

# 1 **Combined Numerical and Experimental Studies on the Dynamic and Quasi-static Failure** 2 **Modes of Brittle Rock**

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10

## 11 **Abstract**

12 Loading rate affects not only the mechanical properties of rock but also the cracking behaviour  
13 or the failure mode. Firstly, experimental studies on the mechanical properties and the related  
14 cracking behaviour are studied based on the Carrara marble specimens containing a pre-  
15 existing flaw under different loading rates. The result shows that both the compressive and  
16 tensile strengths are rate-dependent but with different sensitivities to strain rate. Meanwhile,  
17 the cracking processes of the single-flawed specimens are also found to be rate-dependent with  
18 the aid of an ultra-high-speed video system. Secondly, to investigate the mechanism behind the  
19 experimental phenomena, a material model based on the experimental strength data is proposed  
20 and applied to the numerical model. The influence of the rate-dependent strength on the  
21 cracking behaviour is studied comprehensively by using the 2D and 3D explicit dynamic FEM  
22 code. The numerical results reveal how the rate-dependent strength affects fracturing behaviour  
23 and the failure stress of the single-flawed specimen. The plastic zones do not equal the damaged  
24 zone or the cracks. The dynamic loadings lead to the “X” shaped failure mode of the single-  
25 flawed specimen regardless of the flaw inclination angle. In contrast, the quasi-static loadings  
26 lead to the diagonal failure mode. It indicates that shear displacement dominates the whole  
27 cracking process under dynamic loadings, while it only dominates the failure period under  
28 quasi-static loadings.

29

30 **Keywords:** Rock dynamics; failure criterion; SHPB; crack; numerical simulation; strain rate

## 31        **1. Introduction**

32    In earth sciences, planetary sciences and rock engineering, impact loading or shaking and its  
33    influence on rocks are usually related and discussed, e.g. earthquake, meteorite impact, volcano  
34    eruption, landslide, tunnel rockburst and engineering blasting. The high and ultra-high loading  
35    rate, which can be as high as  $10^4 \text{ s}^{-1}$  in rock blasting and  $10^9$  or higher in meteorite impact, is  
36    encountered in these natural and engineering activities <sup>1, 2</sup>. The slip rate in an earthquake can  
37    be as high as several meters per second<sup>3</sup>, producing an acceleration exceeding  $20 \text{ m/s}^2$ . In this  
38    case, for a limited length of shear contact zone or asperity, the shear strain rate or the  
39    compressive strength rate can be very high. The shear rupture propagation speed can be higher  
40    than the shear wave speed for more than  $2000 \text{ m/s}$  <sup>4-6</sup>. But the generated loading rate decreases  
41    rapidly with the distance. To better understand the rock and lithosphere behaviours under  
42    dynamic loadings, the rate-dependent mechanical properties and failure processes of rocks are  
43    required to evaluate the failure of geological bodies. On the other side, the failure mode of  
44    geological bodies can help to back analyze the force resource (plate movement, earthquake,  
45    volcano eruption or meteorite impact). In engineering, the influence of rock blasting wave on  
46    the neighbouring tunnel, cavern and underground structures cannot be neglected<sup>7, 8</sup>. The  
47    stability of tunnel, cavern and underground structures under severe earthquake or explosions is  
48    also a focus of rock dynamics <sup>9-12</sup>.

49    In solid material mechanics, the mechanical properties of many brittle materials are dependent  
50    on the loading rate. Most construction materials and rocks are investigated to be sensitive to  
51    the loading rate for the mechanical properties <sup>13, 14</sup>. Rock shows its rate-dependence of  
52    mechanical properties ranging from quasi-static regime to the dynamic regime. Strength,  
53    fracture toughness and brittleness of rock or rock-like materials were reported to be affected  
54    by the loading rate, but to different extents <sup>15-25</sup>.

55    The dynamic strength of rocks has been studied by using various laboratory tests, e.g. drop  
56    weight test <sup>26-28</sup>, impact test <sup>29-31</sup>, and split Hopkinson pressure bar test <sup>16, 32-40</sup>. Dynamic  
57    compressive strengths of rock-like materials are more extensively studied than dynamic tensile  
58    strengths, shear strengths and dynamic fracture toughness due to a more complicated  
59    experimental setup needed for these tests. However, these less-studied properties of rocks have  
60    also been found to be loading rate-dependent to various extents <sup>20, 24, 37, 41-44</sup>.

61 Since the material properties influence the cracking behaviour, crack initiation and propagation  
62 should also be rate-dependent. Although this concept is generally accepted, there have been  
63 limited investigations on the cracking behaviour of rocks under dynamic loadings, and on the  
64 influence of rate-dependent mechanical properties on cracking behaviour. Previous literature  
65 on the dynamic crack behaviour generally concentrated on crack speed<sup>45</sup>, fragmentation<sup>46-48</sup>,  
66 fracture toughness<sup>49</sup> and the spalling phenomena<sup>46</sup>. Since intact rock always containing micro-  
67 defects or micro-cracks and rock masses are typically fractured or jointed, the rock specimens,  
68 which is cut open flaws at the center, are usually modelled to study the rock fracturing  
69 processes under different loading conditions. Rock or rock-like specimens with artificial flaws  
70 has been studied by experimental approaches and numerical simulations under quasi-static<sup>50-</sup>  
71 <sup>52</sup> and dynamic loadings<sup>53-57</sup>. The authors have investigated the dynamic cracking behaviour  
72 of marble using a single-flawed specimen with different flaw inclination angle by experimental  
73 methods<sup>58-60</sup>.

74 Carrara marble, which is a commonly used material to physically simulate the mechanical  
75 properties and failure process of rocks or rock masses<sup>61-65</sup>, is used to fabricate the specimens  
76 in the present study. One essential phenomenon associated with the cracking behaviour of such  
77 marble is the appearance of white patches before the initiation of visible macro-cracks under  
78 both quasi-static and dynamic loadings<sup>59, 61, 66, 67</sup>. Some authors also named the white patch  
79 shear belt or white belt. This phenomenon makes the observation of the cracking process more  
80 informative. The origin of the white patch has not been confirmed. It is usually considered the  
81 severely stripped (even crushed) material of marble<sup>66</sup> or the deviation/failure of crystalline  
82 grains in marble<sup>67</sup>. Based on the microscopic investigation by the scanning electron  
83 microscope (SEM) and the nanoindentation technique<sup>68, 69</sup>, the white patch is revealed to  
84 consist of zones with a variety of micro-cracks reducing the nanoindentation hardness and  
85 modulus.

86 Due to the limited sample rate of the data collection system and the high-speed propagation of  
87 the crack, as well as pulverization of specimens after dynamic tests, numerical modelling is  
88 developed to obtain more detailed information of the cracking process. By analyzing the  
89 simulation results, the physical relations between the rate-dependent strength and the dynamic  
90 fracturing modes can be explained in some ways. The dynamic FEM code – AUTODYN is a  
91 widely-used explicit FEM program in material dynamics research. Failure processes of both  
92 ductile and brittle materials have been simulated using this program by various researchers<sup>70-</sup>

93 <sup>74</sup>. Li and Wong (2012) <sup>70</sup> investigated the application of AUTODYN in the simulation of rock  
94 fracturing processes under both quasi-static and dynamic loadings. Huang et al. (2015) studied  
95 the crack coalescence of a sandstone numerically using AUTODYN <sup>75</sup>. The present two-  
96 dimensional numerical study extends the previous works in the material model correlation with  
97 the experimental results and rate-dependent failure criterion.

98 In this numerical study, a strain rate-dependent constitutive model is developed for the Carrara  
99 marble based on the experimental results obtained in quasi-static and dynamic tests. Since the  
100 plastic deformation increase with the strain rate in the dynamic regime, the Drucker-Prager  
101 yield criterion (D-P criterion) has been used and modified to a strain rate dependent failure  
102 envelope here. D-P criterion is a pressure-dependent model for determining the failure of a  
103 material or undergoing plastic yielding, has been applied to various brittle materials, such as  
104 rocks, concrete and polymers <sup>13, 71, 76-79</sup>. To test the validity of the plain-strain assumption in  
105 2D simulation, the 2D simulation results are compared with the 3D simulation results, which  
106 indicates that the 2D simulation can make a balance between simulation accuracy and  
107 efficiency. The numerical results generally agree with the experimental observations with  
108 respect to the fracturing phenomena under different loading conditions. Links between the rate-  
109 dependent mechanical properties and the different cracking processes are made. The results  
110 also verify the feasibility of this material model in the potential application into more practical  
111 rock engineering projects which are involved with high strain rate loadings.

112

## 113 **2. Methodology**

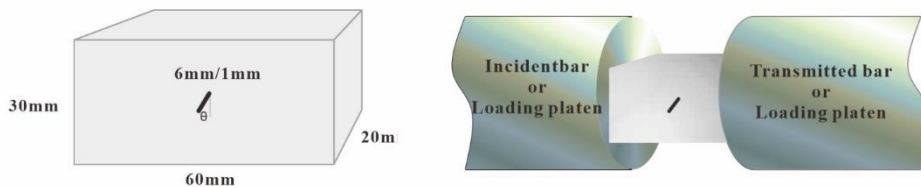
114 To minimize the size effect <sup>80, 81</sup>, specimens, which are fabricated from Carrara marble slabs,  
115 are of the same dimensions and geometries. The dimensions and flaw configuration of the two-  
116 dimensional and three-dimensional numerical models are the same as those of the experimental  
117 specimens. The compressive and tensile strengths under various strain rate from quasi-static to  
118 dynamic regimes are determined from the uniaxial compression tests and Brazilian disc tests  
119 conducted on intact specimens, respectively. The test systems include a hydro-servo  
120 compression machine and a split Hopkinson pressure bar (SHPB) setup for quasi-static and  
121 dynamic tests, respectively <sup>59, 82</sup>. For the data processing, the quasi-static and dynamic strength  
122 is calculated based on the suggested methods by ISRM <sup>83, 84, 38</sup>.

### 123 2.1 Experimental methodology

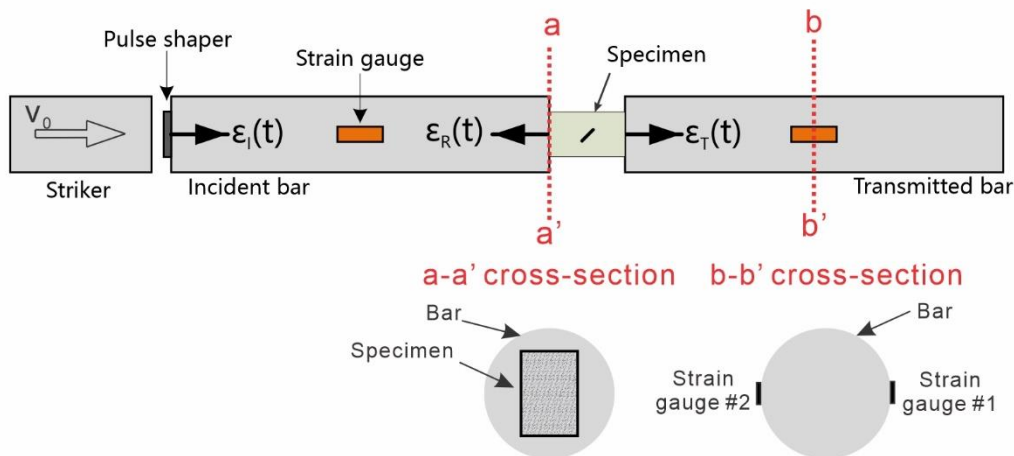
124 2.1.1 Experimental models

125 **Figure 1** shows the dimensions of the rectangular specimen. An abrasive jet cutter cuts the pre-  
 126 existing straight open flaw. The specimens with various flaw inclination angles  $\theta$ , including  $0^\circ$ ,  
 127  $15^\circ$ ,  $30^\circ$ ,  $45^\circ$ ,  $60^\circ$ ,  $75^\circ$ , and  $90^\circ$ , are studied in this paper. The specimen size effect and the  
 128 friction between the specimen and the bar should be considered in evaluating the strength and  
 129 cracking behaviours<sup>25</sup>. To minimize the size and end boundary friction effect to the cracking  
 130 behaviour, the length to width ratio of 2:1 is used, and the different friction coefficients are  
 131 used for quasi-static and dynamic simulations. The length is about 60 mm to make the specimen  
 132 have the ability to undertake a higher strain rate.

**A: Specimen**



**B: SHPB setup**



133

134 **Figure 1** Dimensions of the single-flawed specimen (A) and the setup of SHPB (B)<sup>59</sup>

135 2.1.2 Loading conditions

136 The strain rates applied to the intact specimens to determine specimen strength are shown in  
 137 **Table 1**. The quasi-static compressive strength, which is 90 MPa under a strain rate of  $10^{-4} \text{ s}^{-1}$ ,  
 138 refers to the average value in previously reported literature<sup>61-65</sup> and the tested value in the  
 139 present study. In the tests on the single-flawed specimens, the quasi-static strain rate is  $10^{-5} \text{ s}^{-1}$ ,  
 140 and the dynamic value is around  $150 \text{ s}^{-1}$ .

141 **Table 1** Strain rates applied on the intact specimens in different test types

Test type	Quasi-static Brazilian test	Dynamic Brazilian test	Quasi-static compression test	Dynamic compression test
Minimum strain rate (s <sup>-1</sup> )	10 <sup>-7</sup>	5	10 <sup>-4</sup>	100
Maximum strain rate (s <sup>-1</sup> )	10 <sup>-4</sup>	30	10 <sup>-4</sup>	600

142

143 The setup of the Split Hopkinson pressure bar (SHPB) system is also shown in **Figure 1**. A  
 144 copper pulse shaper, which is clung on the left end of the incident bar, is used to reduce the  
 145 wave dispersion effects and help achieve a better stress equilibrium condition. The high-speed  
 146 video subsystem captures the specimen failure process at a frame rate of 40,000 frames per  
 147 second.

### 148 2.1.3 Stress equilibrium

149 The satisfaction of the stress equilibrium state plays a crucial role in ensuring the data reliability  
 150 of dynamic SHPB tests. All the specimens have achieved the stress equilibrium in the dynamic  
 151 tests according to the ISRM suggested method. Two approaches are used to achieve the stress  
 152 equilibrium condition in a specimen.

153 (1) Relatively short specimens are used. The critical strain rate  $\dot{\epsilon}_m$ , which is the maximum  
 154 allowed strain rate for the stress equilibrium state, is calculated according to the following  
 155 equation<sup>57, 69</sup>:

$$156 \quad \dot{\epsilon}_m = \epsilon_f c_s / \alpha L$$

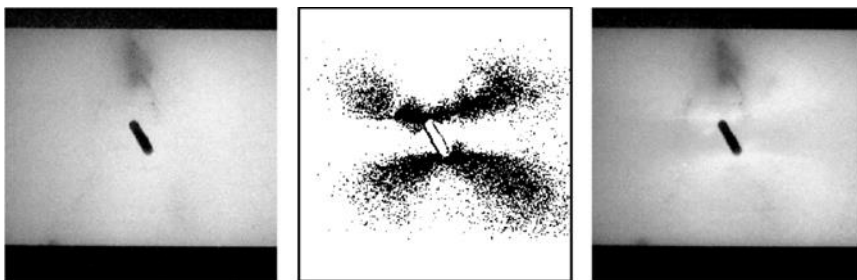
157 where  $c_s$  is the wave velocity in the specimen,  $\epsilon_f$  is the failure strain of the brittle materials,  
 158 and  $L$  is the specimen length. The times of the reflection  $\alpha$  are determined to be six for marble  
 159 specimens, which is a conservative value calculated according to Ravichandran and Subhash  
 160 (1994)<sup>85</sup>.

161 For the present study,  $\dot{\epsilon}_m$  is calculated to be 167 s<sup>-1</sup> for a 60 mm long Carrara marble specimen,  
 162 while it is determined to be 500 s<sup>-1</sup> for a 20 mm long specimen.

163 (2) A pulse shaper is used to promote a stress equilibrium state due to its relatively quick and  
164 easy deformation property<sup>86-90</sup>. It is normally a 1 mm or 2 mm thick copper with a diameter of  
165 20 mm adhering to the end of the incident bar using grease.

#### 166 2.1.4 White patch recognition

167 An image comparison program is used to help observe the subtle development of white patches  
168 in a white marble which is not easily distinguishable by unaided eyes. **Figure 2** shows a  
169 typically processed image by the image comparison software. The black dots represent those  
170 pixels experiencing a colour or greyscale change upon comparing two images.



171

172 **Figure 2** Processed high-speed image (middle) showing the pixel difference between the left  
173 (earlier) and right (later) images (flaw inclination angle 30°)<sup>59</sup>

#### 174 2.2 Numerical models

175 The numerical model creates a new strain rate-dependent Drucker-Prager yield criterion for the  
176 Carrara marble based on the present experimental data. By using this strain rate-dependent  
177 yield criterion, the deformation and cracking processes in the rectangular specimen containing  
178 a single flaw subjected to the dynamic and quasi-static loading tests are simulated and  
179 compared with the experimental results. Quadrilateral isoparametric elements are used under  
180 the plane strain assumption.

181 Since the wave reflection and transmission can happen at the bar-specimen interfaces, the  
182 simulations of a dynamic loading system only considering the specimen or a system  
183 considering the specimen and short loading platen cannot comprehensively represent the SHPB  
184 testing system in previous numerical studies. The length of the incident and transmitter bars  
185 should also be long enough to minimize the undesirable wave superposition. Moreover, to  
186 achieve a satisfactory stress equilibrium, the half wavelength of the incident wave is preferred  
187 to have a size comparable to the bar length in the experimental study. Therefore, a full-size  
188 SHPB model is therefore established instead. For simplicity of the numerical calculation, the

189 impact offered by the striking system is modelled as a time-history stress wave extracted from  
 190 the experimental data. The material properties of the steel and Carrara marble used in the  
 191 numerical models are shown in **Table 2**.

192 **Table 2** Material properties obtained from experimental studies

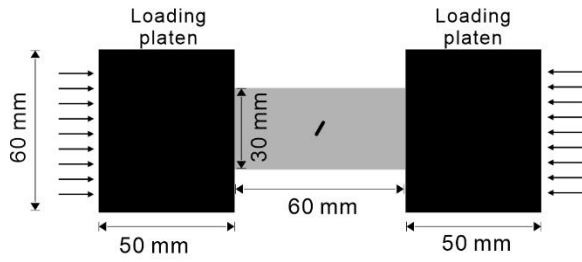
Carrara marble	Poisson's ratio, $\nu$	0.19
	Young's modulus $E$	49 GPa
	Dry density, $\rho$	2.7 g/cm <sup>3</sup>
	Bulk Modulus, $K$	26.3 GPa
	Shear Modulus, $\mu$	20.6 GPa
Maraging steel	Density, $\rho$	8.1 g/cm <sup>3</sup>
	Young's modulus $E$	210 GPa
	Poisson's ratio, $\nu$	0.3
	Bulk Modulus, $K$	175.0 GPa
	Shear Modulus, $\mu$	80.8 GPa
	Yield tensile strength, $\sigma_t$	>1000 MPa

193

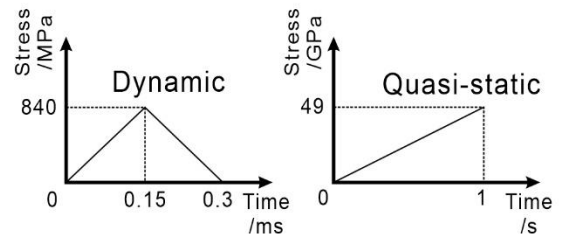
194 The numerical models of the quasi-static and the dynamic tests in the present study are shown  
 195 in **Figure 3**. The geometries and dimensions of specimens in these two kinds of loading  
 196 conditions are identical. Mixed rectangular and triangular mesh grids are used in the numerical  
 197 models. The rectangular grid is applied to mesh the bars or loading platens, while the triangular  
 198 meshing grid is used to mesh the specimen. The size of the mesh grid varies in the model to  
 199 make a balance between the calculation load and the precision. The bars are meshed by the  
 200 coarse grid due to their high strength and elastic modulus. A medium meshing grid is conducted  
 201 to the specimen. The mesh around the flaw is refined again to keep the necessary details of  
 202 stress and strain variation around the flaw tips.



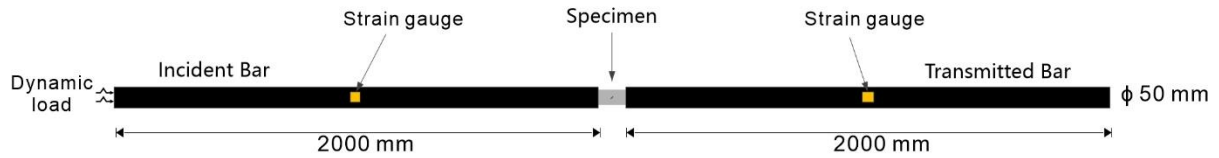
### A: Quasi-static test



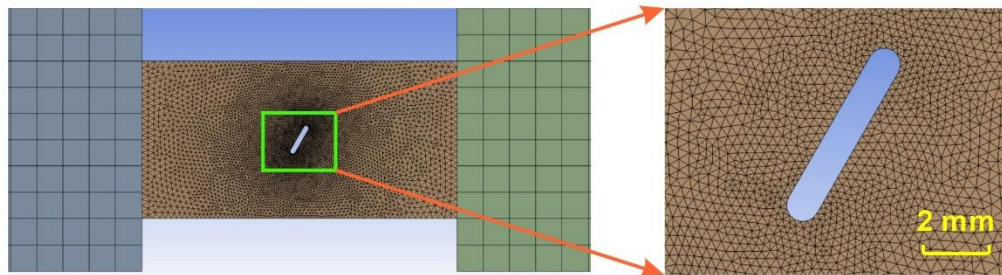
### C: Load-time curve



### B: Dynamic test



### D: Mesh grid



203

204 **Figure 3** Numerical models and boundary conditions of quasi-static (A) and dynamic (B)  
 205 tests as well as the input load-time curves (C) and mesh grid (D)

#### 206 2.2.1 Dynamic Drucker-Prager yield criterion considering strain rate effect

207 The D-P criterion has the form

208

$$\sqrt{J_2} = A + BI_1 \quad \text{Equation 1}$$

209 where  $I_1$  is the first invariant of the Cauchy stress and  $J_2$  is the second invariant of the  
 210 deviatoric part of the Cauchy stress. The constants  $A$  and  $B$  are usually determined from either  
 211 uniaxial or triaxial mechanical tests.

212 In the present study, the piece-wise Drucker-Prager strength criterion (D-P criterion) is used  
 213 for the Carrara marble, while the linear elastic strength criterion is applied to the steel bar since  
 214 the maximum stress in the test is much lower than the elastic limit. Since the marble strength  
 215 generally varies with the strain rate, a strain-rate dependent D-P criterion is more suitable for

216 the dynamic simulation. To avoid the massive computation in dynamic simulation, a constant  
 217 strain rate loading is assumed to simplify the calculation. Thus, only one piece-wise D-P  
 218 strength envelope is needed for a particular strain rate or loading rate, which significantly  
 219 increases computation efficiency. The determination flow chart of the failure criterion is  
 220 illustrated in **Figure 4**.

221 If  $\sigma_t$  is the yield stress in uniaxial tension,

$$\frac{1}{\sqrt{3}}\sigma_t = A + B\sigma_t \quad \text{Equation 2}$$

222 If  $\sigma_c$  is the yield stress in uniaxial compression,

$$\frac{1}{\sqrt{3}}\sigma_c = A - B\sigma_c \quad \text{Equation 3}$$

223 Then we can have the expression of A and B in this type:

$$A = \frac{2}{\sqrt{3}} \left( \frac{\sigma_c \sigma_t}{\sigma_c + \sigma_t} \right) \quad B = \frac{1}{\sqrt{3}} \left( \frac{\sigma_t - \sigma_c}{\sigma_c + \sigma_t} \right) \quad \text{Equation 4}$$

224

225 Under dynamic strain rates, for a specific strain rate,

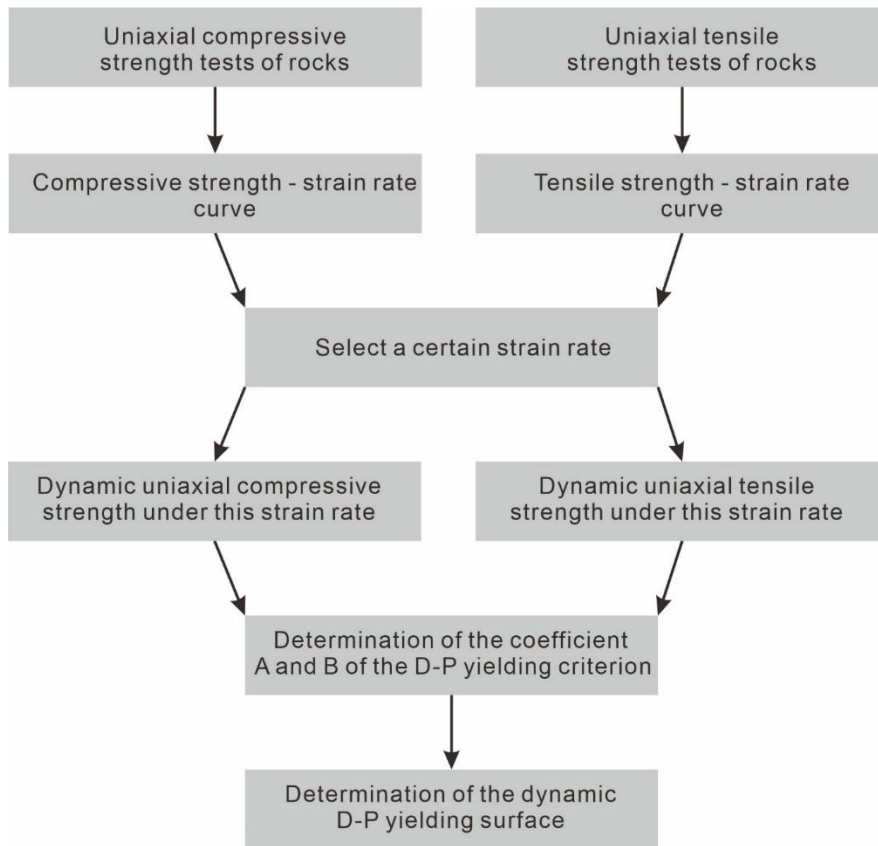
$$A_d = \frac{2}{\sqrt{3}} \left( \frac{\sigma_{cd}(\dot{\epsilon}) \sigma_{td}(\dot{\epsilon})}{\sigma_{cd}(\dot{\epsilon}) + \sigma_{td}(\dot{\epsilon})} \right) \quad B_d = \frac{1}{\sqrt{3}} \left( \frac{\sigma_{td}(\dot{\epsilon}) - \sigma_{cd}(\dot{\epsilon})}{\sigma_{cd}(\dot{\epsilon}) + \sigma_{td}(\dot{\epsilon})} \right) \quad \text{Equation 5}$$

226

227 Then, we have the dynamic Drucker-Prager strength criterion,

$$\sqrt{J_2} = A_d(\dot{\epsilon}) + B_d(\dot{\epsilon})I_1 \quad \text{Equation 6}$$

228 Since only uniaxial dynamic tests have been conducted on the marble in the present  
 229 experimental study, a two-point linear D-P strength envelope is first determined. Then, to avoid  
 230 the irrational strength increase in the linear envelope, the strength envelope becomes a gently  
 231 sloping line when  $I_1$  is higher than a certain level. This line is assumed to be parallel to the last  
 232 segment of the quasi-static piece-wise strength envelope. The final failure envelope is shown  
 233 in **Figure 6**.



234

235 **Figure 4** Flow chart of the dynamic D-P yielding surface determination under different strain  
 236 rates

### 237 2.2.2 Cumulative damage failure model

238 The cumulative damage failure model is commonly utilized to describe the macroscopic  
 239 inelastic behaviour of materials such as ceramics, concrete and rocks, the strength of which  
 240 will be significantly reduced by crushing<sup>91</sup>. The isotropic post-failure response of materials is  
 241 assumed for this failure model. When a cell of the material fails, it can no longer sustain any  
 242 shear or any negative pressure. In progressive crushing, a damage factor ‘*D*’ is used to model  
 243 the reduction of elastic moduli and yield strength of the material. In the standard situation, *D*  
 244 is zero for the material undergoing all plastic deformation, the effective plastic strain of which  
 245 is less than a value named EPS1. When the effective plastic strain increases to a specific value  
 246 EPS2, *D* reaches the maximum value  $D_{max}$  which is less than one. Between EPS1 and EPS2, *D*  
 247 increases linearly with the effective plastic strain.

248 The mathematical expression of *D* is illustrated in **Equation 7**.

$$D = D_{max} \left( \frac{EPS - EPS1}{EPS2 - EPS1} \right) \quad \text{Equation 7}$$

249 A damage function for the damage factor  $D$  is used to model the reduction of material properties  
 250 in the process of crushing. These mechanical properties include the bulk modulus, shear  
 251 modulus and yield strength of the material. For the fully damaged material, the compressive  
 252 strength reaches a residual, while the tensile strength becomes zero. The damaged yield  
 253 strength  $Y_{dam}$  is determined from the original yield strength  $Y$  and factor  $D$  as follows:

$$Y_{dam} = Y(1 - D) \text{ for a positive hydrostatic pressure } p \quad \textbf{Equation 8}$$

$$Y_{dam} = Y\left(1 - \frac{D}{D_{max}}\right) \text{ for a negative } p \quad \textbf{Equation 9}$$

254 In contrast,  $D$  is assumed to have no influence on the bulk modulus  $K$  and shear modulus  $G$  in  
 255 compression. However,  $K$  and  $G$  in tension will decrease to zero progressively when the  
 256 damage is complete or reaches the maximum value.

$$K_{dam} = K \text{ and } G_{dam} = G \text{ for a positive hydrostatic pressure } p \quad \textbf{Equation 10}$$

$$K_{dam} = K\left(1 - \frac{D}{D_{max}}\right) \text{ and } G_{dam} = G\left(1 - \frac{D}{D_{max}}\right) \text{ for a negative } p \quad \textbf{Equation 11}$$

257

### 258 2.2.3 Dynamic and quasi-static loading

259 Typical waveforms obtained from the SHPB test with the use of a pulse shaper have an  
 260 approximately triangular shape of incident wave<sup>59</sup>. Therefore, the triangular loading curve is  
 261 designed to simplify the numerical computation, as shown in **Figure 3C**. The strain rate of this  
 262 loading curve is  $150 \text{ s}^{-1}$  which is the average value of the strain rates used in the present  
 263 dynamic tests on the single-flawed specimens. Within the quasi-static regime, the strain rate  
 264 effect on the mechanical properties is inconspicuous. A representative constant strain rate of  $1$   
 265  $\text{s}^{-1}$  is assumed to increase the computation efficiency.

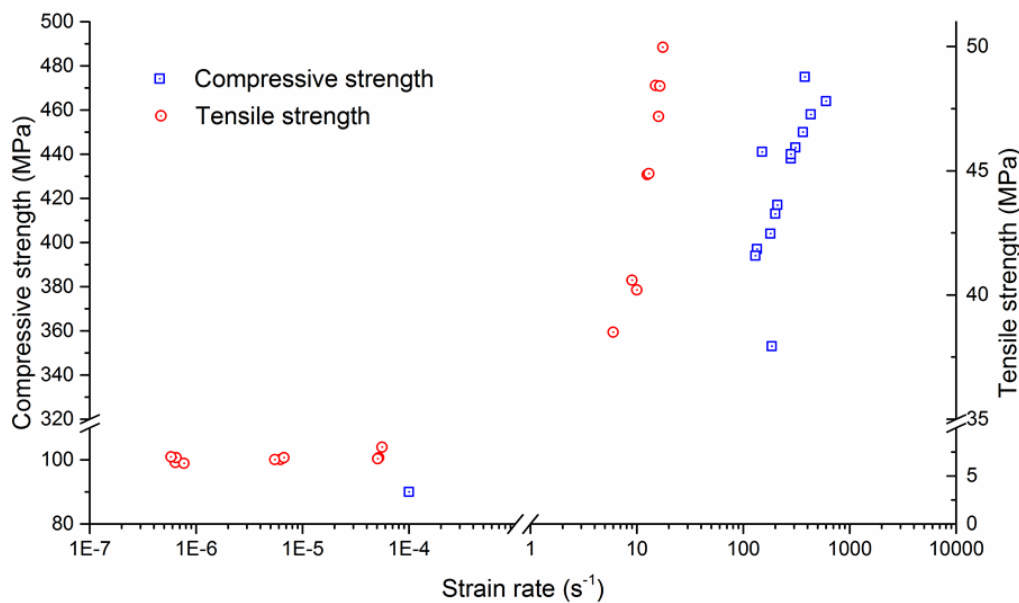
## 266 3. Experimental results

267 Typical fracturing processes of marble generally include the development of the white patches  
 268 and the macro-cracks. The initiation and propagation of white patches, which play an essential  
 269 role in the failure process, are therefore observed and discussed in detail. The fracturing  
 270 processes of single-flawed marble specimens with seven differently-oriented inclination angles  
 271 under both quasi-static and dynamic loading conditions are studied. The present experimental  
 272 and numerical studies focus mainly on a typical specimen possessing a  $30^\circ$  flaw inclination  
 273 angle in detail.

274 3.1 Compressive and tensile strength

275 In this study, the compressive and tensile strengths are first determined by the dynamic SHPB  
276 tests. Afterwards, the dynamic strength corresponding to a particular strain rate is used to  
277 determine the yielding surface under such strain rate. For quasi-static conditions, the piece-  
278 wise D-P yielding surface is derived from a series of triaxial compression tests reported in the  
279 previous literature <sup>92</sup>, and Brazilian tensile tests concluded by the authors.

280 The compressive and tensile strengths of Carrara marble are obtained under various strain rates  
281 in the present study, as shown in **Figure 5**. Since the dynamic SHPB tests on the single-flawed  
282 marble specimens are conducted under strain rates ranging from 120 to 180 s<sup>-1</sup>, a representative  
283 strain rate of 150 s<sup>-1</sup> is selected for the dynamic D-P yielding surface. The average dynamic  
284 uniaxial compressive strength and the average dynamic tensile strength under such strain rate  
285 are 400.5 MPa and 73.1 MPa, respectively.



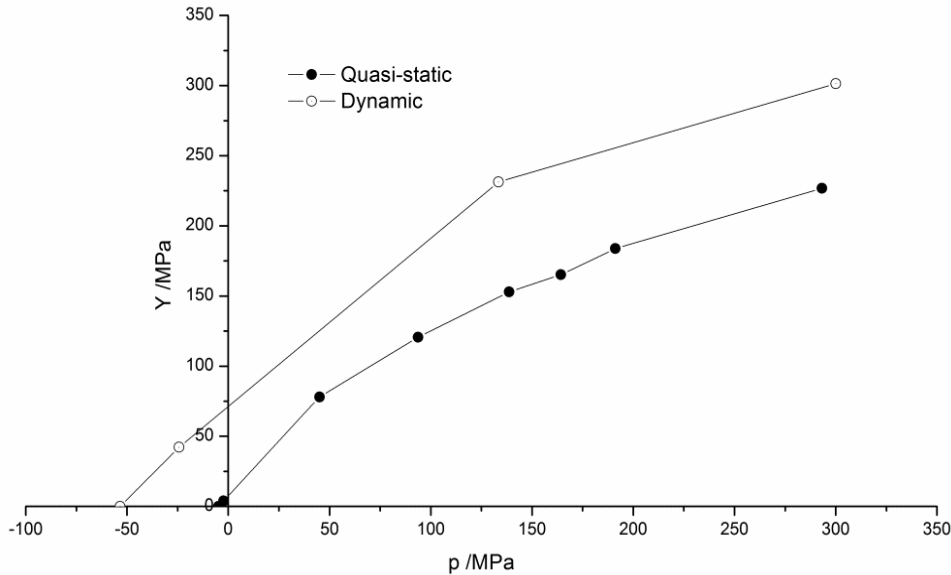
286

287 **Figure 5** Experimental compressive and tensile strengths of Carrara marble under various  
288 strain rates (Note: Breaks are inserted in the axes for better visualization)

289 3.2 Quasi-static and dynamic D-P yielding surfaces

290 The quasi-static D-P yielding surface is derived from the data provided in **Figure 6**. The values  
291 of yielding stresses corresponding to various confining pressures are calculated, based on  
292 which the piece-wise quasi-static yielding surface is constructed. The dynamic D-P yielding

293 envelope is also shown in the same figure. The corresponding strength values under such  
 294 loading conditions are listed in **Table 3**.



295

296 **Figure 6** Piece-wise quasi-static D-P yielding surface and linear dynamic D-P yielding  
 297 surface derived from the experimental data. The definitions of  $p$  and  $Y$  are shown in **Table 3**.

298

299 **Table 3** Stress conditions of Carrara marble under the strain rate of  $150 \text{ s}^{-1}$  ( $I_1$  is the first  
 300 invariant of the Cauchy stress.  $J_2$  is the second invariant of the deviatoric part of the Cauchy  
 301 stress)

Specimen No.	$\sigma_1$ /MPa	$\sigma_2$ /MPa	$\sigma_3$ /MPa	$p = \frac{1}{3}I_1$ /MPa	$Y = \sqrt{J_2}$ /MPa
1	0	0	-73.1	-24.4	42.2
2	400.5	0	0	133.5	231.2

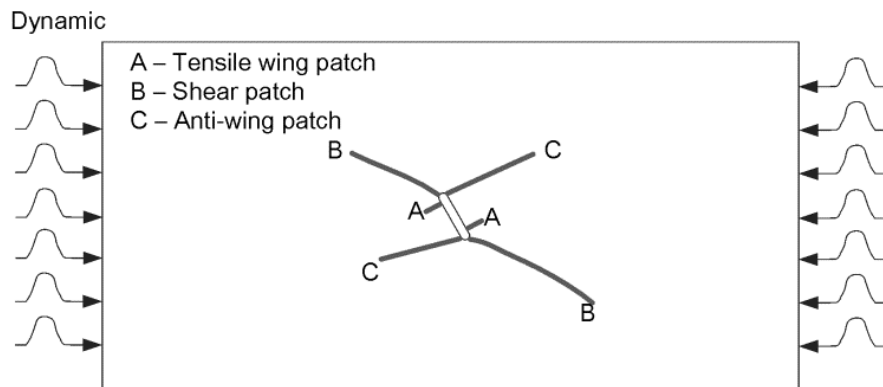
302

### 303 3.3 Fracturing processes of single-flawed marble specimens

#### 304 3.3.1 White patch development

305 White patches develop earlier than the initiation of the macro-cracks in marble. In this section,  
 306 the development of white patches and macro-cracks of the specimen containing an inclined  
 307 flaw ( $\beta = 30^\circ$ ) under both quasi-static and dynamic loadings are described and discussed. The

308 white patches initiated around the pre-existing flaw are also named the “first white patches”.  
 309 Three main types of white patches, namely tensile wing patches, shear patches and anti-wing  
 310 patches (**Figure 7**), are commonly observed in the experiments. The nomenclature of the first  
 311 patches is on the basis of the properties of macro-cracks that the white patches will eventually  
 312 develop into. Wong and Einstein (2009)<sup>93</sup> summarised seven first crack types initiated around  
 313 the single flaw under quasi-static loadings.



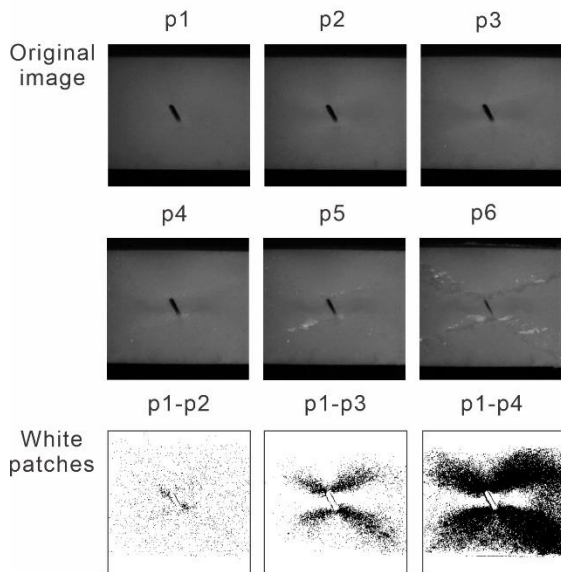
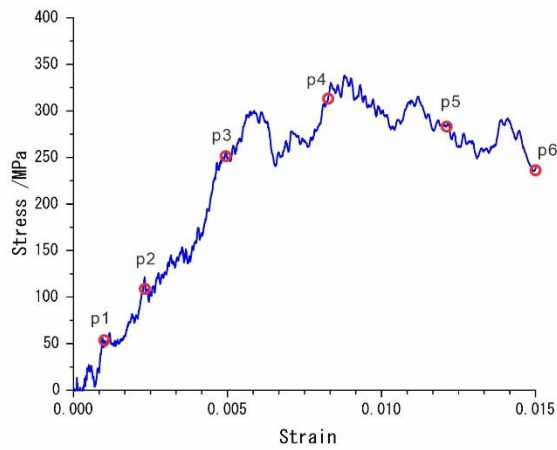
314

315 **Figure 7** Three types of first white patches developed under dynamic loading<sup>59</sup>

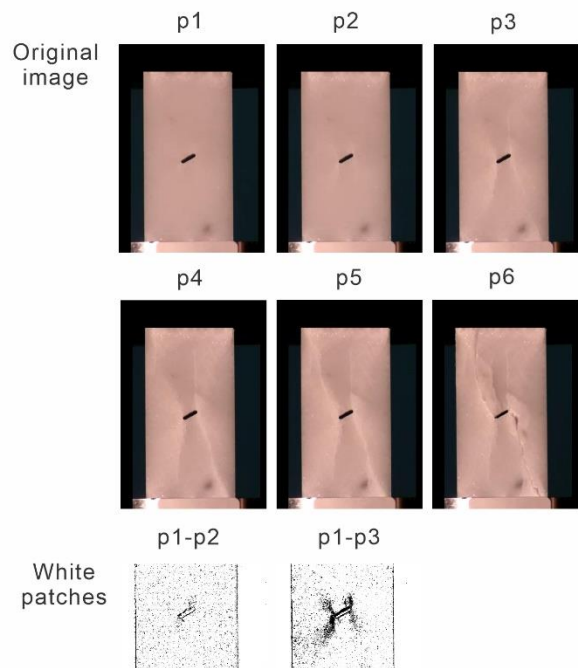
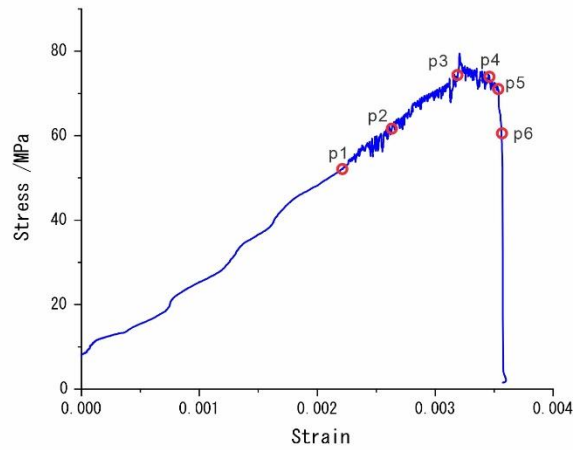
316 For the specimen loaded dynamically, both the tensile wing patches and the shear patches  
 317 initiate almost simultaneously around the flaw tips. For the specimen loaded quasi-statically,  
 318 generally, the first white patches are very similar to those developed under dynamic loadings  
 319 for the initiation position and the propagation direction at the early stage. The first white  
 320 patches also include the tensile wing and shear patches.

321 The anti-wing patches as the secondary white patches also initiate around the flaw soon after  
 322 the initiation of the first white patches under both quasi-static and dynamic loadings (**Figure**  
 323 **8**). The first and secondary patches constitute an ‘X’ shape pattern. This result indicates that  
 324 the influence of the loading rate on the first and secondary white patches at the early stage of  
 325 loading is very limited. However, some subtle differences are observed under these two loading  
 326 conditions, including the time-lapse and the weight of white patches. The white patches of the  
 327 dynamic tests occur at the very early stage of loading, while that of the quasi-static tests occur  
 328 at the second half of loading.

### A: Dynamic loading



### B: Quasi-static loading



329

330 **Figure 8** White patch and crack development in a single-flawed specimen under dynamic (A)  
 331 and quasi-static (B) loadings. Top row – original images. Bottom row – processed images  
 332 obtained from image comparison software (p1-pX means the comparison between p1 and  
 333 pX).

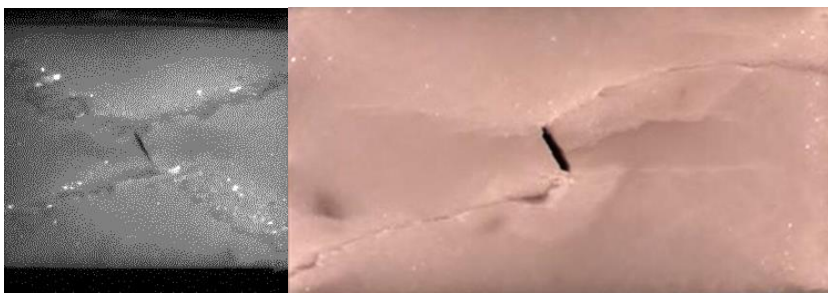
### 334 3.3.2 Macro-cracks

335 In response to further loading, macro-cracks with visible aperture openings initiate along the  
 336 trajectories of the previously initiated white patches. The future-developed macro-cracks share  
 337 the same trajectories with the white patches (**Figure 9**). However, not all the white patches will  
 338 evolve to the macro-cracks, which is generally observed in other single-flawed specimens with  
 339 different inclination angles<sup>59, 82</sup>. In dynamic tests, one pair of the shear patches and another  
 340 pair of anti-wing patches will evolve to an “X” shape crack pattern. No tensile wing patches



341 will develop into any visible macro-cracks. In contrast, in quasi-static tests, only the anti-wing  
342 patches lead to the failure of the specimens.

343 As revealed from additional studies on specimens containing a flaw oriented at different  
344 inclination angles, the failure modes are also found to be significantly different under different  
345 loading rates. Under dynamic loading, two symmetrical pairs of shear cracks with an “X”  
346 shaped failure crack band lead to the specimen failure pattern. In contrast, under quasi-static  
347 loading, two major diagonal macro-cracks evolved from the anti-wing patches will cause the  
348 specimen failure regardless of the inclination angle. The complete crushing of the specimens  
349 appears after the peak stresses of the loading regardless of the loading rates.



350

351 **Figure 9** Failure modes of the specimens under dynamic (left, No. s20111220-4-30M) and  
352 quasi-static (right, No. 20121123-2-30) loadings. The crack tracks are highlighted in the right  
353 image.

354

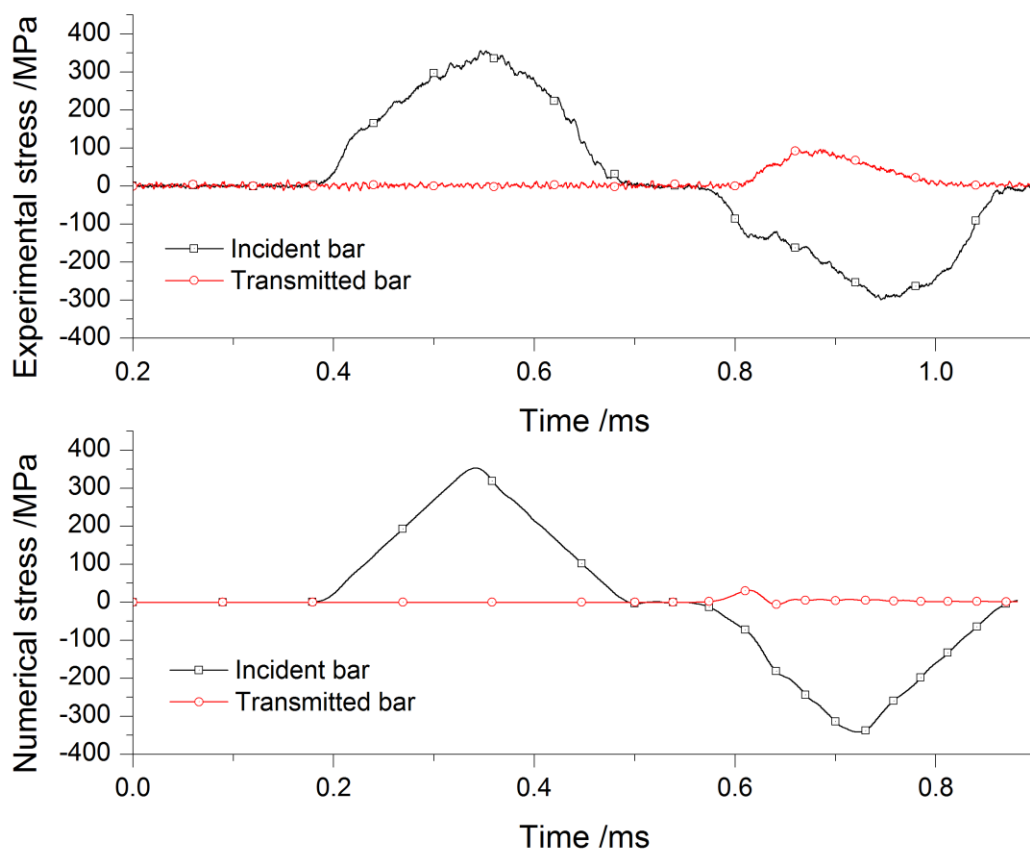
#### 355 **4. Numerical results**

356 In this section, numerical results of the mechanical responses of the specimen based on the  
357 material properties derived from the experiments are discussed. The cracking processes are  
358 illustrated and compared with the experimental results.

##### 359 4.1 Model calibration

360 Two monitoring points are set in the numerical model of dynamic SHPB tests to record the  
361 stress histories of the incident and transmitter bars (**Figure 10**). The numerical stress histories  
362 are compared with the experimental ones. Due to the assumption of the ideal triangular incident  
363 wave in the numerical studies, the numerical stress histories appear to be smoother than the  
364 experimental ones. Generally, the numerical and experimental stress histories are similar to  
365 each other. However, due to the 2-D simulation nature and the lack of dynamic triaxial test  
366 results on the marble, the simulated peak stress (which is related to the compressive strength)

367 of the transmitted wave is slightly less than the experimental one. Therefore, in the present  
 368 study, the numerical result of the strain-stress curve has a lower magnitude compared with the  
 369 experimental one. This shortage will be repaired and optimized by 3D simulation and more  
 370 dynamic triaxial tests to establish a more accurate 3D Drucker-Prager criterion. But the  
 371 dynamic peak stress of the specimen is several times higher than the quasi-static peak stress  
 372 (**Figure 11**), which is consistent with the experimental behaviour. Therefore, in the strain-stress  
 373 curves, the relative position of each cracking event can be considered and analyzed in dynamic  
 374 failure processes.



375  
 376 **Figure 10** Experimental stress waves of the SHPB tests on the single-flawed marble,  $\beta = 30^\circ$   
 377 (top) and simulated stress waves (bottom)

378 4.2 Cracking processes

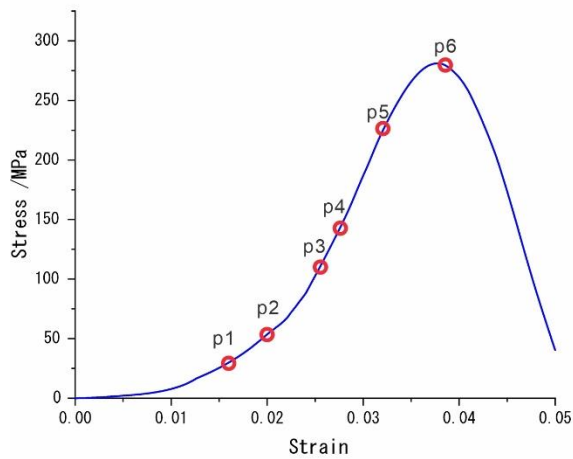
379 In the cumulative damage model, the failure of a cell represents the development of a segment  
 380 of a macro-crack. The damage model is found to be able to simulate the initiation and  
 381 propagation of macro-cracks in the single-flawed specimen. In this model, the EPS1 and EPS2  
 382 are chosen to be 0.001 and 0.03, respectively, according to the experimental results. Other basic

383 material properties are chosen to be identical in the numerical studies of quasi-static and  
384 dynamic tests. A lower friction coefficient at the interface between the specimen and the bar is  
385 used under dynamic loading <sup>94</sup>.

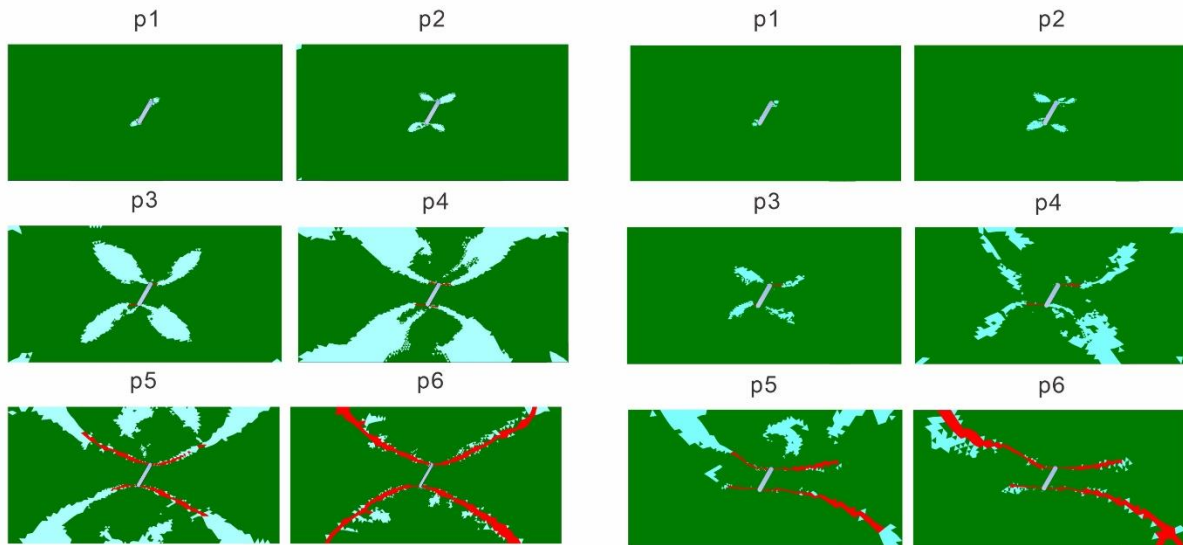
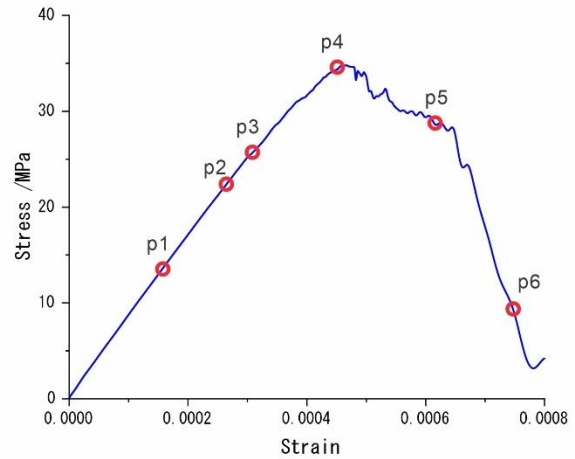
#### 386 4.2.1 Quasi-static loading

387 As shown in **Figure 11B**, the first cracks initiate at the flaw tips have the same positions as the  
388 tensile wing patches observed in the experimental study. With increasing loading, the  
389 propagation of the first cracks is limited, while two anti-wing cracks occur and propagate  
390 symmetrically around the flaw. Subsequently, shear cracks appear nearly at the same position  
391 as the first cracks do, but in different propagation direction. Finally, the development of the  
392 anti-wing cracks forming a diagonal crack band leads to specimen failure. In this period, the  
393 shear cracks propagate to a limited extent. The shear cracks and the anti-wing cracks constitute  
394 a diagonal failure band around the flaw, which agree with the experimental observation of the  
395 cracking processes (**Figure 9**).

**A: Dynamic loading**



**B: Quasi-static loading**



396

397 **Figure 11** Stress-strain curve and the cracking processes under dynamic (A) and quasi-static  
 398 (B) loadings in numerical studies (Red – cracks, Cyan – lower damage zone)

399 4.2.2 Dynamic loading

400 The simulation results of the dynamic test also well agree with the experimental cracking  
 401 processes. **Figure 11A** shows the first cracks of the same specimen loaded dynamically. The  
 402 geometry and position of the first cracks, which are also tensile wing cracks, under dynamic  
 403 loading are similar to those under quasi-static loading. When the loading increases, the  
 404 propagation of the first tensile wing cracks will be suppressed, and it changes its propagation  
 405 direction and turns to be shear. The initiation and growth of the shear cracks and anti-wing  
 406 cracks are favoured then. The shear and anti-wing crack patterns form an “X” shape. In contrast,  
 407 the shear cracks are more developed than those under quasi-static loading. With the increase  
 408 of the dynamic loading, both the shear cracks and anti-wing cracks develop preferentially.  
 409 Consequently, an “X” shaped cracking mode dominates the specimen failure. Apart from the

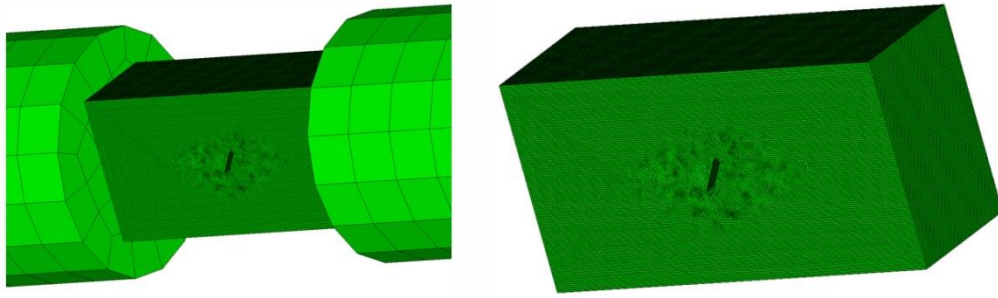
410 cracking behaviour, the simulated failure curves, as well as the failure strain, also show a  
411 similar result with the experiments. The simulated dynamic stress-strain curve is smoother and  
412 has a much higher failure strain, which implies the increase of ductileness under dynamic  
413 loading. This phenomenon is supported by the large-scale spreading of the plastic zone in the  
414 dynamic simulation too. The result indicates the validity of the present numerical model,  
415 including the material model and the failure criterion.

#### 416 4.3 3D simulation

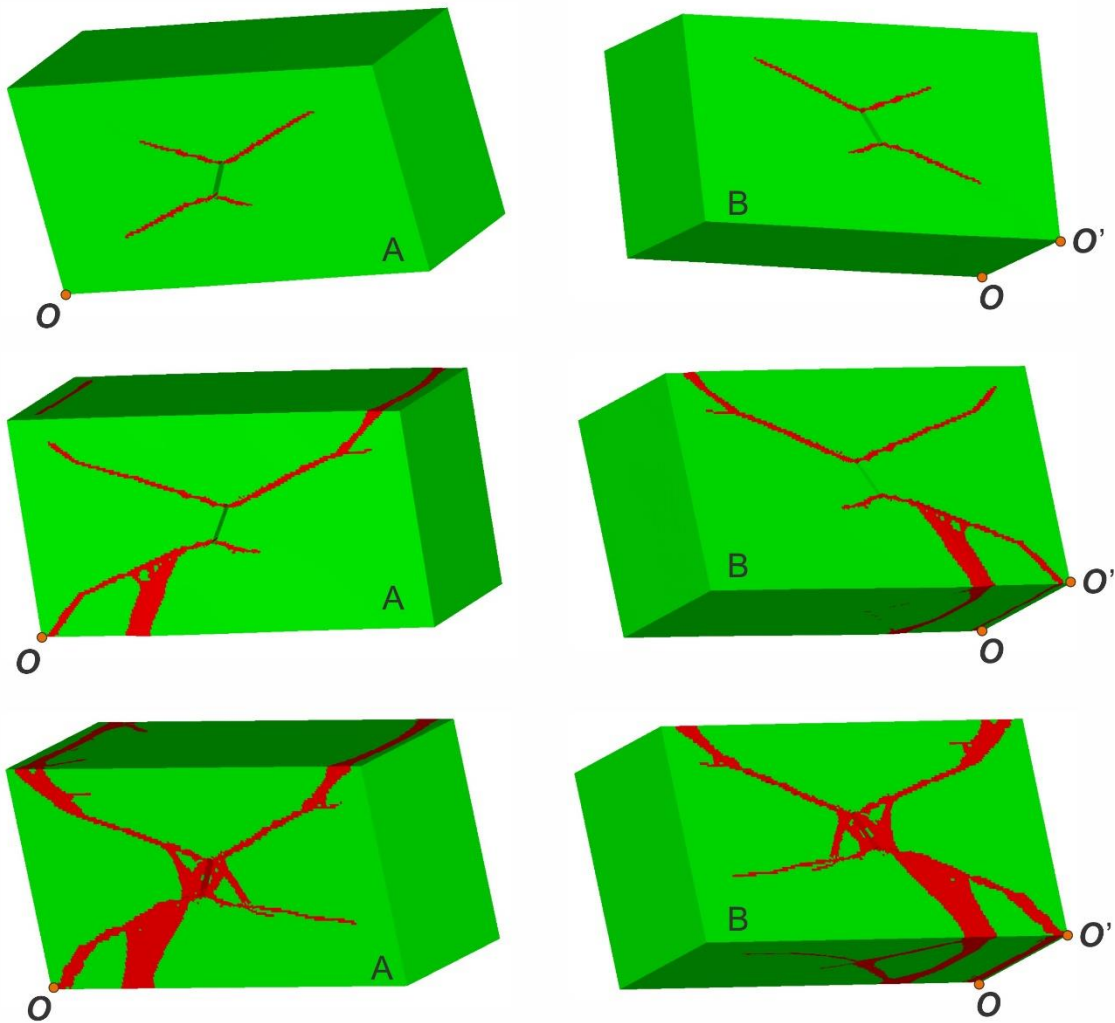
417 In order to verify whether the difference between 2D and 3D simulation result is acceptable, a  
418 3D dynamic simulation is run and compared with the 2D result. The 3D simulation will  
419 consume nearly a hundred times of computation time even though the rectangular element and  
420 the relatively coarse element are used to improve the computation efficiency. The numerical  
421 model is shown in **Figure 12**. The cracking patterns on the specimen surface of the 3D  
422 simulation are generally similar to those of the 2D simulation. The 3D simulation result also  
423 indicates that the plane strain assumption is acceptable in the study of dynamic cracking  
424 processes using 2D FEM simulation.

425 The 3D simulation shows that the X shape failure mode appears not only on the front and  
426 reverse surfaces but also on the top and bottom surfaces. On the top and bottom surface, two  
427 diagonal cracking bands lead to the failure mode. The transparent view of the failure band show  
428 that the failure near the surface or the flaw boundary occurs first, and the side failure plane  
429 occurs consequently. It is caused by the unconfinement stress state on the free surface. The  
430 differences between the 2D and 3D simulations are minor but including (1) Relatively more  
431 branching cracks occur in the 3D simulation; (2) Similar X shape failure band occurs at the  
432 bottom surface in the 3D simulation; (3) The 3D failure zone around the pre-existing flaw is  
433 larger than the 2D condition

A: 3D model of SHPB



B: Simulation result



A: Front surface    B: Reverse surface

434

435 **Figure 12** 3D numerical model and the corresponding cracking processes of the single-flawed  
436 specimen

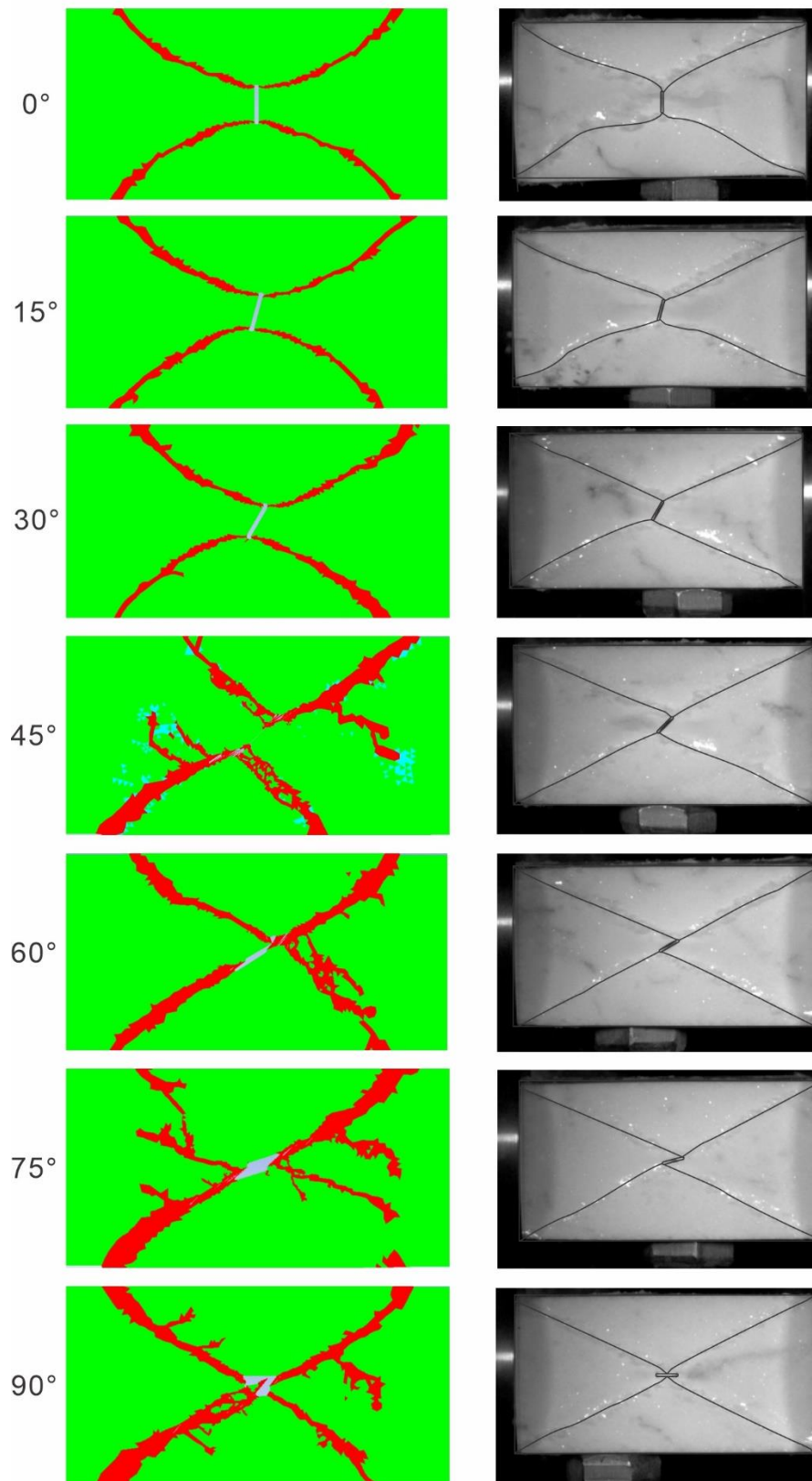
437 4.4 Summary

438 Similar to the experimental observation, the first cracks initiate at the flaw tips. It is due to the  
439 high stress concentration around the flaw tips and the low ratio of tensile strength to  
440 compressive strength for the Carrara marble. Even though the tensile strength has a higher rate-  
441 dependency than the compressive strength, the enhanced dynamic tensile strength is still  
442 significantly less than the dynamic compressive strength. Tensile cracks still appear first. After  
443 the initiation of the first cracks, the shear cracks initiate nearly simultaneously along with the  
444 secondary cracks. The further propagation of the first tensile cracks is suppressed under both  
445 loading conditions. The first tensile cracks evolve to shear cracks under dynamic loadings,  
446 while those under quasi-static loadings have minimal propagation. The former type of  
447 suppression leads to the “X” shaped failure mode under dynamic loadings, and the latter type  
448 of suppression leads to the diagonal failure mode under quasi-static loadings. In **Figure 13**, the  
449 numerical result of the specimens with other flaw inclination angles is compared with the  
450 experimental results under dynamic loading, which is reported in the author’s previous paper.  
451 The comparison shows similar failure modes and proves the universal “X” shape failure mode  
452 under dynamic loading. The “X” shape failure mode shows independence on the flaw  
453 inclination angle for the single flawed specimen under dynamic loadings. This finding may  
454 provide some tips about the analysis of the historical forces on the pre-existing fractures, joints  
455 or faults. The historical loading rate of the forces can also be given by analyzing the crack  
456 patterns.

457

• Left column:  
Numerical result

• Right column:  
Experimental result



458

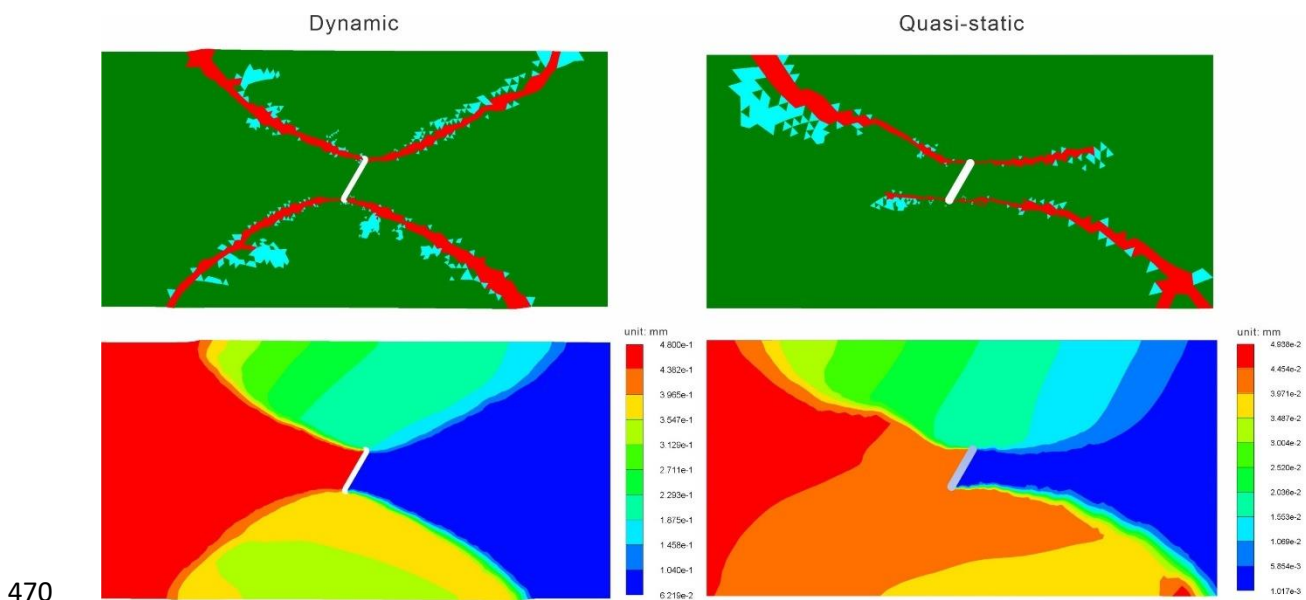
459 **Figure 13** Failure modes of the single-flawed specimens with different flaw inclination  
460 angles in numerical and experimental results<sup>59</sup>



461 **5. Discussion**

462 5.1 Failure mode

463 From both the experimental and numerical results, the development of shear cracks under  
464 dynamic loading is more favoured than that under quasi-static loading. Furthermore, the  
465 remarkable growth of shear cracks with the growth of anti-wing cracks constitutes a  
466 symmetrical “X” shaped failure mode under dynamic loading <sup>59</sup>. From the displacement  
467 contour (**Figure 14**), a significantly higher displacement difference is found in the dynamic  
468 condition compared with the quasi-static condition. The result also indicates higher shear stress  
469 or strain under dynamic loadings, which dominates the failure of the specimen.



470  
471 **Figure 14** Failure mode and displacement contour under dynamic and quasi-static loadings in  
472 the numerical simulations

473 5.2 Suppression of the tensile wing patches

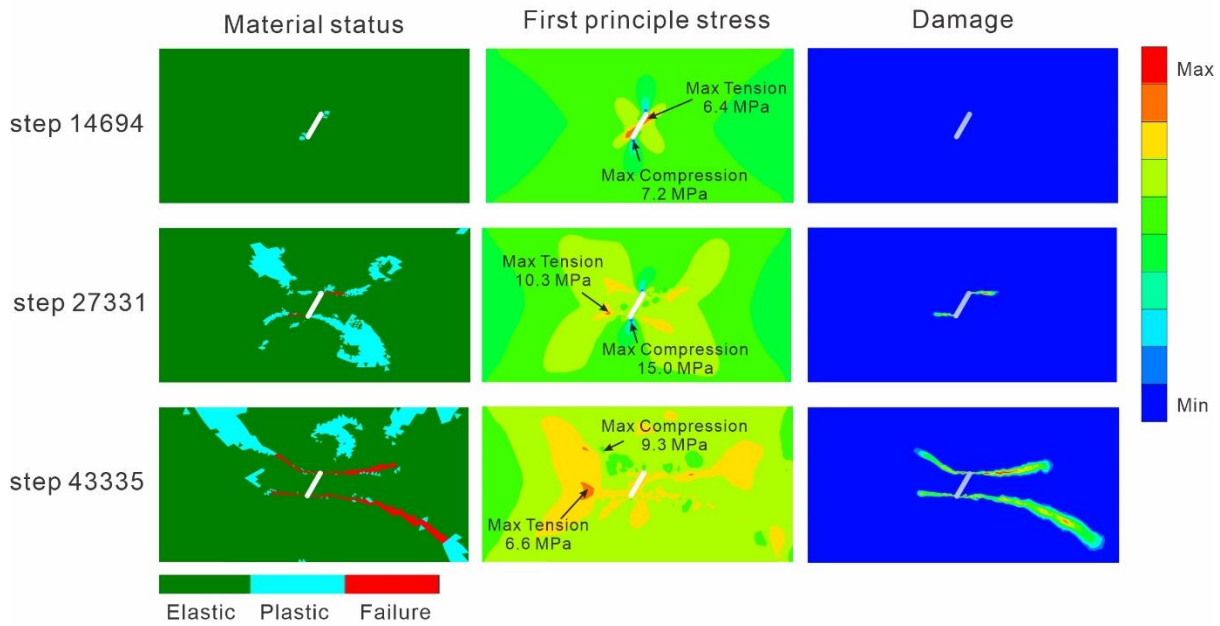
474 The tensile-wing patches are suppressed under dynamic loadings, which might be the result of  
475 a relatively higher tensile-compressive strength ratio under dynamic strain rates than that under  
476 quasi-static strain rates. In other words, the tensile strength increases faster than the  
477 compressive strength with the strain rate. The tests on the compressive dynamic increase factor  
478 (CDIF), which is the dynamic strength value normalized to the average quasi-static strength  
479 value, and tensile dynamic increase factor (TDIF) support this point. From the experimental  
480 results, around the dynamic strain rate of  $150 \text{ s}^{-1}$ , the TDIF is more than twice the CDIF.  
481 Therefore, for the same stress concentration factor, a higher TDIF will suppress the

482 development of the tensile wing cracks. On the other hand, higher TDIF and lower CDIF will  
483 encourage the development of shear cracks.

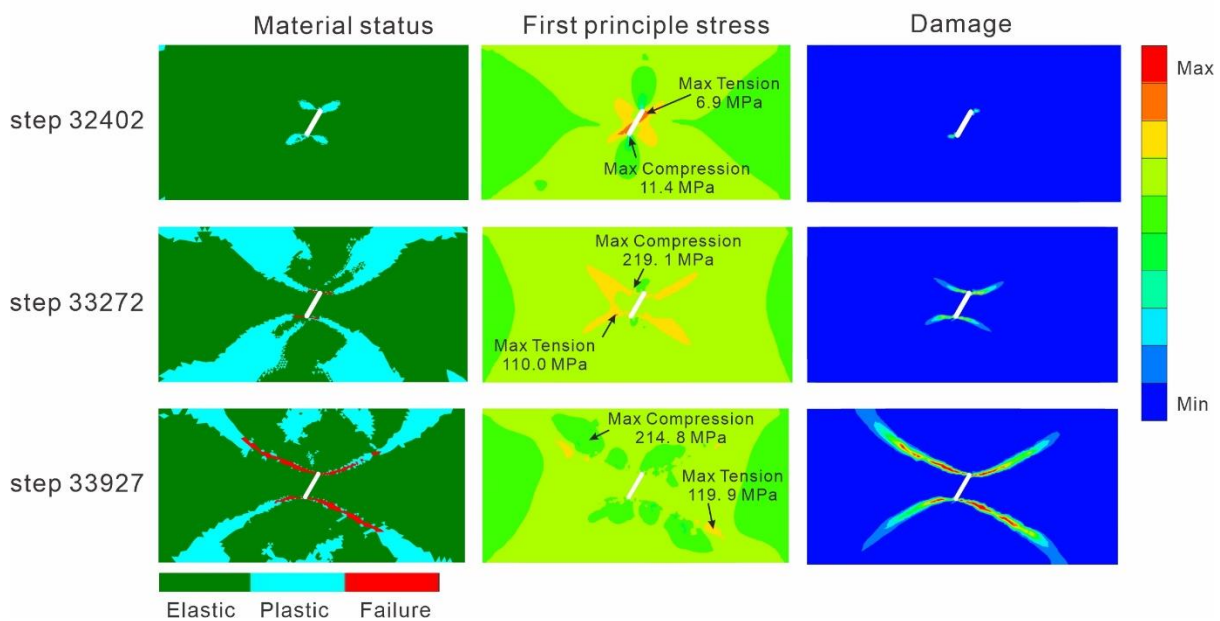
### 484 5.3 Similar crack initiation

485 The stress concentration coefficient around the flaw depends on the geometry of the flaw and  
486 the specimen, not the loading rate and the material mechanical properties before the regional  
487 failure of the specimen. It is one of the reasons that account for the similar fracturing and  
488 deformation phenomena observed under different loading conditions, especially at the early  
489 stage of the cracking associated with the initiation of the white patches in Carrara marble. This  
490 point is also supported by the numerical results of the principal stress, as illustrated in **Figure**  
491 **15**. However, the variation of TDIF and CDIF affects the resistance of propagation of different  
492 types of cracks.

**A: Quasi-static loading:**



**B: Dynamic loading:**



493

494 **Figure 15** First principal stress and damage zone at different loading step in the quasi-static  
 495 (A) and dynamic (B) loading simulation

496 5.4 Principal stress distribution and damage

497 Since both the tensile strength and the compressive strength increase with the strain rate, the  
 498 maximum tensile and compressive stresses inside the specimen also increase significantly,  
 499 especially in the stage of the crack propagation. The maximum dynamic tensile stress can be  
 500 nearly ten times the quasi-static one, which is consistent with the dynamic strength increase

501 factor (**Figure 15**). In the experimental result, the peak stress or strength of the single-flawed  
502 specimens also increases. The numerical result well explains why the failure stress on the  
503 specimen under dynamic loadings is much higher than that under quasi-static loadings.

504 The plastic zones do not always equal the damaged zone or the cracked zone. Early-stage  
505 loading can lead to the regional plastic zone, which is very similar to the white patches from  
506 the aspect of shape. However, since it is less than the EPS1, it is not a damaged zone. When  
507 the loading continues to increase, the stress keeps redistribution. More plastic zones appear and  
508 develop more. Only some of them remain the rising trend and reach the EPS1. These plastic  
509 zones become the damaged zone and finally evolve to the cracks. It means not all the white  
510 patches can develop into the macro-cracks.

#### 511 5.5 Future work

512 The numerical study is performed based on the assumption that the different TDIF and CDIF  
513 at a specific dynamic strain rate affect the cracking behaviour of the marble specimens.  
514 However, other mechanical properties, which may also be strain rate-dependent, can also  
515 possibly affect the cracking behaviour. Further experimental and numerical investigations on  
516 other mechanical properties and their influences on fracturing are warranted. The preliminary  
517 3D simulation also requires that the calculation efficiency should be enhanced by either better  
518 meshing methods or modified dynamic failure criterion. If this problem can be solved, the 3D  
519 simulation will be more applicable than the 2D one to obtain a more accurate result and more  
520 information in the special failure processes.

521

## 522 **6. Conclusions**

523 In the present paper, both experimental and numerical approaches are applied to study the  
524 different mechanical and fracturing behaviour in Carrara marble under quasi-static and  
525 dynamic loadings. Some conclusions and explanations for the rate-dependent differences can  
526 be made below.

527 1) The strain rate effect is significant on both compressive strength and tensile strength, while  
528 the dynamic tensile strength increases several times faster than the dynamic compressive  
529 strength with the strain rate.

530 2) The strain rate plays a decisive role in the development of macro-cracks and the failure  
531 modes of the single-flawed specimens.

532 3) Both the experimental and numerical studies reveal that the development of tensile-wing  
533 cracks is suppressed under the dynamic loadings, which coincides with the higher increase rate  
534 of the dynamic tensile strength. Shear cracks and anti-wing cracks are encouraged under  
535 dynamic loadings leading to the “X” shaped failure of the specimens despite the flaw  
536 inclination angle. Different increase rate between tensile and compressive strengths induces  
537 rate-dependent cracking behaviour.

538 4) The consistency between the experimental observation and the numerical simulation  
539 indicates that this material model and the dynamic D-P failure criterion have a potential  
540 application into more practical rock engineering projects. The 2D simulation is suitable to  
541 obtain reliable data and enough cracking process information compared with the 3D simulation.

542 5) The numerical plastic zones have nearly the same trajectories as the experimental observed  
543 white patches, which supports the origin of the white patch to be plastic zone.

544

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553

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