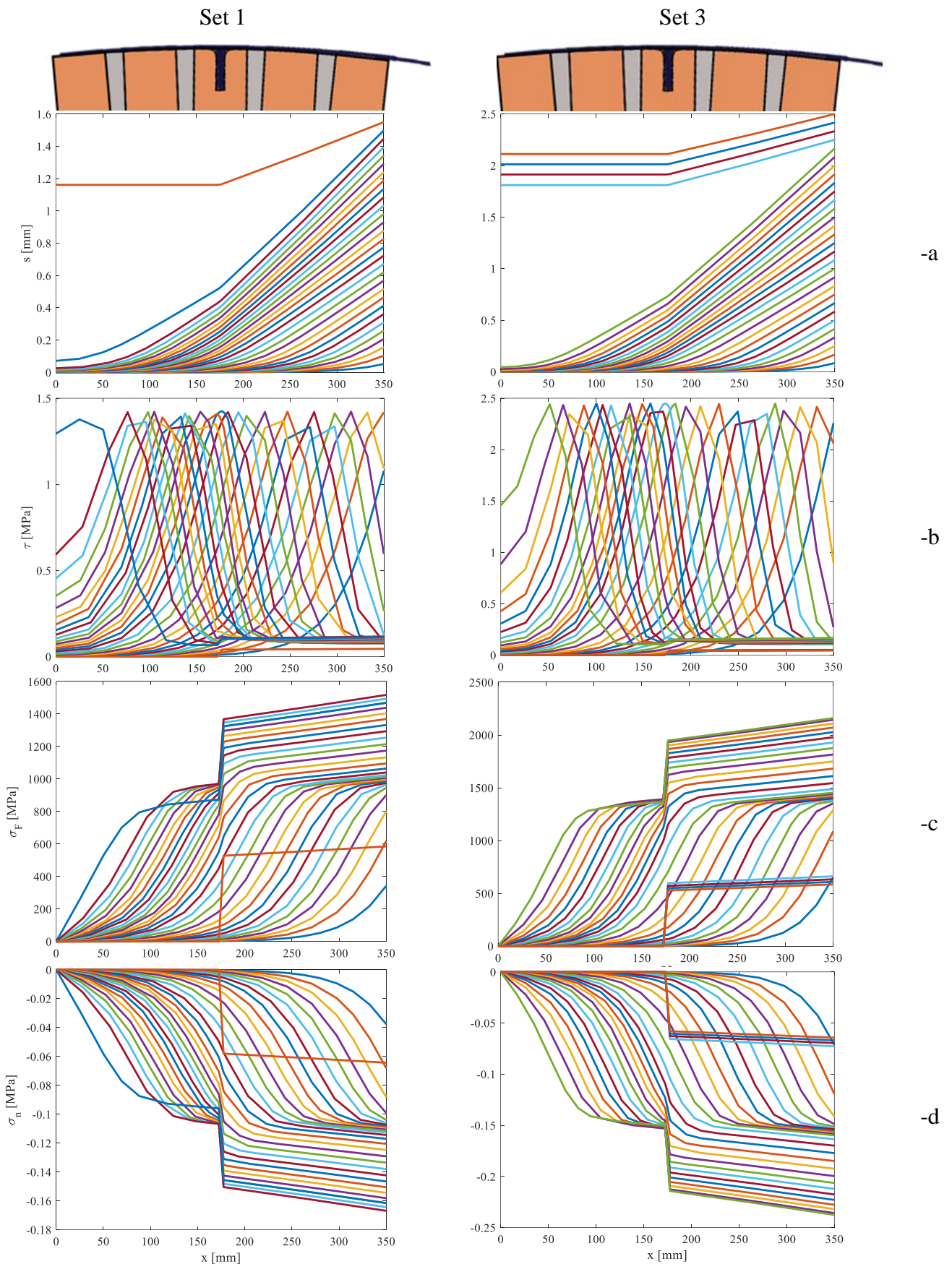


Figure 18: CAE with anchor spike, -a: x-slip diagram -b: x- tangential stress at the FRP substrate interface. -c: x- normal stress on FRP. -d: x- stress normal to the FRP/substrate interface.

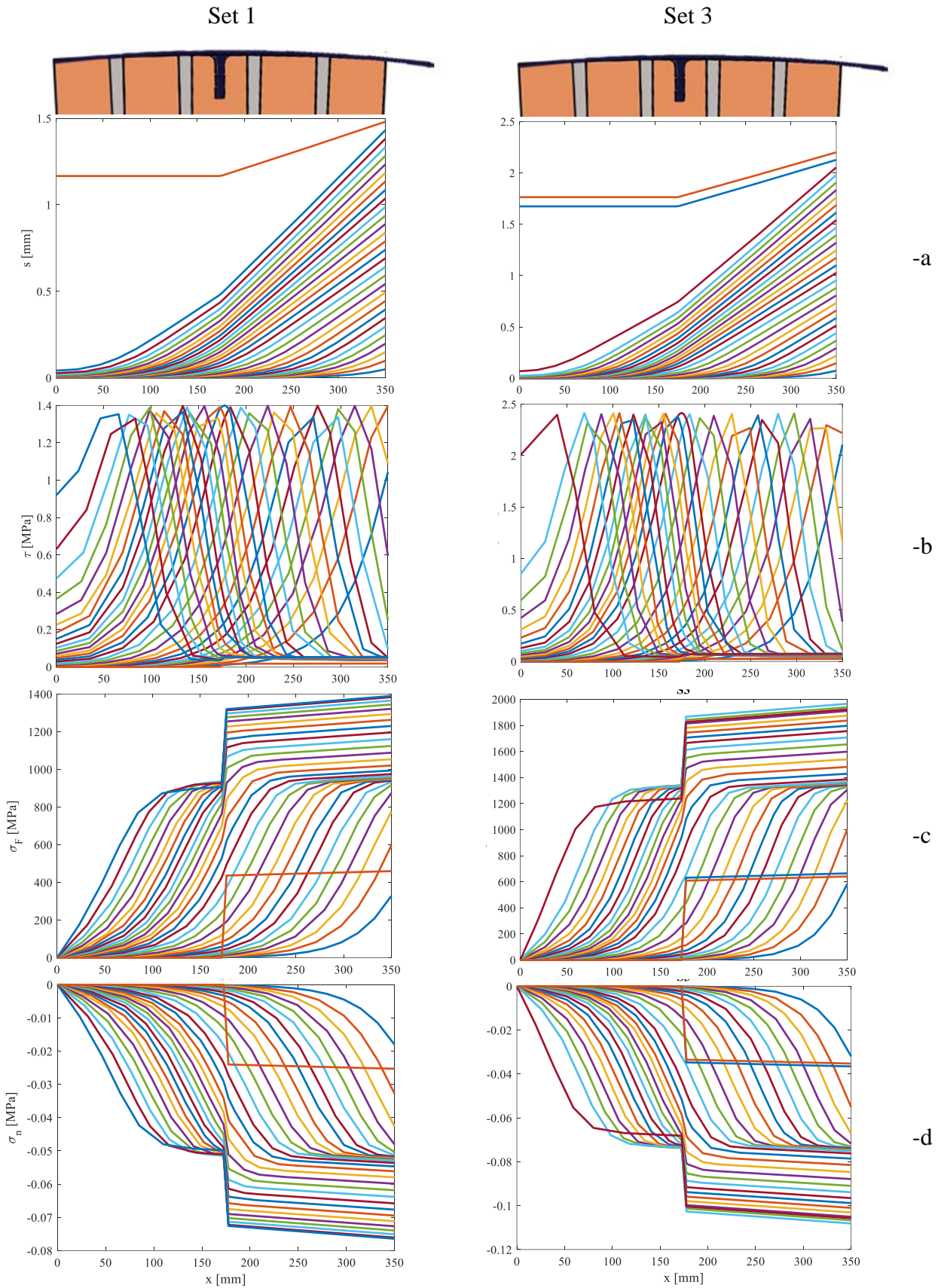


Figure 19: CBE with anchor spike, –a: x-slip diagram –b: x- tangential stress at the FRP substrate interface. –c: x- normal stress on FRP. –d: x- stress normal to the FRP/substrate interface.

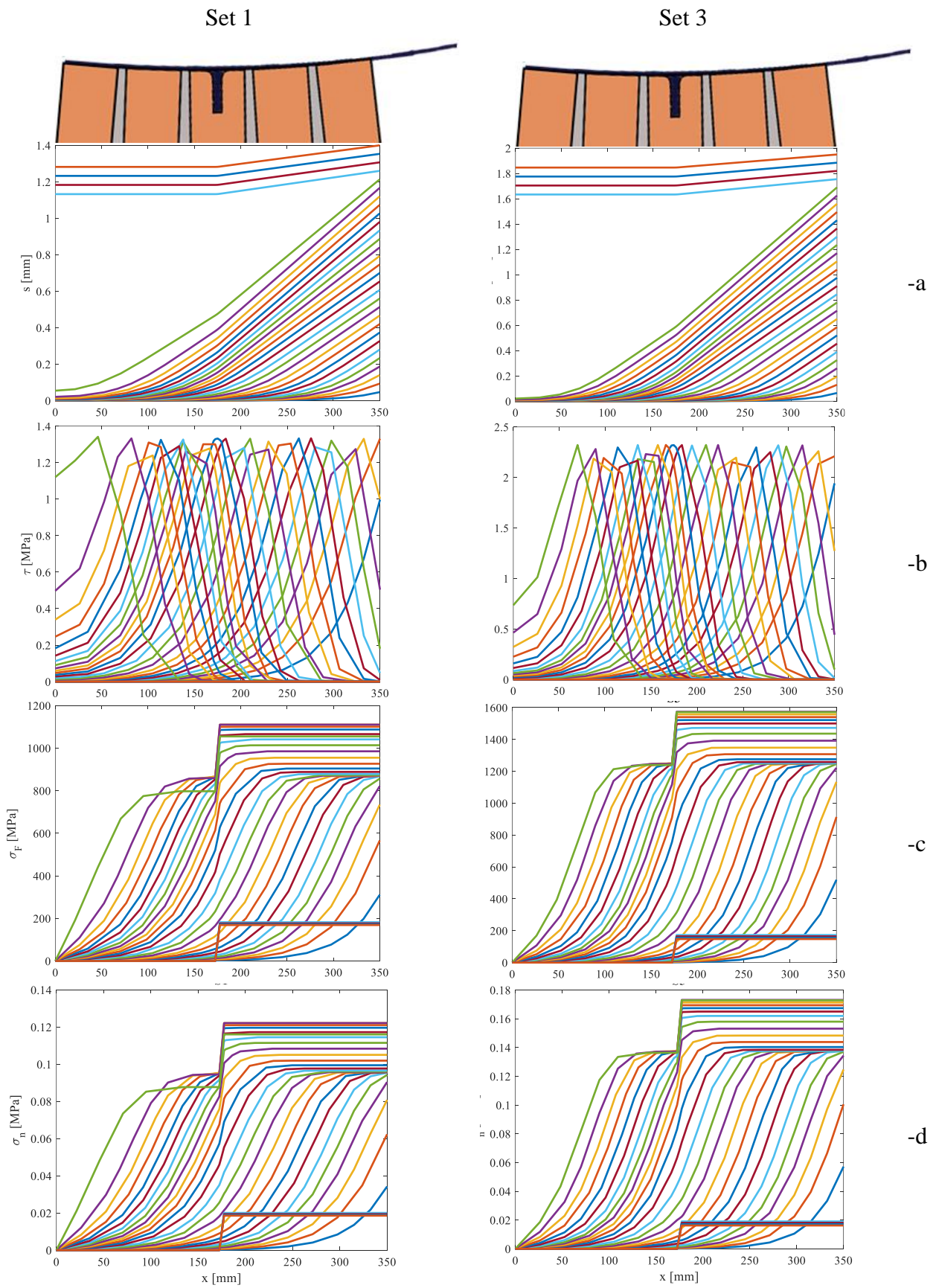


Figure 20: CAI with anchor spike, -a: x-slip diagram -b: x- tangential stress at the FRP substrate interface. -c: x- normal stress on FRP. -d: x- stress normal to the FRP/substrate interface.

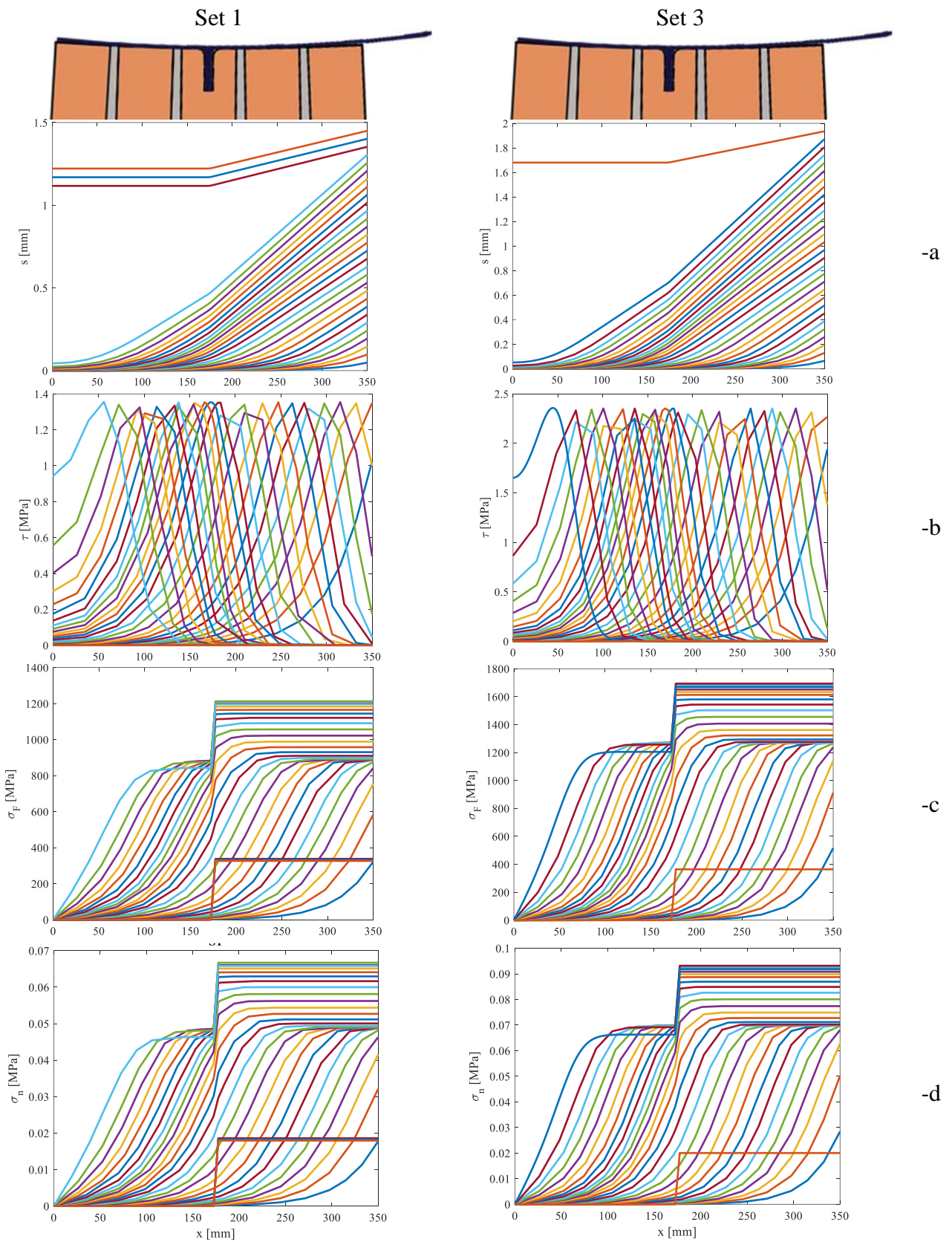


Figure 21: CBI with anchor spike, –a: x-slip diagram –b: x- tangential stress at the FRP substrate interface. –c: x- normal stress on FRP. –d: x- stress normal to the FRP/substrate interface.

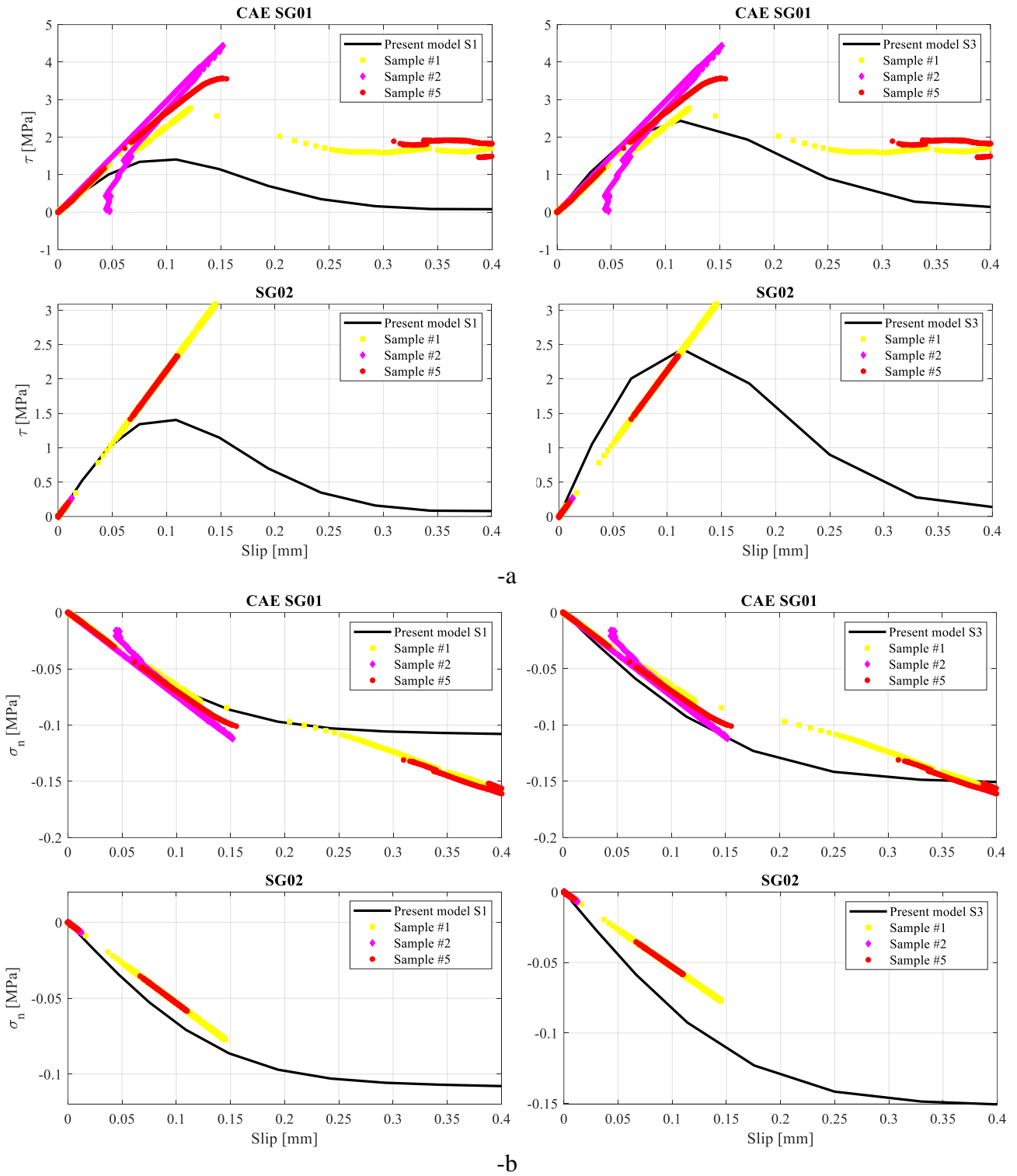


Figure 22: CAE specimen with anchor spike, tangential (-a) and normal (-b) interface stresses. Comparison between numerical prediction and experimental data provided by strain gauges SG01 & SG02. Left column: Set 1 mechanical properties. Right color: Set 3 mechanical properties.

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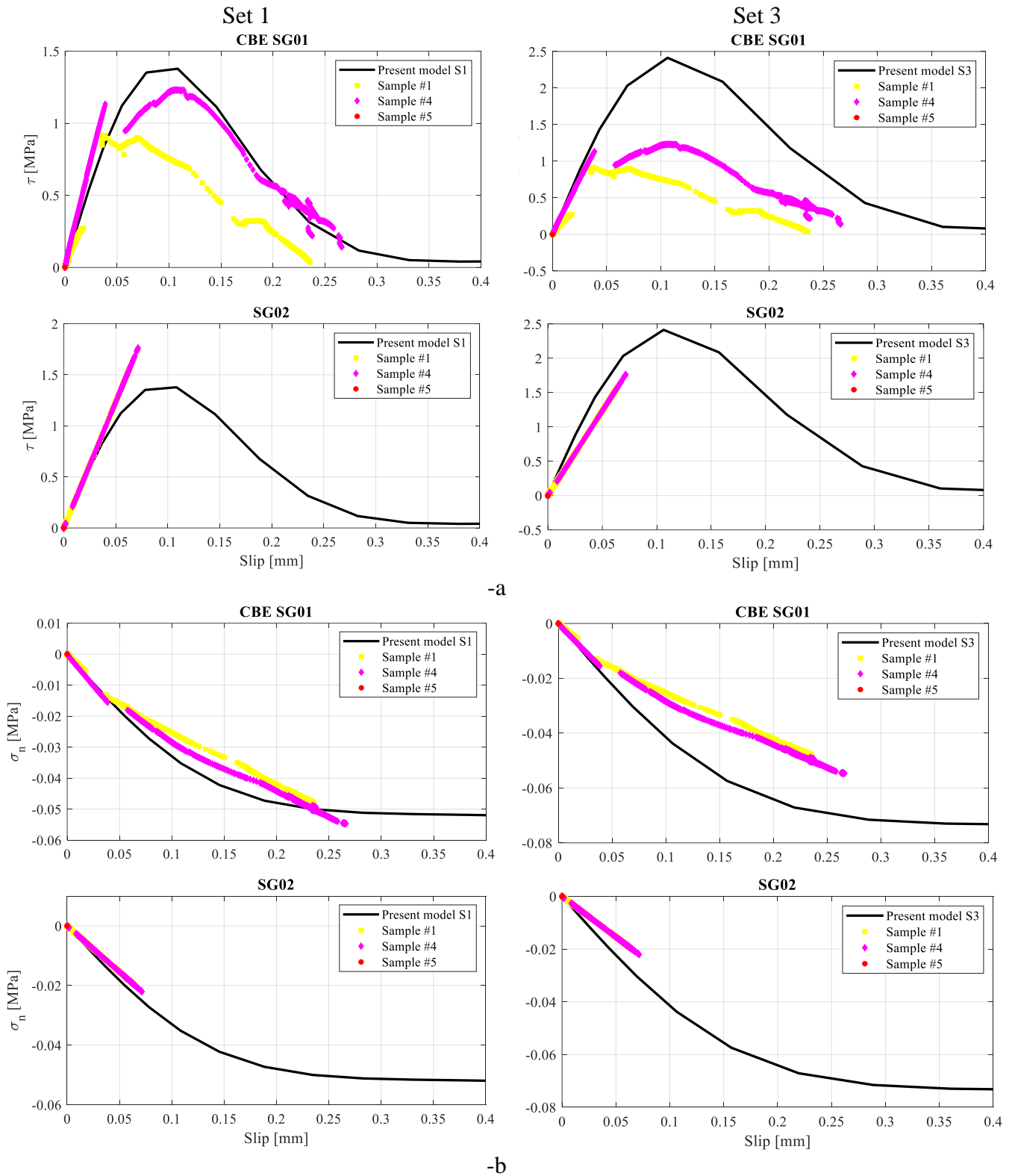


Figure 23: CBE specimen with anchor spike, tangential (-a) and normal (-b) interface stresses. Comparison between numerical prediction and experimental data provided by strain gauges SG01 & SG02. Left column: Set 1 mechanical properties. Right color: Set 3 mechanical properties.

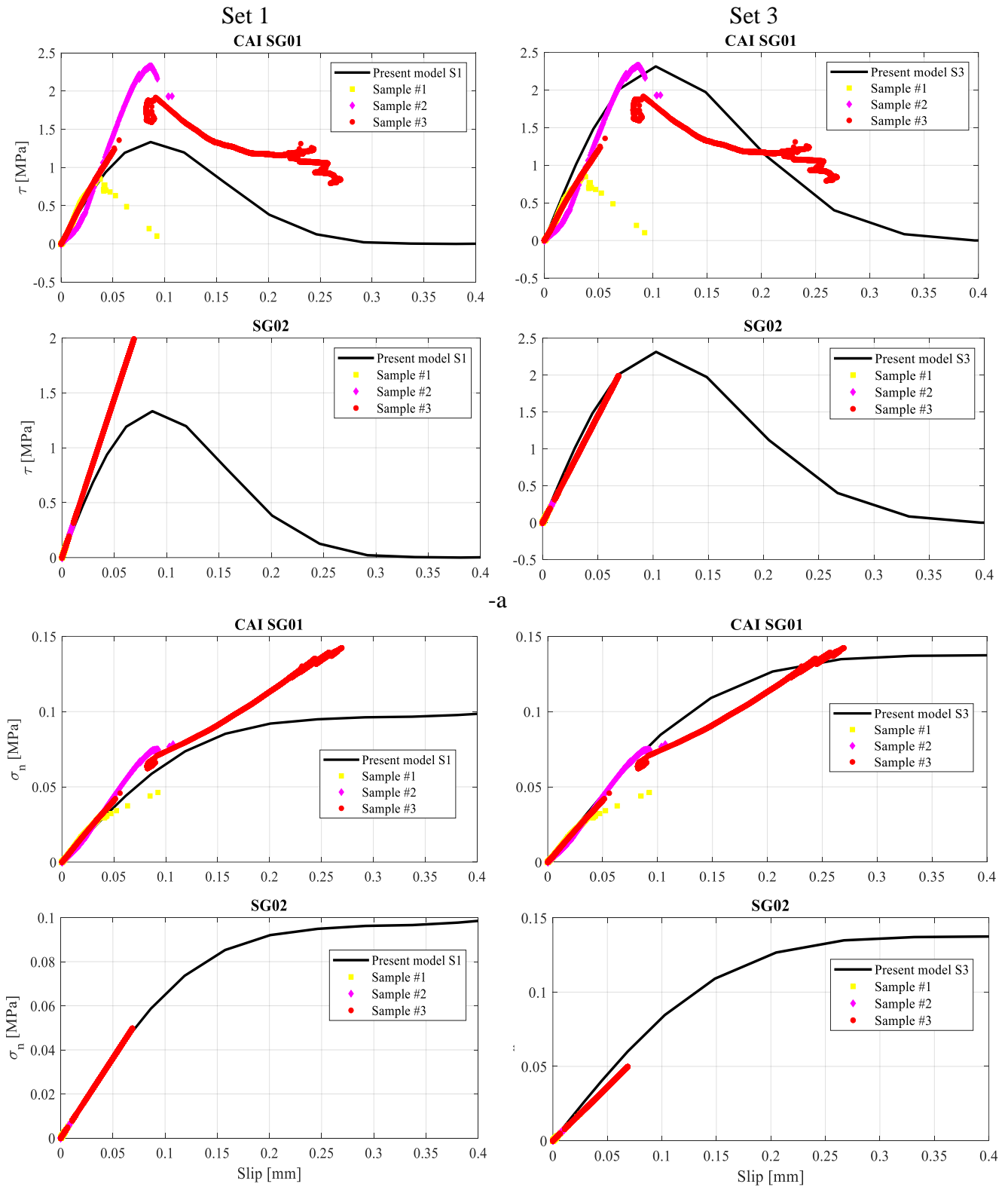


Figure 24: CAI specimen with anchor spike, tangential (-a) and normal (-b) interface stresses. Comparison between numerical prediction and experimental data provided by strain gauges SG01 & SG02. Left column: Set 1 mechanical properties. Right color: Set 3 mechanical properties.

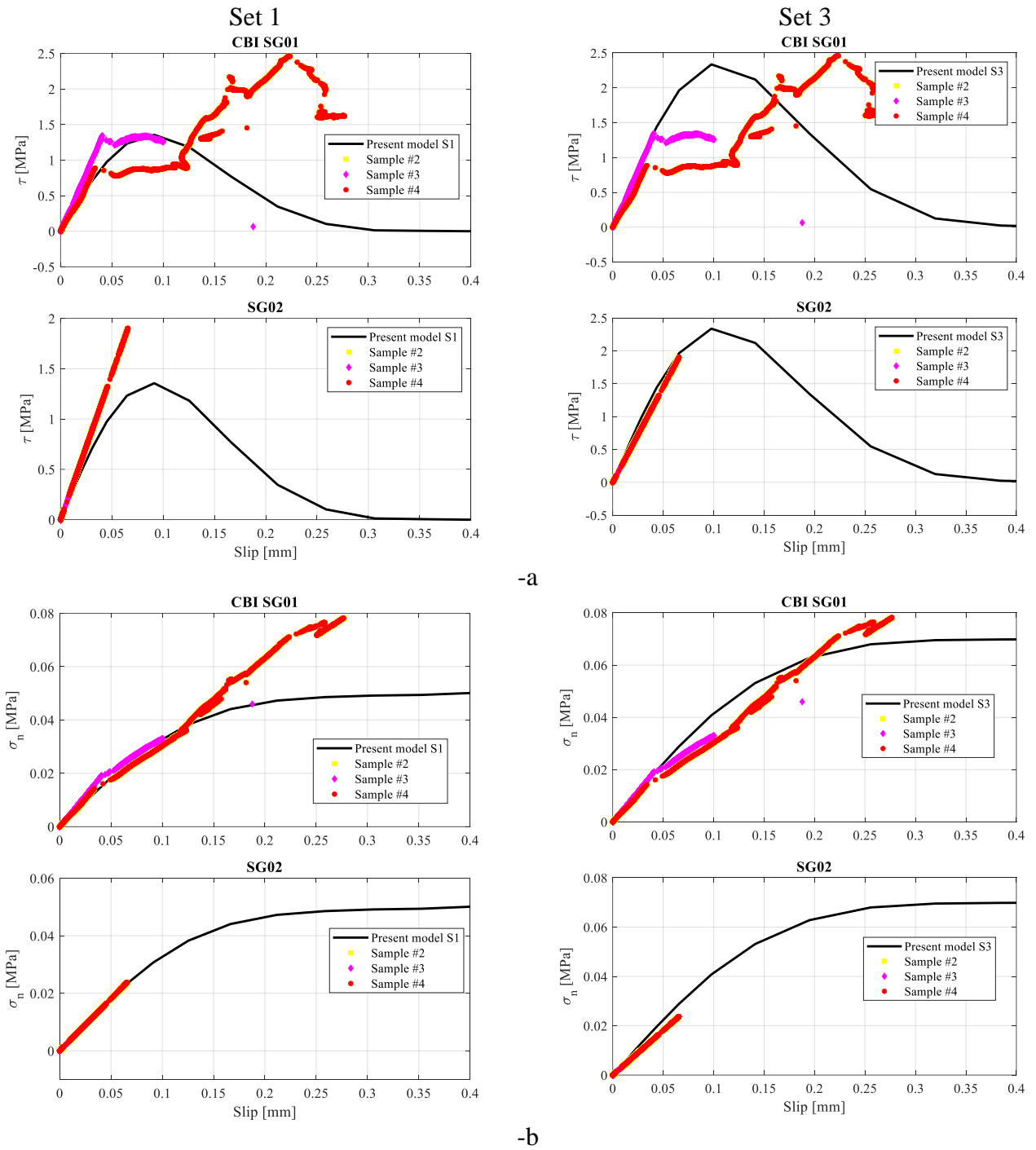


Figure 25: CBI specimen with anchor spike, tangential (-a) and normal (-b) interface stresses. Comparison between numerical prediction and experimental data provided by strain gauges SG01 & SG02. Left column: Set 1 mechanical properties. Right color: Set 3 mechanical properties.

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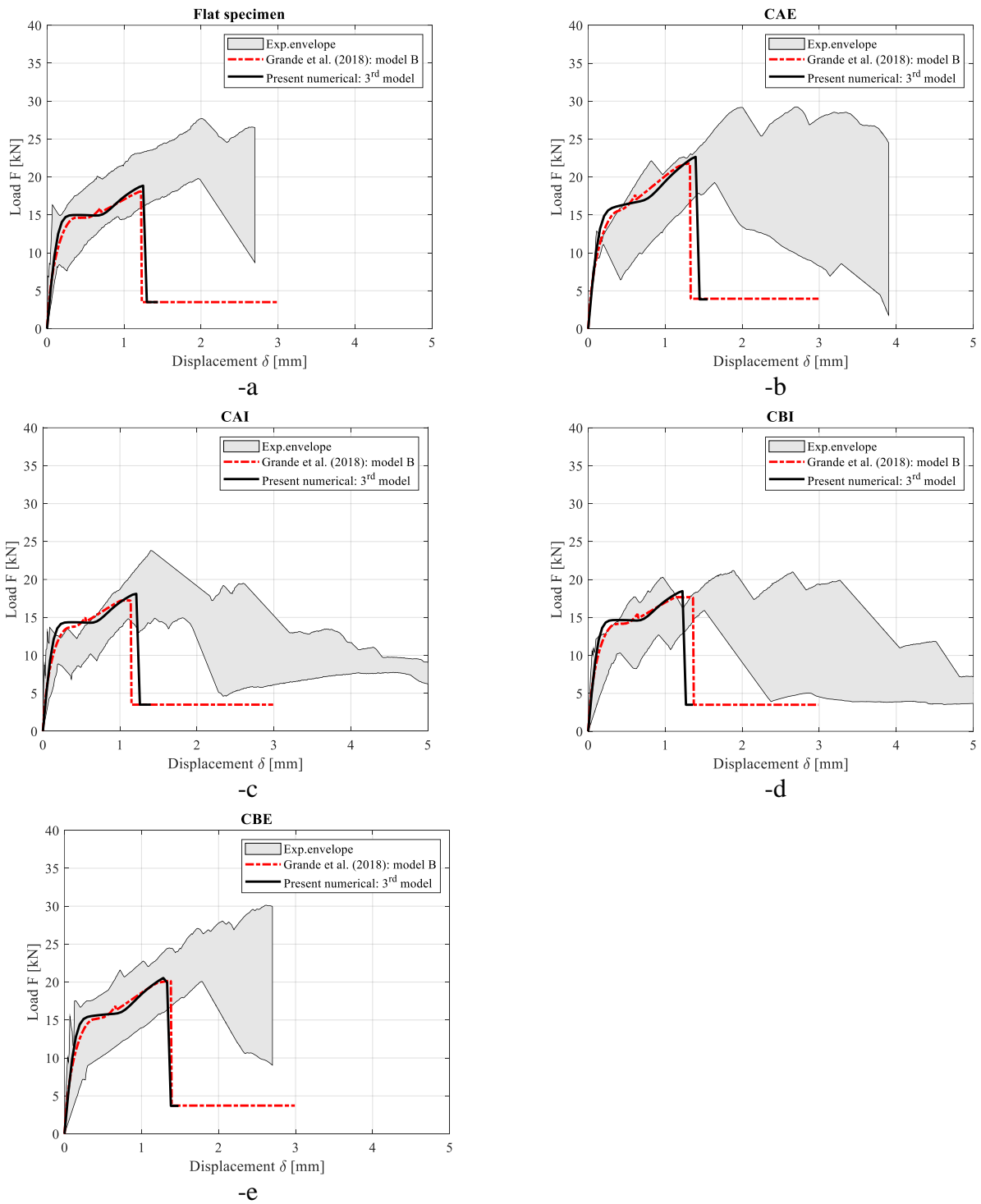


Figure 26: Global load-displacement curves obtained with the present approach assuming the spike relationship reported in Figure 11-d. Comparison among present model results, experimental evidences and previously presented model by Grande et al. [48]—a: Flat. —b: CAE. —c: CAI. —d: CBI. —e: CBE.

506 5. Conclusions

507 Structural rehabilitation of masonry constructions throughout FRPs generally involves elements
508 with a curved configuration, such as arches, vaults, domes, etc. In these cases, the bond behavior of
509 the reinforcing system, and consequently its performance, is particularly influenced by the curvature
510 geometry and the position of the reinforcement (i.e. at the extrados or intrados). Moreover, in order
511 to prevent premature failures due to tensile stresses normal to the direction of reinforcement, arising
512 in case of applications at the intrados, mechanical anchors are nowadays generally employed in
513 current practice. The presence of this additional component further influences the bond behavior of
514 FRPs applied on curved masonry substrates. Indeed, a recent and current field of research just
515 concerns the study of the bond behavior of FRP applied to curved masonry structures throughout
516 both experimental tests and numerical modeling strategies.

517 In this context, the present paper has presented an interface exponential model for simulating the
518 bond behavior of curved masonry pillars reinforced with FRP strips applied to the masonry
519 substrate by both epoxy adhesive and anchor spikes. The proposed model has been based on a
520 relatively simple BVP obtained by assuming for the spike a constitutive behavior under shear forces
521 quantitatively deduced from post processing of numerical data obtained by a FE micro-modelling
522 approach. After a detailed discussion reported in the first part of the paper, the reliability of
523 proposed model has been assessed with reference to experimental cases object of a previous
524 research carried out by the authors.

525 The obtained results have shown the reliability of the proposed model in reproducing the
526 experimental results both concerning global response, analyzed in the paper in terms of force-
527 displacement curves and, also, local behavior in terms of shear stress and normal stress vs. slip
528 curves. The stability of the solution of the BVP has been clearly shown together with the capacity of
529 the proposed model to capture the main phases of the bond behavior prior and after the activation of
530 the spike anchor.

531 In the paper it has been also reported the comparison with the results obtained by the authors
532 throughout other modeling approaches based on both a detailed and simplified finite element
533 approach, object of previous studies. The outcomes emerged from this additional comparison have
534 underlined features useful for further assessing the hypotheses at the basis of the proposed model
535 particularly in terms of constitutive laws of both masonry and reinforcement-masonry interface. In
536 particular, it has been possible to quantitatively deduce a constitutive behavior of the spike anchor
537 under shear forces from a post processing of numerical data obtained by a previous research of the
538 authors, where the debonding process was modelled in ABAQUS FEM software assuming damage
539 propagating in the bulk substrate.

540 In conclusion, the proposed approach:

- 541 • allows to skip the high computational burden required to capture the behaviour of anchored
542 FRP reinforcement.
- 543 • simplifies the geometric problem of the spike anchoring and the related dimensional
544 inaccuracies or mesh refinement problems that can arise using traditional FEM strategies.
- 545 • provides stable solutions during the whole debonding process and accurate results both
546 locally (tangential stress distributions) and globally (force-displacement responses).
- 547 • requires a quick calibration involving few mechanical parameters.
- 548 • is extendable to different spike anchor configurations, provided that the anchoring force is
549 properly calibrated.

- 550 • might help practitioners in a preliminary design phase of the strengthening system by
551 underlining the activation of spikes.

552

553 6. References

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