Ignition, puffing and sooting characteristics of kerosene droplet combustion under sub-atmospheric pressure

Hongtao Zhang^a, Zhihua Wang^{a,*}, Yong He^a, Jun Xia^b, Jincheng Zhang^a, Hua Zhao^b, Kefa Cen^a

^aState Key Laboratory of Clean Energy Utilization, Zhejiang University, Hangzhou 310027, P.R. China ^bDepartment of Mechanical and Aerospace Engineering & Institute of Energy Futures, Brunel University London, Uxbridge UB8 3PH, U.K.

Abstract

The ignition, puffing and sooting characteristics of Chinese RP-3 kerosene droplet burning have been studied using high-speed, OH* chemiluminescence and soot thermal radiation imaging. The experiments were conducted in air at standard temperature and sub-atmospheric pressures ranging from 0.2 bar to 1 bar. The kerosene droplet was supported by a thermocouple tip and ignited by a retractable coiled heating wire. The results showed that the ignition delay time increased with a decrease of the ambient pressure, due to an increased distance between kerosene and oxygen molecules. Steady burning and disruptive burning were identified following the ignition. OH* chemiluminescence images showed a spherical flame and a longer flame standoff distance under a lower pressure. The puffing intensity was observed to be enhanced with a reduction of the ambient pressure, and a decreased pressure was found to lower the sooting emission. *Keywords:* kerosene droplet, ignition, puffing, sooting, sub-atmospheric pressure

^{*}Corresponding author

Email address: wangzh@zju.edu.cn (Zhihua Wang)

1 1. Introduction

A ramjet is an air-breathing high-speed jet engine, which has the advantages of simple structures and high performance [1]. For a ramjet, the stableperformance height is below 20 km. As the flight altitude increases, the ambient pressure at the inlet of the combustion chamber will decrease to 0.1 bar-0.3 bar, due to a reduced air density. This situation will cause ignition difficulties and reduce the stability of the flame, which significantly affects the combustion efficiency and working performance of aircrafts [2]. Therefore, investigating liquid-fuel combustion mechanisms under low-pressure conditions is essential to improve the stability of the combustor under extreme conditions.

Kerosene is a commonly used fuel for ramjets, owing to its high energy den-11 sity and stable thermodynamic characteristics 3. To better understand the 12 combustion characteristics of liquid fuels inside the aero-engine chamber, inves-13 tigations of fuel droplet combustion is necessary and important 4-6. A first 14 key performance indicator of droplet burning is ignition. Law 7 investigated 15 effects of droplet heating on the ignition delay, and ignition was found to occur 16 when the Damköhler number of the system exceeded the ignition Damköhler 17 number. Kadota et al. 8 studied the ignition delay of a single-component 18 droplet at high pressure and found that the ignition delay decreased with an 19 increase in the ambient pressure. 20

In previous studies of a multicomponent fuel droplet, puffing is a common 21 phenomenon observed, characterized as a disruptive burning process 9-11. It 22 is now well known that puffing is due to the nucleation and growth of bubbles 23 inside the droplet, leading to bursting of the fuel droplet under superheating 24 12. It can enhance the internal mixing of the droplet, deform the liquid-air in-25 terface and the droplet shape, and cause secondary fragmentation of the droplet 26 **13**. The ejection of boiled vapor improves fuel-vapor/air mixing, which in turn 27 28 improves combustion. Effects of pressure on the occurrence of microexplosion, which is an extremely intense puffing, has been experimentally studied by Wang 29 et al. 14. Their results show that an elevated pressure advances the time of 30

its occurrence. However, the influence of pressure on puffing intensity is not yet
well understood.

Another important phenomenon related to kerosene droplet combustion is 33 sooting. As soot formed in a droplet flame is controlled by the evaporation of 34 liquid fuel, it is different from that formed in the combustion of gaseous fuel 15. 35 Moreover, the majority of the studies focus on soot characteristics of a burning 36 fuel droplet under atmospheric and elevated ambient pressures. Kadota et al. 37 16 also noted that increasing pressure enlarged the size of soot particles, but 38 did not change their structures at pressures ranging from 0.1 MPa to 3 MPa. 30 For a 1-Propanol droplet, Dakka and Shaw 17 also reported a more intense 40 sooting behaviour at 0.3 MPa and above, while soot was not observed at 0.141 MPa. 42

Although RP-3 kerosene is the most important aviation hydrocarbon fuel in China [18], experimental studies on the combustion characteristics, especially under sub-atmospheric pressure, are inadequate. In this study, high-speed, OH* chemiluminescence and soot thermal radiation imaging were used to study the ignition, puffing and sooting characteristics of Chinese RP-3 kerosene droplet burning under 0.2–1 bar. The effects of the ambient pressure on droplet burning are elaborated in several aspects.

50 2. Experimental setup

Fig. 1 shows a schematic of the experimental setup. The experiments were 51 performed in a 0.091 m³ pressure-controlled stainless-steel chamber, which has 52 four quartz windows for optical measurement. Compared to the droplet size, 53 the chamber volume is large enough so that the ambient-gas influence on droplet 54 combustion can be neglected. To support the droplet and simultaneously mea-55 sure droplet temperature, a 0.1 mm type-S platinum/platinum-rhodium ther-56 57 mocouple was used. A micro-pipette was used to produce a 1.45 ± 0.1 mm fuel droplet and place it on the tip of the thermocouple. The size of the thermocouple 58 is smaller than 10% of the droplet size, so the thermocouple did not significantly 59

influence the burning characteristics during droplet combustion [19]. A coiled
hot wire was used for ignition and withdrawn by an air cylinder immediately
after an ignition succeeded. The ignition time was precisely controlled by a
time-delay relay whose resolution is 0.01 s, and the current flowing through the
hot wire was maintained at about 3.10 A during the ignition.



Figure 1: Schematic of experimental setup

The burning history of a fuel droplet was recorded by two cameras. The first 65 one was a black/white high-speed camera (IDT Y4-S1) at 2000 frames/s with 66 an exposure time of 200 µs fitted with a Sigma 105 mm macro lens. To measure 67 the location of the flame front during the combustion, an intensified charged-68 coupled device (ICCD, PI-MAX4) camera was used to record OH* chemilumi-69 nescence signals, which was equipped with a Nikon PF10545MF-UV lens and 70 a narrowband interference filter centred at 310 nm. The ICCD equipped with 71 a 652 nm narrowband filter was used to record the concentration and distri-72

⁷³ bution information of the burning soot particles [20]. An Agilent 3970A data
⁷⁴ logger was used to collect the temperature inside the droplet at 10 Hz during
⁷⁵ the combustion process.

The droplet diameter was evaluated using Matlab. As the droplet was not spherical due to gravity and puffing, the droplet diameter was calculated by $\pi D^2/4 = A_p$, where A_p is the projected droplet area on the observing direction. The maximum uncertainty in droplet diameter measurements is estimated to be 10% [21].

The fuel used for all the experiments was Chinese RP-3 kerosene. The major

 $_{82}$ components are shown in Tables 1 22.

Saturated hydrocarbons Aromatic hydrocarbons Naphthenes Alkyl Naphthalene Indan & Alkanes Naphthalene Monocyclic Bicyclic Tricyclic Tetralin derivatives Benzenes 52.233.8 0.15.11.30.60.96.0

Table 1: Major components of Chinese RP-3 kerosene (mass fraction)

3. Results and discussion

⁸⁴ 3.1. Pressure influence on ignition delay

Once the igniter was moved to the target location which is 1 mm beneath the droplet, the droplet temperature began to rise, and volatile components of RP-3 on the droplet surface evaporated to form a vapor cloud surrounding the droplet. Ignition was identified by finding a spherical luminous region surrounding the droplet, which exceeded 50% maximum image intensity. The droplet size slightly increased because of the rise of the droplet temperature.

The ignition delay time is defined as the duration between the time when the igniter reaches the target location and when an ignition succeeds. The pressure influence on the ignition delay time is shown in Fig. 2. The ignition delay time increases as the pressure decreases from 1.0 bar to 0.2 bar. One reason is that the RP-3 and O_2 molecule numbers per unit volume decrease with a reduction



Figure 2: Ignition delay time of RP-3 droplet at different pressures

 $_{96}$ of the ambient pressure. This will result in an increased distance between the RP-3 and O₂ molecules, thereby reducing the probability of collisions between them.



Figure 3: OH* chemiluminescence during ignition at 0.2 bar

Fig. 3 shows the OH* chemiluminescence during the ignition at 0.2 bar. It can be seen that the ignition process of the droplet can be divided into two stages. The first stage is the slow oxidation process of RP-3 before the ignition, while in the second stage, an initially premixed flame propagates from the ignition kernel location and later surrounds the droplet. In the first stage, the
RP-3 droplet was heated by the igniter, and the most volatile component began
to evaporate and diffuse towards the location of the igniter. The fuel molecules
then pyrolyzed and oxidized because of the relatively high local temperature.
Among the reactions, the most significant elementary reaction to trigger the
ignition is

$$\mathbf{H} + \mathbf{O}_2 = \mathbf{O}\mathbf{H} + \mathbf{O},\tag{1}$$

which can generate a large amount of active OH* radicals to facilitate the igni-tion of the RP-3 droplet [23].



Figure 4: Time evolution of OH* chemiluminescence intensity at different pressures

Fig. 4 shows the OH^{*} chemiluminescence intensity during the ignition at 111 different pressures. As the pressure decreases, Reaction (1) needs much more 112 time to trigger the chain reaction of RP-3, because of a larger distance between 113 molecules. After the ignition, the OH* chemiluminescence intensity is stronger 114 under a lower pressure than that under a higher pressure, because of a larger 115 amount of evaporated RP-3 molecules surrounding the droplet caused by a 116 longer heating time. Another possible explanation for this is that the quenching 117 rate of OH* chemiluminescence decreased, due to the reduced collisional rate 118

¹¹⁹ between molecules under lower pressure 24.

¹²⁰ 3.2. Pressure influence on burning behaviour

Following the ignition, the burning behaviour of the RP-3 droplet under 121 different pressures were investigated next. As shown in Figs. 577, for all the 122 experiments, the burning of the RP-3 droplet can be divided into two stages: 123 (1) steady combustion; (2) disruptive combustion. Fig. 8 and Fig. 9 com-124 pare the size and temperature histories of the droplet at different pressures. 125 During the steady-combustion stage, the droplet temperature rapidly rose until 126 reaching the boiling temperature of the most volatile component of RP-3. An 127 envelope-shape flame formed surrounding the suspended droplet and the flame 128 size increased gradually. According to the OH* chemiluminescence figures, the 129 droplet burned smoothly without disruptions on the flame front. The luminous 130 yellow region surrounding the droplet indicated the broadband radiant emission 131 from soot 25. The droplet diameter continued to decrease with time, which 132 approximately agrees with the classical D^2 law 26. 133

Following the steady combustion, the RP-3 droplet displayed puffing be-134 haviour, resulting in disruptive burning. Fig. 10 shows the process of nucle-135 ation, multiple bubble formation, growth, merging, and inner circulation, which 136 eventually causes bubble breakup and sub-droplets ejection at 1 bar. Following 137 the steady burning stage, homogeneous nucleation occurred inside the droplet, 138 leading to several small bubbles. Then, bubbles began to grow once their sizes 139 surpassed the critical size, due to the high vapor pressure inside the bubbles 140 and thermal diffusion effects. With inner circulation, all the bubbles eventu-141 ally merged into one big bubble. Meanwhile, the droplet expanded slightly 142 because of the inner boiling. When the big bubble reached the droplet surface, 143 an ejection of fuel vapor and sub-droplets occurs. The distorted and bright 144 flame region indicates that the local equivalence ratio was enhanced owing to 145 the puffing. Puffing repeated multiple times until the extinction of the droplet. 146 Since RP-3 kerosene is a multicomponent fuel, more volatile components 147 will evaporate first at the droplet surface, and the surface temperature will 148



Figure 5: Burning sequences at 1 bar: (a) Images captured by high speed camera; (b) Images of OH* chemiluminescence emission captured by ICCD equipped with 310 nm filter



Figure 6: Burning sequences at 0.6 bar: (a) Images captured by high speed camera; (b) Images of OH* chemiluminescence emission captured by ICCD equipped with 310 nm filter



Figure 7: Burning sequences at 0.2 bar: (a) Images captured by high speed camera; (b) Images of OH* chemiluminescence emission captured by ICCD equipped with 310 nm filter

¹⁴⁹ be dominated by less volatile components with a higher boiling temperature. ¹⁵⁰ Meanwhile, more volatile components remain inside the droplet. Under con-¹⁵¹ tinuous superheating, the inner-droplet temperature will exceed the superheat ¹⁵² limit temperatures of more volatile components. The superheating will cause ¹⁵³ homogeneous nucleation inside the droplet, which leads to puffing.

Pressure effects on flame structure. When the ambient pressure is 1 bar, the 154 flame shape was distorted from the spherical symmetry because of the buoyancy 155 effect. Due to the natural convection, the downstream region was stretched, 156 while the upstream region was compressed. Considering that the droplet size is 157 relatively large, the natural convection effect became prominent, leading to high 158 flow velocities surrounding the droplets. Since the buoyance effect reduces as 159 the ambient pressure decreases, a less-stretched spherical flame was observed at 160 a lower pressure. And the flame front gradually moved away from the droplet 161 surface at a reduced pressure, which characterized as an increasing trend of the 162 horizontal flame standoff ratio, $D_{\rm fl}/D$. The horizontal flame standoff ratios for 163



Figure 8: Droplet size histories at different pressures



Figure 9: Droplet temperature histories at different pressures



Figure 10: The process of nucleation, bubble growth, merging of bubbles, inner circulation, breakup and eventually sub-droplets ejection at 1 bar

different pressures are as follows: $(D_{\rm fl}/D)_{\rm 1bar} = 2.93 \pm 0.22$; $(D_{\rm fl}/D)_{0.6\rm bar} =$ 3.91 ± 0.75 ; $(D_{\rm fl}/D)_{0.2\rm bar} = 6.47 \pm 0.65$, where $D_{\rm fl}$ is measured from OH* chemiluminescence images.

Pressure effects on droplet temperature. Following the ignition, the droplet 167 temperature first rapidly rose for a period and then slowly increased. During 168 the latter stage, the droplet temperature can be considered to be the same as 169 the pressure-dependent saturation temperature of RP-3 kerosene. As the boiling 170 temperature of the droplet decreases with the reduction of the ambient pressure, 171 the ignition and flame temperatures decreased accordingly. As shown in Fig. 9, 172 the boiling temperatures of RP-3 kerosene are within the ranges of: $478.4 \sim 538.7$ 173 K (1 bar), 472.7~529.9 K (0.8 bar), 469.7~513.7 K (0.6 bar), 435.4~487.9 K 174 (0.4 bar) and $390.8 \sim 454.9 \text{ K}$ (0.2 bar), respectively. 175

Pressure effects on puffing. Puffing was observed in all the experiments. As 176 shown in Fig. 8, the oscillations of the droplet size indicate flame disruption 177 due to puffing during the burning. The first puffing for all the cases occurred 178 after the ignition except at 0.2 bar. Due to a longer ignition time and a lower 179 boiling point under the low ambient pressure 0.2 bar, the inner boiling caused 180 ejection of fuel vapor and sub-droplets during the ignition, which facilitated the 181 ignition by enhancing the local equivalence ratio. However, the bubble growth 182 rate is much slower than that during the burning, owing to a lower temperature 183 of the droplet. 184

Moreover, the intensity of puffing was enhanced with the decrease of the ambient pressure. A much more severe distortion of the flame at sub-atmospheric pressure can be found in Figs. 5.77, which is caused by the ejection of vapor and multiple microdroplets. Meanwhile, the OH* chemiluminescence due to sub-droplet burning was captured in the figures.

As the intensity of puffing is controlled by the bubble growth process, which can be divided into three stages: (1) Inertia controlled stage, characterized by a rapid growth rate as a result of the difference between the pressure inside the bubble and the ambient pressure; (2) Transition stage, during which the interface velocity is significantly reduced; (3) Diffusion controlled stage, which is dominated by thermal diffusion effects, leading to a much slower bubble growth rate. During the inertia controlled stage, the bubble growth process can be modelled as 27:

$$R(t) = \left[\frac{2}{3}\frac{\rho_{\rm v}}{\rho_{\rm l}}A(T_0 - T_{\rm B})\right]^{1/2} \cdot t, \qquad (2)$$

198

$$P_{\rm v} - P_{\infty} = \rho_{\rm v} A (T_0 - T_{\rm B}),$$
 (3)

where $\rho_{\rm v}$ is the density of the saturated vapor inside the bubble, $\rho_{\rm l}$ is the density of the liquid, A is a linearization constant, T_0 is the initial temperature at the bubble boundary, $T_{\rm B}$ is the saturation temperature of the liquid, $P_{\rm v}$ is the vapor pressure inside bubble and P_{∞} is the ambient pressure.

Therefore, $P_{\rm v} - P_{\infty}$ in Eq.(3) increases with the reduction of the ambient pressure. Consequently, inertia controlled bubble growth will become more effective at a lower pressure, resulting in intense puffing of the droplet.

206 3.3. Pressure influence on sooting characteristics

As the broadband radiant emission from soot is characterized by the lu-207 minous yellow region above the droplet, sooty flames were observed in all the 208 experiments following the ignition. The presence of luminous sooty areas can 209 be caused by two factors. On the one hand, the upwardly directed natural 210 convection carries soot particles towards downstream. On the other hand, the 211 oxidation rate of soot particles is finite. Therefore, when soot particles pass 212 through the flame, they are heated and oxidized. The reaction takes a finite 213 time to complete and, therefore, extends above the flames 15. And the location 214 and dimension of the luminous yellow region is opposite with those of the fuel 215 and oxidizer zone, as shown in Fig. 57. As convection intensity increases at 216 high ambient pressure, the opening of the flame rear region and soot particle 217 escape will occur. 218

As shown in Fig. 11(a) and Fig. 7 with the reduction of the ambient pressure, the yellow luminous zone was observed to gradually stand further away



0.4 bar 0.6 bar 0.8 bar 1 bar

Figure 11: Sooting characteristics of RP-3 droplet burning at different pressures (a) Images captured by high speed camera; (b) Images of thermal radiation emission of burning soot particles captured by ICCD equipped with 652 nm filter

from the droplet surface and expand because of less convection effects. Mean-221 while, at 0.4 bar and 0.2 bar, the particles having escaped from the flame tip 222 became invisible. And the particle stream appeared darker and denser under a 223 higher ambient pressure. Fig. 11(b) shows the emission originated from thermal 224 radiation of the burning soot particles, the soot particles were found to be con-225 centrated on the edges of the downstream flow at all pressures, which agreed well 226 with the Laser-induced incandescence (LII) results obtained by Vander Wal et 221 al. 28. And the emission intensity decreased with the reduction of the ambient 228 pressure, as shown in Fig. 12. 229



Figure 12: Average emission intensity histories of soot particles thermal radiation at different pressure

The above observations are due to two reasons. First, the distance between molecules increases with the reduction of the pressure, and the probability of collision between small molecules decreases, which makes it difficult to form large molecules. Moreover, due to the weakening of the convection effect at lower pressures, the residence time of soot molecules in the oxidation zone increases, thereby generating less macromolecular polymers.

236 4. Conclusions

The ignition, combustion and sooting characteristics of a Chinese RP-3 kerosene droplet were investigated and compared under different sub-atmospheric ambient pressures. The following conclusions can be drawn from the present study:

1. The ignition delay time of the droplet increased with the reduction of the ambient pressure (1 bar - 0.2 bar), which is caused by an increased distance between fuel and oxidant molecules. Meanwhile, two distinctive stages were identified for the droplet ignition, i.e. a slow oxidation process followed by propagation of an initially premixed flame. According to the comparison of the OH* chemiluminescence, an increase of the time for the first stage contributed to a longer ignition delay time under a lower ambient pressure.

248 2. Following the ignition, the droplet underwent steady and disruptive burning
except at 0.2 bar. With the reduction of the ambient pressure, the buoyance
effect surrounding the droplet was weakened, resulting in a spherical flame
and a larger flame standoff ratio. At the same time, the RP-3 kerosene boiling
temperature decreased from 478.4~538.7K at 1 bar to 390.8~454.9K at 0.2
bar.

²⁵⁴ 3. Puffing was observed in all the experiments. The first disruptive ejection
²⁵⁵ occurred in the ignition stage when the ambient pressure dropped to 0.2 bar.
²⁵⁶ The reduction of the pressure enhanced the intensity of puffing because of
²⁵⁷ effective inertia controlled bubble growth at a lower pressure.

4. Under the ambient pressure below 0.4 bar, the particulates having escaped
from the rear flame region became invisible, and the average emission intensity originated from thermal radiation of the burning soot particles decreased
with the reduction of the ambient pressure, due to an increased distance between molecules and weakened convection effects.

263 Acknowledgement

Financial support from the China Scholarship Council, the National Natural Science Foundation of China (51621005), Fundamental Research Funds for the Central Universities (2020FZZX003-01-01), the Royal Society (IES\R3\193152) and the Engineering and Physical Sciences Research Council (EPSRC; EP/T033940/1) of the UK is gratefully acknowledged.

269 References

²⁷⁰ [1] T. Inamura, M. Takahashi, A. Kumakawa, Combustion characteristics of ²⁷¹ a liquid-fueled ramjet combustor, J. Propul. Power 17 (4) (2001) 860–868.

doi:10.2514/2.5817

- [2] W. L. Luo, Y. Pan, J. G. Tan, Z. G. Wang, Experimental investigation on
 combustion efficiency of the ramjet model at low pressure, J. Propul. Tech.
 31 (3) (2010) 270–275.
- [3] P. Dagaut, M. Cathonnet, The ignition, oxidation, and combustion of
 kerosene: A review of experimental and kinetic modeling, Prog. Energy
 Combust. Sci. 32 (1) (2006) 48–92. doi:10.1016/j.pecs.2005.10.003.
- [4] B. H. Chen, J. Z. Liu, H. P. Li, W. J. Yang, K. F. Cen, Laser ignition and
 combustion characteristics of al/jp-10 nanofluid droplet, J. Therm. Anal.

- [5] T. I. Farouk, Y. C. Liu, A. J. Savas, C. T. Avedisian, F. L. Dryer, Submillimeter sized methyl butanoate droplet combustion: Microgravity experiments and detailed numerical modeling, Proc. Combust. Inst. 34 (1)
 (2013) 1609–1616. doi:10.1016/j.proci.2012.07.074.
- [6] A. Ambekar, A. Chowdhury, S. Challa, D. Radhakrishna, Droplet combustion studies of hydrocarbon-monopropellant blends, Fuel 115 (2014)
 697-705. doi:10.1016/j.fuel.2013.07.056.

²⁸¹ Calorim. 135 (2) (2019) 925–934. doi:10.1007/s10973-018-7393-6.

- [7] C. K. Law, Theory of thermal ignition in fuel droplet burning, Combust.
 Flame 31 (1978) 285–296. doi:10.1016/0010-2180(78)90141-4.
- [8] T. Kadota, H. Hiroyasu, H. Oya, Spontaneous ignition delay of a fuel
 droplet in high pressure and high temperature gaseous environments, Bull.
 JSME 19 (130) (1976) 437-445. doi:10.1299/jsme1958.19.437.
- [9] J. Shinjo, J. Xia, L. C. Ganippa, A. Megaritis, Physics of puffing and microexplosion of emulsion fuel droplets, Phys. Fluids 26 (10) (2014) 103302.
 doi:10.1063/1.4897918.
- [10] D. C. K. Rao, S. Karmakar, S. K. Som, Puffing and micro-explosion behavior in combustion of butanol/jet a-1 and acetone-butanol-ethanol (a-be)/jet a-1 fuel droplets, Combust. Sci. Technol. 189 (10) (2017) 1796–1812.
 doi:10.1080/00102202.2017.1333502.
- [11] M. L. Botero, Y. Huang, D. L. Zhu, A. Molina, C. K. Law, Synergistic
 combustion of droplets of ethanol, diesel and biodiesel mixtures, Fuel 94
 (2012) 342-347. doi:10.1016/j.fuel.2011.10.049.
- [12] M. M. Avulapati, L. C. Ganippa, J. Xia, A. Megaritis, Puffing and micro explosion of diesel-biodiesel-ethanol blends, Fuel 166 (2016) 59–66. doi:
 10.1016/j.fuel.2015.10.107.
- A. Hoxie, R. Schoo, J. Braden, Microexplosive combustion behavior of
 blended soybean oil and butanol droplets, Fuel 120 (2014) 22-29. doi:
 10.1016/j.fuel.2013.11.036.
- [14] C. H. Wang, X. Q. Liu, C. K. Law, Combustion and microexplosion of freely
 falling multicomponent droplets, Combust. Flame 56 (2) (1984) 175–197.
 doi:10.1016/0010-2180(84)90036-1.
- [15] A. L. Randolph, C. K. Law, Influence of physical mechanisms on soot formation and destruction in droplet burning, Combust. Flame 64 (3) (1986)
 267–284. doi:10.1016/0010-2180(86)90145-8.

- [16] T. Kadota, H. Hiroyasu, A. Farazandehmehr, Soot formation by combus-316
- tion of a fuel droplet in high pressure gaseous environments, Combust. 317
- Flame 29 (1977) 67-77. doi:10.1016/0010-2180(77)90094-3. 318
- [17] S. M. Dakka, B. D. Shaw, Influences of pressure on reduced-gravity com-319 bustion of 1-propanol droplets, Microgravity Sci. Tec. 18 (2) (2006) 5–13. 320 doi:10.1007/BF02870978. 321
- [18] Y. Liu, Y. Liu, D. Chen, W. Fang, J. Li, Y. Yan, A simplified mechanistic 322 model of three-component surrogate fuels for rp-3 aviation kerosene, Energy 323 Fuels 32 (9) (2018) 9949-9960. doi:10.1021/acs.energyfuels.8b02094. 324
- [19] E. Mura, R. Calabria, V. Califano, P. Massoli, J. Bellettre, Emulsion 325 droplet micro-explosion: Analysis of two experimental approaches, Exp. 326 Therm. Fluid Sci. 56 (2014) 69-74. doi:10.1016/j.expthermflusci. 327 2013.11.020

328

333

- [20] W. Weng, M. Costa, Z. Li, M. Aldén, Temporally and spectrally resolved 329 images of single burning pulverized wheat straw particles, Fuel 224 (2018) 330 434-441. doi:10.1016/j.fuel.2018.03.101. 331
- [21] D. L. Dietrich, P. M. Struk, M. Ikegami, G. Xu, Single droplet combustion 332 of decane in microgravity: experiments and numerical modelling, Combust.
- Theory Model. 9 (4) (2005) 569-585. doi:10.1080/13647830500256039. 334
- [22] W. Zeng, H. Li, B. Chen, H. Ma, Experimental and kinetic modeling study 335 of ignition characteristics of chinese rp-3 kerosene, Combust. Sci. Technol. 336 187 (3) (2015) 396-409. doi:10.1080/00102202.2014.948620 337
- [23] B. H. Chen, J. Z. Liu, F. Yao, Y. He, W. J. Yang, Ignition delay charac-338 teristics of rp-3 under ultra-low pressure (0.01–0.1 mpa), Combust. Flame 330 210 (2019) 126-133. doi:10.1016/j.combustflame.2019.08.009. 340
- [24] M. Tamura, P. A. Berg, J. E. Harrington, J. Luque, J. B. Jeffries, G. P. 341 Smith, D. R. Crosley, Collisional quenching of ch(a), oh(a), and no(a) in 342

- low pressure hydrocarbon flames, Combust. Flame 114 (3) (1998) 502-514. 343 doi:10.1016/S0010-2180(97)00324-6. 344
- [25] J. B. Wei, B. D. Shaw, Reduced gravity combustion of propanol droplets in 345 oxygen-inert environments, Combust. Sci. Technol. 181 (12) (2009) 1480-346 1494. doi:10.1080/00102200903190851. 347
- [26] C. K. Law, H. K. Law, A d2-law for multicomponent droplet vaporization 348 and combustion, AIAA J. 20 (4) (1982) 522-527. doi:10.2514/3.51103 349
- [27] J. C. Lasheas, L. T. Yap, F. L. Dryer, Effect of the ambient pressure on the 350 explosive burning of emulsified and multicomponent fuel droplets, Symp. 351 (Int.) Combust. 20 (1) (1985) 1761–1772. doi:10.1016/S0082-0784(85) 352
- 80673-1. 353
- [28] R. L. Vander Wal, D. L. Dietrich, Laser-induced incandescence applied to 354 droplet combustion, Appl. Opt. 34 (6) (1995) 1103-1107. doi:10.1364/ 355 AD.34.001103
- 356