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Debonding Mechanism of FRP Strengthened Flat Surfaces: Analytical Approach and Closed Form Solution.

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15 Abstract

Fiber Reinforced Polymer (FRP) composites represent an effective retrofitting strategy for the 16 17 rehabilitation of masonry and concrete structures. The importance of adhesion between support and 18 strengthening material is crucial and great research effort has been aimed at understanding this 19 phenomenon from an analytical perspective. According to interfacial stress analysis, the debonding 20 mechanism may be idealized and studied as an FRP-interface-support system with elastic FRP bonded 21 to a brittle inelastic interface. In the present work, a fully analytical approach is developed to analyse 22 the debonding mechanism of FRP strips applied to flat masonries providing a closed form solution 23 characterized by few parameters governing the mathematical problem. Compared with other 24 analytical methods, the present approach is advantageous in its closed form formulation, which allows 25 the problem to be solved with a limited computational effort in a standard Matlab environment. The 26 approach is benchmarked with two different sets of experimental results taken from the technical 27 literature and from an ongoing investigation carried out by the authors. The results demonstrate the 28 reliability of the method in analysing the debonding of FRP applied on flat masonry prisms.

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31 **1. Introduction**

32 Nowadays it can be stated that Fibre Reinforced Polymer (FRP) composites have replaced traditionally 33 repairing techniques for the strengthening of old or structurally deficient masonries [1]-[3]. FRPs 34 present some common advantages with newly developed strengthening materials, such as FRCM / 35 TRM composites, including ease of installation, design flexibility and high corrosion resistance [1]. In 36 contrast to FRCM / TRM composites, FRPs offer a low weight-to-strength ratio, high stiffness, high 37 tensile strength, thus leading to high ductility and strength enhancements. Although FRCM / TRM 38 materials have some clear advantages in terms of compatibility with ancient supports, their failure 39 mechanisms present some tricky aspects, such as multiple failure modes that involve fabric slippage, 40 detachment from the supports and bundle tensile ruptures. Given the rather low tensile strength of the 41 cementitious matrix, the long-term performance of these materials, especially when the matrix is 42 cracked, is still being evaluated, as well as a complete procedure for deriving significant parameters 43 for their design. Thus, the application of FRPs gained increasing popularity mainly thanks to their high 44 versatility [4]-[7]. At the same time, their wide adoption shed light on one of their major features: the 45 importance of adhesion between support and strengthening material, to which the performance of the entire system is entrusted [8]-[15]. When it comes to existing masonry structures, the quality of the 46 bonding is of fundamental importance as different mechanical and geometric parameters of the 47 48 structure to be reinforced can deeply affect the bond quality of the final installation. For such reason, 49 researchers all over the world, focused on discovering the underlying causes of the loss of adhesion, 50 predicting the final performance of FRP strengthening solutions and improving their adhesion 51 characteristics [16]-[25]. A step towards a better understanding of the debonding phenomenon is 52 represented by the huge number of experimental studies dealing with the debonding of FRPs from 53 brittle supports (i.e. masonry and concrete) [1], [8]-[15]. Traditionally, bonding properties and 54 debonding failures of FRP materials are investigated, even if with variable consistency, through three 55 laboratory set-ups [1], [8]-[15]: (i) single lap, (ii) double lap and (iii) two-block double lap shear tests. 56 Some peculiar conclusions can be drawn from all these investigations: (i) the influence of the type of 57 support is confirmed, being the quality of the bond influenced by both its mechanical and physical 58 properties. Considering the variety of masonry structures, usually built with locally available 59 materials, this point is of crucial interest for the design of FRP reinforcement solutions. (ii) FRP 60 debonding phenomena could be assimilated to a fracture mechanics Mode-II interface bonding loss 61 event. Thus, the damage process is enclosed in a thin or thick layer at the interface between the 62 reinforcing material and the support with decreasing participation of this latter as it moves away from 63 the retrofitting area. The interfacial stress transfer phenomenon can be described fictitiously using a 64 cohesive law or tau-slip relationship. The stress-slip law of the interface bond is usually described by a 65 bi or tri linear model characterized by a linear elastic part upon reaching the peak tangential strength 66 and followed by a descending softening branch until reaching a frictionless state or a residual friction

67 strength [16]-[19],[24]-[26]. Piecewise linear bond-slip laws were found out mainly using strain 68 sensor readings and inverse analysis, which allowed the calibration of their characteristic parameters 69 too (i.e. peak tensile strength, tangential stiffness and post peak behavior). Therefore, the 70 experimental tests confirmed the possibility of reducing the development of nonlinear detachment 71 phenomena to an interface layer and the advantages inherent in this hypothesis encouraged 72 researchers to follow this convenient modeling approach. Such approach was established as the 73 dominant strategy not only for research purposes but also for the design of FRP composite 74 strengthening solutions in current Italian and international guidelines [26]-[32]. Although a lack of 75 consensus on the precise parametric dependence exists among different international guidelines [27]-76 [32], it is generally accepted that the mechanical properties of the support are a direct determinant of 77 the bond-slip properties. Two different but equally worthy approaches are developed in the technical 78 literature: (i) 2D/3D numerical models that consider perfectly bonded FRPs [21],[33],[34] or with 79 elastic interfaces [35]-[37] and, (ii) analytical models [19], [23]-[26]. The first category [21], [33], [34] is 80 often considered as a rough simplification of the adhesion properties of FRPs, which in those 81 researches completely depend on the mechanical properties of the constituent materials. Usually, this 82 approach is coupled with sophisticated material models (i.e. Concrete Damage Plasticity models) and a 83 micro-mechanical modeling strategy that, in the authors' opinion, makes such approaches worth 84 mentioning especially considering their practice-oriented feature. The focus of such works [21],[33]-85 [37] is indeed to provide an in-depth view of the failure modes of masonry assemblies and to explain, 86 regardless of the adhesion quality, the implications of secondary parameters on the final performance 87 of the installation, namely: geometry of the specimens, tensile and compressive properties of the 88 constituent materials and their post-peak characteristics and damage propagation in the different 89 components (e.g. mortar joints, bricks). Regardless of the complexity of the model which varies 90 according to the strategy and hypothesis adopted, the aforementioned approaches offer several 91 advantages such as the possibility to calibrate new bond-slip relationships, to compare the currently 92 available ones using pseudo-experimental data generated by the FE models or to simplify the study of 93 multiple related parameters [21],[33]-[36]. Another example of the possibilities that such approaches 94 offer is represented by a simplified 1D FE model implemented into a Matlab environment proposed in 95 [16] and later validated in [17],[18] against different experimental results, some of them on curved 96 masonry pillars. The model simulates an FRP strip applied to a masonry substrate by means of a set of 97 non-linear axial and shear springs, in which the latter are characterized by a bi-linear bond-slip 98 interface law and the relationship between shear and normal springs typically obeys a Mohr-Coulomb 99 behaviour. Apart from the extremely low computational burden, the model is easily generalizable -as 100 already anticipated- to complex geometries such as curved supports [18]. Closed form analytical 101 approaches represent a second branch of research in this context, driven by their extremely low 102 computational cost and ease of adoption and utilization by everyone. Worth of special mention is the 103 work presented in [38], in which a closed form analytical solution is presented to predict the

104 debonding behaviour of FRP-to-concrete strengthening. Further to provide a strong theoretical 105 background, the authors in [38], enrich the discussion by furnishing an experimental-based method to 106 identify the interfacial properties. However, the model is developed considering only the specific case 107 of a bilinear frictionless bond-slip law. Similarly, the closed-form solution proposed by [39] is based on 108 a bilinear frictionless bond-slip interface law. Although the mathematical background of the two works 109 is similar, the main difference relies on the generalization of the model proposed by [39] to any bond 110 length. Indeed, considering a long bonding length, the model allows a softening-debonding state, 111 whilst for short bonding lengths, the model allows only a softening state before reaching the failure. A 112 further extension of these works was proposed later in [23], with the adoption of a piecewise linear 113 bond-slip interface law comprising a non-zero residual friction. The model was benchmarked against 114 both short and long bonding lengths and adopted for both Near Surface Mounted (NSM) and Externally 115 Bonded (EB) strengthening solutions.

116 The present work proposes a fully analytical closed-form model in which the interfacial bond-slip 117 relationship is characterized by two branches: (i) a linear elastic phase followed by (ii) an inelastic 118 exponentially decreasing softening behaviour. The advantages with respect to previous closed-form 119 solutions are pivoted to this latter aspect which ensures: (i) a realistic description of the softening 120 branch in a bond-slip law with smoothly decreasing friction as the load increases, (ii) a continuous 121 function representing the softening behaviour during debonding, (iii) few parameters to calibrate the 122 descending branch and finally, (iv) a stable solution even in case of snap backs. Also, the model is 123 intrinsically applicable to any bonding length, allowing the development of three stages: (i) elastic, (ii) 124 mixed elastic/debonding and (iii) debonding.

The present paper is organized as follows: Section 2 discusses the closed-form analytical model here proposed and presents the three possible bonding states: elastic (*Case 1*), mixed elastic-debonding (*Case 2*) and debonding (*Case 3*). Section 3 describes the validation of the analytical model against some experimental studies available in the technical literature and an ongoing collaboration between different Universities. Finally, Section 4 outlines the main conclusions of the present work.

130 **2. The closed-form mathematical model**

The mathematical model herein proposed is developed with reference to a FRP strip externally applied on the flat surface of a specimen representing the structural support of the strengthening system (Figure 1). Moreover, since the equations at the basis of the proposed approach are carried out from equilibrium considerations involving an infinitesimal zone of the FRP and specifically considering a debonding mechanism at the reinforcement/support interface without the occurrence of additional phenomena due to damage of the reinforcement nor of the support, the obtained results do not depend on the material composing the support and the input data are related to the reinforcement and the reinforcement/support interface only. Indeed, the following three assumptions were adopted: (i) the strengthening material behaves as elastic during the whole loading process, (ii) all the nonlinearities concentrate at the interface between FRP strip and masonry support (Figure 1-b) and (iii) the FRP-to-support interface is associated only with a Mode II tangential fracture ruled by a $\tau(x) - s(x)$ curve.

Regarding the FRP, the parameters here accounted for are: Young's modulus, E_{FRP} ; thickness, t_{FRP} ; width, B_{FRP} (see Figure 1-a). On the other hand, regarding the behaviour of the interface, considering the shear stress-slip law showed in Figure 2, the introduced parameters are: bond strength, f_b ; slip at the end of the phase 1, s+ and, the rate of fracture energy in the post-peak phase, G_{II} .



Figure 1: Flat FRP strengthening configuration: geometry of the FRP strengthening (-a) and mathematical interface model in case of flat FRP reinforcement (-b).

147 By imposing the equilibrium along the longitudinal direction on a portion of FRP bonded to the 148 support (Figure 1-b), the following equation is obtained:

$$\frac{d\sigma_{FRP}}{dx} \cdot dx \cdot t_{FRP} \cdot B_{FRP} - \tau(s) \cdot B_{FRP} \cdot dx = 0$$
 Eq. 1

149 Considering the assumption (i) related to the constitutive behaviour of the FRP, normal stress of FRP150 results:

$$\sigma_{FRP} = E_{FRP} \cdot \varepsilon_{FRP}(x) = E_{FRP} \cdot \frac{ds(x)}{dx}$$
Eq. 2

151 where: $\varepsilon_{FRP}(x)$ is the axial strain of FRP and s(x) is the relative displacement between the 152 reinforcement and the support, i.e. the slip.

153 Substituting Eq. 2 into Eq. 1, it results:

$$E_{FRP} \cdot t_{FRP} \cdot \frac{d^2 s(x)}{dx^2} = \tau(s)$$
 Eq. 3

By multiplying both members of Eq. 3 by $\frac{ds(x)}{dx}$, Eq. 4-Eq. 6 are carried out:

$$\frac{1}{2} \cdot \left[2 \cdot \frac{ds(x)}{dx} \cdot \frac{d^2 s(x)}{dx^2} \right] = \frac{\tau(s)}{E_{FRP} \cdot t_{FRP}} \cdot \frac{ds(x)}{dx}$$
 Eq. 4

$$\frac{ds(x)}{dx} = \sqrt{\frac{2}{E_{FRP} \cdot t_{FRP}}} \cdot \int \tau(s) \, ds$$
 Eq. 5

$$\int \frac{ds(x)}{\frac{2}{E_{FRP} \cdot t_{FRP}} \cdot \int \tau(s)ds} = \int dx$$
 Eq. 6

155 In the present study, the nonlinear piecewise relationship depicted in Figure 2 was selected to 156 describe the behaviour of the specimen at the interface between FRP strengthening and support.



Figure 2: Tangential stress-slip relationship at the interface between FRP strengthening and support.

157 The relationship $\tau(s) - s(x)$ is composed of two braches identifying two different phases of the 158 behaviour of the interfaces: Phase 1, where a linear phase characterizes the behaviour of the interface 159 until the peak tangential stress f_b is reached (see Eq. 7); Phase 2 where a nonlinear softening branch 160 characterizes the post-peak behaviour of the interface (see Eq. 8):

$$\tau(s) = ks(x)$$
 Eq. 7

$$\tau(s) = f_b \cdot e^{\frac{-(s-s_*) \cdot f_b}{G_{II}}} = \tau_0 \cdot e^{-\frac{s \cdot f_b}{G_{II}}}, \text{ with } \tau_0 = f_b \cdot e^{\frac{s \cdot f_b}{G_{II}}}$$
Eq. 8

- where: k is the slope of the linear branch; f_b identifies the peak tangential strength at the interface; G_{II} stands for the rate of fracture energy associated with Mode II in the post-peak stage; s_* is the slip value at the end of the phase 1.
- 164 The assumptions at the basis of the proposed approach and the type of selected shear stress-slip law for the interface, allow to identify three possible cases for the behaviour of the specimen. Indeed, 165 depending on the value of the displacement imposed at the loaded edge (at this section it corresponds 166 167 to the slip, here denoted s_0), the three possible cases are: (i) Case 1, Eq. 9 holds for the entire bonding length L_L which behaves as elastic; (ii) Case 2 is characterized by a mixed interface response: only a 168 portion of the bonding length, herein identified as \tilde{L} , behaves as elastic, whilst the other one lies in 169 phase 2, Eq. 10; Case 3, characterized by L_L (i.e. the entire bonding length) behaving in phase 2 (Eq. 170 171 11).

$$0 \le s_0 < \frac{2s_*}{e^{-\alpha L_L} + e^{\alpha L_L}}$$

$$\frac{2s_*}{e^{-\alpha L_L} + e^{\alpha L_L}} \le s_0 \le s_*$$

$$s_0 > s_*$$

$$Case 1$$

$$Eq. 9$$

$$Case 2$$

$$Eq. 10$$

$$Case 3$$

$$Eq. 11$$

172 Where $\alpha^2 = \frac{k}{E_{FRP} \cdot t_{FRP}}$. In the following the solution of the proposed analytical approach is then 173 provided for the three identified cases.

174 **2.1 Case 1**

175 Since *Case 1* is characterized by a maximum value of slip lower than s_* , the phase I of the law $\tau(s) - s(x)$, i.e. the one described by Eq. 6, distinguished the behaviour of the interface along the whole bond 177 length. Thus, after trivial manipulations, Eq. 12 is obtained. The solution of which is expressed in Eq. 178 15 provided that α^2 is expressed as Eq. 13 and replaced in Eq. 14.

$$\frac{d^2 s(x)}{dx^2} = \frac{k}{E_{FRP} \cdot t_{FRP}} \cdot s$$
 Eq. 12

$$\alpha^2 = \frac{k}{E_{FRP} \cdot t_{FRP}}$$
 Eq. 13

$$\frac{d^2 s(x)}{dx^2} - \alpha^2 \cdot s(x) = 0$$
 Eq. 14

$$s(x) = A_1 e^{-\alpha x} + A_2 e^{\alpha x}$$
 Eq. 15

- 179 where A_1 and A_2 are two constants to be determined.
- By imposing the conditions at the loaded edge (i.e., $s(0) = s_0$), A_1 is obtained (Eq. 16) as a function on A_2 :

$$A_1 = s_0 - A_2$$
 Eq. 16

$$A_1 = \frac{s_0}{2} = A_2$$
 Eq. 17

182 The constants A_1 and A_2 (Eq. 17) are obtained by imposing $\sigma_{FRP} = \frac{ds}{dx} \cdot E_{FRP} = E_{FRP} \cdot \alpha \cdot (A_2 e^{\alpha x} - A_1 e^{-\alpha x}) = 0$ when x=0 and then substituting the relationship between A1 and A2 into Eq. 16.

Finally, the slip s(x) and normal stress in the FRP strengthening σ_{FRP} are obtained (Eq. 18 and Eq. 19 respectively):

$$s(x) = \frac{s_0}{2} \cdot (e^{-\alpha x} + e^{\alpha x})$$
 Eq. 18

$$\sigma_{FRP} = E_{FRP} \cdot \frac{ds(x)}{dx} = E_{FRP} \cdot \alpha \cdot \frac{s_0}{2} \cdot (e^{\alpha x} - e^{\alpha x})$$
 Eq. 19

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187 **2.2 Case 2**

188 *Case 2* is characterized by a behaviour of the interface where both phases 1 and 2 coexist on the 189 bonding length L_L . Therefore Eq. 6 is used providing that the integral of $\tau(s)$ described by Eq. 8 is used, 190 as shown in Eq. 20-Eq. 21:

$$\int \tau(s) \, ds = \int \tau_0 \cdot e^{-\frac{s \cdot f_b}{G_{II}}} \, ds = -\frac{\tau_0}{f_b} \cdot G_{II} \cdot e^{-\frac{s \cdot f_b}{G_{II}}} + C_1$$
 Eq. 20

$$\frac{ds}{dx} = \sqrt{\frac{2\tau_0 \cdot G_{II}}{E_{FRP} \cdot t_{FRP} \cdot f_b} \cdot \left(-e^{-\frac{s \cdot f_b}{G_{II}}} + C_1\right)}$$
Eq. 21

191 The solution of Eq. 21 is presented in Eq. 22:

$$atanh\left(\sqrt{1-\frac{e^{-\xi}}{C_1}}\right) = \frac{\sqrt{C_1} \cdot k_1}{2} \cdot x + C_2$$
 Eq. 22

192 where:
$$\xi = \frac{s(x) \cdot f_b}{G_{II}}$$
, while $k_1 = \sqrt{\frac{2\tau_0 f_b}{E_{FRP} \cdot t_{FRP} \cdot G_{II}}}$

193 To find the values of the two constants C_1 and C_2 , the following Initial Condition might be applied, 194 which implies that $\frac{ds(x)}{dx} \cdot E_{FRP} = \tilde{\sigma}_{FRP}$ (Eq. 23-Eq. 24):

$$\frac{ds(x)}{dx} \cdot E_{FRP} = \sigma_{FRP}(\tilde{L}) = \frac{s_0}{2} \cdot \alpha \cdot E_{FRP} \cdot (e^{\alpha \tilde{L}} - e^{-\alpha \tilde{L}}) = \tilde{\sigma}_{FRP}$$
 Eq. 23

$$\frac{ds(x)}{dx} \cdot E_{FRP} = \sqrt{\frac{2\tau_0 \cdot E_{FRP} \cdot G_{II}}{t_{FRP} \cdot f_b}} \cdot \left(-e^{-\frac{s \cdot f_b}{G_{II}}} + C_1\right)$$
Eq. 24

195

196 Finally, the following value of C_1 (Eq. 25) is obtained:

$$C_1 = e^{\xi_0} + \tilde{\sigma}_{FRP}^2 \cdot \frac{t_{FRP} \cdot f_b}{2E_{FRP} \cdot \tau_0 \cdot G_{II}}$$
Eq. 25

197 where $\xi_0 = \frac{s_* f_b}{G_{II}}$. The value of C_2 (Eq. 26) is obtained using Eq. 22 knowing the value of C_1 .

$$C_2 = atanh\left(\sqrt{1 - \frac{e^{\xi_0}}{C_1}}\right)$$
 Eq. 26

198 Then, considering the values of the two constants C_1 and C_2 , it is possible to derive S_L and σ_{FRP} , which 199 are reported in Eq. 27 and Eq. 28, respectively.

$$S_L = -\frac{G_{II}}{f_b} \cdot ln \left\{ C_1 \left[1 - tanh^2 \left(\frac{\sqrt{C_1} \cdot k_1 \cdot \left(L - \tilde{L} \right)}{2} + C_2 \right) \right] \right\}$$
Eq. 27

$$\sigma_{FRP} = \sqrt{\frac{2\tau_0 \cdot G_{II} \cdot E_{FRP}}{t_{FRP} \cdot f_b} \cdot \left(C_1 - e^{-\frac{s(x) \cdot f_b}{G_{II}}}\right)}$$
Eq. 28

200 where \tilde{L} is derived in closed form in Eq. 29:

$$\tilde{L} = \frac{1}{\alpha} \cdot ln\left(\frac{s^* + \sqrt{s^{*2} - s_0^2}}{s_0}\right)$$
Eq. 29

Analyzing the solution obtained for *Case 2*, still considering the data of the generic specimen accounted for *Case 1* (see Figure 3 and Figure 4), it is possible to observe a nonlinear trend of the normal stress of FRP vs. slip at the loaded edge curve. This curve is characterized by an ascending segment with a significant slope until the load point B, a subsequent smooth segment which assumes a descending trend next to the load point C (this point corresponds to the end of *Case 2*). Indeed, considering the trend of slip s along the bond length at the load points B and C (and the corresponding curves σ_{FRP} -x and τ -x) it is evident that: (i) at the load point B the zone of specimen next to the unloaded end is characterized by negligible values of slips (and then of τ), i.e. a limited zone of the specimen is significantly involved in the bond process (the so called effective length); (ii) at the load point C the behavior of the specimen shows an opposite situation in terms of shear stresses at the interface: negligible values characterize the status of a significant zone of the interface in close proximity to the loaded end. The latter occurrence is then responsible for the softening behavior.

213 **2.3 Case 3**

In *Case 3*, i.e. when the phase 2 characterizes the behaviour of the interface along the whole bond length, Eq. 21 is still valid. In this case, the initial condition to be imposed is reported in Eq. 30, where $\xi_0 = \frac{s_0 \cdot f_b}{G_{II}}$.

$$\sigma_{FRP}(s_0) = \sqrt{\frac{2\tau_0 \cdot G_{II} \cdot E_{FRP}}{t_{FRP} \cdot f_b}} \cdot \sqrt{C_1 - e^{\xi_0}} = 0$$
 Eq. 30

217 It is useful to point out that $\sigma_{FRP}(s_0)$ in Eq. 30 represents the normal stress on FRP at the free edge, 218 because in Case 3 phase 2 characterizes the behaviour of the interface along the entire bond length.

By imposing Eq. 30, the value of C_1 is obtained, as indicated in Eq. 31, which in turn is used in Eq. 32 to find C_2 .

$$C_1 = e^{-\frac{S_0 \cdot f_B}{G_{II}}}$$
 Eq. 31

$$C_2 = atanh\left(\sqrt{1 - \frac{e^{-\xi_0}}{C_1}}\right)$$
 Eq. 32

Finally, by knowing the values of the two constants C_1 and C_2 , S_L and σ_{FRP} might be obtained as reported in Eq. 33 and Eq. 34.

$$S_L = -\frac{G_{II}}{f_b} \cdot ln \left\{ C_1 \left[1 - tanh^2 \left(\frac{\sqrt{C_1} \cdot k_1 \cdot L}{2} + C_2 \right) \right] \right\}$$
 Eq. 33

$$\sigma_{FRP}^{L} = \sqrt{\frac{2\tau_0 \cdot G_{II} \cdot E_{FRP}}{t_{FRP} \cdot f_b} \cdot \left(C_1 - e^{-\frac{S_L \cdot f_b}{G_{II}}}\right)}$$
Eq. 34

Analyzing the solution of the generic specimen still considered for the previous cases, also for *Case 3*, it emerges a behavior in terms of σ_{FRP} -S highlighted by the snap-back phenomenon, where both normal stress and slip at the loaded edge reduce. Indeed, considering the generic load point C, it is possible to observe a reduction of shear stresses: the maximum value of shear stresses along the bond length is lower than the shear strength f_b . Consequently, this implies an unloading status of the FRP with a corresponding reduction of both normal stresses and strains.

As underlined in next sections, this phenomenon is generally not observed from standard shear-lap
tests where it is imposed a displacement at the load edge, progressively increased until the failure.

231 The obtained solution of the equations at the basis of the proposed approach was implemented in 232 Matlab [40] with the twofold goal of analysing it in terms of bond behaviour of a generic specimen and, 233 subsequently, of assessing its reliability with reference to some experimental cases derived from 234 literature (see next sections). According to the proposed analytical model, the input data here 235 accounted for the generic specimen then concern its geometry ($B_{FRP}=100$ mm; $t_{FRP}=0.165$ mm; 236 $L_{\rm L}$ =287.5 mm), the Young's modulus of reinforcement ($E_{\rm FRP}$ =250000 MPa) and the parameters of the 237 FRP/masonry interface law (f_b=1.65 MPa; s*=0.05 mm; G_{II}=0.2 N/mm). Introducing these parameters 238 into the proposed analytical model, in particular in Eq. 18 and Eq. 19 referring to the *Case 1*, the results 239 in terms of slip *s* along the bond length at different load steps and in terms of corresponding normal 240 stress of reinforcement σ_{FRP} vs. slip at the loaded edge, are graphically presented in Figure 3. In 241 particular, the point A corresponds to the maximum value of load where the whole interface lies in 242 phase I (i.e. the behavior of the specimen is within *Case 1*). As expected, the maximum value of the slip 243 is attained at the loaded edge where assumes the value s^{*} corresponding to the end of the phase I. The 244 normal stress of the reinforcement at the loaded edge linearly varies until the point A because of the 245 assumption of a linear constitutive law for this component of the strengthening system and the linear 246 shape of the first branch (phase I) of the shear stress-slip law of the interface.

247 Figure 5 and Figure 6 depict slip s and shear stress distribution maps along the bond length obtained 248 with the proposed approach considering four time-steps, namely points A, B, C and D. It is worth 249 mentioning that the dimensions of the elements (i.e. support, interface and strengthening materials) 250 were scaled to improve the readability of the graphs. As expected, the debonding front propagates 251 from the loaded end toward the free end (see Figure 6-A, -B and -C). While at the end of the analysis 252 (i.e. point D), a clear snap-back phenomenon is observed, especially in Figure 5-D. A snap-back is also 253 visible in the global curve of Figure 3-b. Generally speaking and according to authors experience, it can 254 be affirmed that such snap-back is visible in the majority of the practical cases at the transition 255 between Case 2 and Case 3. As a matter of fact, in Case 3 the interface between FRP and support is all 256 subjected to softening, whereas in Case 2 part of the interface (near the free edge) is still in the elastic 257 phase. The procedure proposed solves in closed form a Cauchy problem where s_0 (i.e. the interface

258 displacement at the free edge) is imposed. As a consequence, s_L (i.e. the interface displacement at the 259 loaded edge) is found as output result. It may occur, especially for short bond lengths, that the observed s_L decreases in Case 3 when compared to that observed in Case 2, as a consequence of the 260 261 sudden decrease of the load applied. It is not easy to provide quantitative information on the 262 occurrence of snap-back, because such phenomenon depends on several parameters, such as the 263 interface law adopted (i.e. initial elastic stiffness, tangential strength and fracture energy in mode II) and the geometrical properties of the reinforcement (bond length). The results presented in Figure 4 264 265 in terms of normal stress of FRP and shear stress of the interface, both plotted along the bond length, 266 also underline for *Case 1* a behavior of the specimen characterized for both normal stress of FRP and 267 shear stress of interface by the attainment of the peak value at the loaded edge and a progressive 268 reduction along the remain bond length.



Figure 3: Slip *s* distribution along the bond length at different load steps (-a) and normal stress of reinforcement σ_{FRP} vs. slip at the loaded edge (-b) obtained with the proposed approach.



Figure 4: Distribution of normal stress of FRP (-a) and shear stress (-b) of the interface along the bond length obtained with the proposed approach.



Figure 5: Slip s along the bond length plots considering different load steps.



Figure 6: Shear stress maps at the interface along the bond length obtained with the proposed approach.

277 **3. Validation**

The proposed approach is here validated by considering shear lap experimental tests of single bricksand brick-mortar assemblages strengthened by FRP derived from studies in the technical literature.

In particular, the first set of specimens refers to the experimental investigation carried out in [13]. They consist of clay bricks strengthened on both sides by a FRP strip bonded to the brick surface for a length *LL* equal to 160mm. Shear lap tests were performed by considering four different types of strengthening materials: carbon (CFRP), glass (GFRP), basalt (BFRP) and steel (SRP). These specimens are labelled in the following as: Valluzzi et al.2012-CFRP (Figure 7); Valluzzi et al.2012-GRFP (Figure 8); Valluzzi et al.2012-BRFP (Figure 9); Valluzzi et al.2012-SRP (Figure 10).

Table 1. Parameters accounted for the validation of the proposed approach.								
Label	E _{FRP}	t_{FRP}	B _{FRP}	fb	s*	GII	LL	
	[MPa]	[mm]	[mm]	[MPa]	[mm]	[N/mm]	[mm]	
Valluzzi et al. 2012 – CFRP [13]	233861	0.17	50	2.49	0.016	0.2913	160	
Valluzzi et al. 2012 – GFRP [13]	84251	0.12	50	2.49	0.016	0.2789	160	
Valluzzi et al. 2012 – BFRP [13]	88397	0.14	50	2.49	0.016	0.2789	160	
Valluzzi et al. 2012 – SRP [13]	195054	0.231	50	2.49	0.07	0.2864	160	
Rotunno et al. 2018 – CFRP [14]	250000	0.165	100	1.626	0.005	0.2593	330	

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Figure 7: Valluzzi et al. 2021-CFRP [13]: Validation of the proposed analytical approach with reference to single brick specimens and different reinforcing materials: reference bi-linear law and obtained equivalent nonlinear law (-a) and experimental, numerical and obtained analytical Force-Slip curves (-b).

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Figure 8: Valluzzi et al. 2021-CFRP [13]: Validation of the proposed analytical approach with reference to single brick specimens and different reinforcing materials: reference bi-linear law and obtained equivalent nonlinear law (-a) and experimental, numerical and obtained analytical Force-Slip curves (-b).



Figure 9: Valluzzi et al. 2021-CFRP [13]: Validation of the proposed analytical approach with reference to single brick specimens and different reinforcing materials: reference bi-linear law and obtained equivalent nonlinear law (-a) and experimental, numerical and obtained analytical Force-Slip curves (-b).



Figure 10: Valluzzi et al. 2021-CFRP [13]: Validation of the proposed analytical approach with reference to single brick specimens and different reinforcing materials: reference bi-linear law and obtained equivalent nonlinear law (-a) and experimental, numerical and obtained analytical Force-Slip curves (-b).

296 Taking into account the simple constitutive shear stress-slip bi-linear law proposed in [17] for 297 characterizing the behaviour of FRP/masonry interface layer, also employed in the numerical study 298 carried out in [16], where the above specimens were specifically accounted for validating a numerical 299 modelling approach denoted 1D spring-model (see [16] for details), the equivalent shear stress-slip 300 nonlinear law characterizing the analytical approach here proposed has been derived for each 301 specimen (Figure 7-a, Figure 8-a, Figure 9-a and Figure 10-a). Indeed, it has been assumed for the 302 phase 1 the same slope of the ascending branch and the same value of the bond strength of the bilinear law; for the phase 2 a value of G_{II} equal to the area subtended by the post-peak descending 303 304 branch of the bi-linear law proposed in [16].

305 A further case here considered is derived from the experimental study carried out in [14] and also 306 introduced in [41] for developing numerical analyses throughout an interface numerical model valid 307 also in case of curved substrates (see for [41] details). In this case, while the reinforcement is still a 308 carbon fiber strip, the substrate is a masonry pillar composed of five clay bricks with interposed 309 mortar joints made of lime and cement as binder. The reinforcement is applied on one side only of the 310 pillar for a bond length LL equal to 330 mm. Also, in this case, the constitutive law of the 311 reinforcement/masonry interface (see Figure 11) has been derived according to the procedure 312 accounted for the previous specimens, by considering the bi-linear law proposed in [18].

The results obtained from the analytical model here proposed are compared with the experimental Force-Slip curves (or envelope areas of experimental curves) and presented in Figure 7-b, Figure 8-b, Figure 9-b and Figure 10-b for the first set of specimens and in Figure 11 for the masonry pillar specimen.



Figure 11: Validation of the proposed analytical approach with reference to pillar masonry specimens: reference bi-linear law and obtained equivalent nonlinear law (-a) and experimental, numerical, and obtained analytical Force-Slip curves (-b).

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From the plots clearly emerges the efficacy of the proposed analytical approach in predicting the prepeak phase and the peak load. A good agreement is also observed with respect to the numerical solutions. Nevertheless, while the numerical solutions halt when the equilibrium is no longer satisfied, the analytical solution proceeds by underlying the snap-back phenomenon.

Although in the majority of analysed cases the ultimate experimental displacement results greater than both numerical and analytical ones, as underlined in [16], the post-peak behaviour is generally influenced by phenomena, such as the degradation of the detached zone of the reinforcement, which lead to a greater deformability of the system.

4. Conclusions

This research proposes a fully analytical approach aimed at studying the FRP debonding process from flat brittle surfaces (i.e. masonry prisms). The analytical approach derives from imposing equilibrium considerations involving an infinitesimal zone of the FRP. The approach assumes that all the nonlinearities are concentrated at the interface between FRP, strip and masonry support with FRP-tosupport interfaces associated with a Mode II tangential fracture behaviour only.

Based on the results of the present work, the following conclusions can be drawn:

- The proposed closed-form model considers a debonding mechanism at the
 reinforcement/support interface only.
- The obtained results are decoupled from the material composing the support and the input
 data are related to the reinforcement/support interface only.
- A closed form solution, characterized by few parameters governing the mathematical problem
 (interfacial stress-slip law characterized by an initial linear elastic phase followed by an
 inelastic exponentially decreasing softening behaviour) and a limited computational effort, is
 provided.
- The approach is benchmarked with two sets of experimental investigations: (i) a laboratory
 campaign using different reinforcing materials and (ii) an ongoing experimental and numerical
 research collaboration carried out by the authors.
- The comparisons in terms of force-slip curves clearly highlight the efficacy of the proposed
 analytical approach in predicting the pre-peak phase and the peak load.
- Compared to other numerical approaches, the analytical solution resulted more stable and able
 to capture possible snap-back phenomena.
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